Crater production on Titan and surface chronology

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\textbf{ABSTRACT}

\textbf{Context.} Impact crater counts on the Saturnian satellites are a key element for estimating their surface ages and placing constraints on their impactor population. The \textit{Cassini} mission radar observations allowed crater counts to be made on the surface of Titan, revealing an unexpected scarcity of impact craters that show high levels of degradation.

\textbf{Aims.} Following previous studies on impact cratering rates on the Saturnian satellites, we modeled the cratering process on Titan to constrain its surface chronology and to assess the role of centaur objects as its main impactors.

\textbf{Methods.} A theoretical model previously developed was used to calculate the crater production on Titan, considering the centaur objects as the main impactors and including two different slopes for the size-frequency distribution (SFD) of the smaller members of their source population. A simple model for the atmospheric shielding effects is considered within the cratering process and our results are then compared with other synthetic crater distributions and updated observational crater counts. This comparison is then used to compute Titan’s crater retention age for each crater diameter.

\textbf{Results.} The cumulative crater distribution produced by the SFD with a differential index of $\beta_2 = 3.5$ is found to consistently predict large craters ($D > 50$ km) on the surface of Titan, while it overestimates the number of smaller craters. As both the modeled and observed distributions flatten for craters with $D \leq 25$ km due to atmospheric shielding, the difference between the theoretical results and the observational crater counts can be considered as a proxy for the scale to which erosion processes have acted on the surface of Titan throughout the Solar System age. Our results for the surface chronology of Titan indicate that craters with $D > 50$ km can prevail over the Solar System age, whereas smaller craters may be completely obliterated due to erosion processes acting globally.

\textbf{Key words.} Kuiper belt: general – planets and satellites: individual: Titan – planets and satellites: surfaces

1. Introduction

Titan is the largest Saturnian satellite and the only satellite in the Solar System known to possess a dense atmosphere. It was discovered in 1655 by Christiaan Huygens, but most of its surface features remained veiled until 350 years later, when the \textit{Cassini-Huygens} mission began its exploration of the Saturn system. Between 2004 and 2017 \textit{Cassini} performed 127 close encounters with Titan, collecting data that revealed a complex world with liquid lakes, seas, and an active hydrologic cycle based on methane. In addition, in 2005 the \textit{Huygens} probe completed the first landing on a satellite other than our Moon, providing in situ measurements, such as a detailed profile of the atmosphere of Titan. Before \textit{Cassini} the crater size distribution of Titan was unknown but estimated to be similar to those of the other Saturnian satellites (Lopes et al. 2019). Instead, the mission’s observations uncovered an unexpectedly low number of eroded craters, indicating that geologic processes modify its surface (e.g., Hedgepeth et al. 2020). Following our previous studies on the mid-sized and small Saturnian satellites (Di Sisto & Brunini 2011; Di Sisto & Zanardi 2013; Rossignoli et al. 2019), in this work we present a study of the crater production on Titan generated by centaur objects. In Sect. 2, we describe the surface of Titan and the current knowledge of its crater population. In Sect. 3, we present the method used to predict the crater distribution on Titan and its surface chronology. In Sect. 4, we present our results based on the comparison with the updated crater counts and the surface age of Titan for each crater diameter. In Sect. 5, we present our conclusions.

2. The surface of Titan

2.1. Geological units

Before the \textit{Cassini-Huygens} mission, the composition and features of the surface of Titan were mostly unknown (Lopes et al. 2019), although the presence of lakes or seas of liquid hydrocarbons had already been proposed based on radar observations from Arecibo (Lorenz & Lunine 2005). Thus, the observations made by \textit{Cassini} over more than 13 years provided the first up-close and in-depth study of the satellite. The main instruments to observe the surface of Titan were the Radio Detection and Ranging (RADAR) instrument (Elachi et al. 2004), whose primary goal was to pierce through Titan’s atmosphere and reveal its surface; the Visual and Infrared Mapping Spectrometer (VIMS) (Brown et al. 2004); and the Imaging Science System (ISS) (Porco et al. 2004). Data collected by these instruments helped build a global topographic map of Titan and constrain its surface properties and composition. Titan was revealed to have one of the most diverse and dynamic surfaces of the Solar System, largely altered by erosional and depositional processes that show a latitude variation (Lopes et al. 2020; Hedgepeth et al. 2020). In addition, six major geological units were identified.
(Lopes et al. 2020): plains, which cover 65% of the global area and dominate the mid-latitudes; dunes, which represent 17% of the total surface and dominate the equatorial regions (+30 latitude); hummocky terrains composed of mountain chains and isolated peaks that comprise 14% of the global area; lakes (dry or liquid-filled), which represent 1.5% of Titan’s total surface area and are located near the poles; labyrinths, which have morphologies similar to karstic terrain and cover 1.5% of Titan’s total surface area; and craters, which occupy only 0.4% of Titan’s global area and are almost completely absent near the poles. Based on the location and superposition between these units, Lopes et al. (2019) were able to determine their relative ages. They conclude that the oldest units are the hummocky terrains, while the dunes and lakes are the youngest. Regarding Titan’s surface composition, hummocky and crater units show lower emissivity in radiometric data consistent with water-ice materials, while plains, dunes, lakes, and labyrinths show high emissivity to radar, indicating the presence of organic materials (Lopes et al. 2020).

### 2.2. Cratering counts on Titan

At the end of the Cassini mission, ~69% of the surface of Titan was mapped by synthetic aperture radar (SAR) (Hedgepeth et al. 2020). In this operating mode, radar images with spatial resolutions as low as 350 m were obtained (Lopes et al. 2019). Overall, the surface of Titan was revealed to present a scarcity of impact craters that is consistent with a heavily eroded surface (Neish & Lorenz 2012). In addition, many of the identified craters show evidence for extensive modification by erosive processes, such as fluvial erosion and aeolian infill (e.g., Neish et al. 2013; Lopes et al. 2019). For example, craters Sinlap and Soi are both ~80 km in diameter, but show different degradation states (Fig. 1). Sinlap appears to be young while Soi is extremely degraded. Neish et al. (2013) and Hedgepeth et al. (2020) studied the crater topography on Titan and compared it to that of Ganymede, a similarly sized, airless satellite with relatively pristine craters. Their results show that Titan’s crater depths are 51% shallower than the craters on Ganymede, which is consistent with aeolian infill as the main process of crater modification. Depending on the infilling rate, Neish et al. (2013) state that many craters on Titan may remain undetected due to rapid aeolian infilling.

Hedgepeth et al. (2020) reassessed the crater population obtained in previous studies (e.g., Wood et al. 2010; Neish & Lorenz 2012) using the entire SAR data set, which allowed them to identify 30 additional craters. In total, only 90 certain to possible impact craters have been identified on Titan (Fig. 2), and they are not homogeneously distributed over the surface (Hedgepeth et al. 2020). In fact, 65% of them are found within 30° of the equator in a region dominated by dunes (Neish et al. 2013). The polar regions show a relative scarcity of craters, possibly due to the concentration of liquid lakes near the poles or the enhanced fluvial activity in the higher latitudes, which could erode craters beyond recognition (Neish et al. 2016; Hedgepeth et al. 2020).

### 3. Method

In order to calculate the theoretical production of craters by centaur objects on Titan, we apply a modified version of a method previously developed in a series of works (Di Sisto & Brunini 2011; Di Sisto & Zanardi 2016; Rossignoli et al. 2019). In addition, for this work we consider the results from our latest numerical simulation of the dynamical evolution of transneptunian objects that become centaurs (Di Sisto & Rossignoli 2020). In this section we describe briefly the theoretical cratering method, together with the changes introduced by the new simulation and the considerations of the atmospheric effects on the impactors.

#### 3.1. The impactor population

Following previous works on impact cratering models on the mid-sized and small Saturnian satellites (Di Sisto & Zanardi 2013; Rossignoli et al. 2019), we consider the main impactors to be the centaur objects. These heliocentric bodies have their origin in the transneptunian region and exhibit a transient nature due to perturbations exerted on their orbits by the giant planets. It has been shown that the scattered disk (SD) in the transneptunian region is the subpopulation with the highest probability of having encounters with Neptune and therefore of evolving toward the planetary region of the Solar System (Duncan & Levison 1997; Di Sisto & Brunini 2007). The scattered disk objects (SDOs) that enter the giant planetary zone become centaurs and are likely to collide and produce craters on planets or their satellites. In addition, those centaur objects that enter the zone interior to Jupiter’s orbit become Jupiter-family comets (JFCs). In that region physical effects such as sublimation and splitting have important implications on the evolution of JFCs. Di Sisto et al. (2009) showed that the mean physical lifetime of the JFCs is very short, on the order of a few thousand years, and that those that do survive disintegration and reenter the centaur zone, pass through it very quickly. Thus, impacts by JFCs on the planets and their satellites can be considered negligible in relation to impacts by centaurs from the SD.

In previous papers (Di Sisto & Brunini 2011; Di Sisto & Zanardi 2013, 2016; Rossignoli et al. 2019) the impact cratering rates on the mid-sized and small Saturnian satellites were modeled based on a numerical simulation of the dynamical evolution of SDOs by Di Sisto & Brunini (2007). In that work the authors built an intrinsic model of the SD based on the available observations at that time, and studied the contribution of the SDOs to the centaur population, which represents the main impactor population in our model. Since 2007, many more SDOs were discovered, which motivated a revision and an update of the model. Thus, Di Sisto & Rossignoli (2020) built the intrinsic SD model again including new observations, which were six times as large as the 2007 sample, and analyzed the general dynamical evolution to the centaur zone, obtaining similar results as in Di Sisto & Brunini (2007).

Di Sisto & Rossignoli (2020) also considered new estimations of the number and size distribution of SDOs, which have been improved by the recent discoveries by the Outer Solar System Origins Survey (OSSOS) (Bannister et al. 2018). It has been
argued that the size-frequency distribution (SFD) of transneptunian objects and in particular of SDOs is not accurately modeled by a single power law, but instead presents a break at a diameter \( d \approx 60–100 \) km, varying from a steep slope at greater diameters to a shallow slope for smaller objects (e.g., Bernstein et al. 2004; Fraser & Kavelaars 2009). Based on the analysis of the observations of small SDOs and centaurs by OSSOS, Lawler et al. (2018) found that a break in the SFD is required at \( d \approx 100 \) km (from larger to smaller SDOs). They found a faint end slope of the absolute magnitude \( H \) size distribution of \( \alpha_H = 0.4–0.5 \), which corresponds to a differential size distribution index of \( s = 3–3.5 \). At the bright end, Elliot et al. (2005) obtained a differential size distribution index for SDOs of \( s = 4.7 \). On the other hand, Parker & Kavelaars (2010a,b) determined the maximum total number of SDOs with diameters larger than 100 km to be \( N(d > 100 \text{ km}) = 3.5 \times 10^5 \). Therefore, scaling the SDO population to this number and considering the SFD of SDOs with a break at \( d = 100 \) km and the size indexes mentioned above, we model the cumulative size distribution (CSD) of the impactor population:

\[
N(d > d) = N_0 \left( \frac{1 \text{ km}}{d} \right)^{s_2-1} \quad \text{for} \quad d \leq 100 \text{ km},
\]

\[
N(d > d) = 3.5 \times 10^5 \left( \frac{100 \text{ km}}{d} \right)^{s_2-1} \quad \text{for} \quad d > 100 \text{ km}.
\]

Here \( N_0 = 3.5 \times 10^5 \times 100^{s_2-1} \) by continuity at \( d = 100 \) km, \( s_1 = 4.7 \), and \( s_2 \) is modeled with two limiting values: \( s_2 = 3 \) and 3.5, due to the uncertainty on the faint end slope of the \( H \) distribution.

An additional break to a shallower impactor SFD at diameters of \( d \approx 1–2 \) km has been modeled in recent works, based on the cratering records on Pluto and Charon (e.g., Robbins et al. 2017; Singer et al. 2019). In addition, Morbidelli et al. (2021) modeled a crater production function for Pluto, Charon, Nix, and Arrokoth based on their crater records and proposed a cumulative power law slope for the Kuiper belt objects SFD given by \( N(d > d) \propto d^{q_{KBO}} \) of \(-1.2 < q_{KBO} < -1\) in the \( 0.03 \lesssim d \leq 2 \) km range. However, in the Saturnian satellites the possibility of a break in the impactor SFD at \( d \approx 1–2 \) km is not clear yet, given that there may be other impactor populations at play such as planetocentric objects that prevent a direct association between the observed crater distributions and the impactor population. For this reason, we have not considered this possibility in our model.

### 3.2. The impact process

In order to model the impact crater size-frequency distribution on the surface of Titan produced by centaur objects, we follow the method described in our previous papers (e.g., Di Sisto & Brunini 2011; Rossignoli et al. 2019). The number of collisions on the satellite is computed from the results of the simulation presented in Di Sisto & Rossignoli (2020), which provides the number of encounters of SDOs with Saturn. In order to relate the number of encounters within the Hill’s sphere of the planet to the...
number of collisions on the satellite, we consider a particle-in-a-box approximation that leads to the equation

\[ N_c (> d) = \frac{v_i R_G^2}{v(R_i) R_i^2} 8.5 N (> d), \]  

(2)

where \( v_i \) is the relative collision velocity on Titan, \( v(R_i) \) is the centaurs’ mean relative encounter velocity when they enter Saturn’s Hill sphere (of radius \( R_i \)), and \( R_G \) is the satellite’s collision radius given by \( R_G = R_T (1 + (v_{\text{enc}}/v(R_i))^2)^{1/2} \) to account for the gravitational focusing effect. The factor 8.5 represents the number of encounters with Titan relative to the initial number of particles in the numerical simulation. This number is somewhat smaller than in our previous studies (e.g., Rossignoli et al. 2019), but statistically more representative due to the larger sample of observed objects used to build the SD model. The relative encounter velocity \( v(R_i) \) is obtained from the simulation encounter files (Di Sisto & Rossignoli 2020). For an airless Titan and assuming isotropic impacts, \( v_i = \sqrt{v_T^2 + v_G^2} \), where \( v_T \) is Titan’s orbital velocity and \( v_G \) is the centaurs mean relative velocity when they cross the orbit of Titan. The values of these velocities are listed in Table 1, together with physical data of Titan. However, it should be noted that the atmosphere of Titan reduces both the relative velocity and the diameter of all impactors as they approach the surface. Thus, in the next section we describe how we include these effects in our cratering model.

### 3.3. Atmospheric effects

Before the Cassini-Huygens mission, the general composition of Titan’s atmosphere was known but poorly constrained. The data collected by the Huygens Atmospheric Structure Instrument (HASI) allowed the determination of the temperature and density profiles from an altitude of 1,400 km down to the surface (Fulchignoni et al. 2005). From this data, it was determined that the atmospheric density at the surface of Titan is approximately four times that of the Earth (Neish & Lorenz 2012). Melosh (1989) studied the minimum diameter projectile that can penetrate Titan’s atmosphere at vertical incidence and computed a value for an ice impactor of \( d \approx 1.7 \) km. Lorenz (1997) studied the likely characteristics of the impact crater distribution on Titan before Cassini and predicted that only impactors with \( d > 120 \) m would be able to penetrate Titan’s atmosphere with a significant portion of its incident velocity. Artemieva & Lunine (2003) found that the atmosphere shielded the surface from impactors smaller than 1 km and would even decelerate larger objects. Thus, in order to constrain the amount of atmospheric shielding against impactors, we consider a simple model where the effects of fragmentation, pancaking, deceleration, and ablation are included.

As the impactor traverses the atmosphere, the differential atmospheric pressure exerted on the object may lead to its fragmentation once its characteristic strength is exceeded (Chyba et al. 1993). In this case the fragments of the disrupted object expand away from each other (the impactor “pancakes”), but may continue to be treated as a single collective bow shock until the dissociating impactor has expanded to twice its initial radius (Chyba et al. 1993; Hills & Goda 1993; Engel et al. 1995), a point at which the fragments separate into individual bow shocks.

We model the deceleration of the impactor via the conventional drag equation (Engel et al. 1995)

\[ m \ddot{v} = -\frac{1}{2} C_D \rho(z) v^2 + g_T m \sin \alpha, \]  

(3)

where \( m, v, \) and \( A = \pi d^2/4 \) are the impactor’s mass, relative velocity, and cross section, respectively; \( g_T \) is Titan’s surface gravity (see Table 1); \( \rho(z) \) is the atmospheric gas density (Fulchignoni et al. 2005); and \( C_D = 0.64 \) is the non-dimensional drag coefficient presented in Korycansky & Zahnle (2005) (hereafter KZ05). We consider the most probable impact angle to be \( \alpha = 45^\circ \) with respect to the horizon. For the ablation effect, which causes continuous shedding of the impactor mass as it traverses the atmosphere, the mass variation is given by (KZ05)

\[ \dot{m} = -C_A \rho(z) A v, \]  

(4)

where \( C_A = 0.71 \) is the non-dimensional ablation coefficient for Titan (KZ05) and the impactors are modeled following KZ05 as cylinders of constant density \( \rho_i \) and length \( h = 4m/(\pi \rho_i d^2) \). In the present work we consider the impactors to be made of ice, thus \( \rho_i = 1 \) g cm\(^{-3} \).

In order to include the pancaking effect in the model, the \( W \) term is introduced, resulting in the expression for the variation of the impactor diameter (KZ05)

\[ \ddot{d} = \frac{2 \dot{m}}{\pi \rho_i d^2} + W, \]  

(5)

and

\[ W = C_p \rho(z) v^2, \]  

(6)

where \( C_p = 0.75 \) is the pancaking coefficient as modeled in KZ05. Based on the models presented in Hills & Goda (1993), Chyba et al. (1993), and Artemieva & Lunine (2003) we follow the motion of the impactors through Titan’s atmosphere considering that its effects are negligible for altitudes higher than \( \sim 200 \) km from the surface. As the impactors traverse through the atmosphere they suffer ablation and deceleration. Disruption occurs when the impactor’s strength \( S = 1 \times 10^7 \) dyn cm\(^{-2} \) is exceeded by the atmospheric pressure \( P = \rho(z) v^2 \) (Hills & Goda 1993). The resulting fragments spread while they continue their journey through the atmosphere until they reach the surface or the expanded impactor grows to twice its initial radius. In the latter case we consider a simple approach where the expanded impactor separates into two fragments and the mass of the initial impactor is divided randomly between the two fragments. The transverse dispersion velocity imparted is given by \( v_i = (0.41 \rho(z)/\rho_i)^{1/3} v \) (KZ05) and each fragment can experience successive fragmentations.
3.4. Crater scaling law

In order to obtain the number of collisions on Titan as a function of the impactor diameter, we substitute Eq. (1) into Eq. (2). Considering that impactors are decelerated and ablated and may even be fragmented as they traverse the atmosphere of Titan (see Sect. 3.3), they will reach the satellite’s surface with a reduced size and velocity. Thus, the following step is to relate the final impactor diameter \( d_i \) to the crater diameter it produces. This relation is modeled by the scaling law from Holsapple & Housen (2007), which gives the transient diameter \( D_t \) of a crater generated by an impactor of diameter \( d_i \) through the equation

\[
D_t = K_1 \left[ \frac{g_T d_i}{2 v_i^2} \right]^{\frac{2}{3}} \left[ \frac{\rho_i}{\rho} \right]^{\frac{1}{3}} + K_2 \left( \frac{Y}{\rho_i v_i^2} \right)^{\frac{2}{3}} \left[ \frac{\rho_i}{\rho} \right]^{\frac{11/32}{2}} d_i, \tag{7}
\]

where \( \rho_i = 1 \text{ g cm}^{-3} \) is the impactor density and \( d_i \) and \( v_i \) are respectively its final diameter and vertical velocity when it reaches the surface of Titan. The parameters \( \rho_i, \mu, v, Y, K_1, \) and \( K_2 \) depend on the target material and \( g_T \) is Titan’s surface gravity (see Table 1). Since the outermost layer (∼ 100 km thick) of the surface of Titan is mostly composed of cold water ice (Neish et al. 2013), we consider \( \rho_i = 1 \text{ g cm}^{-3} \) and continue to use the values for the icy Saturnian satellites of \( \mu = 0.38, v = 0.397, \) and \( K_1 = 1.67, \) while we select the value of \( K_2 = 0.8. \) For the target strength \( Y \) we adopt the most commonly used value of the tensile strength of polycrystalline water ice: \( Y = 1 \times 10^7 \text{ dyn cm}^{-2} \) (Manga & Wang 2007).

With Eq. (7) one can determine from the dominant term if, during the impact crater formation, the crater growth is limited by the target’s gravity (first term) or its strength (second term). In the case of Titan, all craters are formed under the gravity regime. The initial compression and excavation stages of the impact process define the transient crater diameter \( D_t \) given by Eq. (7). Then, a final stage of gravity-driven crater collapse takes place and expands the crater to its final diameter (e.g., Collins et al. 2012). This last stage will affect crater sizes differently depending on the initial impact energy, which separates craters into two morphological categories. Smaller and bowl-shaped craters are called simple craters and tend to present a depth to diameter ratio near 1:5 (Melosh & Ivanov 1999). In contrast, above a certain crater diameter a complex crater is formed, with a central peak or morphology categories. Smaller and bowl-shaped craters are called simple craters and tend to present a depth to diameter ratio near 1:5 (Melosh & Ivanov 1999). In contrast, above a certain crater diameter a complex crater is formed, with a central peak or ring, a relatively flat floor, and a smaller depth-to-diameter ratio. The transition from a simple to a complex morphology occurs at a distinct crater size given by (Kraus et al. 2011)

\[
D^* = \frac{g_{\text{gan}}}{g_T} 2 R_{\text{gan}}^*, \tag{8}
\]

where \( g_{\text{gan}} = 1.43 \text{ m s}^{-2} \) is the surface gravity of Ganymede and \( R_{\text{gan}} = 1 \text{ km} \) is the radius of the transit crater between simple and complex craters for Ganymede (Schenk 2002). Thus, following the relation presented in Kraus et al. (2011), the final crater diameter can be obtained with

\[
D = (1.3 \, k) D_t \quad \text{for} \quad D_t \leq D^*/1.3 \, k, \tag{9}
\]

\[
D = D_t (1.3 \, k)^{1/1-\eta} \left( \frac{D_t}{D^*} \right)^{\eta/(1-\eta)} \quad \text{for} \quad D_t > D^*/1.3 \, k, \tag{9}
\]

where \( k = 1.19 \) and \( \eta = 0.04 \) (Kraus et al. 2011).
which is the maximum crater retention age for each crater diameter according to our model:

$$\tau(> D) = t_f \left( 1 - e^{-\frac{N_0(> D)}{N_{e}(> D)}} \right).$$  \hspace{1cm} (11)

Here $t_f = 4.5$ Gyr is the age of the Solar System, $N_0(> D)$ is the satellite’s cumulative number of observed craters for each crater diameter, and $N_{e}(> D)$ is our model’s cumulative number of craters for each crater diameter.

4. Results

In the previous section we describe the method used to calculate the crater distribution on Titan produced by centaur objects over the age of the Solar System. Given the uncertainty in the size distribution for the smaller objects of the centaur source population, in this section we present our results for two limiting values of the $s_2$ index in Eq. (1), $s_2 = 3$ and $s_2 = 3.5$. Our predicted crater distributions are compared with the updated observational crater counts presented in Hedgepeth et al. (2020), where the authors reassessed Titan’s crater population using the entire Cassini SAR data set.

In Fig. 4, we present our predicted cumulative number of craters per square kilometer for both $s_2$ values, together with the observed crater counts from Hedgepeth et al. (2020). Specific results for the case of the largest impactor for both $s_2$ indexes of the impactor size distribution are shown in Table 2.

Table 2. Model results.

| $D^*$ | $v_f$ | $d_m$ | $D_m$ | $v_f$ | $d_m$ | $D_m$ |
|-------|-------|-------|-------|-------|-------|-------|
| 2.11  | 7.29  | 12.88 | 158.12| 7.31  | 19.4  | 227.01|

Notes. Transition crater diameter $D^*$ between simple and complex craters (in kilometers); final vertical collision velocity $v_f$ of the largest impactor (in kilometers per second); largest impactor diameter $d_m$ and largest crater diameter $D_m$ on Titan for both $s_2$ indexes (in kilometers).

As can be seen in Fig. 4, the results obtained with the $s_2 = 3.5$ index present the most similar fit to the observations for craters with diameters $D > 50$ km, with the exception of crater Menrva ($D = 400$ km). A crater of this size may have formed early in the history of Titan, during the initial mass depletion that produced the late heavy bombardment of minor bodies on the planets, a process that is not included in our model. For craters with $D < 50$ km, our model with the $s_2 = 3.5$ index overestimates the number of craters even when a simple fragmentation model is considered. However, including impactor fragmentation and pancaking effects (solid lines in Fig. 4) produces a considerable change in our predicted crater distributions for those craters with $D \leq 25$ km, which flatten, emulating the same behavior as the observed distribution for that diameter range.

According to previous studies on Titan’s atmospheric shielding effects, disruption becomes minimal at crater diameters of $D \geq 20$ km (Neish & Lorenz 2012; Artemieva & Lunine 2003; Korycansky & Zahnle 2005). Considering that only well-preserved small craters may be observable on Titan due to erosion and resurfacing (Wood et al. 2010; Neish & Lorenz 2012), an overestimation of the number of small craters is to be expected. In particular, fluvial processes may be able to erode craters beyond recognition, explaining their scarcity in polar regions (Neish et al. 2016). Provided that several works consider the $s_2 = 3$ index for the size distribution of impactors with $d < 100$ km (e.g., Shoemaker & Wolfe 1982), we find that the crater distribution obtained via this index underestimates the number of craters for all craters with $D \geq 25$ km (see Fig. 4).

In addition, we also studied the crater distribution resulting from the index $s_2 = 2.5$ considered in previous works (Di Sisto & Brunini 2011; Di Sisto & Zanardi 2013; Rossignoli et al. 2019). However, for $s_2 = 2.5$ the results fall under the observed distribution for almost all crater diameters $D$, and thus have not been included in Fig. 4.

4.1. Comparison with other predicted crater distributions

In order to compare our results to other estimations of the crater size distribution on Titan we consider the models presented in Artemieva & Lunine (2005) (hereafter AL05) and KZ05. The results obtained in these papers were analyzed and compared thoroughly by Wood et al. (2010) and Neish & Lorenz (2012). The AL05 and KZ05 models both included impactor disruption by Titan’s atmospheric shielding and used the impact rate presented in Zahnle et al. (2003). However, KZ05 considered a constant impact rate, while AL05 assumed a 1/$t$ dependence over time. The linear fit in Fig. 3 allows us to obtain the current cratering rate on Titan and ease the comparison between our model and the cratering rates for Titan presented in Zahnle et al. (2003) and Dones et al. (2009). The slope value $C$ (see Sect. 3.5) enables us to compute, for the past 3 Gyr, a cratering rate on Titan for craters with diameters $D > 10$ km of $5.45 \times 10^{-15}$ yr$^{-1}$ for an airless Titan and of $1.28 \times 10^{-15}$ yr$^{-1}$ if the atmospheric effects are included, all considering $s_2 = 3.5$ in Eq. (1). The corresponding value presented in Table 4 of Zahnle et al. (2003) for the case A impactor population, where the size-number distribution of impactors is inferred from craters on the Galilean satellites, is $6 \times 10^{-15}$ craters per year with an uncertainty factor of four, while in Dones et al. (2009) the present-day cratering rate at an airless Titan for the case A impactor population is $3.4 \times 10^{-15}$ yr$^{-1}$. Another point of comparison between the KZ05 and AL05 models are their crater scaling laws, which was determined by Neish & Lorenz (2012) to be the aspect where the most important differences between the models resided. On the one hand, AL05 con-
Fig. 5. Our model’s predicted cumulative number of craters per square kilometer as a function of crater diameter for $s_2 = 3.5$, compared to results by AL05 and KZ05 for a 4.5 Gyr surface. The blue triangles correspond to crater counts from Hedgepeth (2020).

considered the projectile density $\rho_i = 1$ g cm$^{-3}$ (equivalent to our model), a scaling law for a water target, and $\eta/(1 - \eta) = 0.176$ for the scaling exponent in Eq. (9). On the other hand, in KZ05 the projectile density value was $\rho_i = 0.5$ g cm$^{-3}$, the scaling law was for a sand target, and $\eta/(1 - \eta) = 0.13$. Therefore, these models led to two distinct predicted crater distributions on Titan, with a difference that ranges between a factor of two for small craters to a factor of 30 for craters of 1000 km (Neish & Lorenz 2012). With respect to Titan’s surface chronology Lorenz et al. (2007), Wood et al. (2010), and Neish & Lorenz (2012) found that the results from KZ05 are most compatible with a crater retention age of 1 Gyr, while the crater distribution obtained in AL05 is more consistent with a crater retention age of 200 Myr (Wood et al. 2010; Neish & Lorenz 2012). Figure 5 shows our results for the $s_2 = 3.5$ index with the model that includes impactor fragmentation, together with the results from AL05 and KZ05 for a 4.5 Gyr surface. As can be seen, the crater distributions from AL05 and KZ05 considering craters formed over the entire Solar System age overestimate the number of craters. For this reason, Neish & Lorenz (2012) presented the predicted crater distributions from AL05 and KZ05 adopting a crater retention age of 200 Myr and 1 Gyr, respectively (Fig. 6). In contrast, our model with the $s_2 = 3.5$ index is able to predict consistently those craters with $D > 50$ km considering craters formed over the entire Solar System age and without requiring a complete global resurfacing between 200 Myr and 1 Gyr ago. Instead, our crater retention age calculation (Fig. 7) shows that craters with $D > 50$ km may be as old as the Solar System, which supports the idea that Titan could be a primordial object. On the other hand, craters with $D < 10$ km reach ages of $\sim 1$ Gyr. These results are similar to those presented in Lorenz & Lunine (1996), which suggest that craters with $D = 20$ km could be preserved for $\sim 2$ Gyr, while craters with $D > 50$ km are not likely to be eroded to the point of being undetectable.

Crater retention ages presented in Wood et al. (2010) and Neish & Lorenz (2012) are based on the best fit between the AL05 and KZ05 predicted crater distributions and the observed crater counts. This method leads to a global crater retention age that considers a normalized age for all crater sizes. Instead, our results for the surface age of different Saturnian satellites (Di Sisto & Zanardi 2016; Rossignoli et al. 2019) show that the crater retention age is dependent on the crater size. Larger craters tend to be preserved for a longer time and may be detectable even if affected by erosive processes, while smaller craters may be eroded beyond recognition on short timescales. On Titan smaller craters may be more easily obliterated by erosive processes such as fluvial modification (Forsberg-Taylor et al. 2004), although atmospheric shielding effects and the uncertainty in crater counts for $D < 20$ km prevent an accurate quantification of this type of erosion (Neish et al. 2016). Nevertheless, our results for Titan’s crater retention age show the same size dependence as was found for other Saturnian satellites (Di Sisto & Zanardi 2016; Rossignoli et al. 2019). Thus, our size-dependent chronology results may help to constrain different erosion processes acting on Titan’s surface.

4.2. Apex-antapex asymmetry

The hemispheric distribution of craters produced on a synchronously rotating satellite has been studied in several works (Shoemaker & Wolfe 1982; Horedt & Neukum 1984; Zahnle et al. 2001) and is predicted to be denser on the leading side of the satellite for the case of heliocentric impactors. However, for most of the satellites in the outer Solar System no distinct asymmetries have been found (e.g., Zahnle et al. 2001; Dones et al. 2009; Kirchoff & Schenk 2010). Wood et al. (2010) analyzed the hemispheric distribution of the 49 impact craters on Titan detected at the time and found that 63% of the total number of craters occurred on the leading side. This exact percent-
age is still valid when analyzing the updated set of 90 observed craters listed in Hedgepeth et al. (2020). Following the method presented in Shoemaker & Wolfe (1982) and using the velocities listed in Table 1, we find that the expected leading/trailing asymmetry depends strongly on the hemispheric difference in the modeled mean impact velocity at the top of the atmosphere. Our results using Shoemaker & Wolfe (1982) derivations give mean impact velocities at the leading and trailing sides of 13.9 km s\(^{-1}\) and 7.17 km s\(^{-1}\), respectively, which results in a cratering rate asymmetry of 15:1. In contrast, if we use the leading and trailing sides mean velocities modeled in KZ05, which are 10.9 km s\(^{-1}\) and 9.4 km s\(^{-1}\), respectively, we obtain a cratering rate asymmetry of 4.9:1, very similar to the 4:1 ratio obtained in KZ05 and the 5:1 ratio from Lorenz (1997). Nevertheless, even the lowest predicted cratering rate asymmetry is higher than the observed 1:7:1 ratio. This difference may be due to a number of factors. On the one hand, there is the possibility that Titan has not rotated synchronously for the age of the Solar System. On the other hand, a planetocentric population of impactors such as fragments from the putative breakup of the Hyperion parent body (Farinella et al. 1990) may have altered the expected hemispheric cratering asymmetry, as Titan may be able to accrete \(\approx 78\%\) of Hyperion’s ejecta (Dobrovolskis & Lissauer 2004).

5. Conclusions

In the present work we modeled the impact crater distribution on Titan generated by centaurs over the Solar System age. In order to obtain the number of impacts on Titan we used the results from an updated simulation of the dynamical evolution of SDOs and their contribution to the centaur population (Di Sisto & Rossignoli 2020). The CSD of SDOs was modeled considering the current uncertainties in the size distribution for the smaller objects of the impactor population. Thus, we considered two limiting values for the differential power law index for objects with \(d < 100\) km, \(x_1 = 3\) and \(x_2 = 3.5\), and our results for the impact crater distribution are presented for both of these values. In addition, we incorporated a simple model for the atmospheric shielding of impactors where the effects of fragmenta-

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