RADAR IMAGING COMPLEX WITH SAR AND ASR FOR AEROSPACE VEHICLE

The subject of this study in the article is the algorithms for radio monitoring of the Earth in various views from aerospace transport. The goal is to design a structural diagram of a radio complex that can operate simultaneously in two modes: modified antenna aperture synthesis (SAR) and aperture synthesis (RAS), in accordance with algorithms synthesized using the maximum likelihood method. The modified SAR mode allows obtaining high-resolution radio images in the observation angle range ±(20°...50°) from the direction to the nadir. A method for combining a modified PCA algorithm is used, which differs from the classical imaging algorithm by the possibility of obtaining a higher spatial resolution, the payment for this is the complication of the signal processing algorithm associated with the implementation of decorrelating filters that expand the spectrum of received signals in each receiving path, and the RAS mode, which allows imaging using passive or active radar principles. The passive RAS mode provides for the imaging in the observation angle range of ±20° from the nadir based on the results of processing signals of its own broadband radio-thermal radiation, and active mode – in the same observation angle range, but using the broadband noise signal of the backlight. An important result in the formation of a radio image in the specified viewing area when using the active mode of the RAS is that the images are close in physical content, namely, proportional to the specific effective reflection surface of the underlying surface. Additionally, a distinctive feature of the synthesized algorithms is the use of wideband probing signals and, accordingly, the same input paths of receivers, which makes it possible to increase the signal-to-noise ratio of the output effect. Conclusions. The scientific novelty of the results obtained is as follows: a structural diagram of the radio complex was developed on the basis of algorithms synthesized using the maximum likelihood method. For the formation of a radio image in the radio complex, a combination of SAR and RAS (with two modes of operation) is implemented. This implementation is important, since it allows obtaining high-resolution images in the observation angle range of ±50° from the direction to the nadir. It is advisable to place the complex on airplanes, helicopters and spacecraft (preferably those that move in low orbits).

Keywords: radar imaging; synthetic aperture radar; aperture synthesis radar; signal processing algorithm; ultrawideband signal processing.

Introduction

Motivation. High information content about the Earth's surface can be obtained using optical and radar surveys [1, 2] from an aerospace carrier. At the same time, radar images are much more informative in conditions of limited optical visibility due to cloudiness, smoke, darkness, etc. This creates the prerequisites for the widespread use of aerospace-based radars capable of imaging. However, it should be noted that to ensure high spatial resolution (comparable to the resolution of images in the optical range), it is necessary to develop and implement complex algorithms for processing radio signals [3 - 7], as well as algorithms for their filtering [8, 9]. In particular, filtering algorithms are necessary because radio images obtained by active radars are usually distorted by a high level of speckle-noise.

Today, two main signal processing algorithms are used to ensure high spatial resolution of radar images. These are synthesis of antenna aperture algorithm implemented by synthetic aperture radar systems (SAR) [10, 11], and an aperture synthesis algorithm implemented in passive aperture synthesis radar (ASR) [12-14]. Each has its own advantages and disadvantages. Thus, SAR allows obtaining high spatial resolution of the image in the range of observation angles ±(20°...50°) from the direction to the nadir [15-17], but does not provide a high resolution in the range of ±20° from the nadir. ASR, on the other hand, allows us to get an image in the range of ±20° from the nadir. Therefore, as shown in [18, 19], it is advisable to combine these radio systems into a single complex capable of forming radar images in a swath of ±50° from nadir.

State of the Art. It should be noted that significant advances in the field of increasing the spatial resolution of images are obtained due to the ability to form and process wide- and ultra-wideband signals [20-22].
modern radioelement base makes it possible to design ultra-wideband radars. Such radars can be multi-position [23, 24], single-position with probing through obstacles [25-27], single-position but with multifrequency signal processing [28], single-position with very short pulse signals [29]. Unlike all of the above, the designed complex is supposed to work not with ultra-short pulse signals, but with ultra-wideband noise signals [20-22]. The latter is very important for meeting the modern requirements of Green Engineering [30].

**Objectives.** The report develops the works [3, 15-19, 31] and proposes the synthesis of a radio complex with the processing of wideband spatio-temporal signals to obtain radar images in the range of observation angles ±50° from nadir with high spatial resolution.

Purpose of the article: designing and research an aerospace-based radio complex capable of forming high spatial resolution radio images in the sector of observation angles ± 50°.

**Radar Imaging Algorithms**

**A. Modified SAR Imaging Algorithm**

We will solve the problem of synthesizing a signal processing algorithm for forming images in the field of view ±(20°...50°) from the nadir. For this, we will use a unified recording [32-34] of a set of wideband probing signals

\[
 s(t) = \text{Re}\exp(-j\varphi_0) \times \\
 \sum_{n=1}^{N} B_n(t) \exp\left(j2\pi f_0(t-(n-1)T_r)\right),
\]

where \( B_n(t) \) is the complex envelope of the \( n \)-th pulse \( (n=1..N) \) in the set, which, for example, can describe linear frequency modulation (Chirp)

\[
 B_n(t) = \Pi(t-(n-1)T_r)\exp\left[j\pi\Delta F^{-1}(t-(n-1)T_r)^2\right];
\]

\( \Pi(t-(n-1)T_r) \) is the radio pulse envelope;

\( \Delta F \) is the bandwidth;

\( T \) is the signal duration;

\( T_r \) is the pulse repetition period in the set;

\( f_0 \) is the central frequency of the emitted signal;

\( t \) is the time;

\( \varphi_0 \) is the initial phase;

\( j \) is the complex unit.

The expression for \( B_n(t) \) can have other forms of notation [35, 36], which does not significantly affect the physical properties of the chirp signal.

We will specify the width of the passband \( \Delta F \) during the simulation; we only note that it must satisfy the wideband condition [37-39]. It should be noted that today there are several options for defining the concept of broadband (at different levels of spectrum falloff from its maximum value) [40-42] and ultra-wideband (in the context of using ultrashort pulses) [43, 44] signals. The choice of a specific criterion of broadband is advisable to carry out based on the specific task under consideration. In fig. 1 it is shown the geometry of the problem, which is necessary for solving the problems of synthesis of signal processing algorithms.

In fig. 1 introduced the following designations: \( V \) is the flight speed; \( t \) is the time; \( t_0 \) is the beginning of synthesis; \( R(t, \cdot) \), \( R_p(t, \cdot) \) are the current distances from the phase center of the transmitting antenna to the element of the underlying surface and from this element to the \( p \)-th element of the antenna system \( (p = 1..Q) \);

\( A_p \) are the elements of the receiving antenna; \( A_{tg} \) is the transmitter antenna; \( r \) is the radius vector from the projection of the phase center of the transmitting antenna on the underlying surface to the elementary area \( D \) with coordinates \((x_C, y_C)\); \( h \) is the flight altitude of the aircraft.

![Fig. 1. Geometry of the radar imaging problem](image)

The signal reflected by a single element of the underlying surface, the coordinates of which are determined by a vector \( \vec{r} \) (Fig. 1), can be written in the following form

\[
 s_s(t,h,r) = G_p(t) G_r(t) \exp(-j\varphi_0) \times \\
 \sum_{n=1}^{N} B_n(t) \exp\left[j2\pi f_0(t-(n-1)T_r-t_d(h,r))\right],
\]

(2)
where \( h \) is the flight altitude of the aerospace carrier; 
\( t_d (h, \mathbf{r}) \) is the signal propagation delay time; 
\( \hat{G}_t (t - t_d (h, \mathbf{r})) \) is the complex radiation pattern of the transmitting antenna in the direction of the surface with coordinates \( \mathbf{r} \).
The signal reflected within the irradiation zone of the underlying surface, which entered the receiving antenna, can be written as follows

\[
s(t) = \exp(\mathbf{j} \varphi) \int_{\mathcal{D}} \hat{G}_r (t, \mathbf{r}) F(\mathbf{r}) s_\delta (t, h, \mathbf{r}) \, d\mathbf{r},
\]
where \( F(\mathbf{r}) \) is the specific complex reflection coefficient of the underlying surface element with coordinates \( \mathbf{r} \);
\( \hat{G}_r (t, \mathbf{r}) \) is the complex radiation pattern of the receiving antenna in the direction of the underlying surface element with coordinates \( \mathbf{r} \) (limits the irradiation zone of the underlying surface). Phase \( \varphi \) uncontrollable, but it can be neglected in coherent signal processing.

We can write the observation equation in the form of an additive mixture of signal (3) and noise

\[
u(t) = \text{Re} \{ s(t) \} + n(t),
\]
where \( n(t) \) is the internal noise of the receiver (white Gaussian noise with zero mean).

The signal processing algorithm in the modified SAR (MSAR) is based on the maximum likelihood method [45] from the solution of the likelihood equation with respect to the optimal recovery of the specific effective reflection surface of the surface elements \( \sigma(\mathbf{r}) \), which can be considered as a radar image,

\[
\frac{\delta \ln p(u(t) | \sigma(\mathbf{r}))}{\delta \sigma(\mathbf{r}')} = \frac{\delta}{\delta \sigma(\mathbf{r}') \ln [k(\sigma(\mathbf{r}))]} \times \exp \left( \frac{1}{2} \int_{0}^{T} W(t_1, t_2, \sigma(\mathbf{r})) u(t_1) u(t_2) \, dt_1 \, dt_2 \right) = 0,
\]
where \( \frac{\delta}{\delta \sigma(\mathbf{r}') \ln [k(\sigma(\mathbf{r}))]} \) is the variational derivative of the specific effective reflection surface of the underlying surface;

\( W(t_1, t_2, \sigma(\mathbf{r})) \) is the inverse function to the correlation function of the observation equation.

Solution (5) can be reduced to the following algorithm for optimal signal processing in MSAR (the transition to complex envelope processing is taken into account):

\[
\hat{I}_{\text{MSAR}} (\mathbf{r}) = \int_{0}^{T} \hat{U}(t) \hat{S}_\delta (t, h, \mathbf{r}) \, dt^2,
\]
where \( \hat{U}(t) \) is the complex observation envelope;
\( \hat{S}_\delta (t, h, \mathbf{r}) \) is the complex conjugate complex envelope of the signal (2). The radar image, which is formed in accordance with this algorithm, will be complex. Therefore, to display it, the modulus or square of the modulus is usually calculated [45] and it is this function that is taken for the assessment of the radar image, i.e.

\[
\hat{I}_{\text{MSAR}} (\mathbf{r}) = \left( \int_{0}^{T} \hat{U}(t) W(t_1, t_2, \sigma(\mathbf{r})) \hat{S}_\delta (t_1, h, \mathbf{r}) \, dt_1 \, dt_2 \right)^2.
\]

For the first time, the problem of synthesis of an algorithm for optimal recovery of the specific effective scattering surface and its solution with obtaining the MSAR algorithm were developed by V. K. Volosyuk [46].

Algorithm (6) corresponds to the following basic signal processing operations: forming the envelope of the reference signal \( \hat{S}_\delta (t, \mathbf{r}) \) for different range sections (the number of range sections is determined by a specific signal shape and antenna directivity patterns for transmission and reception); setting of the decorrelation filter, the impulse response of which is determined by the function \( W(t_1, t_2, \sigma(\mathbf{r})) \) (in the optimal form, this function is adaptive, since it requires a priori knowledge of \( \sigma(\mathbf{r}) \)); formation of observation envelopes \( \hat{U}(t, \mathbf{r}) \) for different parts of the range; performing double correlation processing operations in accordance with (6).

It should be noted that the most difficult operation in (6) is the formation of the function \( W(t_1, t_2, \sigma(\mathbf{r})) \). However, in practice, it can be realized as the average value of \( \sigma(\mathbf{r}) \), which is typical for a certain type of underlying surface. In some cases, filtering can be neglected and go to a quasi-optimal signal processing algorithm. For this, it is necessary to introduce the following replacement in (6) \( W(t_1, t_2, \mathbf{r}) = \delta(t_1 - t_2) \). In this case, the signal processing algorithm is reduced to the classical algorithm used in SAR [47, 48]

\[
\hat{I}_{\text{SAR}} (\mathbf{r}) = \left( \int_{0}^{T} \hat{U}(t) \hat{S}_\delta (t, h, \mathbf{r}) \, dt \right)^2.
\]

Note that algorithm (6) allows obtaining high spatial resolution for viewing angles \( \pm(20^\circ ... 50^\circ) \) from the direction to the nadir.
The block diagram that implements the MSAR algorithm will be given below as part of the complex for the formation of radar images.

B. Active ASR Imaging Algorithm

For the first time, the problem of synthesizing a signal processing algorithm in an active aperture synthesis system was formulated and partially solved in [15]. Its peculiarity lies in the fact that it is not the effective temperature of the underlying surface that is subject to recovery, as is typical for passive aperture synthesis systems, but the specific effective scattering surface, similar to active synthesis systems, for example, in MSAR. This makes it possible to form radar images of the same physical entity by different synthesis systems. Active systems of aperture synthesis are of particular importance when solving problems of forming radar images within ±20° of the nadir.

We will synthesize a signal processing algorithm in such a system. To achieve a high resolution in spatial coordinates, we will use a wideband signal, the model of which can be represented as a Gaussian process with zero mean as follows

\begin{equation}
F_{\text{max}} \int_{F_{\text{min}}} s(t) = 2 \int F_{\text{j}2\pi f} N(j2\pi f) \exp(j2\pi f t) df,
\end{equation}

where \( F_{\text{max}} \) and \( F_{\text{min}} \) are the limits of operating frequencies;

\( N(j2\pi f) \) is the random spectral density of the complex amplitude;

\( t \) is the time;

\( f \) is the frequency.

As it is usually done in aperture synthesis systems, to receive the reflected signal, we will use an \( N \)-element antenna array with a non-equidistant spatially distributed arrangement of the phase centers of individual antennas. The position of the phase centers of the elements of the antenna array is determined by the Monte Carlo method.

According to fig. 1 let us determine the signal propagation delay time from the transmitter antenna to an elementary section of the underlying surface and then to the \( i \)-th \((i = 1..N)\) receiver antenna (the phase center is characterized by the end of the radius vector \( r'_i = (x'_i, y'_i) \), the beginning of which is at the phase center of the transmitter antenna)

\begin{equation}
t_{\text{d},i}(r, r'_i) = (R(t, \cdot) + R_1(t, \cdot))c^{-1},
\end{equation}

where \( R(t, \cdot) = \sqrt{h^2 + (x'_C - Vt)^2 + y'^2_C} \),

\( R_1(t, \cdot) = \sqrt{h^2 + (x'_C - Vt - x'_{di})^2 + (y'_C - y'_{di})^2} \),

\( (x'_i, y'_i) \) are the coordinates of the phase center of the \( i \)-th element of the antenna system, \( i = 1..N \), \( (x'_{di}, y'_{di}) \) are the coordinates of the receiving aperture of the \( i \)-th element of the antenna system, which are counted from the end of the vector \( r'_i \).

Signal at the output of the \( i \)-th element of the receiving antenna system

\begin{equation}
s_i(t, r'_i) = \int_{-\infty}^{\infty} \int_{\text{F}_{\text{min}}} f_{\text{C}}(2\pi f, r) \mathcal{G}(2\pi f, Vt, r) \mathcal{N}(2\pi f) \times \times \exp\left(2\pi f \left(t - t_{\text{d},i}(r, r'_i)\right)\right) df \ dr'_i dr \int_{-\infty}^{\infty} \int_{\text{F}_{\text{min}}} ,
\end{equation}

where \( \mathcal{G}(2\pi f, Vt, r) \) is the complex radiation pattern of the transmitter, which is recalculated to the elements of the underlying surface;

\( f_{\text{C}}(2\pi f, r) \) is the complex reflection coefficient of an elementary section of the underlying surface centered at the end of the vector \( r \).

We rearrange the observation equation at the output of each element of the antenna system by a vector:

\begin{equation}
u(t) = \left\{s_i(t, r'_i) + n_i(t)\right\}_{i = 1}^{N},
\end{equation}

where

\begin{equation}
n_i(t) = \int_{\text{F}_{\text{min}}}^{\text{F}_{\text{max}}} N_{\text{n},i}(j2\pi f) \exp(j2\pi f t) df ,
\end{equation}

is the internal noise of the \( i \)-th receiving channel,

\( N_{\text{n},i}(j2\pi f) \) is the random spectral density of the complex amplitude of the noise in the \( i \)-th channel.

The physical nature of noise is such that it is a zero-mean Gaussian process. There is no cross-correlation of noise between different channels, because in different channels the noise is statistically independent.

Full information about the observation is contained in the matrix of correlation functions

\begin{equation}
R_{ij}(t_1, t_2, \sigma(r)) = \left\{u(t_1) u^T(t_2)\right\},
\end{equation}

where "\( T \)" is the transposition operator;

\( \sigma(r) \) is the radar cross section of the underlying surface.
Substituting (9) and (11) in (10), and (10) in (12) and rewrite the matrix of correlation functions in the following form

\[
\begin{align*}
\mathbf{R}(t_1, t_2, \sigma) &= \left\{ \mathbf{u}^T (t_1) \mathbf{u} (t_2) \right\} = \\
&= \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \sigma(f, r) \exp\left(j2\pi f (t_1 - t_2)\right) \times \\
&\times N(2\pi f) \tilde{G}(2\pi f, \nu t_1, \nu r) \tilde{G}^\ast(2\pi f, \nu t_2, \nu r) \times \\
&\times \exp\left[-j2\pi f c^{-1}\left[ \mathbf{R}(t_1, \cdot) - \mathbf{R}(t_2, \cdot) \right] + \\
&\quad + \mathbf{R}_r(t_1, ...) - \mathbf{R}_r(t_2, ...) \right] \times \\
&\quad \times \mathbf{d}r_r \mathbf{d}r_r \mathbf{d}f \mathbf{d}f \\
&+ \mathbf{I}^N \int_{F_{\text{min}}}^{F_{\text{max}}} N_n(2\pi f) \exp\left(j2\pi f (t_1 - t_2)\right) \mathbf{d}f, \quad (13)
\end{align*}
\]

where \( \mathbf{I}^N \) is the unit dimension matrix \( N \times N \);

\( N(2\pi f) \) and \( N_n(2\pi f) \) are the spectral power densities of the signal and noise (we assume that the spectral power densities of the noise are the same in value in all channels, which is determined by the thermodynamic temperature of the input paths);

\( \sigma(f, r) \) is the frequency-dependent specific effective scattering area of the underlying surface.

Function \( \sigma(f, r) \) act as a radar image both at each of the individual frequencies, and in integral form at all frequencies.

Expression (13) is a mathematical model of the relationship between the statistical characteristics of the observation and the radar image.

We can find the signal processing algorithm by solving the likelihood equation

\[
\frac{\delta}{\delta \sigma(r')} \ln \mathbf{p}(\mathbf{u}(t) | \sigma(f, r)) = \frac{\delta}{\delta \sigma(r')} \ln k(\sigma(f, r)) - \\
- \frac{\delta}{\delta \sigma(r')} \left\{ \frac{1}{2} \int \left[ \mathbf{u}(t_1) - \langle \mathbf{u}(t_1) \rangle \right]^T \times \\
\mathbf{W}(t_1, t_2, \sigma(f, r)) \left[ \mathbf{u}(t_2) - \langle \mathbf{u}(t_2) \rangle \right] \mathbf{d}t_1 \mathbf{d}t_2 \right\} = 0, \quad (14)
\]

where \( k(\sigma(f, r)) \) is the function that depends on the radar image; \( \langle \rangle \) are the statistical averaging brackets;

\( \mathbf{W}(t_1, t_2, \sigma(f, r)) \) is a matrix that is inverse to the matrix of correlation functions, and which is found from the solution of an integral matrix equation

\[
\mathbf{T}(\tau) = \sum_{i, l = 1}^{N} \int_{F_{\text{min}}}^{F_{\text{max}}} \mathcal{F}\left\{ u_i(t, r) \right\} \mathcal{F}^{-1}\left\{ u_i(t, r) \right\} \mathbf{d}f \quad (16)
\]

where \( \mathcal{F}\left\{ \right\} \) and \( \mathcal{F}^{-1}\left\{ \right\} \) are the operators of the direct and inverse Fourier transform.

Similarly to (15) in (16), signal decorrelating filters are not applied. The initial effect of the radar that implements (16) will be a function proportional to the radio brightness temperature of the underlying surface.
Fig. 2. Radiocomplex Block Diagram
Radiocomplex Block Diagram

According to the formulated task, we will design a structural diagram of the radio complex, which will provide the formation of a radar image within ±50° of the nadir. For this, it is necessary to implement signal processing algorithms (6) and (15).

In fig. 2 it is shown a block diagram of the complex for the formation of radar images.

In fig. 2, the following designations have been introduced: $A_{i\text{MSAR}}$ is the modified SAR transmitter antenna; $A_{i\text{AASR}}$ is the antenna of the AASR transmitter; $A_i$ (i = 1..N) are the receiving elements of the antenna system (all are involved in the operation of the AASR, and some, along the direction of movement of the aircraft, are involved in the work of the MSAR); ID is the block of initial data for the formation of the probing signal; $T_{i\text{MSAR}}$ and $T_{i\text{AASR}}$ are the transmitters of MSAR and AASR systems; $I_{i\text P}$ is the receiver input path with frequency filtering; WLNA are the wideband low-noise amplifiers; Mix is the mixer with a filter for suppressing higher harmonics; $F_{i\text I}$ is the block for filtering (separating) signals by frequencies; $f_{\text{MSAR}}$ and $f_{\text{AASR}}$ are the operating frequencies of MSAR and AASR systems for radiated signals; designations $F_{i\text{MSAR}}$ and $F_{i\text{AASR}}$ indicate that the signals of the MSAR and AASR systems in the i-th receive channel are transferred to intermediate frequencies; R is the block for dividing signals by range to sections of the underlying surface (uses a priori information about the flight altitude h from the H block); $V_{i\text{F-FR}}$ is the block of the Volosyuk-Fourier-Fresnel transformation; PD is the a priori data block; G is the heterodyne; RSG is the reference signal generator for MSAR; ID is the block of initial data; SM is the signal multiplication unit (one input is direct, and the other is complex conjugate, which is indicated by the corresponding sign *); $E_{i\text J}$ is the block of multiplication by a complex exponent, the argument of which is proportional to the distance between the i-th and 1-th antennas; $I_\Sigma$ is the integrator; $\Sigma$ is the adder; $W_m$ is the signal decorrelation unit; PhD is the phase detector; CSM is the complex signal multiplier; AS is the block for calculating the square of the module; $\hat{I}_{i\text{AASR}}$ is the assessment of the radar image at the AASR output; $\hat{I}_{i\text{AASR}}$ is the evaluation of the radar image at the output of the signal processing channel from the i-th and 1-th antennas for the k-th (k = 1..K) range in the AASR; $\hat{I}_{i\text{AASR}}$ is the estimation of the radar image for the k-th range in the AASR; $I_{i\text{MSAR}}$ is the evaluation of the radar image at the MSAR output; $I_{i\text{MSAR}}$ is the estimation of the radar image for the m-th (m = 1..M) range in MSAR; $C_{i\text{k}}$ is the formation channel for $\hat{I}_{i\text{AASR}}$; $C_{i\text{MSAR}}$ is the formation channel for $\hat{I}_{i\text{MSAR}}$; $C$ is the computer; Mem is the memory for storing images; Disp is the display for displaying the image.

The complex works as follows. The transmitters $T_{i\text{MSAR}}$ and $T_{i\text{AASR}}$ generate and radiate signals through the antennas $A_{i\text{MSAR}}$ and $A_{i\text{AASR}}$ towards the underlying surface. The signals reflected by the surface are fed to the elements $A_i$ of the receiving antenna system. All N elements of the receiving antenna system are involved in the operation of the AASR, and some of these elements (located along the direction of movement of the aircraft) are involved in the operation of the MSAR. Next, let’s look at how AASR works. The signals from the outputs of the antennas go to the blocks $I_{i\text P}$, where they are amplified in WLNA, transferred to the intermediate frequency in the mixers Mix, which include filters for suppressing the higher components in the signal spectrum, then the signals go to the signal separation unit by frequency $F_{i\text I}$ and then to the range separation units R. Observation from the given range sections is fed to the $C_{i\text k}$ blocks, where the Volosyuk-Fourier-Fresnel transformation, multiplication of complex conjugate processes and compensation of the observation phase difference in $E_{i\text J}$, which occurs due to the different paths of the reflected signals to the i-th and j-th antennas, are performed. After integration, one sample of the image $\hat{I}_{i\text{AASR}}$ is formed.

Combining the outputs of the blocks $C_{i\text k}$ for all possible pairs of antennas provides the formation of a radar image at the output of the block $C_{i\text k}$ of the $\hat{I}_{i\text{AASR}}$ for k-th (k = 1..K) section of the range. After combining in the adder all the range elements into a line of the image of the observation area within ±20° from the nadir, it is fed to computer C and then stored in Mem and can be displayed on the Disp display.

Some AASR blocks (antennas, mixers, frequency separators) are common to MSAR as well. Further, the signals at frequencies $F_{i\text{MSAS}}$ and filtered by range are fed to the $C_{i\text{MSAR}}$ blocks for forming images in the zone ±(20°...50°) from the nadir. In these blocks, whitening of signals in filters $W_m$, phase detection in PhD blocks and correlation processing in CSM and I,
as well as the formation of a square of the image module in AS are performed. At the outputs of \( \Phi_{\text{MSAR}} \) \( m \) element of the radar image \( \hat{I}_{\text{MSAR}} \) is formed for the \( m \)-th section of the range. It should be noted that the number of ranges for MSAR and AASR may differ. Images of different ranges of range are combined into a line for evaluating the radar image \( \hat{I}_{\text{MSAR}} \) and fed to a computer.

It should be noted that the issues of the accuracy of image formation are often investigated in the problems of synthesizing radio complexes. These questions are beyond the scope of this article, but they can be solved both by using optimization methods, when the so-called potential values of accuracy are obtained [45], and directly from navigation [51-53] or other data [54-56], by indirect methods of image processing.

**Conclusions**

In this work, the structure of a radar complex for forming an image of the underlying surface in a swath of \( \pm 50^\circ \) from the direction to the nadir is synthesized for the first time. The peculiarity of the complex is as follows:

- the presence of two modes of operation - MSAR and AASR, which respectively provide survey in the range of angles \( \pm 50^\circ \) and \( \pm 20^\circ \) from the nadir, and their combined use gives a continuous image in the swath of \( \pm 50^\circ \) from the nadir;
- the use of ultra-wideband probing signals and, accordingly, the same channels of the signal processing receiver, allows obtaining high image resolution in spatial coordinates.

It is assumed that this complex can be used from aircraft or space carriers.

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ФОРМУВАННЯ ЗОБРАЖЕНЬ З РСА ТА РАС

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Предметом вивчення в статті є алгоритми радіомоніторингу Землі у широкій зоні огляду з аерокосмічного транспорту. Метою є розробка структурної схеми радіокомплексу, який може працювати одночасно у двох режимах: модифікованого синтезування апerture антен (РСА) і апertureного синтезу (РАС), відповідно до алгоритмів, синтезованих методом максимальної правдоподібності. Режим модифікованого РСА дозволяє отримати радіозображення високого розрізнення у діапазоні кутів спостереження ±(20°…50°) від напрямку в надир. Використовується метод поєднання модифікованого алгоритму РСА, який відрізняється від класичного алгоритму формування зображення можливістю отримання більш високої просторової роздільної здатності, платою за це є ускладнення алгоритму обробки сигналів, пов’язане з реалізацією декорелюючих фільтрів, які розширюють спектр прийнятих сигналів в кожному прямому тракті, та режиму РАС, який дозволяє формувати зображення за допомогою принципів пасивної або активної радіолокації. Пасивний режим РАС передбачає побудову зображення у діапазоні кутів спостереження ±20° від надир за результатами оброблення сигналів власного широкосмугового радіотелетрансмісійного каналу, а активний – у цьому ж діапазоні кутів спостереження але з використанням широкосмугового шумового сигналу підвішувача. Важливим результатом у формуванні радіозображень у зазначеній зоні огляду при використанні активного режиму роботи РАС є те, що зображення є близькими за фізичним змістом, а саме пропорційні питомій ефективній поверхні відбиття підстильної поверхні. Крім того, відмінною рисою синтезованих алгоритмів є використання широкосмугових зондуючих сигналів і, відповідно, таких же вхідних трактів приймачів, що дозволяє підвищити відношення сигнал/шум вихідного ефекту. Висновки. Наукова новизна отриманих результатів полягає в наступному: було розроблено структурну схему радіокомплексу на основі алгоритмів, синтезованих методом максимальної правдоподібності. Для формування радіозображення в радіокомплексі реалізовано поєднання РСА і РАС (з двома режимами роботи). Дана реалізація має важливе значення, адже дозволяє отримувати зображення високої розрізнівальної здатності у діапазоні кутів спостереження ±50°.
отражения подстилающей поверхности. Кроме того, отличительной чертой синтезированных алгоритмов формирования радиоизображений в этом же диапазоне углов наблюдения, но с использованием широкополосного шумового сигнала подсветки. Важным результатом в формировании радиоизображения в указанной зоне обзора при использовании активного режима работы РАС является то, что изображения близки по физическому содержанию, а именно пропорциональны удельной эффективной поверхности отражения подстилающей поверхности. Кроме того, отличительной чертой синтезированных алгоритмов является использование широкополосных зондирующих сигналов и, соответственно, таких же входных трактов приемников, что позволяет повысить отношение сигнал / шум выходного эффекта. Выводы.

Активный режим РАС реализован на основе алгоритма с дробной решеткой, реализованный методом максимального правдоподобия. Для формирования радиоизображения в радиокомплексе реализовано сочетание РСА и РАС (с двумя режимами работы). Данная реализация имеет важное значение, поскольку позволяет получать изображения высокой разрешающей способности в диапазоне углов наблюдения ±50° от направления в надир. Комплекс целесообразно размещать на самолетах, вертолетах и космических аппаратах (желательно тех, которые движутся по низким орбитам).

**Ключевые слова:** формирование радиоизображений; радар с апертурой антенны; алгоритм обработки сигнала; обработка сверхширокополосного сигнала.
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