Vorticity production through rotation, shear, and baroclinicity

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\textbf{ABSTRACT}

\textbf{Context.} In the absence of rotation and shear, and under the assumption of constant temperature or specific entropy, purely potential forcing by localized expansion waves is known to produce irrotational flows that have no vorticity.

\textbf{Aims.} Here we study the production of vorticity under idealized conditions when there is rotation, shear, or baroclinicity, to address the problem of vorticity generation in the interstellar medium in a systematic fashion.

\textbf{Methods.} We use three-dimensional periodic box numerical simulations to investigate the various effects in isolation.

\textbf{Results.} We find that for slow rotation, vorticity production in an isothermal gas is small in the sense that the ratio of the root-mean-square values of vorticity and velocity is small compared with the wavenumber of the energy-carrying motions. For Coriolis numbers above a certain level, vorticity production saturates at a value where the aforementioned ratio becomes comparable with the wavenumber of the energy-carrying motions. Shear also raises the vorticity production, but no saturation is found. When the assumption of isothermality is dropped, there is significant vorticity production by the baroclinic term once the turbulence becomes supersonic. In galaxies, shear and rotation are estimated to be insufficient to produce significant amounts of vorticity, leaving therefore only the baroclinic term as the most favorable candidate. We also demonstrate vorticity production visually as a result of colliding shock fronts.

\textbf{Key words.} magnetohydrodynamics (MHD) – turbulence – Galaxies: magnetic fields – ISM: bubbles

1. Introduction

Turbulence in the interstellar medium (ISM) is believed to be driven by supernova explosions. Such events inject sufficient amounts of energy to sustain turbulence with rms velocities of $\sim 10\,\text{km/s}$ and correlation lengths of up to $100\,\text{pc}$ (Beck et al., 1996). Simulations of such events can be computationally quite demanding, because the bulk motions tend to be supersonic and the flows involve strong shocks in the vicinity of individual explosion sites, as was seen early on in two-dimensional simulations (Rosen & Bregman, 1995). Nevertheless, such simulations are able to reproduce a number of physical phenomena such as the observed volume fractions of hot, warm, and cold gas (Rosen et al., 1996, Korpi et al., 1999), the statistics of pressure fluctuations (Mac Low et al., 2003), the effects of the magnetic field (de Avillez & Breitschwerdt, 2005), and even dynamo action (Gressel et al., 2008, Gissinger et al., 2009, Hanasz et al., 2009). These simulations tend to show the development of significant amounts of vorticity, which is at first glance surprising. Indeed, each supernova drives the gas radially outward and can roughly be described by radial expansion waves. In such a description, turbulence is forced by the gradient of a potential that consists of a time-dependent spherical blob at random locations. Obviously, such a forcing is irrotational, so no vorticity is produced.

Earlier work of Mee & Brandenburg (2006) showed that under isothermal conditions only the viscous force can produce vorticity and that this becomes negligible in the limit of large Reynolds numbers or small viscosity. In principle, vorticity can also be amplified akin to the dynamo effect by the $\nabla \times (\mathbf{u} \times \mathbf{\omega})$ term, which is analogous to the induction term in dynamo theory, where $\omega$ plays the role of the magnetic field. However, neither this nor the viscosity effect were found to operate – even at numerical resolutions of up to $512^3$ meshpoints. This disagreed with subsequent simulations by Federrath et al. (2010), who solved the isothermal inviscid Euler equations with irrotational forcing using the FLASH CODE. They found significant vorticity generation in proximity to shocks where some kind of effective numerical viscosity must have acted.

Given that under isothermal conditions, only viscosity can lead to vorticity production, one must ask whether numerical viscosity or effective viscosity needed to stabilize numerical codes might have contributed to the production of vorticity in some of the earlier works. Indeed, it is possible that the directional operator splitting used in the FLASH CODE may have been responsible for spurious vorticity generation in the work of Federrath et al. (2010) (R. Rosner, private communication). On the other hand, when cooling and heating functions are included to perform more realistic simulations of the ISM, vorticity could be produced by the baroclinic term. Furthermore, even in the isothermal case, in which the baroclinic term vanishes, vorticity could be produced if there is rotation and/or shear.

The baroclinic term results from taking the curl of the pressure gradient term and is proportional to the cross product of the gradients of pressure and density. This term can play an important role when the assumptions of isothermality or adiabaticity are relaxed. Indeed, the baroclinic term can also be written as the cross product of the gradients of entropy and temperature. This formulation highlights the need for non-ideal effects, because in the absence of any other heating or cooling mechanisms, the entropy is just driven by viscosity. Again, it is not obvious that in the absence of additional heating and cooling much vorticity can be produced. On the other hand, it is clear that viscous heating must be significant even in the limit of vanishing viscosity.
because the velocity gradients can be very large, especially in shocks. Of course, the assumption about additional heating and cooling is not realistic for the interstellar medium and will need to be relaxed. Finally, there are the effects of rotation and shear, that can contribute to the production of vorticity even in the absence of baroclinicity.

The goal of this paper is to study the relative importance of the individual effects that contribute to vorticity production. It is then advantageous to restrict oneself to simplifying conditions that allow one to identify the governing effects. An important simplification is the restriction to weakly supersonic conditions so that shocks and other sharp structures can still be resolved with just a uniform and constant viscosity. We also neglect the effects of stratification which can only indirectly contribute to vorticity production. In fact, a constant gravitational acceleration drops out when taking the curl. Only in the non-isothermal and non-isentropic case can gravity contribute to vorticity production. In fact, a constant gravitational acceleration.

2. Preliminary considerations

We recall that in the absence of baroclinicity, rotation, and shear, the curl of the evolution equation of the velocity is given by (see, e.g., Mee & Brandenburg [2006])

\[
\frac{\partial \omega}{\partial t} = \nabla \times (u \times \omega - \nu \nabla \times \omega) + \nu \nabla \cdot G,
\]

where \( \nu \) is the kinematic viscosity (assumed constant) and \( G = 2S \Omega \nabla \ln \rho \) is a part of the viscous force that has non-vanishing curl even when the flow is purely irrotational. Here,

\[
S_{ij} = \frac{1}{2}(u_{i,j} + u_{j,i}) - \frac{1}{2} \delta_{ij} u_{k,k}
\]

is the traceless rate of strain matrix, and commas denote partial differentiation. The \( G \) term breaks the formal analogy with the induction equation. It is convenient to express the resulting rms vorticity in terms of the typical wavenumber \( k_w \) of vortical structures which we define as

\[
k_w = \omega_{\text{rms}} / u_{\text{rms}}.
\]

We monitor the ratio \( k_w / k_t \), where \( k_t \) is the adopted nominal forcing wavenumber. In Mee & Brandenburg [2003], the resulting vorticity, expressed in terms of the ratio \( k_w / k_t \), was found to be zero within error bars. This result is compatible with the idea that the \( \nu \nabla \times G \) term in Equation 1 is insignificant for vorticity production. By contrast, in vortical turbulence and at moderate values of the Reynolds number, \( k_w / k_t \) is found to be of the order of unity [Brandenburg, 2001], although one should expect a mild increase proportional to the square root of the Reynolds number as this number increases.

2.1. Rotation

Rotation leads to the addition of the Coriolis force, \( 2\Omega \times u \), in the evolution equation for the velocity. Taking the curl, we obtain the vorticity equation (1) with two additional terms, both proportional to \( \Omega \), so we have

\[
\frac{\partial \omega}{\partial t} = \ldots - 2\Omega \nabla \perp \cdot u_{\perp} + 2\Omega \cdot \nabla u_{\perp},
\]

where the dots denote the other terms in Equation 1 that we discussed already. In order to estimate the production of vorticity, one could derive an evolution equation for the enstrophy density, \( \omega^2 \), by multiplying the right-hand side of Equations 1 and 4 by \( \omega \), and use a closure assumption for the resulting triple correlations. However, it is then difficult to obtain a useful prediction for \( \omega_{\text{rms}} \), because the right-hand side of such an equation would necessarily be proportional to \( \omega \) and would therefore vanish unless \( \omega_{\text{rms}} \) was different from zero to begin with. Instead, we estimate \( \omega_{\text{rms}} \) by computing the rms value of \( \Omega \omega / \partial t \) and replacing it by \( \omega_{\text{rms}} / \tau_{\Omega} \), where \( \tau_{\Omega} \) is a typical time scale of the problem. This leads to

\[
\omega_{\text{rms}} \approx 2\Omega \tau_{\Omega} \left( (\nabla \perp \cdot u_{\perp})^2 + (\nabla \parallel u_{\parallel})^2 \right)^{1/2},
\]

where \( \nabla \perp \) and \( \nabla \parallel \) denote derivatives in the directions perpendicular and parallel to the rotation axis and \( u_{\perp} \) is the velocity vector perpendicular to the rotation axis. Using Cartesian coordinates where \( \Omega \) points in the \( z \) direction, we have

\[
\omega_{\text{rms}} \approx 2\Omega \tau_{\Omega} \left( (u_{x,x} + u_{y,y})^2 + u_{z,z}^2 \right)^{1/2}.
\]

We expect \( \tau_{\Omega} \) to be comparable to the turnover time, \( \tau = (u_{\text{rms}} k_t)^{-1} \). We expect the rms values of the velocity derivative term in Equation 6 to be comparable to the rms velocity and some inverse length scale. Typically, one would expect it to be proportional to \( u_{\text{rms}} k_t \), although, again, there can be an additional dependence on the square root of the Reynolds number. However, for fixed Reynolds number, and not too rapid rotation, we expect \( \omega_{\text{rms}} \) to increase linearly with the Coriolis number, i.e.,

\[
\text{Co} = 2\Omega \tau, \quad \text{where} \quad \tau = (u_{\text{rms}} k_t)^{-1}.
\]

Thus, we expect \( k_w / k_t = \text{St}_{\Omega} \text{Co} \), where we have defined an effective rotational Sroubal number,

\[
\text{St}_{\Omega} = \tau_{\Omega}^\text{eff} u_{\text{rms}} k_t.
\]

We regard this as a fit parameter that will emerge as a result of the simulations. We have here introduced the quantity \( \tau_{\Omega}^\text{eff} \), where \( \tau_{\Omega}^\text{eff} / \tau_{\Omega} \) is given by the ratio of the velocity gradient terms divided by \( u_{\text{rms}} k_t \). However, for larger values of Co there may be a departure from a linear dependence between \( k_w / k_t \) and Co. (We note that, apart from a possible \( 4\pi \) factor, the Coriolis number is just the inverse Rossby number.) One aim of this paper is therefore to verify this dependence from simulations and to determine empirically the value of \( \tau_{\Omega} \).

2.2. Shear

In the presence of linear shear with \( u_s = (0, Sx, 0) \), the evolution equation for the departure from the mean shear attains additional terms, \( -u_i \nabla \cdot u - u \cdot \nabla u_s \). This implies a dependence of \( \omega_{\text{rms}} \) on \( S \), analogous to the \( \Omega \) dependence discussed above. In components form, this means that

\[
\omega_{\text{rms}} \approx S T_S \left( (u_{x,x} + u_{y,y})^2 + u_{z,z}^2 + O(xu'') \right)^{1/2},
\]

which is quite similar to Equation 5, except that in the penultimate term in angular brackets the indices are now interchanged, i.e. we now have \( u_{x,y} \) instead of \( u_{y,x} \). In analogy to \( \tau_{\Omega} \), we define \( \tau_S \) as a typical time scale of the problem and we expect it to be again related to the turnover time \( \tau \). The \( O(xu'') \) denotes the presence of additional terms that are proportional to \( x \) and to second derivatives of \( u \). However, when adopting the shearing box approximation with shearing-periodic boundaries [Goldreich & Lynden-Bell, 1965; Wisdom & Tremaine, 1988].
each point in the $xy$ plane is statistically equivalent. We would therefore not expect there to be a systematic $x$ dependence, which corresponds to the assumption of Galilean invariance that is sometimes used in the study of turbulent transport coefficients in linear shear flows (Sridhar & Subramanian 2009). We will postpone the possibility of additional terms until later. Since we expect $\tau_S$ to be comparable to $\tau = (u_{rms} k_t)^{-1}$, the rms vorticity should be proportional to the shear parameter,

$$\text{Sh} = S / u_{rms} k_t,$$

although for large values of $|\text{Sh}|$ we may expect departures from a linear dependence. Determining this dependence is another aim of this paper. Again, a linear dependence is characterized by the values of $\tau_S$ and $\tau_S^{\text{eff}}$, where, in analogy with the previous case with rotation, the ratio $\tau_S^{\text{eff}} / \tau_S$ is given by the derivative term in Equation (9), normalized by $u_{rms} k_t$. A convenient non-dimensional measure of the value of $\tau_S^{\text{eff}}$ is what we call the shear Strouhal number,

$$\text{St}_S = \frac{\tau_S^{\text{eff}} u_{rms} k_t}{c_s},$$

which can be determined provided there is a range in Sh over which $\omega_{rms}$ increases linearly with Sh.

The study of vorticity production by rotation and shear is quite independent of thermodynamics and can in principle be studied even in the incompressible case. However, in the present paper we study this effect in the weakly compressible case of low Mach numbers and under the assumption of an isothermal equation of state, where the baroclinic term vanishes.

### 2.3. Baroclinicity

As mentioned in the introduction, the baroclinic term, proportional to $\nabla \rho \times \nabla p$, emerges when taking the curl of the pressure gradient term, $\rho^{-1} \nabla p$. This term can also be written as

$$\rho^{-1} \nabla p = \nabla h - T \nabla s,$$

where $h$ and $s$ are specific enthalpy and specific entropy, respectively, and $T$ is the temperature. Thus, we have

$$\frac{\partial \omega}{\partial t} = \ldots + \nabla T \times \nabla s. \quad (13)$$

In order to study the effect of the baroclinic term, it is useful to look at the dependence of the mean angle $\theta$ between the gradients of $s$ and $T$, defined via

$$\sin^2 \theta = \langle (\nabla T \times \nabla s)^2 \rangle / \langle (\nabla T)^2 \rangle \langle (\nabla s)^2 \rangle. \quad (14)$$

An important aspect is then to study first the dependence of the rms values of the gradients of $s$ and $T$. We can do this by looking at a one-dimensional model where, of course, $\theta = 0$.

Next, we need to determine $\theta$ from three-dimensional simulations. The hope is then that we can express baroclinic vorticity production in the form

$$k_x / k_t = \text{St}_{\text{base}} (\nabla T)_{\text{rms}} (\nabla s)_{\text{rms}} \sin \theta / u_{\text{rms}}^2 k_t^2. \quad (15)$$

On dimensional grounds we expect the product of $(\nabla T)_{\text{rms}}$ and $(\nabla s)_{\text{rms}}$ to be of the order of $u_{\text{rms}}^2 k_t^2$, and so a possible ansatz would be

$$k_x / k_t = \text{St}_{\text{base}} \sin \theta, \quad (16)$$

where we have subsumed the scalings of $(\nabla T)_{\text{rms}}$ and $(\nabla s)_{\text{rms}}$ in that of an effective baroclinic Strouhal number $\text{St}_{\text{base}}^{\text{eff}}$.

An important issue is the fact that viscous heating leads to a continuous increase of the temperature. As a result, the sound speed changes and it becomes then impossible to study the behavior of the system in a steady state. In order to avoid this inconvenience, we add a volume cooling term that is non-vanishing when the local sound speed $c_s$ is different from a given target value, $c_{0 \text{d}}$. Thus, in the presence of finite thermal diffusivity $\chi$, and with a cooling term governed by a cooling time $\tau_{\text{cool}}$, our entropy equation takes the form

$$T \frac{D s}{D t} = 2 \nu \nabla^2 + \nabla \cdot (c_p \rho \chi \nabla T) - \frac{1}{\tau_{\text{cool}}} (S_{\text{eff}}^2 - S_{\text{0 \text{d}}}^2), \quad (17)$$

where $c_s$ is the adiabatic sound speed. We assume a perfect gas so that $S_{\text{eff}}^2 = (\gamma - 1) c_p T$, where $\gamma = c_p / c_v = 5 / 3$ for a monatomic gas, and $c_p$ and $c_v$ are the specific heats at constant pressure and constant volume, respectively. The value of $\tau_{\text{cool}}$ can have an influence on the results, so we need to consider different values. We express $\tau_{\text{cool}}$ in terms of $c_{0 \text{d}}$ and $k_t$, and define the nondimensional quantity $\text{St}_{\text{cool}} = \tau_{\text{cool}} c_{0 \text{d}} k_t$.

### 3. The model

In this paper we solve the continuity equation for the density $\rho$,

$$\frac{D \ln \rho}{D t} = - \nabla \cdot \mathbf{u}, \quad (18)$$

together with the momentum equation for the velocity $\mathbf{u}$,

$$\frac{D \mathbf{u}}{D t} = - \rho^{-1} \nabla p - 2 \Omega \times \mathbf{u} - S_{\text{0 \text{d}}} \mathbf{y} + \nabla \phi + \mathbf{F}_{\text{visc}}, \quad (19)$$

where $D / D t = \partial / \partial t + (\mathbf{u} + \mathbf{u}^S) \cdot \nabla$ is the advection operator with respect to the sum of turbulent flow $\mathbf{u}$ and laminar shear flow $\mathbf{u}^S$, $p$ is the pressure, $\phi$ is the forcing potential, and

$$\mathbf{F}_{\text{visc}} = \rho^{-1} \nabla \cdot (2 \nu \mathbf{S}), \quad (20)$$

is the viscous force, where $\mathbf{S}$ was defined in Equation (2). The forcing potential is given by

$$\phi (\mathbf{x}, t) = \phi_0 \ N \exp \left\{ \left[ - |\mathbf{x} - \mathbf{x}_c(t)|^2 / R^2 \right] \right\}, \quad (21)$$

where $\mathbf{x} = (x, y, z)$ is the position vector, $\mathbf{x}_c(t)$ is the random forcing position that changes abruptly after a time interval $\Delta t$. $R$ is the radius of the Gaussian, and $N$ is a non-dimensional factor proportional to $\Delta t^{-3 / 2}$. This ensures that the amplitude of the correlation function of $\phi$ is independent of $\Delta t$. Thus, we choose

$$N = \sqrt{R^3 / c_{0 \text{d}} \Delta t}. \quad (22)$$

Since $N$ is non-dimensional, the prefactor $\phi_0$ has the same dimension as $\phi$, which is that of velocity squared. We consider two forms for the time dependence of $\mathbf{x}_c$. First, we take $\mathbf{x}_c$ such that the forcing is $\delta$-correlated in time. In that case, $\Delta t$ is equal to the length of the time step $\delta t$. Alternatively, we choose a finite forcing time $\delta t_{\text{force}}$ that defines the interval during which $\mathbf{x}_c$ remains constant, after which the forcing changes again abruptly. Thus,

$$\Delta t = \max (\delta t, \delta t_{\text{force}}) \quad (23)$$

equal to $\delta t$ in the $\delta$-correlated case or equal to $\delta t_{\text{force}}$ in the case of finite correlation time.

The work of Mee & Brandenburg (2006) showed that the peak of the energy spectrum depends on the radius $R$ of the Gaussian. Indeed, the Fourier transform of $\exp(-r^2 / R^2)$ is also a Gaussian with $\exp(-k^2 / k_t^2)$, where

$$k_t = 2 / R. \quad (23)$$
In the following we use this as our definition of \( k_f \) and check a posteriori that this is close to the position of the peak of the energy spectrum. In the following, we characterize our simulations in terms of the ratio \( k_f/k_1 \), and consider values between 2 and 10.

We use the PENCIL CODE\(^1\) which is a non-conservative, high-order, finite-difference code (sixth order in space and third order in time) for solving the compressible hydrodynamic and hydromagnetic equations. We adopt non-dimensional variables by measuring speed in units of a reference sound speed, \( c_0 \), and length in units of \( 1/k_1 \), where \( k_1 \) is the smallest wavenumber in the periodic domain. This implies that the non-dimensional size of the domain is \((2\pi)^3\).

In order to study the effects of rotation and shear, we ignore entropy effects and restrict ourselves to an isothermal equation of state with constant sound speed \( c_0 \). This means that \( \rho^{-1} \nabla p \) reduces to \( c_0^2 \nabla \ln \rho = - \nabla h \), which has vanishing curl. Here, \( h = c_0^2 \ln \rho \) is the relevant enthalpy in the isothermal case. On the other hand, in order to study the effects of baroclinicity, we do need to allow the entropy to vary, so we also need to solve Equation (17), and study the dependence of \( k_{\omega}/k_f \) on the Mach number,

\[
\text{Ma} = \frac{u_{\text{rms}}}{c_0}.
\]  
(24)

In order to characterize the degree of turbulence, we define the Reynolds number based on the energy-carrying scale, corresponding to the typical wavenumber where the spectrum peaks, i.e.

\[
\text{Re} = \frac{u_{\text{rms}}}{\nu k_f}.
\]  
(25)

For vortical turbulence, this definition is known to be a good measure of the ratio of the resulting turbulent viscosity divided by the molecular diffusivity (Yousef et al. 2003). The two numbers, \( \text{Ma} \) and \( \text{Re} \), can be varied by changing \( \nu \) and/or the strength of the forcing. In all cases we use \( \chi = \nu \). Another input parameter is the forcing Strouhal number

\[
\text{St}_{\text{force}} = \frac{\tau_{\text{force}} u_{\text{rms}}}{\nu k_f},
\]  
(26)

which is zero for \( \delta \)-correlated forcing and equal to about 0.3 in cases with finite correlation time. These are also the values used by Mee & Brandenburg (2006).

In the following we also consider kinetic energy and enstrophy spectra, \( E_k(k) \) and \( E_{\omega}(k) \), respectively. They are normalized such that (Lesieur 1990)

\[
\int E_k(k) \, dk = \frac{1}{2} \langle u^2 \rangle, \quad \int E_{\omega}(k) \, dk = \frac{1}{2} \langle \omega^2 \rangle,
\]  
(27)

where \( \frac{1}{2} \langle u^2 \rangle \) and \( \frac{1}{2} \langle \omega^2 \rangle \) are kinetic energy and enstrophy, respectively. For comparison we also consider spectra of enthalpy, \( E_h(k) \), which are normalized such that \( \int E_h(k) \, dk = \frac{1}{2} \langle h^2 \rangle \).

Throughout this paper we assume periodic boundary conditions, except that in the presence of shear we employ shearing-periodic boundary conditions where the \( x \) direction is periodic with respect to positions in \( y \) that shift with time, i.e.

\[
f(-\frac{1}{2} L_x, y, z, t) = f(\frac{1}{2} L_x, y + L_x St, z, t),
\]  
(28)

where \( f \) represents any one of our four dependent variables \((u, \rho)\). This boundary condition was first proposed by Goldreich & Lynden-Bell (1965) and has been routinely used in local simulations of accretion disk turbulence (Hawley et al. 1995). Note, however, that recent work of Regev & Umurhan (2008) and Bodo et al. (2008) called attention to the possibility of problems with the shearing sheet approximation when the size of the perturbations is large. In somewhat weaker form, this problem also applies to a non-shearing periodic box. Indeed, we shall keep this in mind when interpreting some of the results presented below.

4. Results

We begin by studying the effect of rotation. In Figure 1 we plot time-averaged kinetic energy and enthalpy spectra, \( E_{K}(k) \) and \( E_{\omega}(k) \), respectively. Note that rotation has a tendency to move the peak of \( E_{K}(k) \) to the left of the nominal value of \( k_f \). However, at the Reynolds number of 25 shown here, there is no inertial range, but in all cases, the energy spectra show a clear viscous dissipation range, suggesting that these runs are sufficiently well resolved. At somewhat larger Reynolds number or smaller forcing wavenumber, earlier work of Mee & Brandenburg (2006) began to show a short \( k^{-2} \) subrange. Such a slope is predicted for shock turbulence

\[
\begin{align*}
E_{K}(k) & \sim k^{-a} \\
E_{\omega}(k) & \sim k^{-b}
\end{align*}
\]

Fig. 1. Time-averaged kinetic energy and enthalpy spectra for two values of the Coriolis number for \( \text{Re} = 25 \) and \( \text{St}_{\text{force}} = 0.4 \). The two straight lines give the slopes \(-2 \) and \(-3 \), respectively. In both cases we have \( k_f/k_1 = 4 \).

Fig. 2. Dependence of \( k_{\omega}/k_f \) on \( \text{Co} \) for three groups of runs: group 1 with \( \text{Re} = 15, k_f/k_1 = 10, \text{St}_{\text{force}} = 0.3 \); group 2 with \( \text{Re} = 25 \) and \( \text{St}_{\text{force}} = 0.4 \); and group 3 with \( \text{Re} = 30 \) and \( 150, k_f/k_1 = 4, \text{St}_{\text{force}} = 0 \).

\[
\begin{align*}
\text{St}_{\text{force}} & = 0, \text{Re} = 150 \\
& = 30, \text{Re} = 150 \\
& = 0.4, \text{Re} = 130 \\
& = 25, \text{Re} = 130 \\
& = 0.3, \text{Re} = 15 \\
& = 0, \text{Re} = 15
\end{align*}
\]

\[
\begin{align*}
\text{Co} & = 1.35 \\
& = 0.15
\end{align*}
\]

\[
\begin{align*}
slope & = 1
\end{align*}
\]

\[
\begin{align*}
k_{\omega}/k_f & = 0.01 \\
& = 0.1 \\
& = 1.0 \\
& = 10.0 \\
& = 100.0
\end{align*}
\]

\[
\begin{align*}
0.01 & = 0.1 \\
& = 1.0 \\
& = 10.0 \\
& = 100.0
\end{align*}
\]
overlap for Co.

Fig. 3. Dependence of $k_w/k_l$ on Re for Co $\approx 1$, $k_l/k_1 = 2$, and $St_{\text{force}} = 0$ ($\delta$-correlated forcing).

Fig. 4. Time-averaged enstrophy spectra, $E_\omega(k)$, compared with $k^2 E_K(k)$, for Re = 25, $St_{\text{force}} = 0.4$, and three values of the Coriolis number. The curves of $k^2 E_K(k)$ are close together and overlap for Co = 0.01 (dotted) and 0.15 (dashed), so it becomes a single dash-dotted line. The $k^{-3}$ slope is shown for comparison. In all three cases we have $k_l/k_1 = 4$.

Table 1. Root-mean-squared values of components of the velocity derivative tensor, normalized by $u_{\text{rms}} k_l$, as well as the three diagonal components of the $\langle u_i u_j \rangle$ tensor for 4 values of Co.

| Co  | Components of the Velocity Derivative Tensor | $\langle u_i u_j \rangle_{\text{rms}} / u_{\text{rms}} k_l$ |
|-----|---------------------------------------------|-------------------------------------------------|
|     | $\langle \nabla \cdot u \rangle_{\text{rms}} / u_{\text{rms}} k_l$ | $0.11$                                           |
|     | $u_{x,x} / u_{\text{rms}} k_l$               | $0.64$                                           |
|     | $u_{x,y} / u_{\text{rms}} k_l$               | $0.64$                                           |
|     | $u_{x,z} / u_{\text{rms}} k_l$               | $0.64$                                           |
|     | $u_{y,x} / u_{\text{rms}} k_l$               | $0.64$                                           |
|     | $u_{y,y} / u_{\text{rms}} k_l$               | $0.64$                                           |
|     | $u_{y,z} / u_{\text{rms}} k_l$               | $0.64$                                           |
|     | $u_{z,x} / u_{\text{rms}} k_l$               | $0.64$                                           |
|     | $u_{z,y} / u_{\text{rms}} k_l$               | $0.64$                                           |
|     | $u_{z,z} / u_{\text{rms}} k_l$               | $0.64$                                           |
|     | $\langle u_x^2 \rangle / u_{\text{rms}}^2$   | $0.33$                                           |
|     | $\langle u_y^2 \rangle / u_{\text{rms}}^2$   | $0.33$                                           |
|     | $\langle u_z^2 \rangle / u_{\text{rms}}^2$   | $0.33$                                           |

Kadomtsev & Petviashvili (1973), and it has also been seen in the irrotational component of transonic turbulence (Porter et al., 1998).

In Figure 2 we plot the dependence of $k_w/k_l$ on Co for three groups of runs: group 1 with Re = 15, $k_l/k_1 = 10$, $St_{\text{force}} = 0.3$; group 2 with Re between 25 and 130, $k_l/k_1 = 4$, $St_{\text{force}} = 0.4$; and group 3 with Re = 30 and 150, $k_l/k_1 = 2$, $St_{\text{force}} = 0$. In all cases we use 128$^3$ mesh points, average the results over at least 200 turnover times and, in some cases, even several thousand turnover times. It turns out that for $St_{\text{force}} \neq 0$ a linear relationship between $k_w/k_l$ and Co is a good approximation for Co $\leq 10$, where $k_w/k_l \approx 0.03$ Co, i.e. $St_\Omega = 0.03$. Furthermore, we see from Table 1 that the normalized velocity derivative terms are all about 0.5, so the root of the sum of their squares is slightly larger than unity, corresponding to $\tau_{\Omega} / \tau_\Omega \approx 1.3$. For Co $> 10$ the value of $k_w/k_l$ seems to saturate at about unity.

A similar result is also found for $St_{\text{force}} = 0$, except that there remains a spurious contamination of vorticity even for small values of Co, a limit in which we expect to observe no vorticity production. By varying the value of Re, while keep-
Finally, we consider the possibility of vorticity generation by the baroclinic term. In a preparatory step we study first the dependence of the product $\langle \nabla T \rangle_{\text{rms}} \langle \nabla \phi \rangle_{\text{rms}}$ on both $\text{Ma}$ and $\text{Re}$ in a one-dimensional model. In all cases we vary the strength of the forcing amplitude in the range $1 \leq \phi_0/c_s \leq 500$ for different values of viscosity and cooling time. As we increase the value of $\phi_0$, the Reynolds number increases for a given value of the viscosity. For small values of $\phi_0$, the Mach number also increases linearly, where the ratio of $\text{Ma}/\text{Re}$ increases with increasing viscosity. However, for larger values of $\phi_0$ there is saturation and $\text{Ma}$ no longer increase with $\phi_0$.

Furthermore, in the range where $\text{Ma}$ still increases linearly with $\phi_0$, the rms value of the entropy gradient increases, but it also saturates when $\text{Ma}$ saturates. The rms value of the temperature gradient, however, decreases with increasing values of $\phi_0$, but this seems to be a special property of the one-dimensional model that is not borne out by the three-dimensional simulations where it stays approximately constant.

Remarkably, the results are fairly independent of the cooling time, except that the break point where $\langle \nabla s \rangle_{\text{rms}}$ saturates occurs for smaller values of $\phi_0$ as we increase the cooling time; see Figure 6. This break point is also related to the point where the Mach number saturates, as can be seen from Figure 7. However, for longer cooling times there can be longer transients, making it difficult to obtain good averages. Therefore we focus, in the rest of this paper, on the case of shorter cooling times using $\text{St}_{\text{cool}} = 0.2$. Another remarkable result is that the normalized value of $\langle \nabla T \rangle_{\text{rms}} \langle \nabla s \rangle_{\text{rms}}$ is always of the order of about $10^{-3}$, independent of resolution, cooling time, or the value of the viscosity.

Most of the basic features of the one-dimensional model are also reproduced by two- and three-dimensional calculations. Two-dimensional simulations have the advantage of being easily visualized and are therefore best suited for illustrating vorticity production by the baroclinic term. In Figure 8 we demonstrate that vorticity production is associated with the interaction between the fronts of different expansion waves. In this example we chose $\delta_{\text{force}} \phi_0/c_s = 0.1$, so the first expansion wave is launched at $t = 0$ and the second one at $tc_{\phi_0}/R = 0.1$. The top row of Figure 8 shows that at $tc_{\phi_0}/R = 0.09$, i.e. just before launching the second expansion wave, the baroclinic term and the vorticity are still just at the noise level. At that time the most pronounced feature is the discontinuity between the Gaussian expansion waves in the periodic domain. This leads to negligibly weak vorticity, and no baroclinic term. At $tc_{\phi_0}/R = 0.11$,

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**Table 2.** Similar to Table 1 but for the case with shear.

| Sh  | $\langle \nabla s \rangle_{\text{rms}}/c_s^2$ | $\langle u_2 \rangle/u_{\text{rms}}$ | $\langle u_1 \rangle/u_{\text{rms}}$ | $\langle u_3 \rangle/u_{\text{rms}}$ | $\langle u_2^2 \rangle/u_{\text{rms}}^2$ | $\langle u_1^2 \rangle/u_{\text{rms}}^2$ | $\langle u_3^2 \rangle/u_{\text{rms}}^2$ |
|-----|----------------------------------------|---------------------------------|---------------------------------|---------------------------------|---------------------------------|---------------------------------|---------------------------------|
| $\phi_0/c_s = 1$ | 0.00 | 0.01 | 0.02 | 0.03 | 0.04 | 0.05 | 0.06 |
| 0.1 | 0.10 | 0.20 | 0.30 | 0.40 | 0.50 | 0.60 | 0.70 |
| 0.2 | 0.30 | 0.40 | 0.50 | 0.60 | 0.70 | 0.80 | 0.90 |
| 0.3 | 0.40 | 0.50 | 0.60 | 0.70 | 0.80 | 0.90 | 1.00 |
| 0.4 | 0.50 | 0.60 | 0.70 | 0.80 | 0.90 | 1.00 | 1.10 |
| 0.5 | 0.60 | 0.70 | 0.80 | 0.90 | 1.00 | 1.10 | 1.20 |

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**Fig. 6.** Dependence of the rms values of temperature and entropy on $\phi_0$ for $\nu/c_s R = 1$ and $\text{St}_{\text{cool}} = 0.2$ (top panel), 0.6 (middle panel), and 2 (bottom panel).

**Fig. 7.** Dependence of Mach number and the rms value of entropy on $\phi_0$ for $\nu/c_s R = 1$ and $\text{St}_{\text{cool}} = 0.2$, 0.6, and 2 (solid, dotted, and dashed line types, respectively).

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(Regev & Umurhan, 2008; Bodo et al., 2008), which may be responsible for spurious vorticity generation. On the other hand, there is vorticity generation even for large scale-separation ratios, $k_\perp/k_1 = 10$; see the dash-dotted line in Figure 5. This suggests the possibility of a more general problem that would not go away even in the limit of small eddies and small values of $\text{Sh}$. Nevertheless, there is a clear rise of $k_\perp/k_1$ when $|\text{Sh}| > 0.1$, which is in agreement with our expectations outlined in Section 2.2. However, the slope in this relationship is rather steep, $\text{St}_S \approx 6$. The velocity derivative terms are only slightly larger than in the case with rotation, corresponding to $\tau_\perp/\tau_S \approx 1.5$; see also Table 2. Tentatively, this suggests that for comparable values of $\text{Co}$ and $\text{Sh}$, $\tau_\perp \gg \tau_S$. On the other hand, given that even for small values of $\text{Sh}$ there is spurious vorticity generation, we cannot be certain that the results are reliable for larger ones either. The case with shear must therefore remain somewhat inconclusive.
the effect of the second expansion wave becomes noticeable in visualizations of both \((\nabla T \times \nabla s)_z\) and \(\omega_z\), while our visualizations of \(T\) and \(s\) barely show the second expansion wave. At \(tc_{s0}/R = 0.14\), the first expansion wave is clearly no longer circular, which is obviously associated with the second expansion wave that is now quite pronounced in our visualizations of both \(T\) and \(s\).

In order to have a more accurate quantitative determination of vorticity production, we now consider three-dimensional models. In Figure 9 we show the dependence of various quantities on \(\phi_0\) for \(St_{cool} = 0.2\) and \(\nu/c_0 R = 1\). In all cases we use 128\(^3\) mesh points and average the results over between 20 and 70 turnover times. Note that here \(Re \approx 0.05 \phi_0/c_{s0}^2\). Given that \(Re\) depends inversely proportionally on \(\nu/Rc_{s0}\), we can also write \(Re \approx 0.05 \phi_0 R/c_{s0} \nu\). The Mach number saturates at about \(Ma = 3\), and the rms value of the entropy gradient increases up until this point. Given that the rms value of the temperature gradient also stays approximately constant, we find a weak increase of \((\nabla T)_{rms}(\nabla s)_{rms}\). The value of \((\nabla T \times \nabla s)_{rms}\) is always found to be a certain fraction below this value, resulting in a typical baroclinic angle of about 45 degrees; see the third panel of Figure 9.

Finally, the amount of vorticity production in terms of \(k_{v}/k_l\) is about 0.3 for \(\phi_0/c_{s0}^2 > 20\). For smaller values, on the other hand, there is an approximately linear increase with \(k_{v}/k_l \approx 0.014 \phi_0/c_{s0}^2\).

The possibility of spurious vorticity is easily eliminated in this case by looking at enstrophy spectra; see Figure 10, where we compare \(E\omega(k)\) with \(k^2 E_k(k)\). All spectra fall off rapidly with increasing \(k\). Thus, even though the initial vorticity generation occurred evidently at the smallest available scales, once the flow becomes fully developed, most of the enstrophy resides at scales equal to or larger than the driving scale. Furthermore, the spectra of \(E\omega(k)\) and \(k^2 E_k(k)\) are close together, suggesting that the vorticity is close to its maximal value.

5. Applications

The level of vorticity that is produced in the usual case of solenoidal forcing of the turbulence is such that \(k_{v}/k_l \approx 1\) (see, e.g., [Brandenburg, 2001]). For turbulence whose driving force has finite correlation time (\(St_{force} = 0.3\), for example), and small values of \(Re\), we have \(k_{v}/k_l = O(1)\) when \(Co \gtrsim 10\). However,
for larger values of Re, the turbulence becomes vortical already for smaller values of Co. Comparing with the galaxy, we have $\Omega \approx 10^{-15} \text{ s}^{-1}$, $u_{\text{rms}} = 10 \text{ km/s}$, and an estimated correlation length of about 70 pc, so $k_t = 3 \times 10^{-20} \text{ cm}$, so Co = 0.07, which is rather small. Thus, rotation may not be able to produce sufficient levels of vorticity. Given that in galaxies with flat rotation curves, $S \approx -\Omega$, shear should not be very efficient either. However, the Mach numbers are undoubtedly larger than unity in the interstellar medium, so this should lead to values of $k_\omega/k_t \approx 0.3$, which is the saturation value found in Figure 9. Given that one of the reasons for studying the production of vorticity is the question of dynamo action, we should point out that such values of $k_\omega/k_t$ are large enough for the small-scale dynamo. Large-scale dynamo action should be possible in galaxies as well, because of their large length scales, but it suffers from the well-known problem of a small growth rate. It then remains difficult to explain large-scale magnetic fields in very young galaxies (Beck et al., 1996).

The question of vorticity generation is also important in studies of the very early Universe, where phase transition bubbles are believed to be generated in connection with the electroweak phase transition (Kajantie & Kurki-Suonio, 1986; Ignatiu et al., 1994). Here the equation of state is that of a relativistic fluid, $p = \rho c^2/3$, where $c$ is the speed of light. Thus, there is no baroclinic term and no obvious source of vorticity. However, the relativistic equation of state may be modified at small length scales, but it is not clear that this can facilitate significant vorticity production.

6. Conclusions

The present work has demonstrated that vorticity production is actually quite ubiquitous once there is rotation, shear, or baroclinicity. This implies that the assumption of potential flows as a model for interstellar turbulence might be of academic interest and could only be realized under special conditions of weak forcing, weak rotation, and no shear. In galaxies, however, the shear and Coriolis number are well below unity, leaving only the baroclinic term as a viable candidate for the production of vorticity. This agrees with early work of Korpi et al. (1999b), who analyzed the production terms in supersonic, supernovadriven turbulence quantitatively; see also Glasner et al. (1997), who showed that on long enough time scales significant vorticity can also be produced for subsonic flows. We have also observed how vorticity is mainly produced close to shock fronts. This motivates a more detailed investigation of these zones as the next step in the study of vorticity generation in the interstellar medium. It should also be pointed out that the baroclinic term corresponds to the battery term in the induction equation (Kulsrud et al., 1997; Brandenburg & Subramanian, 2005). Thus, when studying the possibility of dynamo action, this battery term provides an intrinsic and well defined seed for the dynamo and should therefore be included as well.

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