Research Article

Numerical Research of the Submerged High-Pressure Cavitation Water Jet Based on the RANS-LES Hybrid Model

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1. Introduction

In recent years, with the implementation of energy-saving and environmental protection policies, the traditional cleaning technology for large-scale equipment with high pollution such as shot blasting is gradually eliminated. The high-pressure water jet cleaning technology has the characteristics of good processing quality, high efficiency, and environmentally friendly, which is gradually applied to the cleaning process of large-scale marine and chemical equipment [1].

The submerged high-pressure water jet is commonly accompanied by intensive cavitation; when the cavitation number is low and the bubble concentration is high, it is difficult to accurately measure the internal flow field by means of PIV or LDV. Therefore, it is of great significance to develop the numerical simulation technology of the submerged high-pressure water jet. After long-term research, the current CFD method has been able to accurately predict the velocity and pressure field of the submerged jets [2–9]. However, due to the complex phenomena involved in the cavitation jet, the model used in the calculation still needs to be further improved. At present, the main problems to be solved include mass transfer, turbulence, and compressibility. The mass transfer problem mainly corresponds to the modeling of cavitation dynamics. The turbulence problem is related to the accurate prediction of vortexes of various scales in the submerged jet, while compressibility will affect the periodic growth of cavitation unsteady characteristics such as shedding and collapse [10].
Currently, the most commonly used mass transfer models of cavitation include the Singhal model [11], Sauer model [12], and Zwart–Gerber–Belamri (ZGB) model [13]. The above models are based on the instantaneous pressure field to calculate the corresponding evaporation and condensation coefficients, which are converted into mass transfer rate and related to the mass and momentum transfer equations of each phase, so they are widely used in the multiphase flow model based on the Euler method. Hsiao et al. [14] modeled the cavitation process in another way. The Euler–Lagrangian method was used to calculate the flow field of the cavitation flow. The position of the gas core in the flow field was tracked under the Lagrange coordinate. In this way, the limitation of homogeneous flow was avoided, and the multiscale distribution of cavitation bubbles was taken into consideration. At the same time, the aggregation and fragmentation of cavitation bubbles were considered in this kind of model. However, the Eulerian–Lagrangian method requires massive computational resources; as a result, it is not widely used at present. Cheng et al. [15] proposed a new Euler–Lagrange cavitation model based on the Rayleigh–Plesset (R-P) equation, taking into account the noncondensable gas. In this model, the Euler method is used to solve the global flow field, and the Lagrangian method is used to track the migration of noncondensable gas bubbles into the vortex core. Based on the simplified R-P equation, the connection between local gas concentration and its effect on cavitation is modeled, and the mass source terms in the original Schnerr–Sauer (S-S) cavitation model are modified. Xi et al. [16] simulated the cavitating flow and pressure fluctuation in the tip region based on the delayed detached eddy simulation (DDES). The high-speed photography and transient pressure measurements were employed to capture the cavitation structures and pressure fluctuation. The numerical results showed a reasonable agreement with the available experiments.

In the research of submerged jet turbulence, different turbulence models were used to simulate free jet and impinging jet with different Reynolds numbers. Yang et al. [17] employed the SST $k$-$\omega$ turbulence model to capture the near-wall characteristics of the flow field within the impellers and diffusers, which has the advantage to predict the flow separation under adverse pressure gradients. Yang et al. [18] used ANSYS CFX 17.0 to do the numerical calculations. The SST $k$-$\omega$ turbulence model is selected, which is capable of solving turbulence parameters very close to the blades. Miltner et al. [19] carried out numerical simulation and experimental research on the submerged swirling jets. Various RANS models were used to calculate the flow field to obtain the velocity distribution at different positions of the jet. At the same time, LDV technology was used to measure the flow field. By comparing the velocity and turbulent kinetic energy distribution obtained by the simulation and experiment, it was found that the standard $k$-$\varepsilon$ model has a better prediction for the detailed characteristics of various regions of the jet. Saddington et al. [20] studied the supersonic jet of an axisymmetric contraction nozzle, found that the calculation results of the RNG $k$-$\varepsilon$ model were in good agreement with the experimental results, and accurately predicted the position and intensity of the shock wave generated by the supersonic jet. Jaworski and Zakrzewska [21] used the inclined blade impeller to form the impinging jet to the bottom in the agitator and carried out the numerical calculation and experimental research on the impinging jet at the same time. It was also found that the $k$-$\varepsilon$ model was the most accurate for the calculation of the averaged velocity and turbulent kinetic energy of the flow field. Birkby and Page [22] used the $k$-$\varepsilon$ model to simulate the undexpanded supersonic jet. In the simulation process, the pressure-based solver was used to iterate, and the compression performance was modified to accurately predict the position of the Mach disk. Wang et al. [23] used large eddy simulation to simulate the heat transfer process of the impinging jet, accurately obtained the heat transfer process of the jet cooling device through numerical calculation, and proposed a design method of the jet cooling device based on numerical simulation. Tip leakage vortex (TLV) flow is a common flow phenomenon in the axial-flow hydraulic machinery. High-efficiency simulation of TLV is still not an easy task because of the complex turbulent vortex-cavitation interactions. Wang et al. [24] evaluated the newly developed MST turbulence model in predicting the TLV flows.

However, in the case of high-pressure submerged jet, especially when intensive cavitation occurs, the turbulence model has a great effect on the prediction accuracy of the mass transfer phenomenon, while the research on the relation between the turbulence model and the cavitation formation process in the high-pressure submerged jet is lacking. The high computational cost of the LES model hinders the promotion of large eddy simulation technology in practical applications, so there is no discussion. The RANS model can no longer meet the accuracy requirements in some research and application fields, the RANS-LES hybrid model is to solve the boundary layer with the RANS model, and the separated flow is solved through the LES models. In this way, the accuracy of the flow field solution is improved while avoiding the excessive number of grids caused by the boundary layer analysis. In this paper, three different turbulence models that represent the RANS and RANS-LES hybrid model are used, and the probability for the models to simulate the microscale vortex structures is compared. By analyzing the relation between the small-scale vortex and the cavitation formation, the probability for these models to simulate the high-pressure cavitation jet with strong shear stress is discussed. The mechanism and the characteristics of the vortex and cavitation cloud evolution process of the submerged high-pressure jet are revealed.

2. Theoretical Models

2.1. Multiphase Model. Cavitation jet belongs to gas-liquid two-phase flow. In order to reduce the amount of calculation, the Euler model is usually used for the calculation of gas-liquid two-phase flow when the individual motion characteristics of the discrete phase are not concerned. In this paper, the mixture model is used to calculate the mixed-phase flow field, and the control equation is as follows [25]:

\[ \frac{\partial}{\partial t} \rho + \nabla \cdot (\rho \mathbf{u}) = 0 \]

\[ \frac{\partial}{\partial t} \rho \mathbf{u} + \nabla \cdot (\rho \mathbf{u} \mathbf{u}) = -\nabla p + \nabla \cdot \mathbf{T} + \rho \mathbf{g} \]

\[ \frac{\partial}{\partial t} \rho e + \nabla \cdot (\rho e \mathbf{u}) = \nabla \cdot (\mathbf{u} \cdot \mathbf{T}) - \rho (e + P) \frac{\partial \mathbf{u}}{\partial t} \]

\[ \frac{\partial}{\partial t} \rho \phi + \nabla \cdot (\rho \mathbf{u} \phi) + \nabla \cdot (\phi \mathbf{u} \mathbf{u}) = \nabla \cdot (\mathbf{u} \cdot \mathbf{T}) \]

where $\rho$ is the density, $\mathbf{u}$ is the velocity, $p$ is the pressure, $\mathbf{T}$ is the stress tensor, $\rho e$ is the total energy, $\mathbf{g}$ is the body force, $\phi$ is the volume fraction, and $\gamma$ is the specific heat ratio.
\[
\frac{\partial}{\partial t}(\rho_m) + \nabla \cdot (\rho_m \mathbf{v}_m) = 0,
\]
\[
\frac{\partial}{\partial t}(\rho_m \mathbf{v}_m) + \nabla \cdot (\rho_m \mathbf{v}_m \mathbf{v}_m) = -\nabla p + \nabla \cdot \left[ \mu_m (\nabla \mathbf{v}_m + \nabla \mathbf{v}_m^T) \right] + \rho_m \mathbf{g} + \nabla \cdot \left( \sum_{k=1}^{n} \alpha_k \mu_k \mathbf{v}_{dr,k} \mathbf{v}_{dr,k}^T \right),
\]

where \( \mathbf{v}_m \) is the mass-averaged velocity, \( \rho_m \) is the density of the mixture,
\[
\rho_m = \sum_{k=1}^{n} \alpha_k \rho_k,
\]
and \( \mu_m \) is the viscosity of the mixture, which is defined as follows:
\[
\mu_m = \sum_{k=1}^{n} \alpha_k \mu_k,
\]
where \( n \) is the number of phases, \( \mathbf{v}_m \) is the volume force, and \( \mathbf{v}_{dr,k} \) represents the slipping velocity of phase \( k \).

2.2. Cavitation Model. When the mixture model is used to calculate the cavitation multiphase flow, the expression of the gas volume fraction transport equation is as follows [26]:
\[
\frac{\partial}{\partial t}(\rho \alpha) + \nabla \cdot (\rho \mathbf{v}_g \alpha) = R_e - R_e,
\]
where \( R_e \) and \( R_c \) are evaporation and condensation rates, and their values can be calculated according to the cavitation model.

At present, the commonly used cavitation models are mainly derived from the Rayleigh–Plesset equation. The cavitation model used in this paper to calculate the phase transition is the Zwart–Gerber–Belamri model, and the mass transfer expression of the model is as follows:
\[
R_e = F_{\text{vap}} \frac{3 \alpha_{\text{vuc}} (1 - \alpha_c) \rho_c}{R_B} \left( \frac{2 P_g - P}{3 \rho_l} \right),
\]
\[
R_c = F_{\text{cond}} \frac{3 \alpha_{\text{puc}}}{R_B} \left( \frac{2 P - P_g}{3 \rho_l} \right),
\]
where \( R_B \) is the radius of the cavity, \( \alpha_{\text{vuc}} \) is the volume fraction of the gas core in the liquid, \( F_{\text{vap}} \) is the evaporation coefficient, and \( F_{\text{cond}} \) is the condensation coefficient. Considering the influence of turbulence on cavitation, the expression of the pressure threshold for cavitation occurrence is as follows:
\[
P_v = P_{\text{sat}} + \frac{1}{2} \rho_l k_t,
\]
where \( \rho_l \) and \( k_t \) represent the liquid phase density and liquid phase turbulent kinetic energy, respectively, and the coefficient \( c \) adopts the default value of 0.39.

2.3. Turbulence Model

2.3.1. RANS Models. Although the scale of the cavitation jet nozzle is relatively small, the nozzle outlet velocity is extremely high, so the Reynolds number of the submerged jet is large. The reasonable choice of the turbulence model has a great influence on the accuracy of the calculation results of the cavitation jet. This paper uses Reynolds time average (RANS) and hybrid (RANS-LES) models to perform numerical calculations on submerged cavitation jets. The calculation results are compared with experimental results, and the analysis is processed. The turbulence flow structure in the shear layer is compared for different turbulence models, and the applicability of the RANS model to calculate the cavitation jet is evaluated. In the RANS model, SST \( k-\omega \) has a good calculation effect on the vortex structure, so this paper will choose this model as the representative of the RANS model to simulate the cavitation jet, whose expression is as follows [27]:
\[
\frac{\partial}{\partial t}(\rho k) + \frac{\partial}{\partial x_j}(\rho k u_j) = \frac{\partial}{\partial x_j}(\Gamma_k \frac{\partial k}{\partial x_j}) + G_k - Y_k + S_k,
\]
\[
\frac{\partial}{\partial t}(\rho \omega) + \frac{\partial}{\partial x_j}(\rho \omega u_j) = \frac{\partial}{\partial x_j}(\Gamma_{\omega} \frac{\partial \omega}{\partial x_j}) + G_{\omega} - Y_{\omega} + D_{\omega} + S_{\omega},
\]
where \( G_k \) is the generation term of turbulent kinetic energy, \( G_{\omega} \) is the generation term of dissipation rate \( \omega \), \( Y_k \) and \( Y_{\omega} \) represent the dissipation of \( k \) and \( \omega \) caused by turbulence, \( \Gamma_k \) and \( \Gamma_{\omega} \) are the effective diffusivity of \( k \) and \( \omega \), and \( \sigma_k \) and \( \sigma_{\omega} \) are the turbulence Prandtl number of \( k \) and \( \omega \), respectively. \( \mu_t \) represents the turbulent viscosity, which is calculated by the following formula:
\[
\mu_t = \frac{\rho k}{\omega} \max \left( \left( \frac{1}{\alpha^*} \right), \left( \frac{SF_2}{\alpha \omega} \right) \right),
\]
\[
F_2 = \tanh \left( \Phi_2 \right),
\]
\[
\Phi_2 = \max \left[ \frac{2 \sqrt{K}}{0.09 \omega y^2} \frac{500 \mu}{\rho y^2 \omega} \right].
\]
2.3.2. RANS-LES Hybrid Models. With the continuous development of CFD technology, the RANS model can no longer meet the accuracy requirements in some research and application fields, and the high computational cost of the LES model hinders the promotion of large eddy simulation technology in practical applications. In order to solve the above problems, the RANS-LES hybrid model has gradually gained more attention in recent years. Among them, the separation vortex (DES) model is currently the most widely used hybrid model. This model was first proposed by Spalart [28] in 1997, so the original version of the DES model is also called DES97. The main idea of the DES model is to solve the boundary layer with the RANS model, and the separated flow is solved through the LES models. In this way, the accuracy of the flow field solution is improved while avoiding the excessive number of grids caused by the boundary layer analysis. To avoid the problem of the grid-induced separation (GIS), the boundary layer is further treated, and the model is called delayed detached eddy simulation (DDES).

(1) Detached Eddy Simulation. The expression of the turbulence energy dissipation term of the DDES model used in this paper is as follows:

\[ Y_k = \beta^* \kappa \omega F_{\text{DES}}, \]

\[ F_{\text{DES}} = \max \left( \frac{L_d}{\Delta_{\text{max}}} (1 - F_{\text{DDES}}), 1 \right), \]

\[ F_{\text{DDES}} = \tanh \left[ \left( \frac{C_d \tau}{r_d} \right)^{C_d_\alpha} \right], \]

\[ r_d = \frac{\nu_t + \nu}{k^2 \nu^2 \left( 0.5 \left( S^2 + \Omega^2 \right) \right)}, \]

where \( C_d = 20, C_d_\alpha = 3, S \) is the strain tensor size, \( \Omega \) is the vorticity tensor, and \( k = 0.41 \).

(2) Stress-Blended Eddy Simulation. In the process of studying the mixed model, a stress-mixed vortex model (SBES) appeared after the DES model. The shielding function used in this type of model is the same as the DES form. The SBES model can directly realize RANS and stress mixing between LES:

\[ \tau_{ij}^{\text{SBES}} = f_{r_{\text{SDES}}} \tau_{ij}^{\text{RANS}} + (1 - f_{r_{\text{SDES}}}) \tau_{ij}^{\text{LES}}, \]

where \( \tau_{ij}^{\text{RANS}} \) and \( \tau_{ij}^{\text{LES}} \) represent the stress tensor of RANS and LES parts.

3. Mesh and Boundary Conditions

Cavitation jet nozzles usually adopt special structures to improve the cavitation performance. Commonly used structures include convergent-divergent nozzles, organ tube nozzles, Helmholtz nozzles, and center body nozzles. Considering practicability and ease of processing, the first two nozzles are more widely used in high-pressure water jet cleaning, oil extraction, and peening. The current research mainly focuses on a convergent-divergent nozzle, and the nozzle structure is shown in Figure 1. The specific parameters are as follows: the convergent angle \( \alpha = 13.5° \), length of the convergent section \( L_1 = 5 \text{ mm} \), throat length \( L_2 = 4 \text{ mm} \), throat diameter \( d = 1 \text{ mm} \), outlet section length \( L_3 = 4 \text{ mm} \), and outlet expansion angle \( \theta = 40° \).

Figure 2 shows the calculation domain of the submerged jet. The incoming flow should develop fully in the pipe before reaching the nozzle, so the pipe at the nozzle inlet is extended by 260 mm upstream. In order to reduce the influence of entrainment and outlet reversed flow on the core area of the jet, the calculation boundary should be as far away as possible from the nozzle outlet. According to the literature, the diameter of the calculation domain should be greater than 100 times the diameter of the nozzle throat. In this paper, the diameter of the submerged water calculation domain is 200 mm. A cylinder is with a length of 500 mm, and the nozzle diameter used in the current calculation is 1 mm. The calculation domain size meets the above requirements.

Yun et al. [29] found that when the mesh resolution improves, the predicted velocity distributions and the cavitation patterns are in better agreement with the experimental results. Moreover, only when the mesh resolution reaches a certain high degree can the predicted cavitation results reflect the effects of the vortex stretching in the unsteady cloud cavitation. The computational domain is divided into structured grids using ANSYS ICEM, and the core area of the jet and the shear layer near the nozzle exit are refined. The computational domain grid is shown in Figure 3. In order to minimize the requirement for the calculation resource, the mesh size is gradually increased from the nozzle outlet to the downstream.

GIS has been a problem for DES models since the DES97 model; in the current research, different meshes are used to test the effect of the grid on the simulation results. The GIS characteristics of both DES and SBES models are investigated by refining the mesh gradually in the region of the throat and the outlet section. The total grid numbers are shown in Table 1.

Figure 4 shows the skin friction coefficient on the surface of the nozzle divergent section. It can be seen that the skin friction coefficient varies with the change of the grid number, while the result by SBES and DDES has few differences when the mesh is not very fine. Since the DDES model has a delayed mechanism to protect the boundary from GIS, the turbulence viscosity in the boundary layer is kept in the range for the RANS simulation. The result reveals that the current used SBES model also treats the boundary...
layer well in a RANS way. When the mesh is refined and the cell number is larger than 13 million, the skin friction coefficient of SBES differs from that of DDES. The result shows that the boundary layer of the DDES model tends to separate. With the refinement of the mesh, the variation of skin friction coefficient at the nozzle outlet becomes less, which remains almost unchanged when the mesh is larger than 19 million. Finally, the mesh with a cell number of around 19 million was used for the later simulation research studies.

According to the water domain structure shown in Figure 3, combined with the experimental conditions, the calculation boundary conditions are set. The calculation domain is meshed based on the finite volume method, and the pressure-velocity coupling method is used to solve the problem. The volume fraction transport equation uses the first-order upwind difference, the transient term uses the second-order central difference, the continuity equation, the momentum equation, and the turbulence control equation are discretized using the second-order upwind difference scheme. The inlet of the high-pressure pipe is set according to the corresponding cavitation number. The reference pressure in the calculation is 101,325 Pa. The outlet is set as the pressure outlet, the gauge pressure value is 0 MPa, and the other boundaries are set to non-slip wall. Since the cavitation jet is an unsteady phenomenon, transient calculations are used in all calculation examples. Due to the high velocity at the nozzle exit, in order to control the Courant number in the calculation, the calculation time step should be reduced as much as possible, which is set as 0.00001 s. The calculation accuracy is $1 \times 10^{-6}$, and the maximum number of iterations in each time step is 20 times. The total number of steps is 2000 steps.

4. Results and Discussion

4.1. Effect of Turbulence Models and Mesh. The choice of the turbulence model determines the accuracy of the velocity field, pressure field, velocity and pressure pulsation, and other physical quantities. Therefore, it is necessary to select a suitable turbulence model before conducting numerical simulation studies of cavitation jets. This paper uses the time-average model (RANS) and two hybrid models (RANS-LES) to simulate high-pressure water cavitation jets. The simulated boundary conditions are set according to the experimental conditions. The inlet pressure of the high-pressure pipeline is set to 20 MPa, and the corresponding cavitation number is $\sigma = 0.005$.

Figure 5 shows the instantaneous velocity distribution of submerged jets calculated by three different turbulence models. The RANS model used is the SST $k-\omega$ model, and the hybrid model uses DDES and SBES based on the SST $k-\omega$ model. According to the calculated velocity field, it can be seen that the jet velocity field predicted by the above models is basically reasonable. The velocity in the core area near the nozzle exit is the highest, the velocity gradually decays along the axis, and the velocity decays to zero when it reaches 150 mm downstream of the nozzle. At the same time, under the action of the viscous force of the shear layer, the high-speed jet entrains the surrounding fluid, and the process of
velocity attenuation from the nozzle to the downstream is accompanied by radial diffusion. The SST $k$-$\omega$ model has a smooth transition from the core area to the periphery due to the time-averaged effect, which can basically reflect the spatial distribution of the submerged jet velocity under this working condition. In the results calculated by the DDES models, the high-speed core area extends from the nozzle to about 30 mm downstream. At the same time, when the velocity field decays, there is no smooth contour but a fuzzy gradient boundary, which is consistent with the dissipation of large-scale vortexes to small-scale vortexes under the action of viscous forces in the turbulent flow field. Compared with the DDES model, the velocity field calculated by the SBES model can reflect the small-scale vortex structure at the edge of the shear layer from the nozzle exit, and the core high-speed zone is intermittently distributed along the axis. The intermittent distribution indicates that the model can better reflect the unsteady velocity pulsation of the cavitation jet flow field.

Figure 6 shows the calculated void volume fraction distribution of the submerged jet. It can be seen that the shape of the void cloud obtained by the last two turbulence models is relatively similar on the whole and is close to the real cavitation form of the high-pressure submerged jet, while the cavitation calculated by the SST $k$-$\omega$ model is only distributed in the 10 mm long interval of the nozzle outlet, which is quite different from the real situation observed in the experiment, indicating that the SST $k$-$\omega$ model is affected by the time-average processing and cannot accurately capture the shear vortex structure and the low-pressure vortex core caused by it. Comparing the three turbulence models, it can be seen that the cavitation range and concentration in the shear layer near the nozzle outlet increase sequentially. The cavitation cloud calculated by the DES model is a discontinuous cluster distribution, and the cavitation cloud calculated by the SBES model is distributed continuously.

At present, the mainstream explanation for the mechanism of submerged jet cavitation is that a Rankine combined vortex is formed between the high-speed jet and the static submerged water under the action of shear. The vortex of this structure is composed of two parts inside and outside. The velocity and pressure value are lower near the center, and the cavitation occurs when the central pressure of the Rankine vortex is lower than the saturated vapor pressure of the liquid. Therefore, whether the cavitation process of the high-pressure water submerged jet can be accurately predicted has put forward higher requirements for the accuracy of the vortex structure prediction. Figure 7 shows the vorticity distribution in the jet flow field calculated by the compared models. It can be seen that the vorticity distribution is similar to the distribution of the cavitation cloud. This is because the vortex and turbulence affect the cavitation in the flow field. The vorticity calculated by the SST $k$-$\omega$ model is mainly concentrated in the nozzle outlet section. The area with a larger vorticity value is basically the same as...
Figure 6: Vapor volume fractions by different turbulence models.

Figure 7: Vorticity of the jet by different turbulence models.
the cavitation area, and the transition of the vorticity iso-
surface is smooth. The vorticity calculated by the DES
models clearly reflects the characteristics of turbulence. The
vorticity boundary of the nozzle outlet section is relatively
smooth, and the transition begins at about 10 mm down-
stream. A vortex appears around the shear layer, and the
vortex scale increases gradually along the mainstream di-
rection. The results of the SBES model are similar to those of
the DES model, and the difference is that the position where
the vortex begins to appear is closer to the nozzle.

Through calculation and comparison, it is found that the
SST $k$-$\omega$ model is affected by the time-average processing
when simulating the submerged high-pressure water cavi-
tation jet, and the corresponding scale of turbulence and
vortex structure cannot be obtained. Therefore, the calcu-
lated local pressures are average values, which affect the
capture of cavitation phenomena. The two hybrid models are
ideal for the calculation results of high-pressure submerged
jets. They can capture the vortexes generated by the viscous
force of the shear layer, as well as the merger and dissipation
of the vortexes. The shape, size, and development period of
the cavitation cloud are reasonable compared to the ex-
periment. Comparing the two hybrid models, the SBES
model has a better probability to capture details in the shear
layer.

Besides the GIS phenomenon, mesh size is found to have
great effect on the vapor distribution in shear cavitation
flows. Figure 8 shows the isosurface of 0.5 vapor volume
fraction with different turbulence models and mesh sizes. It
can be found that the shape of the cavity changes with the
refinement of the mesh, and more small vortex structures
can be seen in the case of the fine meshes. Generally, the
result of the SBES model improves more by refining the
mesh. For the hybrid models, the region where the vortex is
solved is decided by the shielding function, which defines the
region where the turbulence viscosity is modeled. Figure 9
shows the isosurface of the vorticity colored by eddy vis-
cosity ratio. It can be found that the eddy viscosity ratio is
decreased when the mesh is refined. Comparing the result of
the same mesh with different models, one can find that the
result of the SBES model shows lower eddy viscosity, and the
jet transfers faster to turbulent under the effect of the shear
stress when the jet flows from the nozzle outlet to
downstream.

4.2. Validation of the Simulation Result. Before analyzing the
high-pressure water submerged cavitation jet using the
current simulation model, a nozzle with a 40° expansion
angle is taken for validation. The numerical calculation
method and its accuracy are verified by comparing the simu-
lation results with the high-speed photography results.
Figure 10 shows the comparison of the cavitation mor-
phology obtained by high-speed photography and numerical
simulation of the selected nozzle with different turbulence
models. It can be found from the figures that the shape of the
cavitation cloud predicted by the SST $k$-$\omega$ model is obviously
different from the reality. The two kinds of hybrid models
can provide good results, which are similar to the high-speed
photograph result both in shape and scale. Observing the
high-speed image, small-scale vortex structures can be seen
from the region close to the nozzle outlet. Comparing the
results by the two types of hybrid models, it can be seen that
SBES provides more detailed shear vortexes near the nozzle
outlet. As a result, the later numerical investigation on the
high-pressure cavitation jet is conducted using the SBES
model.

4.3. Vortex and Cavitation Characteristics. High-pressure
water submerged jets usually have a relatively high central
velocity. When the high-speed liquid rushes into still sub-
merged water, there is a large velocity gradient at the
boundary layer. The boundary layer is filled with shear stress
and reverses pressure difference. For small-scale vortexes,
because the pressure at the center of the vortex is low,
cavitation will occur when the pressure at the center of the
vortex is lower than the saturated vapor pressure at the
ambient temperature where the liquid is located. When the
jet keeps shearing submerged water at a high speed, the shear
layer will always be full of Rankine vortexes, and cavitation
in the low-pressure zone will occur continuously, and the
bubbles will travel downstream with the mainstream.
Figure 11 shows the velocity vector diagram at the exit of the
angular nozzle. It can be clearly seen from the figure that the
velocity in the core area is as high as 200 m/s, while sub-
merged water at the periphery is almost stationary. Shearing
layer thickness at the exit of the throat is on the order of
micrometers, which indicates that there must be a strong
velocity gradient in the shear layer. Under the action of
viscous force and reverse pressure difference, the phe-
omenon of K-H instability appears. The shear layer is
rapidly destabilized from the ideal shear layer, and the fluids
on both sides begin to exchange momentum to form a vortex
structure. The process of vortexes propagating downstream
is accompanied by a pairing process. Small-scale vortexes in
the shear layer continue to increase through merging, so the
thickness of the shear layer continues to increase, and the jet
velocity field also diffuses.

Figure 12 shows the distribution of various physical
quantities near the shear layer of the nozzle outlet section.
From the cavitation vapor volume fraction in Figure 12(a), it
can be seen that the cavitation of the nozzle starts from the
junction between the throat and whistle. The cavitation
bubbles are mainly distributed along the boundary of the
shear layer near the exit of the nozzle. Affected by pressure
fluctuations, cavitation near the shear layer is also accom-
panied by the collapse of a large number of bubbles. The
shock wave generated during the collapse aggravated the
collapse of the surrounding cavitation. The alternate gen-
eration and collapse of the cavitation formed the develop-
ment process of the cyclical growth, shedding, and collapse
of the cavitation cloud as captured by the high-speed
photography experiment. Figure 12(b) shows the mass
transfer rate between vapor and liquid phases. A positive
value represents the occurrence of cavitation, while a neg-
ative value represents the collapse of cavitation bubbles. It
can be seen that, except for the exit of the throat, the area
where cavitation occurs is basically circular in cross section, which corresponds to the vortex ring formed by shear in the space. Figure 12(c) shows the distribution of the vortex criterion $Q$ value, which is defined as follows:

$$Q = \frac{1}{2} (\|\Omega\|^2 - \|E\|^2) = \frac{1}{2} (\Omega_{ij} \Omega_{ij} - e_{ij} e_{ij}),$$  \hspace{1cm} (11)

where $\Omega_{ij}$ and $e_{ij}$ represent the vorticity tensor and strain rate tensor, respectively; when $\|\Omega\|^2 - \|E\|^2 > 0$, it means that the effect of the local fluid rotation overwhelms the strain.

It can be seen that the $Q$ value in the shear layer alternates between positive and negative. Under the effect of the shear stress, the viscous fluid must have a strain rate, and the positive $Q$ value represents the formation of vortexes. The characteristics of the vortexes are gradually increased streamwise. On the one hand, decrease of the core velocity in the downstream leads to a decrease in shear stress. On the other hand, the scale of the vortexes gradually increases during the vortex pairing process. Figure 12(d) shows the distribution of turbulent kinetic energy near the shear layer. The area with higher turbulent kinetic energy corresponds to a stronger vortex. Comparing the volume fraction of bubbles, it can be found that the turbulent kinetic energy is the highest at the gas-liquid interface, indicating that, in such areas, the flow field disturbance is increased due to the momentum exchange between the two phases.

In industrial applications, jets of different pressures and flows correspond to different application conditions. When nozzles, pipelines, and other systems are determined, it is generally necessary to select a reasonable operating pressure according to the requirements. When the pressure upstream of

Figure 8: Shape of vapor with different meshes and turbulence models.

Figure 9: Isosurface of vorticity colored by eddy viscosity ratio.
the jet is changed, the pressure field, velocity field, and turbulence intensity in the jet will all change accordingly. In order to study the characteristics of the submerged cavitation jet under different pressure supplies, the aforementioned nozzle with an outlet angle of 40° was used to calculate the jet flow field under different pressures. Figure 13 shows the isosurface with a vapor volume fraction of 0.2. Through comparison, it can be seen intuitively that, with the increase of pressure, the distribution range of the cavitation cloud shows an expanding trend in both axial and radial directions, while the shape of the vacuole cloud is similar.

Figure 14 shows the distribution of vapor volume fraction inside and near the outlet of the nozzle under different cavitation numbers. The angle of the convergent section at the inlet of the angular nozzle throat is designed to be 13.5° so that the fluid can enter the throat section smoothly. The transition structure effectively weakens the flow separation and vortex generation at the entrance of the throat, so there is almost no cavitation inside the nozzle. Cavitation starts to appear at the exit of the throat, which increases gradually with the decrease of the cavitation number. When $\sigma = 0.0203$, cavitation mainly occurs at the...
Figure 12: Distribution of variables at the angular nozzle outlet. (a) Vapor volume fraction. (b) Mass transfer rate. (c) Q-criteria. (d) Turbulence kinetic energy.

Figure 13: Isosurface of vapor volume fraction of the jet under different cavitation numbers. (a) $\sigma = 0.0203$. (b) $\sigma = 0.0101$. (c) $\sigma = 0.0068$. (d) $\sigma = 0.0051$. 
connection between the outlet of the throat and the whistle. At the outlet of the whistle, there is basically a low-concentration cavitation group. When \( \sigma \approx 0.0101 \), the cavitation cloud extends to the whistle outlet, but the cavitation cloud in the external flow field is still intermittently distributed, shedding off from the outlet and moving downstream. When \( \sigma \approx 0.0068 \), the length of the cavitation cloud generated in the whistle increases and extends to the downstream divergent section until it sheds off under the influence of the large-scale vortex. When the cavitation number is further reduced to \( \sigma \approx 0.0051 \), the cavitation is not limited to the inside of the nozzle because the shear layer with large viscous force is also expanded, and cavitation occurs within a certain distance from the nozzle outlet.

Figure 15 shows the jet velocity field under different cavitation numbers. The shape of the velocity field is similar to that of the vapor isosurface. Because the ambient pressure of the submerged jet remains unchanged, the turbulence intensity decreases with the decrease of the cavitation number. When the cavitation number is reduced from 0.0203 to 0.0051, the range of the submerged jet increases from about 60 mm to more than 120 mm. At high cavitation number conditions, the velocity core area in the nozzle exit section is relatively smooth, and the transition occurs downstream of the nozzle, while at low cavitation number, due to the high Reynolds number, the jet transitions inside the whistle. It accelerates the diffusion of the downstream flow field and makes the cavitation cloud more widely distributed.

Since the RANS-LES hybrid model has large fluctuations in the velocity and pressure in the high-turbulence area near the nozzle outlet, here, results of 400 time steps are time-averaged during calculation to obtain the averaged velocity field. Figure 16 shows the radial distribution curve of jet velocity at different positions downstream of the nozzle under different cavitation numbers. Generally speaking, the jet velocity decays along the radial direction in all cases, which is similar to the curve shape in the experiment. Since the cavitation number here is adjusted by adjusting the pump pressure upstream of the nozzle, as the cavitation number decreases, the jet pressure gradually increases, and the nozzle outlet velocity also shows an increasing trend. When the cavitation number \( \sigma \approx 0.0203 \), the maximum velocity at 10 mm downstream of the nozzle is 100 m/s. At this position, the jet has passed the turning section, and there is no obvious core area. When the cavitation number \( \sigma \approx 0.0051 \), the maximum velocity at 10 mm downstream of the nozzle is 140 m/s, which is at the initial stage of the jet. The jet velocity first remains constant from the center along the radial direction and then begins to decrease. As the cavitation number continues to decrease, the characteristics of the core area at \( x = 10 \) mm become more obvious, and the maximum core velocity continues to increase.

The velocity under different cavitation numbers shown in Figure 16 is changed to a dimensionless form, where \( u^* = u/ u_0 \) and \( y^* = y/ y_{0.5} \). Figure 17 shows the radial distribution curve of the nondimensional axial velocity. It can be seen that the jet velocity distribution at different positions conforms to the characteristics of self-similarity, and the dimensionless velocity distribution curves under different cavitation numbers basically coincide, indicating that the cavitation number has little effect on the dimensionless distribution of jet velocity in the radial direction. According to the results of the PIV experimental data analysis, the radial distribution of the submerged cavitation jet velocity conforms to the exponential function. By Gaussian fitting, the dimensionless velocity distribution curve can be described as \( u^* = e^{-(r^*/1.5)^2} \). According to the comparison between the
Figure 15: Velocity magnitude under different cavitation numbers. (a) $\sigma = 0.0203$. (b) $\sigma = 0.0101$. (c) $\sigma = 0.0068$. (d) $\sigma = 0.0051$.

Figure 16: Continued.
fitting function and the numerical calculation results in Figure 17, it can be seen that the formula is in good agreement with the calculation results, indicating that the fitting formula in the PIV experiment is applicable under different cavitation numbers.

5. Conclusions

In the current research, the submerged cavitation jet with different cavitation numbers is numerically investigated, and the effect of the turbulence model on the simulation results...
of the cavitation jet is analyzed. Three different turbulence models that represent the RANS and RANS-LES hybrid models are compared. The probability for the models to predict the vortex with small scale and the cavitation is evaluated in detail. The mechanism of the cavitation generation in the intensive shear layer of the submerged jet is revealed according to the simulation result, and the effect of the cavitation number on the characteristics of the submerged cavitation jet is analyzed as well. The following conclusions were obtained through research:

(1) The calculation probability for small-scale vortices has a great influence on the prediction accuracy of cavitation in the submerged jet. The SST model is affected by the time-averaged process, which cannot predict the small-scale vortices in the jet shear layer. Representative models of RANS-LES include the DES model and the SBES model. These two models can effectively capture the vortices in the shear layer, and the SBES model can obtain more turbulence details than the DES model. The result of the simulation under different cavitation numbers using the SBES model agrees well with the experimental result.

(2) In the case of a low cavitation number, there exists an extensive shear layer at the exit of the corner nozzle, and small-scale vortices will continue to form in the shear layer. According to the comparison of different physical parameters near the nozzle outlet, it is found that cavitation starts at the interface between the throat and the nozzle and mainly distributes along the boundary of the shear layer near the nozzle exit. The phase transition from liquid to cavitation is in a ring-shaped distribution, mainly located in the low-pressure area of the vortex center.

(3) According to the simulation results, it is found that, as the cavitation number decreases, the size and volume fraction of the cavitation bubbles increase significantly. The scale of the jet velocity field increases obviously with the decrease of the cavitation jet, while the profiles of the velocity distribution are similar to each other. The nondimensional velocity distribution along the radial direction shows a perfect similarity trend under each cavitation number, which follows a fitted exponential function.

Data Availability

The curved data used in this paper are available from the corresponding author upon request.

Conflicts of Interest

The authors declare that they have no conflicts of interest.

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