1. Introduction

In 2016, Google introduced the Global Navigation Satellite System (GNSS) Analysis Tools. Back in Android nougat, developers implemented application program interfaces (APIs) to retrieve raw measurements of the GNSS from Android devices. The GNSS Analysis Tools are currently ready to process and analyze the measurements. Android runs on over 2 billion devices and Android Phones are produced by many different manufacturers. The main goal of the new tools is to enable device manufacturers to realize in detail how well the GNSS receivers work in each device design and how can improve the GPS/GNSS performances. However, the publicly available tools additionally attach considerable importance to the research and application development [1].

Most of the GPS receivers that provide raw data can measure pseudo-range and carrier-phase, generally, the accuracy of carrier-phase is much more accurate than pseudo-range and the pseudorange data are very noisy so that the non-noisy pseudo-range with high accuracy can be achieved when recalculated depending on the carrier-phase, this recalculation process is called smoothing pseudo-range.

2. The most common smoothing pseudo range that depend on carrier-phase is Hatch filter algorithm. The Hatch filter need only the observation for the current time and the observations of the previous time. They do not need any additional observations and sensors, for that reason the authors used this method because it is most method applicable for GPS Smartphone.

3. GPS Observables

Positioning by a GPS receiver based on the geometrical principles of the predicted satellite positions from the navigation message and the distances between the user antenna position and the GPS satellite transmitter, which is determined by measuring the transit time of the satellite signals. In principle, modern GPS receivers can determine different observation variables. However, not every GPS receiver has the capability of measuring all of the following observables:

1) Pseudo-distance (pseudo-range)
2) The carrier-phase measurement in one or two-frequency.

Besides, the difference in distance between receivers and two orbit positions of the same satellite can be determined by measuring the Doppler shift of the incoming carrier phase in single or dual frequency receivers [2, 3].
4. Smoothing Hatch filter

The pseudo-range measurements generally contain high noise while the carrier-phase contains less noise and having high constancy. The smoothing of the measurements is one of the methods used to reduce the noise of the pseudo range to obtain accurate measurements as much as possible [4-5]. This used to estimate precise positions using single receiving units.

The most common smoothing pseudo range that depends on carrier-phase is the Hatch filter algorithm. Because the Hatch filter needs only the observation for the current time and the observations of the previous time, they do not need any additional observations and sensors [6], for that reason the authors used this method because it is the most method that can be applied for single-phase GPS Smartphone.

Typically used to reduce or smooth the errors of the effect of noise and multipath in pseudo-range measurements by using the corresponding high accuracy carrier phase measurements, the Hatch smoothing filter is represented by the following equation [7]:

$$\hat{R}(s, k) = \frac{n-1}{n} \hat{R}(s, k-1) + \left( \phi(s, k) - \phi(s, k-1) \right) + \frac{\hat{R}(s,k)}{n}$$  (1)

Where:

- $\hat{R}(s,k)$: Code measurement for Satellite s at a time n
- $\phi(s,k)$: Carrier phase measurement
- s: Satellite.
- n: Time
- $\hat{R}(s,k-1)$: smoothed code for the previous epoch.
- The initial value for the algorithm starts with $\hat{R}(s, 1)=R(s, 1)$.

Eq. (1) (the Hatch filter) depends on the current epoch measurements and the previous estimate, without any mathematical models or sensors to eliminate errors. Generally, it is used for single frequency low-cost GPS receivers. The filter depends on constancy variations in carrier phase measurement thus if the cycle-slip occurs in carrier phase the filter must be initialized [6, 7].

5. Cycle slips

In the case of phase measurement, the ambiguity value of a given epoch is estimated where the ambiguity represents the main problem, which depends on the continued reception of the signal between the receiver and the satellite. The cycle slips occurs if the receiver losses tracking of the satellite then the satellite reappears again. In other words, a new guessing value will be given and the cycles will be counted from the beginning because of the interruption in the received signal. The result of the cycle slips on the phase measurement is represented as an integer number of cycles. Finding and correcting the cycle slips ensures that the correct ambiguity value is obtained [6].

6. GPS data observations

The data from the GPS receiver was collected using a Huawei P10 device. The specifications of the receiver and antenna that gathered from RINX file are as follows:

| Serial Number | Type     | Ver.     |
|---------------|----------|----------|
| Receiver      | 7EX0217923002483 | HUAWEI VTR-L29 |
| Antenna       | 7EX0217923002483 | VTR-L29 |

Table 1: Specifications for the receiver and antenna.

Data were taken for 41 minutes and 24 seconds with sampling interval 1-second using Geo++ Android application and above the building of the department of civil engineering, University of Technology campus without obstructions preventing visibility to the sky and leading to multipath errors.

7. Experimental Results

In the observation time, eleven satellites were obtained (G5, G7, G8, G9, G11, G13, G17, G23, G27, G28, and G30) and represented by elevation angle in degree and epochs in seconds as shown in Figure 1.

The measured pseudo-range observation without smoothing of the eleven satellites is represented by pseudo-range in meters and epochs in seconds as shown in Figure 2.

Figure 1: Number of satellite and elevation angles
Figure 2: Observations without smoothing

In Figure 2, the satellite number G27 has been dropped because the elevation angle of the satellite is below the elevation mask of 10°. The raw measurements of the pseudo-range observations were used to estimate the location by the least-squares observation method, using a single point positioning (epoch by epoch) method. The error residuals of the locations estimated from raw data from the most probable value of the local location in (NEU) are illustrated in Figures 3-7. The estimated errors represented by errors in meters and time in seconds.

Figure 3 represents all the residuals raw data in the ENU direction with time, Figure 4 illustrates the errors in the easting direction with time. Figure 5 depicts the errors in the northing direction with time. Figure 6 shows the errors in the up direction with time, and Figure 7 represents the horizontal positioning error of raw data in the Easting and North directions.

It is clear that Figure 3,4,5, and Figure 6 have high noisy and errors in the easting direction ranging from -17m to 11m, errors in the northing direction ranging from -17m to 12m, and errors in the up direction ranging from -50 m to 30 m.

The processing of pseudo-range with the Hatch filter, and estimate the positions as in raw pseudo-range epoch by epoch single point positioning the error results as shown in Figures 8-12.
Figures 8-12 express that the noisy behavior of the raw data have highly reduced after applying smoothing Hatch filter, and the smoothed data having errors in the easing direction ranging from -9m to 0.5 m, errors in the northing direction ranging from -6m to 0.5m, and errors in the up direction ranging from -10 m to 15 m, except that the residuals in the northing and up directions, are jumped up to 27 m in northing and 43 m in up. After investigating the reasons of the jump in the smoothed data, the authors found that it is the result of cycle-slip, and MATLAB software is used for GPS cycle-slip processing [8], to detect that cycle-slip is shown clearly in Figure 13.

As already mentioned earlier, the filter is depending on the constancy variations in carrier phase measurement, so that if the cycle-slip occurs in carrier phase the filter must be initialized when computing smoothed pseudo-range. Figure 1 illustrates that satellite G17 elevation has dropped below the elevation mask, and Figure 13 represents the cycle slips detection of G17, for that reason the initial value for the algorithm in equation (1) must be initialized from start, and the filter at that point will give result worse than raw data.

8. Conclusion
The pseudo-range measurements generally contain high noise when taking measurements while the carrier phase contains less noise and more constant. The raw data are highly noisy and having errors in the easing direction ranging from -17m to 11m, errors in the northing direction ranging from -17m to 12m, and errors in the up direction ranging from -50 m to 30 m. The noisy behavior of the raw data have highly reduced after applying smoothing Hatch filter, and the smoothed data having errors in the easing direction ranging from -9m to 0.5 m, errors in the northing direction ranging from -6m to 0.5m, and errors in the up direction ranging from -10 m to 15 m, except we have jumped in the residuals in
the northing and up directions, that jumps up to 27 m in northing and 43 m in up. The cause of the jumps that satellite G17 elevation has dropped below the elevation mask and that presents cycle slips, the cycle slips will change the initial value for the filter algorithm and must be initialized from start, and the filter at that point will give result worse than raw data. The result also concluded that the Hatch filter can be used to detect and identify the location of cycle slips.

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