Terahertz-supported Untrusted UAV-Relaying: Secrecy Energy Efficiency Maximization via Trajectory and Communication Co-design

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Abstract—Unmanned aerial vehicles (UAVs) and Terahertz (THz) technology are envisioned to play paramount roles in next-generation wireless communications. Hence, this paper presents a novel secure UAV-assisted mobile relaying system operating at THz bands for data acquisition from multiple ground user equipments towards a destination. We assume that the UAV-mounted relay may act, besides providing relaying services, as a potential adversary called the untrusted UAV relay. To safeguard end-to-end communications, we present a secure two-phase transmission strategy with cooperative jamming. Then, we formulate an optimization problem in terms of a new measure — secrecy energy efficiency (SEE), defined as the ratio of achievable average secrecy rate to average system power consumption, which enables us to obtain the best possible security level while taking UAV’s inherent flight power limitation into account. This optimization problem leads to a joint design of key system parameters, including UAV’s trajectory and velocity, communication scheduling, and power allocations. Since the formulated problem is a mixed-integer non-convex optimization, and computationally intractable, we propose alternative algorithms to solve it efficiently via greedy/sequential block coordinated descent, successive convex approximation, and non-linear fractional programming techniques. Numerical results demonstrate significant SEE performance improvement of our designs when compared to other known benchmarks.

Index Terms—UAV communications, THz technology, untrusted relaying, physical layer security, trajectory design, resource allocation, convex optimization.

I. INTRODUCTION

The unmanned aerial vehicle (UAV) has recently been recognised as one of the major technological breakthroughs to be pervasively applied in 5G-and-beyond wireless communication networks supporting massive machine-type communications, internet of things (IoT), and artificial intelligent (AI) empowered communications [1]–[3]. Thanks to the unique characteristics of agility, on-demand swift deployment, versatility, and channel superiority amongst the other potentialities, UAV-aided wireless communications have recently attracted a great deal of research [4]–[7]. Despite numerous advantages, the open nature of air-ground (AG) links inevitably makes such systems vulnerable to malicious attacks such as eavesdropping. Accordingly, the security and confidentiality of such promising wireless communication systems are of utmost concern and undeniable requirements. To protect the confidentiality of UAV communications against hostile entities, one promising technique is the physical layer security (PLS) that uses the characteristics of wireless channels and applies communication techniques to combat attacks without complex encryption. A number of works have found leveraging the PLS in UAV-aided communications plausibly effective [8]–[22]. For example, PLS has been exploited in a wireless-powered UAV-relay system to combat eavesdropping via maximizing secrecy rate by a joint design of UAV’s position and resource allocation [10]. Other efforts were made to maximize the average secrecy rate (ASR) via joint trajectory and communication design for UAV-standalone wireless system [11]–[13], for double-UAV with external jamming [14]–[16], and for secure UAV-relaying scenarios [17]–[22]. The majority of previous research has deemed the UAV to be a fully authorized and legitimate communication node in UAV-aided relaying applications. However, when the UAV behaves as an untrusted relay, which is called untrusted UAV-relay (UUR), with the capability of information eavesdropping while assisting end-to-end communications (see [23], [24]), the system design becomes quite challenging and entirely different from the existing body of research.

Further, energy efficiency is another imperative need for UAV-aided communications due to UAVs’ inherent constraints on size, weight, and power (SWAP). Typically, the small-scale rotary-wing UAVs are powered via limited on-board batteries, leading to a restrictive operational lifetime, which undoubtedly impacts their overall system performance. Nonetheless, UAVs’ flight endurance, if properly designed, can be enhanced to a considerable extent [25]. Several works have studied the secrecy performance of UAV-aided systems considering the propulsion energy consumption constraint [26]–[29]. In [26], the authors have investigated ASR maximization for a cooperative dual-UAV secure data collection with propulsion energy limitation. Exploring the problem of secrecy energy efficiency (SEE) maximization for UAV-aided wireless systems is another research path [27]–[29]. The authors have designed both trajectory and resource allocation for the energy-efficient secure UAV communication system with the help of a multi-antenna UAV-jammer in [27]. Some appropriate system designs have been conducted for the SEE improvement of a single UAV-relay system [28], and a UAV-swarm multi-hop relaying scenario [29]. It is worth pointing out that all the aforementioned designs have only aimed to combat external terrestrial eavesdroppers.
On the other hand, owing to the ultra-broad bandwidth at the terahertz (THz) frequency range (0.1-10 THz), THz transmission has been acknowledged as a promising technology capable of catering an explosive growth of user demand of higher mobile traffic for future wireless systems. However, THz links incur severe path loss and high susceptibility to environmental blockage, and molecular absorption [1], [5], [30], which limit signal propagation distance and coverage range. To overcome the hindrances, one possible solution might be to explore UAV-aided communications in THz links. Notably, in the context of THz-UAV systems, few initial research studies have thus far been conducted. The coverage probability of the UAV-THz downlink communications was analyzed in [31], while [32] has explored a similar non-security scenario with a focus on minimizing communication delay by a joint design of the UAV’s location and power control. When it comes to security issues of such high-frequency systems, despite the widely-acknowledged improved resiliency against eavesdropping of THz links, the authors of [33] have characterized the possibility of eavesdropping attacks for such systems. Needless to mention that even considering negligible information leakage towards the external malicious eavesdroppers through THz transmissions, the scenarios involving untrusted relays, particularly the UAV systems, may still be vulnerable to eavesdropping. The appropriate design for such systems has yet to be understood; therefore, one needs to design novel frameworks to enable the efficient deployment of THz-UAV wireless systems.

I-A Our contributions

To the best of our knowledge, this is the first work addressing the energy-efficient secure design of a THz-UAV wireless communication system to guarantee confidentiality and transmission secrecy with the least system power consumption. Our detailed contributions are summarized below.

- We present an UUR-enabled wireless communication system for data collection from multiple ground user equipments (UEs) towards the base station (BS) over THz-based AG links. We adopt a secure two-phase transmission strategy using destination-assisted cooperative jamming to improve security.
- Then, we formulate a maximin optimization problem in terms of a new measure secrecy energy efficiency (SEE), defined as the ratio of achievable ASR to average system power consumption. This optimization problem leads to a joint design of key system parameters, including UUR’s trajectory and velocity, communication scheduling, and network transmission power allocations.
- Since the optimization problem is originally intractable due to non-convexity, we decompose it into four subproblems and then solve each via successive convex approximation (SCA) or Dinkelbach fractional programming techniques. Further, we propose two computationally efficient algorithms according to the sequential and maximum improvement (MI) based block coordinate descent (BCD) approaches with guaranteed convergence to at least a suboptimal solution. We also conduct computational and complexity analysis and show that our solution can be obtained in polynomial time order, making it applicable to the energy-hungry UAV-based scenarios.
- We conduct extensive simulations to verify the analyses and demonstrate the effectiveness of our proposed designs in terms of SEE compared to some other benchmarks, without communication resource allocation design or trajectory and velocity optimization and ignoring flight power consumption. We also investigate the impact of some fundamental setting parameters such as the flight mission time and the molecular adsorption factor on the overall system secrecy performance.

The rest of the paper is organized as follows. Section II introduces system model and formulates the problem of interest. In Section III, we present efficient iterative algorithms to solve the optimization problem, followed by numerical results and discussions given in Section IV. Finally, we draw conclusions in Section V.

II SYSTEM MODEL AND PROBLEM FORMULATION

We consider a UAV-enabled wireless communication system for data collection from a set of $K$ ground UEs towards a BS via a UAV-assisted mobile amplify-and-forward (AF) relay, as shown in Fig. 1. Here we assume that there are no reliable direct links from UEs to BS (see [24], [34] and references therein), and all nodes are equipped with a single antenna\(^1\), operating in half-duplex mode. Therefore, a UAV-relay is employed to assist end-to-end communications [18]; nonetheless, the UAV-relay may not be fully authorized to access collected confidential information and may conduct malicious eavesdropping, i.e., an UUR [23]. Thus secure data transmission is in demand.

Without loss of generality, we consider 3D Cartesian coordinate system, where the BS’s horizontal coordinate is located at the origin $q_0 = [0, 0] \in \mathbb{R}^{1 \times 2}$, and the ground UEs with horizontal coordinates $q_k = [x_k, y_k] \in \mathbb{R}^{1 \times 2}$ for $\forall k \in K$, where $K = \{1, 2, \cdots, K\}$, are randomly distributed in a circular annulus region with the inner radius $R_1$ and outer radius $R_2$ and the coordinates are assumed to be known in

\(^1\)This work can be readily extended to consider multiple-antenna transceivers and/or massive multiple-input multiple-output scenarios. Such extensions will be considered in our future research.
prior. Here, \( R_1 \) is considered to be the largest distance at which a reliable uplink transmission can be obtained, while beyond \( R_1 \) in our case implies no direct link between UE and BS. Further, \( R_2 \) indicates the boundary of the permitted flying region for the UAV to provide communication service.

We also consider that UAV flies from and back to the same specific point over the region of interest for a duration of \( T \) seconds in order to provide relaying services to all UEs with fairness. This specific point may refer to the checkup point wherein the UAV gets recharged and physically examined to maintain its service. Assuming that UAV flies at a fixed altitude \(^2\) \( H \) meters whose instantaneous horizontal coordinate and velocity is represented by \( q(t) = [x(t), y(t)] \) and \( v(t) = \frac{dq(t)}{dt} \), respectively, where \( 0 \leq t \leq T \). For the ease of analysis, we adopt the time-slotted system such that the flight duration \( T \) is equally discretized into \( N \) sufficiently small time slots of duration \( \delta t = \frac{T}{N} \). Hence, the UAV’s horizontal location at time slot \( n \in N = \{1, \ldots, N\} \) can be approximated by \( q[n] = [x[n], y[n]] \). This assumption is valid when \( \delta t \exp(-\alpha_f d_{ku}[n]) \leq H \), wherein \( \delta t \) denotes the maximum UAV’s displacement per time slot.

II-A Channel model

We assume that the AG links are over THz channels, which are mainly affected by both free space spreading loss and the molecular absorption according to [32]. Further, as per widely used assumption in the body of existing UAV literature, e.g., [11], [27], [35], the Doppler effect due to the UAV mobility is also considered to be perfectly compensated for ease of exposition in the sequel. Therefore, assuming that at each time slot \( n \) the channel state information is regarded static due to adequately small \( \delta t \), we adopt the line-of-sight (LoS) dominant time-varying THz channel power gain model, similar to [32], between the UUR and any UE \( k \in K \) as

\[
h_{uk}[n] = \beta_0 \exp\left(-\alpha_f d_{ku}[n]\right) d_{ku}[n]^{-2}, \quad \forall n
\]

where \( d_{ku}[n] \) denotes the Euclidean distance between the UUR and the \( k \)-th UE, given by

\[
d_{ku}[n] = \sqrt{||q[n] - q_k||^2 + H^2}, \quad \forall n
\]

Note that the multiplicative term \( \exp(-\alpha_f d_{ku}[n]) \) in (1) is the indication of excessive path loss of THz links due to water vapor molecular absorption effect\(^3\), wherein \( \alpha_f \) is the frequency and chemical composition of air dependent adsorption factor [38]. It should be also pointed out that the main cause of absorption loss in THz frequency ranges is the water vapor molecules that causes discrete, but deterministic loss to the signals in the frequency domain. Further, \( \beta_0 \triangleq \frac{C}{4\pi f^2} \) denotes the reference channel power gain at unit distance, wherein \( C \) is the speed of light, \( f \) is the operation frequency. Likewise, the THz channel power gain between the UUR and the BS can be written as \( h_{bu}[n] = \beta_0 \exp\left(-\alpha_f d_{bu}[n]\right) d_{bu}[n]^{-2} \), wherein \( d_{bu}[n] = \sqrt{||q[n] - q_b||^2 + H_b^2}, \quad \forall n \).

II-B Constraints on user scheduling, power, UAV’s mobility

We adopt the time division multiple access (TDMA) protocol for multiuser relaying services, wherein UUR serves at most one enabled UE at \( n \)-th time slot, while the other ground UEs keep silence. Therefore, letting \( \zeta_k[n] \) be a binary user scheduling variable for UE \( k \in K \) at time slot \( n \in N \), we have the user scheduling constraints

\[
C1: \quad \zeta_k[n] \in \{0, 1\}, \quad \forall k, n
\]

\[
C2: \quad \sum_{k \in K} \zeta_k[n] \leq 1,
\]

where \( \zeta_k[n] = 1 \) if user \( k \) is scheduled at time slot \( n \), and \( \zeta_k[n] = 0 \), otherwise. Further, the transmit powers of the UUR, the BS, and \( k \)-th user in time slot \( n \), denoted respectively as \( p_u[n], p_b[n], \) and \( p_k[n] \), are generally subject to average and peak transmit powers given as

\[
C3: \quad \frac{1}{N} \sum_{n=1}^{N} \sum_{k=1}^{K} \zeta_k[n] p_k[n] \leq p_u^{ave},
\]

\[
C4: \quad 0 \leq p_b[n] \leq p_b^{max}, \quad \forall k, n
\]

\[
C5: \quad \frac{1}{N} \sum_{n=1}^{N} p_k[n] \leq p_u^{ave},
\]

\[
C6: \quad 0 \leq p_k[n] \leq p_k^{max}, \quad \forall n
\]

\[
C7: \quad \frac{1}{N} \sum_{n=1}^{N} p_b[n] \leq p_b^{ave},
\]

\[
C8: \quad 0 \leq p_b[n] \leq p_b^{max}, \quad \forall n
\]

where sets \( \{p_u^{ave}, p_b^{ave}, p_k^{ave}, \forall k\} \) and \( \{p_u^{max}, p_b^{max}, p_k^{max}, \forall k\} \) represent the corresponding average and maximum power constraints of the network nodes.

The mechanical power consumption of the energy-limited UAV due to high demand of propulsion energy for aerial operation with fixed altitude level flight can be approximately given by [25]

\[
P_f[n] = P_0 \left(1 + \frac{3||\nabla[n]||^2}{\Omega^2 R_u^2} \right) + P_i \left(1 + \frac{||\nabla[n]||^4}{4\rho^2} - \frac{||\nabla[n]||^2}{2\rho^2}\right), \quad \forall n
\]

wherein \( \nabla[n] \) is the UAV’s instantaneous velocity at time slot \( n \), \( P_0 \) and \( P_i \) are two constants representing UAV’s blade profile power and induced power in hovering mode,
respectively, $\Omega_u$ and $R_u$ are the UAV’s blade angular velocity in Radian per second (rad/s) and its rotor radius in meter (m), $d_f, \rho, s,$ and $A$ indicate the unit-less fuselage drag ratio, air density in kg/m$^3$, rotor solidity, and rotor disk area in m$^2$, respectively. Further, the average rotor induced velocity in hovering is shown as $v_0$. And we have the average flight power consumption constraint as

$$C9 : \frac{1}{N} \sum_{n=1}^{N} P_f[n] \leq \bar{P}_{lim},$$

(12)

wherein $\bar{P}_{lim}$ indicates the UAV’s average propulsion power budget, which is proportional to the UAV’s onboard battery capacity. Therefore, it should be required that the total consumed propulsion energy by the UAV over $N$ time slots be less than such limit in order for network functioning. Further, the considered scenario should be subject to UAV’s mobility constraints in terms of initial and final locations for cyclic path, in-flight maximum displacement per time slot for satisfying channel invariant assumption, and permitted flying region as

$$C10 : \mathbf{q}[0] = \mathbf{q}[N] = \mathbf{q}_f,$$

$$C11 : \mathbf{q}[n+1] = \mathbf{q}[n] + \mathbf{v}[n]\delta t, \ \forall n \setminus N$$

$$C12 : \|\mathbf{v}[n]\| \leq v_u^{\text{max}}, \ \forall n$$

$$C13 : \|\mathbf{v}[n+1] - \mathbf{v}[n]\| \leq a_u^{\text{max}}, \ \forall n \setminus N$$

$$C14 : \|\mathbf{q}[n] - \mathbf{q}_0\| \leq R_2, \ \forall n$$

(13)

wherein $\mathbf{q}_f$ indicates UAV’s initial and final location per flight, $v_u^{\text{max}}$ and $a_u^{\text{max}}$ are the UAV’s maximum speed and acceleration, respectively.

II-C Secure transmission strategy, problem formulation

For the purpose of wireless security, we adopt a secure two-phase transmission strategy with destination-assisted cooperative jamming (DBCJ) technique similar to [11], [18], [24]. In the first phase, at each time slot $n$, the scheduled UE sends confidential information to UUR, and simultaneously the BS jams UUR. In the second phase, UUR forwards the received signals using AF relaying protocol to the BS.

Under such setting, given the equally shared communication bandwidth $B$ Hz, the achievable end-to-end instantaneous data rate in bits-per-second (bps) from the $k$-th UE at time slot $n$ is

$$R_b^k[n] = \zeta_b[n]B\log_2\left(1 + \frac{p_k[n]g_{ku}[n]p_u[n]g_{bu}[n]}{(p_k[n]+p_u[n])g_{bu}[n]+p_k[n]g_{ku}[n]+1}\right).$$

(14)

Then the UUR may overhear the confidential information with an achievable wiretap secrecy rate per Hz at time slot $n$ as

$$R_u^k[n] = \zeta_u[n]B\log_2\left(1 + \frac{p_k[n]g_{ku}[n]}{p_u[n]g_{bu}[n]+1}\right),$$

(15)

wherein $g_{ku}[n] \triangleq \frac{h_{ku}[n]}{N_0}$ and $g_{bu}[n] \triangleq \frac{h_{bu}[n]}{N_0}$, and $N_0 \triangleq B\sigma_u^2(\text{b})$ indicate the power of additive white Gaussian noise (AWGN) at the receivers, wherein $\sigma_u^2$ and $\sigma_b^2$ which are assumed equal for simplicity, denote the power spectral density (PSD) at the UUR and the BS.

We adopt the ASR as one of the key secrecy metrics and the ASR of $k$-th UE at time slot $N$ is

$$\bar{R}_{sec}^k = \frac{1}{N} \sum_{n=1}^{N} \left[ \frac{1}{2} \left( R_b^k[n] - R_u^k[n] \right)^+ \right] \text{ bps}$$

(16)

wherein $(x)^+ \triangleq \max\{x, 0\}$, and the ratio $\frac{1}{2}$ is due to the fact that secure transmission is done in two phases of equal duration at each time slot. The achievable average information bits can securely be exchanged between $k$-th UE and BS is $B_{sec}^k = \delta_1 \sum_{n=1}^{N} \bar{R}_{sec}^k[n]$.

To fully exploit the capability of aerial platforms for communication, the limited energy resource must be considered in system design. In practice, the UAV’s propulsion power consumption is much higher than those used for UEs’ signal transmission, BS’s jamming, and receiver processing. Hence, we approximate the network’s total power consumption mainly from UAV’s propulsion only. Consequently, for the secrecy metric, we define secrecy energy efficiency (SEE) of the proposed scheme for the $k$-th UE as the ratio of the achievable ASR to the approximated average system power consumption,

$$\text{SEE}^k(\zeta, Q, P) \triangleq \frac{\bar{R}_{sec}^k}{\frac{1}{N} \sum_{n=1}^{N} P_f[n]}, \text{ bits/Joule}$$

(17)

wherein the user scheduling set $\zeta = \{\zeta_k[n], \forall n, k\}$, UAV’s location and velocity set $Q = \{\mathbf{q}[n], \mathbf{v}[n], \forall n\}$, and network transmit power set $P = \{P_u = \{p_u[n], \forall k, n\}, P_b = \{p_b[n], \forall n\}\}$ are the involving parameters.

Remark 1. It is worth pointing out, for later analysis, that we use normalized metrics, i.e., the numerator and denominator of (17) divided by $B$ and $\bar{P}_{lim}$, respectively, to balance well numerical values and both metrics in $\text{SEE}$.

To design the network to obtain the best performance among UEs and provide fairness support to all UEs given UAV’s stringent on-board battery, we maximize UEs’ minimum SEE by

$$\begin{align*}
(P) : \max_{\zeta, Q, P} \min_{k \in K} \text{SEE}^k(\zeta, Q, P) \\
\text{s.t.} \ C1 - C14,
\end{align*}$$

(18)

We note that the problem $(P)$ is a mixed-integer non-convex optimization problem, which is too hard to solve optimally. The non-convexity is mainly due to the non-concave objective function with respect to (w.r.t) the optimization variables, and also having the non-smoothness operator $(\cdot)^+$ and the non-convex constraints $(C1), (C3),$ and $(C9).$ Indeed, the major challenge in solving (P) arises from the binary user scheduling constraint C1 and the highly coupled optimization variables in the objective function in fractional form. To make it tractable, we first remove the operator $(\cdot)^+$ from the numerator of the objective function, since the value of the objective function should be non-negative at the optimal point; otherwise, one can set, e.g., $P_k = 0, \forall k$ and get zero SEE performance without modifying the original problem. Nonetheless, having at least a differentiable objective function, the problem is still non-convex, thereby no standard approach to solve it efficiently. To remedy this issue, we first handle the binary constraint
as per the approach in [39], by relaxing C1 into continuous constraint. Then, we propose some computationally efficient algorithms to iteratively solve a sequence of approximated convex sub-problems by adopting several techniques such as block coordinated descent (BCD), successive convex approximation (SCA), and nonlinear fractional Dinkelbach programming, discussed below.

III PROPOSED ITERATIVE SOLUTION

In this section, we split the problem (P) into four subproblems with different blocks of variables, then solve each block by block, while keeping the other blocks unchanged. Specifically, we delve into solving the joint user scheduling and transmit power optimization problem in the form of

\[
\begin{align*}
(P_1) & : \max_{\psi, \mathbf{P}_k} \psi \\
& \text{s.t.} \sum_{n=1}^{N} \tilde{\zeta}_k[n] \ln \left(1 + \frac{C_n \tilde{p}_k[n]}{\tilde{p}_k[n] + D_{k,n} \tilde{\zeta}_k[n]} \right) - \tilde{\zeta}_k[n] \ln \left(1 + B_{k,n} \frac{\tilde{p}_k[n]}{\tilde{\zeta}_k[n]} \right) \geq \frac{\psi}{\lambda_1}, \quad \forall k \\
& \frac{1}{N} \sum_{n=1}^{N} \sum_{k=1}^{K} \tilde{p}_k[n] \leq \tilde{p}_k^{\text{ave}}, \\
& 0 \leq \tilde{p}_k[n] \leq \tilde{p}_k^{\text{max}}, \quad \forall k, n \\
& \sum_{k \in K} \tilde{\zeta}_k[n] \leq 1, \quad \forall n \\
& 0 \leq \tilde{\zeta}_k[n] \leq 1, \quad \forall k, n
\end{align*}
\]

where

\[
\lambda_1 = \frac{1}{2 \ln 2 \sum_{j=1}^{N} P_j[n]} \quad B_{k,n} = \frac{g_{ku}[n]}{p_b[n] g_{ub}[n] + 1} \\
C_n = g_{bu}[n] p_u[n] \quad D_{k,n} = \frac{g_{bu}[n] p_u[n] + p_b[n]}{g_{ku}[n] + 1}.
\]

Note that the constraint (19a) should be met with equality at the optimal point; otherwise, the value of the objective function in problem (P1) can still be increased by increasing \( \psi \), which violates the optimality. The subproblem (P1) is still non-convex due to non-convexity of the constraint (19a) and for general \( N \), it is indeed NP-hard. Therefore, we cannot solve it efficiently. To handle (19a), we first present Lemma 1 below.

**Lemma 1.** Let’s define the bivariant functions \( Z_1(x, y; a, b) \triangleq x \ln(1 + \frac{ay}{x + b}) \) and \( Z_2(x, y; c) \triangleq x \ln(1 + \frac{y}{x}) \) over the domain \( x, y > 0 \) with the positive constants, i.e., \( a, b, c > 0 \). Both \( Z_1 \) and \( Z_2 \) are jointly concave w.r.t the variables \( x \) and \( y \). Additionally, the inequality below near the given point \((x_0, y_0)\) always holds with tightness:

\[
Z_2(x, y; c) \leq x_0 (1 + \frac{cx_0}{y_0}) + \left( \ln \left(1 + \frac{y_0}{x_0} \right) - \frac{cy_0}{x_0 + cy_0} \right) (x - x_0) + \left( \frac{cx_0}{x_0 + cy_0} \right) (y - y_0) \triangleq f_1^{\text{ub}}(x, y; x_0, y_0, c),
\]

(20)

**Proof.** Please see Appendix A.

Using Lemma 1, it can be identified that both Terms I and II in (19a) are concave w.r.t the optimization variables \( \psi, \tilde{\zeta}_k[n] \) and \( \mathbf{P}_k \), since the summation operator preserves the convexity. The non-convexity of the left-hand-side (LHS) expression is in the form of concave-minus-concave. Then using (20) and applying the SCA technique, we approximate the non-convex constraint with the corresponding approximate convex one at each iteration. Given the local point \((\mathbf{P}_k^{(l)}, \tilde{\zeta}_k^{(l)})\) in the \( l \)-th iteration, we can reformulate (P1) as follows.

\[
(P_1.1) : \max_{\psi, \tilde{\zeta}_k[n], \mathbf{P}_k} \psi \\
\text{s.t.} \sum_{n=1}^{N} \tilde{\zeta}_k[n] \ln \left(1 + \frac{C_n \tilde{p}_k[n]}{\tilde{p}_k[n] + D_{k,n} \tilde{\zeta}_k[n]} \right) - \tilde{\zeta}_k[n] \ln \left(1 + B_{k,n} \frac{\tilde{p}_k[n]}{\tilde{\zeta}_k[n]} \right) \geq \frac{\psi}{\lambda_1}, \quad \forall k \\
\frac{1}{N} \sum_{n=1}^{N} \sum_{k=1}^{K} \tilde{p}_k[n] \leq \tilde{p}_k^{\text{ave}}, \\
0 \leq \tilde{p}_k[n] \leq \tilde{p}_k^{\text{max}}, \quad \forall k, n \\
\sum_{k \in K} \tilde{\zeta}_k[n] \leq 1, \quad \forall n \\
0 \leq \tilde{\zeta}_k[n] \leq 1, \quad \forall k, n
\]

Since the reformulated problem (P1.1) is convex w.r.t the optimization variables \( \{\psi, \tilde{\zeta}_k[n], \mathbf{P}_k\} \), it can be solved efficiently via CVX using the interior-point method [40]. Having solved this subproblem, we can then obtain the optimized value of \( \mathbf{P}_k = \{p_k[n] = \tilde{p}_k[n] \tilde{\zeta}_k[n], \forall k, n \} \). Further, once the solution of the overall algorithm is obtained, we can reconstruct the corresponding binary solution of \( \tilde{\zeta}_k \), according to the method in [39], or using \( \tilde{\zeta}_k[n] = \{\tilde{\zeta}_k[n] \geq 0.5\}, \forall k, n \).

**Remark 2.** The formulated convex optimization model given in (P1.1), though being convex, cannot be directly accepted by CVX, as it does not follow the disciplined convex programming (DCP) ruleset required. Given that the relative entropy function \( E_{\text{rel}}(x, y) = x \log(x/y), x, y > 0 \) is convex and accepted
by CVX, we can rewrite concave function $Z_1(x, y; a, b)$ (or the equivalent expression in the constraint (19a)), as

$$Z_1(x, y; a, b) = \frac{1}{b} \left[ (y + bx) \ln \left( 1 + \frac{ay}{y + bx} \right) - \frac{1}{a} \right] \times ay \ln \left( 1 + \frac{ay}{y + bx} \right) \right] \equiv - \left( \frac{1 + a}{ab} \right) E_{rel} \left( y + bx, (a + 1)y + bx \right) - \left( \frac{1}{ab} \right) E_{rel} \left( (a + 1)y + bx, y + bx \right),$$

where the equality (a) follows from the following relations between different form of logarithmic functions and the convex relative entropy function given by

$$x \ln \left( 1 + \frac{y}{x} \right) = -E_{rel}(x, x + y), \quad (23)$$

$$x \ln \left( 1 + \frac{x}{y} \right) = E_{rel}(x + y, y) + E_{rel}(y, x + y), \quad (24)$$

wherein (23) and (24) are jointly concave and convex w.r.t the joint variables $(x, y)$ over $x, y > 0$, respectively.

In terms of computational cost, here we have $(2NK + 1)$ optimization variables and $(K(2N + 1) + 1)$ convex constraints. Assume the convergence accuracy of SCA algorithm employed for solving this sub-problem is $\varepsilon_1$, the complexity of solving approximated sub-problem (P1.1) can be attained as $O \left( (2NK + 1)^2(K(2N + 1) + 1)^{1.5} \log_2 \left( \frac{1}{\varepsilon_1} \right) \right)$.

**III-B Relaying power optimization**

The corresponding sub-problem for optimizing UUR’s relaying power can be rewritten, introducing the slack variable $\psi$, as

$$(P2) : \max_{\psi, P_a} \psi$$

s.t. $\sum_{n=1}^{N} \left[ \lambda_{k,n} \ln \left( 1 + \frac{E_{k,n}p_{a,n}}{p_a[n] + F_{k,n}} \right) \right] - G_k \geq \psi, \quad \forall k \quad (25a)$

$$= \frac{1}{N} \sum_{n=1}^{N} p_{a}[n] \leq p_a^{ave},$$

$$0 \leq p_{a}[n] \leq p_a^{max}, \forall n \quad (25c)$$

where $\lambda_{k,n} = \frac{\tilde{c}_k[n]}{2 \ln 2 \sum_{n=1}^{N} \tilde{r}_{f,n}[n]}$, $E_{k,n} = p_k[n]g_{k,n}[n]$ and $F_{k,n} = \frac{p_k[n]g_{k,n}[n] + p_b[n]g_{b,n}[n] + 1}{g_{b,n}[n]}$

$$G_k = \frac{N}{2} \sum_{n=1}^{N} \tilde{c}_k[n] \log_2 \left( 1 + \frac{p_a[n]g_{a,n}[n]}{p_a[n]g_{a,n}[n] + 1} \right), \quad \forall k, n$$

Note that sub-problem (P2) is a convex optimization problem due to having an affine objective function and all convex constraints, following from Lemma 2 introduced below.

**Lemma 2.** Let define the function $f_2(x; a, b, c, d)$, $x \geq 0$ with positive constant values $a, b, c, d > 0$ as

$$f_2(x; a, b, c, d) \triangleq \ln \left( 1 + \frac{ax + b}{cx + d} \right)$$

$$\triangleq \ln \left( 1 + \frac{a}{c} - \frac{ab - bc}{c^2x + cd} \right),$$

where $f_2(x)$ is concave subject to the condition $ad \geq bc$, following from the fact that the function $\ln(1 + qx), q \geq 0, x > 0$ is concave w.r.t $x$, whose extended-value extension is non-decreasing and $h(x) = -\frac{1}{x}$ is also concave; therefore, $(f \circ g)(x)$ is concave. Note that the last equality of (26) represents the understandable reformulation of the function $f_2(x; a, b, c, d)$ by the CVX optimization toolbox. We also stress that for any given point $x_0$, there is a unique convex function $f_2^b(x; x_0, a, b, c, d)$ defined as

$$f_2^b(x; x_0, a, b, c, d) \triangleq \ln(1 + \frac{ax_0 + b}{cx_0 + d}$$

$$+ \frac{(ad - bc)(x - x_0)}{(cx_0 + d)(b + d + (a+c)x_0)},$$

such that $f_2^b(x; x_0, a, b, c, d)$ serves as a global lower-bound of $f_2(x)$, i.e., $f_2(x) \geq f_2^b(x; x_0, a, b, c, d)$ [41].

Consequently, one can solve subproblem (P2) efficiently using CVX. Here, we have $(N + 1)$ optimization variables and $(N + K + 1)$ convex constraints. Assuming the convergence accuracy of interior-point algorithm employed for solving this convex problem with logarithmic cone is $\varepsilon_2$, the complexity cost of solving sub-problem (P2) can be obtained as $O \left( (N + 1)^2(N + K + 1)^{1.5} \log_2 \left( \frac{1}{\varepsilon_2} \right) \right)$.

**III-C Jamming power optimization**

Keeping the other variables unchanged and taking the slack variable $\psi$, the BS’s jamming power optimization sub-problem is given as

$$(P3) : \max_{\psi, P_b} \psi$$

s.t. $\sum_{n=1}^{N} \lambda_{k,n} \ln \left( 1 + \frac{H_{k,n}}{p_b[n] + I_n} \right) - \ln \left( 1 + \frac{J_{k,n}}{p_b[n] + K_n} \right) \geq \psi, \quad \forall k \quad (28b)$

$$= \frac{1}{N} \sum_{n=1}^{N} p_b[n] \leq p_b^{ave},$$

$$0 \leq p_b[n] \leq p_b^{max}, \forall n \quad (28c)$$

where $H_{k,n} = g_{k,n}[n]p_{k}[n]p_{b,n}[n]$, $J_{k,n} = \frac{p_{k}[n]g_{k,n}[n]}{g_{b,n}[n]}$, $K_n = \frac{1}{g_{b,n}[n]}$, and $I_{k,n} = \frac{p_{k}[n]g_{k,n}[n] + p_{b,n}[n]g_{b,n}[n] + 1}{g_{b,n}[n]}$. Notice that subproblem (P3) is non-convex due to non-convex constraint (28b), which is in the form of convex-minus-convex according to [15, Lemma 1]. Therefore, we apply SCA such that for a given local point $P_b^{(i)}$ in $l$-th iteration. We approximate
the first convex term with the global underestimator concave expression and obtain the convex reformulation as

\[
(P3.1) : \max_{\psi, P_b} \psi \quad \text{s.t.} \quad \sum_{n=1}^{N} \lambda_{k,n} \left[ f_3(p_b[n]; p_{n,b}^{(l)}, H_{k,n}, I_{k,n}) - \ln \left( 1 + \frac{H_{k,n}}{p_{n,b}^{(l)} + I_{k,n}} \right) \right] \geq \psi, \quad \forall k \quad (28c) & (28d)
\]

wherein

\[
f_3(p_b[n]; p_{n,b}^{(l)}, H_{k,n}, I_{k,n}) = \ln \left( 1 + \frac{H_{k,n}}{p_{n,b}^{(l)} + I_{k,n}} \right) - \frac{H_{k,n}}{p_{n,b}^{(l)} + H_{k,n} + I_{k,n}} (p_b[n] - p_{n,b}^{(l)}). \quad (30)
\]

Since subproblem (P3.1) is convex, we can solve it efficiently using CVX. Here, we have \( N + 1 \) optimization variables and \((N + K + 1)\) convex constraints. Assuming the accuracy of SCA algorithm for solving this problem is \( \varepsilon_3 \), the complexity of solving approximated sub-problem (P3.1) can, therefore, be represented as \( O \left( (N + 1)^2 (N + K + 1)^{1.5} \log_2 \left( \frac{1}{\varepsilon_3} \right) \right) \).

### III-D Joint trajectory and velocity optimization

Now, we optimize the trajectory \( q \) and velocity \( v \) of the UUR while keeping the transmit power allocation and user scheduling sets \((P, \mathcal{Q})\) fixed. Therefore, the corresponding subproblem can be given as

\[
(P4) : \max_{q} \min_{k \in \mathcal{K}} \frac{\bar{R}^k_{sec}(q, v)}{P_f(v)} \quad \text{s.t.} \quad C9 - C14 \quad (31b)
\]

wherein \( \bar{P}_f(v) = \frac{1}{N} \sum_{n=1}^{N} P_f[n] \). In order to solve subproblem (P4), we should maximize every single fractional terms of \( \left\{ \frac{R^k_{sec}(q, v)}{P_f(v)} \right\}, \forall k \) subject to the given constraint (31b). In light of this, let \( \lambda^* \) be the maximum SEE of sub-problem (P4) with solution set \((q^*, v^*)\) given by

\[
\lambda^* = \max_{q, v \in \mathcal{F}} \min_{k \in \mathcal{K}} \frac{\bar{R}^k_{sec}(q, v)}{P_f(v)} \quad \text{s.t.} \quad C9 - C14
\]

\[
\min_{k \in \mathcal{K}} \frac{\bar{R}^k_{sec}(q^*, v^*)}{P_f(v^*)}, \quad (32)
\]

wherein \( \mathcal{F} \) represents the feasible set spanned by the constraint (31b). Applying nonlinear fractional Dinkelbach programming theory [42], the objective function of problem (P4) can be equivalently transformed into a subtractive version such that the optimal value of \( \lambda^* \) can be achieved iff

\[
\max_{q, v \in \mathcal{F}, k \in \mathcal{K}} \frac{\bar{R}^k_{sec}(q, v)}{P_f(v)} - \lambda^* \bar{P}_f(v) = \min_{k \in \mathcal{K}} \frac{\bar{R}^k_{sec}(q^*, v^*)}{P_f(v^*)} - \lambda^* \bar{P}_f(v^*) = 0, \quad (33)
\]

Thus, we can optimize the equivalent problem to obtain the optimal solution of \( Q \), via solving the reformulated problem as

\[
(P4.1) : \max_{q, v} \min_{k \in \mathcal{K}} \bar{R}^k_{sec}(q, v) - \lambda^{(m)} \bar{P}_f(v) \quad \text{s.t.} \quad (31b)
\]

\[
\text{wherein } \lambda^{(m)} = \min_{v \in \mathcal{V}} \bar{R}^k_{sec}(q^{(m)}, v^{(m)}) \text{ showing the value of } \lambda \text{ in the } m \text{-th iteration of the Dinkelbach algorithm. Reformulated problem (P4.1) is still non-convex due to non-convex objective function and constraint (C9) which can be dealt with as follows.}
\]

By introducing the slack variables \( \psi \) and \( \mu = \{\mu[n]\}_{n=1}^{N} \) such that

\[
\mu[n] = \left( \sqrt{1 + \frac{\|v[n]\|^4}{4v_0^4}} - \frac{\|v[n]\|^2}{2v_0^2} \right)^{1/2}, \quad \forall n \quad (35)
\]

we can relax the problem (P4.1) to the one with the approximately equivalent but enjoying concave objective function as

\[
(P4.2) : \max_{q, v, \mu} \psi - \lambda^{(m)} \omega \quad \text{s.t.} \quad C10 - C14
\]

\[
\omega \leq \bar{P}_{fin}, \quad \mu[n] \geq 0, \quad \forall n \quad (36d)
\]

\[
\mu^2[n] + \frac{\|v[n]\|^2}{v_0^2} \geq \frac{1}{\mu^2[n]}, \quad \forall n \quad (36e)
\]

\[
\bar{R}^k_{sec} \geq \psi, \quad \forall k \quad (36f)
\]

wherein \( \omega \approx P_f(v) = \frac{1}{N} \sum_{n=1}^{N} P_f[n] \), with \( \{P_f[n], \forall n\} \) serving as a global convex upper-bound of (11), defined as

\[
P_f[n] = P_0 \left( 1 + \frac{2\|v[n]\|^2}{Q_n^2 R_n^2} \right) + 2I_0 \beta A \|v[n]\|^2 + P_1 \mu[n], \quad (37)
\]

Note that constraint (36e) must be met with equality at the optimal point, because \( \mu[n] \) can be otherwise decreased, resulting in an increase of the value of the objective function, which of course, violates the optimality. Plus, we also point out that the objective function, the constraints C10 – C14, and (36c) are now convex. However, the problem (P4.2) is still unsolvable due to the generated extra non-convex constraints (36e) and (36f). Note that the LHS expression of (36e); i.e., summation of norm-square components, is jointly convex w.r.t the variables \( \mu[n] \) and \( v[n] \). Owing to the fact that the right-hand-side (RHS) of (36e) is convex, since the second derivative of the inverse-square function \( \frac{1}{\mu^2[n]} \) is non-negative; therefore, by replacing the LHS with the corresponding global concave lowerbound using first-order Taylor expansion at the local given point \( (\mu_n^{(m)}, v_n^{(m)}) \) with superscript \( m \) indicating the iteration index of fractional Dinkelbach programming, we can reach the approximate convex constraint, associated with (36e), as

\[
-(\mu_n^{(m)})^2 + 2\mu_n^{(m)} \mu[n] + \frac{1}{v_0^2} \times \left( -\|v_n^{(m)}\|^2 + 2\nu_n^{(m)} v^1[n] \right) \geq \frac{1}{\mu^2[n]}, \quad \forall n \quad (38)
\]
Now, we deal with the last non-convex constraint (36f) by introducing the slack variables $s = \{s_k[n], \forall k, n\}$, $r = \{r_k[n], \forall k, n\}$, and $w = \{w[n], \forall n\}$, rewriting problem (P4.2) as

(P4.3) \[ \max_{\psi, \mu, q, v, s, r, w} \psi - \lambda^{(m)} \omega \]

s.t. \[(36b), (36c), (36d), (38) \]

\[
\frac{1}{2} N \ln 2 \sum_{n=1}^{N} \left[ \ln \left( 1 + \frac{k_0 s_k[n]}{k_3 w[n]} + k_1 w[n] + \epsilon \right) \right] - \ln \left( 1 + \frac{k_2 s_k[n]}{k_3 w[n]} \right) \geq \psi, \quad \forall k, n \]

\[
\left( \frac{1}{2} N \right) \sum_{n=1}^{N} \left( \frac{k_2}{k_1} s_k[n] + k_1 w[n] \right) \leq \psi, \quad \forall k, n \]

By introducing the slack variables $u = \{u_k[n], \forall k, n\}$, and using Lemma 3, we can approximate the non-convex problem (P4.3) with a more tractable reformulation given as

(P4.4) \[ \max_{\psi, \mu, q, v, s, r, w, u} \psi - \lambda^{(m)} \omega \]

s.t. \[(39b), (39e), (39f) \]

\[
\frac{1}{2} N \ln 2 \sum_{n=1}^{N} \zeta_k[n] \left( f_{41}^{lb} (r_k[n], w[n]; t_k[n], w^{(m)}[n], k_u[k], k_1) - \ln \left( 1 + \frac{k_2 s_k[n]}{k_3 w[n]} \right) \right) + f_{42}^{lb} (w[n]; w^{(m)}[n]) \geq \psi, \quad \forall k, n \]

\[
\frac{1}{2} N \sum_{n=1}^{N} \left( \frac{k_2}{k_1} s_k[n] + k_1 w[n] \right) \leq \psi, \quad \forall k, n \]

where $\{r_k[n], w^{(m)}[n], v_k[n], u_k[n]\}$ are the value set of slack variables $(r, w, u)$ in the $m$-th iteration of Dinkelbach’s algorithm. Finally, since the last constraint is non-convex, we apply [11, Lemma 3] to approximate it with the corresponding convex constraint using the SCA approach, and obtain an approximate convex reformulation of (P4.4) as

(P4.5) \[ \max_{\psi, \mu, q, v, s, r, w, u} \psi - \lambda^{(m)} \omega \]

s.t. \[(44b), (44c), (44d), (44e) \]

\[
- \frac{\|q^{(m)}[n]\|^2}{2} + 2 \left( \frac{q^{(m)}[n] - q_k}{q[n]} \right) \leq \psi, \quad \forall n \]

\[
\|q[n] - q_k\|^2 + H^2 \geq \frac{u_k[n]}{\epsilon}, \quad \forall k, n \]

We remark the fruitful lemma below.

Lemma 3. Let define the bivariate functions $f_{41}(x, y; a, b)$ and $f_{42}(x, y; c, d)$, and univariate functions $f_{43}(x; e)$ and $f_{44}(x; c)$ with positive constants $a, b, c, d, p, r > 0$ as

\[
f_{41}(x, y; a, b) = \ln \left( 1 + \frac{1}{ax + by} \right), \quad f_{43}(x; p) = x^2 \exp(px), \]

\[
f_{42}(x, y; c, d) = \ln \left( 1 + cx^{-1} + dy^{-1} \right), \quad f_{44}(x; r) = \ln \left( 1 + \frac{r}{x} \right). \]

We have the following tight inequalities

\[
f_{41}(x, y) \geq f_{41}(x_0, y_0) - \frac{a(x-x_0)}{(a x_0 + b y_0)(a x_0 + b y_0 + 1)} \]

\[
- \frac{b(y-y_0)}{(a x_0 + b y_0)(a x_0 + b y_0 + 1)} \geq f_{41}^{lb}(x, y; x_0, y_0, a, b), \quad (40) \]

\[
f_{42}(x, y) \geq f_{42}(x_0, y_0) - \frac{c y_0(x-x_0)}{x_0(c y_0 + d x_0 + x_0 y_0)} \]

\[
- \frac{d x_0(y-y_0)}{y_0(c y_0 + d x_0 + x_0 y_0)} \geq f_{42}^{lb}(x, y; x_0, y_0, c, d), \quad (41) \]

\[
f_{43}(x) \geq f_{43}(x_0) + x_0 \exp(x_0)(x_0 + 2)(x-x_0) \]

\[
\frac{\exp(3x_0)}{x_0} x_0 \geq f_{43}^{lb}(x; x_0, p), \quad (42) \]

\[
f_{44}(x) \geq f_{44}(x_0) - \frac{r(x-x_0)}{x_0(x_0 + r)} \geq f_{44}^{lb}(x; x_0, r), \quad (43) \]

Proof. Please see Appendix B.
Algorithm 1: Proposed Dinkelbach-based algorithm to approximately solve subproblem (P4)

Result: $q^*, v^*$

Initialize feasible point $(q^{(0)}, v^{(0)})$ and slack variables, set iteration index $m = 0$, then $\psi^{(m)} = R_{qc}^k(q^{(m)}, v^{(m)})$, $\psi^{(m)} = P_f(v^{(m)})$, define $\lambda^{(m)} = \frac{\psi^{(m)}}{\psi^{(m)}}$, and set Convergence = false;

while not Convergence do

Given $(\lambda^{(m)}, q^{(m)}, v^{(m)})$, solve (P4.5) using (45a), then obtain $(\psi^{(m+1)}, \omega^{(m+1)}, q^{(m+1)}, v^{(m+1)})$;

Calculate $\lambda^{(m+1)}$, then $F = \psi^{(m)} - \lambda^{(m+1)} \omega^{(m)}$;

if $|F| \leq \varepsilon_2$ then

$q^* = q^{(m+1)}$, $v^* = v^{(m+1)}$;

Convergence = true;

end

$m \leftarrow m + 1$;

end

Algorithm 2: Overall sequential based proposed iterative algorithm for SEE maximization (SEE-Seq)

1: Initialize a feasible point $(q^i, v^i, P_u^i, P_b^i, P_k^i, \zeta^i)$, and let iteration index $l = 0$;

2: Repeat:

3: Solve (P1.1) using (21), updating $P_k^{(l+1)}$ and $\zeta^{(l+1)}$; $l \leftarrow l + 1$;

4: Given $(P_k^{(l+1)}, \zeta^{(l+1)})$, solve (P2) using (25a), updating $P_u^{(l+1)}$; $l \leftarrow l + 1$;

5: Given $(P_u^{(l+1)}, \zeta^{(l+1)}, P_b^{(l+1)})$, solve (P3.1) using (29a), updating $P_b^{(l+1)}$; $l \leftarrow l + 1$;

6: Given $(P_k^{(l+1)}, P_u^{(l+1)}, P_b^{(l+1)})$, run Algorithm 1 with $(q^i, v^i)$, updating $q^{(l+1)} \leftarrow q^*$ and $v^{(l+1)} \leftarrow v^*$; $l \leftarrow l + 1$;

7: Until fractional increase of objective function in (18) gets below the threshold $\varepsilon_1$;

8: Return: $(Q^{opt}, P^{opt}, \zeta^{opt}) \leftarrow (Q^{(l)}, P^{(l)}, \zeta^{(l)})$;

III-E Overall algorithms and complexity discussion

Having obtained an efficient optimization model for each sub-problem in the previous section, we are now ready to propose iterative algorithms based on sequential block optimization and maximum improvement (MI) or the so-called greedy optimization introduced in [43], summarized in Algorithm 2 and Algorithm 3, respectively. The former is simpler to implement and requires less computations at each iteration. The latter converges faster thanks to a large step-size at each iteration and implementation via parallel computation capability; otherwise, it maybe too expensive. It can be mathematically proved that both algorithms are guaranteed to converge to at least a suboptimal solution. Since the feasible solution set of $(P)$ is compact and its objective value is non-decreasing over iteration index $l$ (a similar explanation also applies to the inner Dinkelbach algorithm over the iteration index $m$), and that the optimal value of minimum SEE is upper bounded by a finite value from the communications engineering perspective [11]. In terms of computational complexity, given $L$ and $M$ be the maximum convergence iteration of the outer overall BCD-SCA algorithm and the inner fractional sub-algorithm, Algorithms 2 and 3 have the complexity of approximately $O \left( L(NK)^{2.5} \left( \log_2 \left( \frac{1}{\varepsilon_2} \right) + L N^{3.5} \log_2 \left( \frac{1}{\varepsilon_3} \right) \right) \right)$ and $O \left( L(NK)^{3.5} \max \left( \log_2 \left( \frac{1}{\varepsilon_1} \right), M \log_2 \left( \frac{1}{\varepsilon_3} \right) \right) \right)$. Both are in polynomial time order and applicable to the UAV scenarios.

IV Numerical results and discussion

In this section, we provide some numerical simulations to evaluate the secrecy performance of the proposed THz-UUR scheme, and demonstrate the effectiveness of our proposed design in comparison with some benchmarks. Unless otherwise stated, all simulation parameters, adopted from the literature such, are given in Table I. Since the initial feasible point is important to use the proposed BCD-SCA-Dinkelbach based algorithms and significantly impacts their convergence performance, we explain how we can obtain initial feasible UAV’s trajectory, velocity, transmission powers, and user scheduling. The initial UAV’s trajectory is assumed to be circular centered at the BS’s location with radius $R_u = ||q_b - q_f||$, provided that UAV’s instantaneous velocity constraint C12 is satisfied, and $T \geq T_{min} \triangleq \frac{2 R_u}{\Delta v_{max}}$, where $T_{min}$ is the minimum required time for circular trajectory. However, if $T_{min} > T \geq T_{cyc} \triangleq \frac{2 R_u}{\sqrt{\Delta v_{max}}}$ (i.e., at least cyclic trajectory was possible with minimum required time $T_{cyc}$), then one could use any cyclic shape as long as C10 – C14 are satisfied. Here, we consider a Piriform trajectory with discretized equations given by $y^i = A_y \left( 1 - \sin \left( \frac{t}{\Delta v_{max}} \right) \right) \cos \left( t \right)$ and $x_u = R_u \left( \sin \left( t \right) + 1 \right)/2$.
in which \( t_{3 \times N} \) indicates the linearly spaced vector in \( [\frac{\pi}{2}, \frac{3\pi}{2}] \). Further, the constant \( A_y \) can be obtained efficiently via a simple 1D search in the range of \([R_u, 0]\) or simply set to zero. The UAV’s initial velocity vector \( v^i \) is then followed by \( v^\| = \mathbf{Q}^{[n+2]} - n \), \( \forall n \in N \) and \( v^\| = v^i[N - 1] \).

Having obtained an initial feasible UAV’s trajectory and velocity (\( Q^i = \{ q^i, v^i \} \)), we can set the initial UUR’s relaying power and the BS’s jamming transmission power as \( P^i = \{ p^i[n] = p^i_{ave}, \forall n \} \) and \( P^b = \{ b^i[n] = \beta_{ave}, \forall n \} \), respectively. The UEs’ initial transmit powers are set as

\[
P^i_k = \begin{cases} 
  p^i_k[n] = \begin{cases} 
  p^i_{ave}, & \zeta_k[n] = 1 \\
  0, & \zeta_k[n] = 0
  \end{cases}, \\
  \forall k, n
\end{cases}
\]

and the UEs are scheduled equally (e.g., \( \frac{N}{2} \) times each), i.e., \( \zeta^i \) is obtained such that the constraint \( C1 \) holds.

After identifying the initial feasible point for the iterative optimization algorithms, we consider different benchmark schemes, all of which are detailed below and labelled in the following figures, to demonstrate the superiority of our proposed minimum SEE-based optimization algorithms.

- **SEE-Seq**: minimum Secrecy Energy Efficiency optimization scheme using the Sequential BCD-based subproblem maximization as given in Algorithm 2.
- **SEE-MI**: minimum SEE optimization scheme based on the Maximum Improvement subproblem maximization as given in Algorithm 3.
- **SEE-FTrj**: Fixed Trajectory and velocity, i.e., \( Q = \{ q^i[n], v^i[n], \forall n \} \) based minimum SEE optimization scheme using the MI-BCD approach via jointly optimizing the transmit power allocations and user scheduling, i.e., \( P_k, P_u, P_b, \) and \( \zeta \).
- **SEE-FFPow**: Jointly designing the trajectory and velocity of the UUR via solving the corresponding sub-problem, i.e., optimizing \( Q \), while keeping the power allocations and user scheduling parameters fixed, i.e., setting them equal to the initial feasible values.
- **ASR-Seq**: Optimizing minimum ASR given in (15) while ignoring the UUR’s flight power limit using the Sequential BCD approach to iteratively improve \( Q, P, \zeta \).

Fig. 2 depicts the convergence of the proposed iterative algorithms. We can see that both benchmark schemes SEE-FTrj and SEE-FFPow converge quickly; however, they can only achieve significantly lower SEE performance than the proposed trajectory using joint design of power control and user scheduling. Specifically, SEE-MI converges relatively faster than SEE-Seq, i.e., 13 against 28 iterations, at the cost of slightly lower minimum SEE than that of its counterpart. However, they both achieve approximately 68% SEE improvement, while SEE-FTrj and SEE-FFPow have 15.1% and 25.5%, respectively.

Fig. 3 illustrates UUR’s trajectories using different optimization algorithms. We see that the optimized trajectories are much more complicated than the initial circular one with the counterclockwise direction. Notice that UUR should fly towards UEs’ locations to obtain data with low power. This, in turn, can significantly increase the chance of information leakage due to a stronger wiretap link and less effective BS’s jamming. Thus, UUR prefers to stay not too far from the BS. Overall, we see that the path planning makes UUR adjust trajectory through the best possible path, efficiently forming the distances between the UUR, selected UEs, and the BS such that the trade-off in the channel conditions for the friendly jamming transmission in the first phase as well as the aerial relaying in the second phase of transmission improve the secrecy performance. Further, we observe that the SEE-based trajectories are smoother than that of the ASR-Seq scheme.
Fig. 3: Designed trajectory for the SEE optimization according to different algorithms. $T = 10s$ is set.

Fig. 4: Designed trajectory for the SEE optimization according to different algorithms. $T = 5s$ is set.

implying possibly a lower flight power consumption of UUR. The SEE optimization demands this in contrast to the ASR-Seq design where the UUR’s velocity might harshly fluctuate for the minimum ASR (mASR) improvement if required. We also note that when the initial circular trajectory is impossible due to significantly low mission time, e.g., $T = 5s$, and owing to the UAV’s physical system limitations, the crucial task of path-planning can be efficiently designed as shown in Fig. 4. It should be mentioned that the curve belonging to the “SEE-FTrj” represents the initial feasible cyclic trajectory based on the Piriform, and the other curves illustrate the optimized UUR’s trajectory according to the different algorithms.

Fig. 5 illustrates mASR and the average flight power consumption profile according to different SEE algorithms.

Fig. 5: Comparison between mASR and AFPC against iteration index for different algorithms.

until convergence after 30 iterations. We also note that this scheme can achieve slightly higher mASR performance than our proposed schemes but at the cost of significantly lower SEE (43.13 Mbits/Joule).

Fig. 6 is plotted to demonstrate how the UAV’s velocity (Vel.) and the instantaneous flight power consumption (IFPC) can be adjusted over time for SEE improvement using different algorithms. We observe that all SEE algorithms, except “SEE-FTrj”, make UUR fly with roughly less speed variation for a relatively more extended period of time (e.g., from 3s to 8s) to satisfy mission requirements as well as improve the SEE performance. However, due to having complicated function of IFPC w.r.t the UAV’s velocity given in (11), UUR starts at a high initial speed to fast reach the targeted location, but not at maximum speed or hovering for the sake of efficient power consumption purposes.

Fig. 6: UAV’s instantaneous velocity and flight power consumption profile according to different SEE algorithms.

Fig. 7 illustrates the joint power allocation and user scheduling vs. time for different algorithms. The sub-figure 7d rep-
Fig. 7: Transmit power allocation and user scheduling against time for different schemes.

Fig. 8: Effect of THz molecular absorption factor on the SEE performance of the SEE-MI scheme for different mission time.
This was formally posed as a challenging mixed-integer non-convex nonlinear maximin optimization problem. We then embarked on tackling the nonconvexity of the formulated problem and proposed low-complex BCD-SCA-Dinkelbach based iterative algorithms to solve it suboptimally with guaranteed convergence. Simulation results confirmed the fast convergence of our proposed algorithms, demonstrated significant SEE performance improvement than the other benchmarks, and provided insightful results in the optimized system parameters such as UUR’s trajectory and velocity pattern as well as communication resource allocations, including transmit power profiles and user scheduling. Also, the effects of mission time, and molecular absorption factors arising from the THz links on the system SEE performance have been examined. As future work, we will deeply investigate the dynamic topology of aerial platforms with more practical THz channel modeling leveraging benefits of extreme directional beamforming for intelligent UUR systems.

**APPENDIX A**

**PROOF OF LEMMA 1**

Computing the Hessian matrices of given functions yields

\[
H_1 = \nabla^2 (Z_1) = \begin{bmatrix}
\frac{e^2}{(x+cy)^2} & -\frac{y^2}{y - x} \\
-\frac{y^2}{y - x} & \frac{e^2}{(x+cy)^2}
\end{bmatrix},
\]

\[
H_2 = \nabla^2 (Z_2) = \begin{bmatrix}
\frac{a b y^2 c}{\sigma_2} & \frac{a b y^2 c}{\sigma_2} \\
\frac{a b y^2 c}{\sigma_2} & -\frac{a b y^2 c}{\sigma_2}
\end{bmatrix},
\]

wherein \(\nabla^2(\cdot)\) denotes the hessian operator, and \(\sigma_1 = \frac{a b y^2 c}{\sigma_2}, \sigma_2 = (y + bx)^2((a + 1)y + bx)^2, \) and \(\sigma_3 = 2(1 + a)y + (a + 2)bx.\) One can verify that both matrices \(H_1\) and \(H_2\) are negative semidefinite \((H_{1/2}(z) < 0)\), thereby functions \(Z_1\) and \(Z_2\) are jointly concave w.r.t \(x\) and \(y\). Accordingly, there exist global over-estimators for concave functions in (20), following the first-order convexity condition law [41]. The proof is completed.

**APPENDIX B**

**PROOF OF LEMMA 3**

Computing the gradients of given functions w.r.t \(x\) and \(y\) yields

\[
\nabla f_{41}(x, y) = \begin{bmatrix}
\frac{a}{(a + b)x + (a + b)y + 1} \\
\frac{a}{(a + b)x + (a + b)y + 1}
\end{bmatrix},
\]

\[
\nabla f_{42}(x, y) = \begin{bmatrix}
\frac{c y}{(c + d)x + cy} \\
\frac{c y}{(c + d)x + cy}
\end{bmatrix},
\]

\[
\nabla f_{43}(x) = x e^x (e x + 2),
\]

Further, calculating the Hessian of the functions \(f_{41}(x, y)\) and \(f_{42}(x, y)\) and the second order derivative of \(f_{43}(x)\), we can reach

\[
H_{41} = \nabla^2 f_{41}(x, y) = \begin{bmatrix}
\frac{a^2}{b} & \frac{a b}{b^2} \\
\frac{a b}{b^2} & \frac{b^2}{b^2}
\end{bmatrix},
\]

\[
H_{42} = \nabla^2 f_{42}(x, y) = \begin{bmatrix}
\frac{c y (c + d)x + cy}{(c + d)x + cy} & \frac{c y (c + d)x + cy}{(c + d)x + cy} \\
\frac{c y (c + d)x + cy}{(c + d)x + cy} & \frac{c y (c + d)x + cy}{(c + d)x + cy}
\end{bmatrix},
\]

\[
\nabla^2 f_{43}(x) = e^x (e^x + 4 e x),
\]

We can verify that the first-order and second-order determinants of \(H_{41}\) and \(H_{42}\) are all non-negative, therefore, the Hessian matrices are positive semi-definite \((H_{41/2}(z) > 0)\), indicating that functions \(f_{41}(x, y), f_{42}(x, y)\) are jointly convex w.r.t \(x\) and \(y\). Further, the convexity of \(f_{43}(x, r)\) and \(f_{44}(x, p)\) follows from (B.6) and [15, Lemma 1]. Given these functions are all convex, one can use the first-order Taylor expansions at points \(x_0\) and \(y_0\) to reach the global tight lower-bounds and inequalities in Lemma 2. The proof is completed.

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