Analytical modeling for rapid design of bistable buckled beams

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**HIGHLIGHTS**

- A generic model of double-clamped bistable buckled beams under center and off-center point force actuation based on the Euler–Bernoulli beam theory.
- Analytical formulas of a bistable beam's critical behavioral values that characterize its snap-through properties, which give rise to a rapid and computationally efficient design method of double-clamped bistable buckled beams.
- An analysis of the influence of design parameters on the snap-through characteristics of bistable buckled beams, with results validated by finite element analysis (FEA) simulations.

**ABSTRACT**

Double-clamped bistable buckled beams demonstrate great versatility in various fields such as robotics, energy harvesting, and microelectromechanical system (MEMS). However, their design often requires time-consuming and expensive computations. In this work, we present a method to easily and rapidly design bistable buckled beams subjected to a transverse point force. Based on the Euler–Bernoulli beam theory, we establish a theoretical model of bistable buckled beams to characterize their snapthrough properties. This model is verified against the results from a finite element analysis (FEA) model, with maximum discrepancy less than 7%. By analyzing and simplifying our theoretical model, we derive explicit analytical expressions for critical behavioral values on the force-displacement curve of the beam. These behavioral values include critical force, critical displacement, and travel, which are generally sufficient for characterizing the snap-through properties of a bistable buckled beam. Based on these analytical formulas, we investigate the influence of a bistable buckled beam's key design parameters, including its actuation position and precompression, on its critical behavioral values, with our results validated by FEA simulations. Our analytical method enables fast and computationally inexpensive design of bistable buckled beams and can guide the design of complicated systems that incorporate bistable mechanisms.

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and high structural simplicity and consume relatively little power when incorporated into mechanical systems, which are potentially desirable for aerospace applications, e.g. energy absorbing [16] when aerospace devices are subjected to foundation excitation [17, 18] or sudden unbalance [19]. These desirable properties suggest more dedicated efforts be put into investigating these mechanisms.

Among various types of bistable mechanisms, double-clamped bistable buckled beams (Fig. 1) have drawn the attention of many researchers, thanks to their remarkable manufacturability and versatility [10, 20, 21]. Early on, Vangbo [22] studied the nonlinear behavior of bistable buckled beams under center actuation with a Lagrangian approach under geometric constraints and was able to express both the bending and the compression energy associated with the snap-through motion with buckling mode shapes of the bistable beam. This method was verified by Saif [3], who also extended the method to tunable micromechanical bistable systems. Qiu et al. [23] then explored the feasibility of this method on double curved beams (i.e. two centrally-clamped parallel beams). Moreover, an analytical expression of the relationship between the force applied at the beam’s center and the corresponding displacement was derived, making the characterization and design of the bistable mechanism easier. Nevertheless, most efforts were put into the modeling of bistable buckled beams under center actuation, while only a few works [24, 25] have tackled the modeling of bistable beams under off-center actuation. Still, off-center actuation possesses unique behavioral properties that make it suitable for many applications. For example, compared to center actuation, off-center actuation usually requires a smaller actuation force but features a longer actuation stroke [26, 27]. In addition, off-center actuation schemes highly pertain to applications with geometric constraints at the mid-span position of the beam [28]. In this paper, we extend the work of Vangbo [22] to bistable buckled beams under off-center actuation to facilitate their design process based on theoretical analysis.

The design of bistable mechanisms largely relies on their snap-through characteristics. Typically, the snap-through characteristics of a bistable structure can be primarily represented by three behavioral values, i.e. the critical force $F_{cr}$ and the critical displacement $w_{cr}$ at the bistable mechanism’s switching point, as well as the travel $w_{tr}$ at the stable equilibrium point, as shown in the force-displacement curve in Fig. 2. These critical behavioral values are determined by design parameters, i.e. the geometry (including the length, width and thickness of the slender beam), precompression [22], actuation position [25, 29], and boundary conditions [26, 30]. Due to the coupled complicating between snap-through characteristics and design parameters, it is often challenging to design a bistable structure efficiently. So far, lots of efforts aiming at efficiently designing bistable mechanisms have been made. Camescasse et al. [25, 31] investigated the influence of the actuation position on the response of a precompressed beam to actuation force both numerically and experimentally, based on the elastica approach. A semi-analytical method for analyzing bistable arches, which involves numerically extracting critical points from bistable arches’ force-displacement curves, was also presented in previous works [32, 33]. Due to the intrinsically strong nonlinearity of bistable mechanisms, existing models are rather complicated and usually involve differential equations that could only be solved semi-analytically or even numerically. Recently, Bruch et al. [34] developed a fast, model-based method for centrally actuated bistable buckled beams, which, however, requires heavy computation with finite element analysis (FEA) methods. Thus, rapidly and efficiently designing bistable mechanisms, especially those under off-center actuation, remains a huge challenge.

In this work, we develop a method for the rapid design of double-clamped bistable buckled beams. Similar to Vangbo’s work [22], the Lagrangian approach is adopted in the theoretical model to determine the contribution of each buckling mode shape under geometric constraints. Through analyzing and simplifying the theoretical model, explicit analytical expressions of the critical force, critical displacement, and travel are obtained. Moreover, based on the presented model, a detailed analysis of the influence of design parameters, including actuation position and precompression, on the snap-through characteristics of the beam is presented and validated by an FEA model. Thus, given a set of design parameters, our analytical formulas can output the critical behavioral values in real-time, consistent with results from FEA simulations which usually take hours on the same machine. Specifically, the contributions of this work include:

1) A generic model of double-clamped bistable buckled beams under center and off-center point force actuation based on the Euler-Bernoulli beam theory;

2) Analytical formulas of a bistable beam’s critical behavioral values that characterize its snap-through properties, which give rise to a rapid and computationally efficient design method of double-clamped bistable buckled beams;

3) An analysis of the influence of design parameters on the snap-through characteristics of bistable buckled beams, with results validated by FEA simulations.

The structure of the paper is as follows: the bistable system is described in Sect. 2; the theoretical model of bistable buckled beams is presented and simplified in Sect. 3; the explicit analytical expressions of the beam’s snap-through characteristics are derived in Sect. 4; our main results and discussions are showcased in Sect. 5, followed by conclusions and future work in Sect. 6.

Fig. 1. Clamped–clamped bistable buckled beam. a) Non-loaded straight beam, b) beam in its buckled configuration with an actuating force $F$ applied at the location $C$. 

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2 Description of the system

Here we consider a clamped-clamped and initially straight elastic beam, as shown in Fig. 1. The original length, width, thickness, and Young’s modulus of the beam are denoted as \(L_0\), \(b\), \(h\), and \(E\), respectively. Under a compressive axial load \(P\), one of the beam’s terminals moves towards the other, resulting in a first-mode buckling shape with initial rise \(w_{\text{rise}}\) (i.e. the initial displacement of the beam’s mid-span). The distance between the two terminals of the beam after buckling, what we refer to as the span, is denoted as \(L\); the difference between the original length and the span is denoted as \(d_0\) (i.e. \(d_0 = L_0 - L\)). Moreover, the cross-sectional area of the beam and the second moment are denoted as \(A = bh\) and \(I = \frac{bh^3}{12}\), respectively.

The system’s two-dimensional reference frame is chosen such that the \(x\)-axis coincides with the line connecting the two ends of the beam after it is axially compressed, while the \(w\)-axis is set perpendicular to the \(x\)-axis at one end of the beam, as shown in Fig. 1. A point force \(F\) in the \(w\) direction is applied vertically to the buckled beam at a selected location \(C\). The ratio \(\delta = (x_C - x_A)/(x_B - x_A)\) is the parameter that indicates the position at which \(F\) is applied to the beam.

3 Modeling and analysis

In this section, a theoretical model of bistable buckled beams is derived and subsequently simplified. This model allows for characterizing the snap-through properties of a bistable buckled beam and enables the derivation of analytical expressions of the beam’s important snap-through characteristics.

3.1 Theoretical model

According to Euler’s buckling model of a double-clamped slender beam, when the axially compressed beam is undisturbed (i.e. \(F = 0\)), its behavior can be described with the following differential equation

\[
\begin{align*}
\frac{d^4w}{dx^4} + n^2 \frac{d^2w}{dx^2} &= 0, \\
\frac{d^2w}{dx^2} &= \frac{P}{EI}, \\
w(0) &= w(L) = w'(0) = w'(L) = 0.
\end{align*}
\]

The eigenvalues of this homogeneous Strum–Liouville problem can be denoted in form of \(n_iL\), and these eigenvalues satisfy the equation

\[
1 - \cos(n_iL) = \frac{1}{2} n_iL \sin(n_iL).
\]

The eigenvalues give rise to a series of nontrivial eigenfunctions of Eq. (2)

\[
w_i(x) = \left\{ \begin{array}{ll}
C_i[1 - \cos(n_i x)], & i = 0, 2, 4, ..., \\
C_i[1 - \cos(n_i x) - \frac{2}{n_iL} n_i x - \sin(n_i x)], & i = 1, 3, 5, ..., 
\end{array} \right.
\]

\[
\delta = \frac{(x_C - x_A)}{(x_B - x_A)} = \frac{\frac{d_0}{d_p}}{L} = \frac{d_0}{d_p} = \frac{d_0}{d_p} = 0.
\]

When a force \(F\) is applied to the beam, its displacement \(w(x)\) can be described as a superposition of these eigenfunctions

\[
w(x) = \sum_{i=0}^{\infty} A_i w_i(x),
\]

where the set of amplitudes \(A_i\) minimizes the energy of the system under the constraint of the beam’s current length, \(L + d_0 - d_p\). \(d_p\) refers to the contraction from the axial load \(P\) and is given as \(d_p = PL/(EA)\). Thus, we have the following equation

\[
L + d_0 - d_p = \int_0^L \sqrt{1 + [w'(x)]^2} dx \approx \int_0^L \left(1 + \frac{[w'(x)]^2}{2}\right) dx.
\]

Combining Eqs. (4) and (5), we have

\[
g(\delta) = \sum_{i=0}^{\infty} \frac{A_i^2 (n_iL)^2}{4} - (d_0 - d_p)L = 0.
\]

The energy of the system can be written as

\[
U(\delta, F) = \frac{EI}{2} \int_0^L [w''(x)]^2 dx + Fw(\delta L) + \frac{Pd_p}{2}.
\]

\[
= \frac{EI}{4L^3} \sum_{i=0}^{\infty} A_i^2 (n_iL)^4 + F \sum_{i=0}^{\infty} A_i w_i(\delta L) + \frac{Pd_p}{2},
\]

where the three terms refer to the bending energy of the beam, the potential energy of the force, and the compression energy, respectively. In Vangbo’s work, the parameter \(\delta\) in the

Fig. 2. Characteristic force–displacement \(F-w\) curve of bistable buckled beams. a Under center actuation, b under off-center actuation.
second energy term is always set to 0.5 as the force is applied at the beam’s center; in this work, however, we allow \( \delta \) to vary in order to account for off-center actuation.

Therefore, we solve for the \( A_i \) that minimize \( U \) in Eq. (7) and conform to the constraint specified by Eq. (6). We introduce a Lagrange multiplier \( \lambda \) in order to find the equilibrium state of the beam under a force \( F \). We consider

\[
K(\dot{A}) = U(\dot{A}, F) - \lambda g(\dot{A}).
\]

The solutions \( A_i \) should satisfy

\[
\frac{\partial K}{\partial A_i} = 0 \quad \text{and} \quad \frac{\partial K}{\partial \dot{A}_i} = 0.
\]

Solving Eq. (9), with \( \lambda \) chosen in the same way as in Vangbo’s work [22], we have

\[
A_i = \frac{2FL^3w_i(\delta L)}{EI(n_iL)^2[(\eta L)^2 - (n_iL)^2]} \quad \text{with} \quad \eta^2 = \frac{P}{EI}.
\]

Combining Eq. (10) and the constraint given by Eq. (6), we can determine the magnitude of \( F \) when given a value of the parameter \( \eta \):

\[
F(\eta) = \frac{EI \sqrt{d_0 - d_0^2} L^3}{L^3 \left( \sum_{i=0}^{\infty} \frac{w_i^2(\delta L)}{(n_iL)^2[(\eta L)^2 - (n_iL)^2]} \right)}.
\]

with \( d_0 = \frac{PL}{E A} \).

Also, combining Eqs. (4) and (10), we have

\[
w(\eta) = \frac{2F(\eta)L^3}{EI} \sum_{i=0}^{\infty} \frac{w_i^2(\delta L)}{(n_iL)^2[(\eta L)^2 - (n_iL)^2]}.
\]

Equations (10)-(12) characterize the connections among the actuating force \( F \), the beam’s displacement \( w \), and the axial load \( P \) (\( P = \eta^2 EI \)) applied to the beam from side walls. Importantly, the obtained force-displacement curve can be used to characterize the mechanical properties of the bistable buckled beam.

3.2 Reduced model

As largely mentioned in Refs. [23, 24], the first two modes of buckling, \( w_0(x) \) and \( w_1(x) \), have predominant contribution in the beam’s displacement \( w(x) \) in both center and off-center actuation scenarios [24]. Thus, we can make the approximation that \( w(x) = A_0w_0(x) + A_1w_1(x) \) and write

\[
F(\eta) = \frac{EI \sqrt{d_0 - d_0^2} L^3}{L^3 \left( \sum_{i=0}^{\infty} \frac{w_i^2(\delta L)}{(n_iL)^2[(\eta L)^2 - (n_iL)^2]} \right)} + \frac{u_i^2(\delta L)}{(n_iL)^2[(\eta L)^2 - (n_iL)^2]}.
\]

\[
w(\eta) = \frac{2F(\eta)L^3}{EI} \left\{ \frac{w_0^2(\delta L)}{(n_0L)^2[(\eta L)^2 - (n_0L)^2]} \right\}.
\]

Moreover, recall that \( P(\eta) = \eta^2 EI \) and that we have

\[
P_0 = n_0^2 EI \quad \text{with} \quad n_0L = 2\pi,
\]

\[
P_1 = n_1^2 EI \quad \text{with} \quad n_1L = 2.86\pi,
\]

(15)

where \( P_0 \) and \( P_1 \) represent the axial compressive load of the first-mode and second-mode buckling, respectively. Note that the switching of the beam always features an axial load greater than \( P_0 \) but not exceeding \( P_1 \) [22].

4 Analytical expressions of the snap-through characteristics

Generally, the three critical behavioral values, \( F_{cr} \), \( w_{cr} \), and \( w_{cr} \), on the force-displacement curve are sufficient for characterizing a bistable buckled beam and facilitating its design. Given the significance of these behavioral values, it is worthwhile to develop explicit analytical expressions for each of them.

4.1 Critical force

The magnitude of \( F_{cr} \) can be considered the maximum of the function \( F(\eta) \) in Eq. (13) when \( 2\pi < \eta L < 2.86\pi \),

\[
F_{cr} = \max[F(\eta)] \quad \text{with} \quad 2\pi < \eta L < 2.86\pi.
\]

To simplify Eq. (16), we take advantage of the fact that the thickness of the beam \( h \) is much smaller than \( \sqrt{d_0 L} \) [22, 24, 31]. Therefore, we have

\[
d_0L = \frac{I}{A} (\eta L)^2 = \frac{1}{12} h^2(\eta L)^2 \ll d_0 L.
\]

Hence, we can assume a simplified version of Eq. (16)

\[
F_{cr} = \frac{EI \sqrt{d_0 L}}{L^3} \max[P(\eta)] \quad \text{with} \quad 2\pi < \eta L < 2.86\pi,
\]

\[
P(\eta) = \sqrt{\frac{u_i^2(\delta L)}{(2\pi n_iL)^2[(\eta L)^2 - (2\pi n_iL)^2]} + \frac{u_i^2(\delta L)}{(2.86\pi n_iL)^2[(\eta L)^2 - (2.86\pi n_iL)^2]}},
\]

\[
1
\]

Notice that we have

\[
w_0(\delta L) = 1 - \cos(2\pi \delta),
\]

\[
w_1(\delta L) = 1 - \cos(2.86\pi \delta) - \frac{2}{2.86\pi} [2.86\pi \delta - \sin(2.86\pi \delta)].
\]

It can be observed from Eqs. (18) and (19) that \( \argmax[P(\eta)] \) (denoted as \( \hat{\delta} \)) and thus \( \max[P(\eta)] \) are only dependent on the parameter \( \delta \). In other words, we can denote \( \max[P(\eta)] \) on \([2\pi, 2.86\pi]\) as a function of \( \delta \), written as \( F_{cr}(\delta) \). So we have

\[
F_{cr} = \frac{EI \sqrt{d_0 L}}{L^3} F_{cr}(\delta).
\]

To obtain an analytical form of \( F_{cr} \), we vary \( \delta \) from 0.15 to 0.5, the scope of this parameter within our consideration (note that by symmetry, we only need to consider one half of the beam), and calculate the corresponding values of \( F_{cr} \). We then apply curve-fitting to obtain an analytical relationship between \( F_{cr}(\delta) \) and \( \delta \). \( F_{cr}(\delta) \) as a function of \( \delta \) is visualized in Fig. 3 and presented in Eq. (21) with the error of fitting less than 7%. \( F_0 \) is chosen as a degree-4 polynomial to ensure relatively high accuracy and acceptable complexity of the model. Note that this curve-fitting can be reperformed to improve the accuracy of the final result or
to reduce the complexity of the model.

The analytical expression of the critical force $F_{cr}$ at a precompressed beam’s switching point can be written as Eq. (21). Note that the minimal critical force is achieved when $\delta$ is equal to 0.37 (or 0.63)

$$F_{cr} = \frac{EI}{L^3} F_0(\sigma),$$

$$\sigma = \min(\delta, 1 - \delta),$$

$$F_0(\sigma) = 50588\sigma^3 - 69285\sigma^3 + 36606\sigma^2 - 8894.5\sigma + 914.9. \; \text{(21)}$$

### 4.2 Critical Displacement

The critical displacement $w_{cr}$ can also be written in form of an analytical expression of the basic parameters. From Eqs. (14) and (21), we have

$$w_{cr} = 2\sqrt{d_0L} F_0(\sigma) \left\{ \frac{u_0^2(\delta L)}{(n_0L)^2[\{(\eta L)^2 - (n_0L)^2]\]} + \frac{u_0^2(\delta L)}{(n_0L)^2[\{(\eta L)^2 - (n_0L)^2]\]} \right\}. \; \text{(22)}$$

Since we have shown that $\eta$ only depends on $\delta$, we can conclude that $w_{cr} = 2\sqrt{d_0L} w_0(\delta)$ by substituting the bulk of Eq. (22) with $w_0$, some function of $\delta$. To obtain an analytical form of $w_0$, we vary the parameter $\delta$ from 0.15 to 0.5 and calculate the corresponding values of $w_0(\delta)$. $w_0$ as a function of $\delta$ is displayed in Fig. 3 and its analytical form is shown in Eq. (23) after some change of variables. The analytical expression of $w_{cr}$ is written as follows

$$w_{cr} = 2\sqrt{d_0L} w_0(\sigma),$$

$$\sigma = \min(\delta, 1 - \delta),$$

$$w_0(\sigma) = -7.155\sigma^4 + 2.872\sigma^3 + 4.339\sigma^2 - 1.538\sigma + 0.0832. \; \text{(23)}$$

Again, the curve-fitting can be reperformed for alternative analytical expressions of $w_0$. Moreover, it is important to note that the critical displacement is primarily dependent on $L$, $d_0$, and $\delta$, a result consistent with that of Bruch et al. [35] but obtained with a different method.

### 4.3 Travel

The initial shape of an axially compressed beam can be approximated using the cosine curve featured in the expression of $w_0(x)$. Thus, we have $w_{cr} = w_{rise}|1 - \cos(2\pi\delta)|/2$ by definition of the travel, where $w_{rise}$ is the initial rise of the beam’s midpoint, determined by the degree of compression. Considering Eq. (5), since we have shown that $d_0 \ll d_0$, we can ignore the term $d_0$ and approximate $w_{rise}$ from the following relationship

$$L + d_0 \approx \int_0^L \left\{ 1 + \frac{|w_{	ext{rise}}(x)|^2}{2} \right\} \, dx$$

with $w_{\text{rise}}(x) = \frac{w_{\text{rise}}}{2} \left[ 1 - \cos \left( \frac{2\pi x}{L} \right) \right]. \; \text{(24)}$

It can be calculated from Eq. (24) that $w_{\text{rise}} = 2\sqrt{d_0L}/\pi$, and so we have

$$w_{\text{rise}} = \frac{2\sqrt{d_0L}}{\pi} \left[ 1 - \cos(2\pi\delta) \right]. \; \text{(25)}$$

One significant observation from Eqs. (23) and (25) is that the value of $w_{cr}/w_{\text{rise}}$ only depends on the parameter $\delta$. The key insight is that when designing a precompressed bistable mechanism, the possible constraints on these two behavioral parameters may uniquely determine its optimal actuation position.

### 5 Results and discussions

In this section, we consider a double-clamped bistable buckled beam with its parameters given in Table 1. All of the parameters above remain unchanged throughout this section unless otherwise stated.

#### 5.1 Model validation

To validate our model, we compare our results, the $F - w$ and $P - w$ curves for both center and off-center actuation of a bistable buckled beam, with quasi-static force-displacement curves from ABAQUS. The bistable beam was modeled in three dimensions and its geometry was meshed with quadrilateral shell elements (i.e. S4R) that can capture large deformation behaviors. The buckling of the beam was introduced by an initial imperfection in the lateral direction. In our analytical model, Eqs. (13) and (14) combined give rise to the $F - w$ characteristic, while the relationship $P = \eta^2EI$ and Eq. (14) combined yield the $P - w$ curve.

##### 5.1.1 Center actuation

With $\delta$ set to 0.5, the $F - w$ and $P - w$ curves of the beam are graphed and compared to data from an FEA model, as shown in Fig. 4. In this figure, the solid black line represents the result from the our model while the circles depict the FEA simulation data. Two series of simulation data are presented: (1) the red circles represent the snap-through motion from the top stable equilibrium state to the bottom one, as depicted in Fig. 1(b); (2) the blue circles correspond to the motion in the opposite direction.

In both diagrams, points $c1$ and $c2$ represent the beam’s two stable equilibrium states that feature first-mode buckling ($P = P_n, \eta L = 2\pi$). Point $c1$ (or $c2$) corresponds to its unstable equilibrium state that features second-mode buckling ($P = P_1, \eta L = 2.86\pi$). Points $b1$ and $b2$ are the switching points.

There is a neat agreement between the actuating force $F$ and the compressive force $P$ calculated from our model and from the
FEA model, with errors bounded within 7% and 6%, respectively. Note that the greatest discrepancy occurs around the switching points, where the critical force is modeled fairly accurately, while the critical displacement from our model is larger than that calculated from the FEA model. This means our model suggests a premature snap-through of the bistable beam.

5.1.2 Off-center actuation

Under an off-center actuation ($\delta = 0.37$), the $F - w$ and $P - w$ curves of the beam are shown in Fig. 5. In the same manner, the solid black curves represent our analytical model while the red (downward) and blue (upward) circles come from the FEA simulation results.

Contrary to the center actuation, the off-center actuation from the two directions results in two distinct branches in the $F - w$ curve, as shown in both diagrams. This indicates that the switching of the beam involves a branch jump [24]. Similarly, $a1$ and $a2$ are the two stable equilibrium points ($P = P_0, \eta L = 2\pi$). $c1$ and $c2$ both represent the unstable equilibrium state of the beam ($P = P_1, \eta L = 2.86\pi$), approached when the beam is actuated by an off-center force from its two different stable positions. Points $b1$ and $b2$ are the switching points of the bistable beam.

The results from our analytical method are consistent with the FEA simulation data. Errors on the $F - w$ and $P - w$ curves with respect to the FEA results are bounded within 2% and 5%, respectively. The small magnitudes of these errors greatly demonstrate the validity of our model.

5.2 Influence of design parameters on snap-through characteristics

To facilitate the rapid design of bistable buckled bistable beams, we discuss the influence of a bistable beam’s key design parameters on its snap-through characteristics, namely its critical force, critical displacement, and travel. These results are also verified by an FEA model.

5.2.1 Actuation position

The impact of $\delta$ on the critical force is visualized in the $F_{cr} - \delta$ curve in Fig. 6(a). As the parameter $\delta$ is varied from 0.15 to 0.85, the corresponding values of critical force are calculated.

From Fig. 6(a), it can be observed that the minimal critical force is obtained when the beam is actuated around the position where $\delta = 0.37$ (or the symmetric position where $\delta = 0.63$). Interestingly, since the influence of actuation position on critical force can be assumed independent of other design parameters, as made evident in Eq. (21), any precompressed beam tends to obtain its minimal critical force when its actuation position is given by $\delta = 0.37$ (or $\delta = 0.63$). This finding pertains to applications that require the actuating force to be small.

Moreover, the $w_{cr} - \delta$ relationship is captured in Fig. 6(c). As the actuation position moves from the beam’s endpoint to its midpoint, the critical displacement increases, with its increment rate increasing. Note that the displacement is calculated with respect to the $x$-axis.

Lastly, when the design parameters of the beam are held constant, the mathematical relationship between the travel $w_{tr}$ and $\delta$ simply features the cosine function discussed in Sect. 4.3, as shown in Fig. 6(e).

As depicted in the Fig. 6(a, c, e), the $F_{cr} - \delta$, $w_{cr} - \delta$, and $w_{tr} - \delta$ curves generated from our model are also compared with those from the FEA model. In addition, the relative errors are presented in Fig. 6(b, d, f). The relative errors of $F_{cr}$ and $w_{cr}$ with respect to the FEA simulation data are both bounded within 4%. The critical displacement $w_{cr}$ calculated from our model matches excellently with the FEA simulation, even though the relative error is fairly notable when the actuation position largely deviates from the beam’s center. Within this range, the relative error of $w_{tr}$ is less informative and is not shown in Fig 6(d).

However, in most applications, the actuation position parameter $\delta$ falls within the range $[0.37, 0.63]$ [24, 28, 35], where the errors of $w_{cr}$ are bounded within 8%. Therefore, our model can be considered generally feasible and accurate.

5.2.2 Precompression

In order to increase the applicability of the following analys-

![Fig. 4](image-url)

Fig. 4. Center actuation: FEA results and comparison to our model. a Actuating force $F$ vs. displacement $w$. b Compressive force $P$ vs. displacement $w$. 

Table 1 Geometric and material parameters of the beam.

| Parameter  | Value  |
|------------|--------|
| Length $L_0$ (mm) | 14.9   |
| Width $b$ (mm)    | 3.0    |
| Thickness $h$ (mm) | 0.132  |
| Precompression $d_0$ (mm) | 0.3    |
| Young’s modulus $E$ (GPa) | 3.0    |
is, we define a parameter $r = \frac{d_0}{L_0}$ that denotes the precompression rate of a bistable buckled beam. Therefore, using the expressions $d_0 = r L_0$ and $L = (1 - r)L_0$, we derive the relationships among $r$ and the behavioral parameters of a bistable beam. Applying Eqs. (21), (23), and (25), we can obtain the relationships between $F_{cr}$, $w_{cr}$, and $w_{tr}$ and the precompression rate $r$. These relationships are given as $F_{cr} \propto \sqrt{r(1-r)}$, $w_{cr} \propto \sqrt{r(1-r)}$, and $w_{tr} \propto \sqrt{r(1-r)}$.

These mathematical relationships are demonstrated with a bistable beam with design parameters given in Table 1 and with $\delta$ set to 0.43. As shown in Fig. 7(a, c, e), all of the three values increase as the precompression rate increases, with their increment rates decreasing. Again, as shown in Fig. 7(b, d, f), the errors of our analytical model remain small (less than 9% for both $F_{cr}$ and $w_{tr}$), with the exception of the critical displacement $w_{cr}$ when the precompression rate $r$ is very large. The enlarged error of $w_{tr}$ when the precompression rate is large is due to the violation of the small-deflection hypothesis assumed in our model. The error of $w_{tr}$, however, is bounded within 15% when $r$ falls in the range $[0, 0.1]$, which indicates that our model still greatly applies to most circumstances [10, 20, 27].

### 6 Conclusions and future work

We have proposed a mechanism that can easily and efficiently characterize the response of a double-clamped bistable buckled beam to point force actuation. Based on the Euler-Bernoulli beam theory, we have established a theoretical model of bistable buckled beams and their behavior under an actuat-

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**Fig. 5.** Off-center actuation: FEA results and comparison to our model. a) Actuating force $F$ vs. displacement $w$. b) Compressive force $P$ vs. displacement $w$.

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**Fig. 6.** Effect of actuation position $\delta$ on the critical behavioral values. a) Critical force $F_{cr}$, b) error of $F_{cr}$ compared with FEA results, c) critical displacement $w_{cr}$, d) error of $w_{cr}$ compared with FEA results. Note that a part of data is not shown here. e) Travel $w_{tr}$, f) error of $w_{tr}$ compared with FEA results.
There are several directions in which we can extend our work. One of our most interesting future directions is optimization. For instance, minimizing the total energy consumption of a bistable buckled beam’s snap-through motion makes it possible to adopt more compact actuators in an integrated system. In addition, a possible extension of the present work involves building models of bistable buckled beams with other boundary conditions. Most importantly, given the complicated relationships among their design parameters and snap-through characteristics, it is worthwhile to propose a computational pipeline that designs bistable buckled beams with the specified critical behavioral values. In conclusion, we believe that our proposed analytical model is a significant step towards the fast and computationally inexpensive design of bistable buckled beams, which will be easily incorporated into more and more mechanical systems (e.g., shape memory alloys [36, 37]) actuated bistable mechanisms).

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