Research Article

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Study on the influence of structural form and parameters on vibration characteristics of typical ship structures

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Abstract: For the plate or multicabin structure of ships, the vibration transmission characteristics have a great correlation with the structural mode. When the excitation force frequency get close to the natural frequency of the structure, strong vibration and sound radiation will occur. Therefore, it is very important and meaningful to study the vibration characteristics and protection methods of ship structures. Based on the classic finite element method, this article studies the influence of structural forms and parameters on the vibration characteristics of typical ship structures. Taking the multicabin structure as a benchmark structure model, the influence of the structural form on vibration mode and transmission characteristics of the target deck and other cabin decks in the multicabin structure was analyzed. Then, without changing the original structural layout, the effects of different structural parameters on the vibration mode and transmission characteristics were analyzed. Finally, the vibration protection process of the ship structure was formed. The results of this study can provide methodological basis and data reference for relevant research in future.

Keywords: ship, vibration characteristics, protection method

1 Introduction

As the main means of transport on the sea, the comfort and quietness of ships have been paid more attentions. The vibration and noise level of a ship is one of the important indicators to measure its comfort level. Therefore, the prediction of ship vibration and noise is of great significance to ship design. However, the application of composite materials in ship structure is still very limited, which is not as common as aircraft and automobile. We also want to let scholars carry out more research in this field through this study, including use of composite materials for vibration and noise reduction.

In the research on ship vibration, Shan et al. [1] took a hospital ship as an example to analyze its vibration characteristic, and provided a reference for the design and construction of hospital ship. Wang et al. studied the vibration response of a wide and fat large ship under wave excitation [2], and they analyzed the hydroelastic behavior of a river–sea-going ship hull through experiments and numerical calculations to verify the accuracy and effectiveness of the nonlinear hydroelastic time domain calculation method [3]. Du et al. [4] conducted a series of experiments on a full-scale river icebreaker to investigate the characteristics of the hull vibration caused by ice. They found that the icebreaking operation of the river icebreaker leads to severe local vibration of the grillage on the main deck. McVicar et al. [5] studied the influence of the duration of the slam force on the vibration response of a high-speed catamaran. Choi et al. [6] used energy flow analysis methods to perform vibroacoustic calculations, and analyzed the noise radiated by turbulent flow on the complex underwater structure. Zhang et al. [7] investigated the influence of water jet propulsion on the local vibration of the grillage structure of high-speed ships. Han et al. [8] considered the noise characteristics of the duct and studied the indoor noise of typical cabins of naval vessel. Tang et al. [9] investigated the longitudinal vibration of a trimaran in regular waves.
Sun et al. [10] applied the time-variant reliability method to effectively evaluate the reliability of the ship's grillage structure. Gong and Zhang [11] used the local cabin finite element method (FEM) model and one cabin FEM model to accurately analyze the grillage structure mode. Wang et al. [12] proposed a simplified method for grillage structure as the statistical energy method to improve calculation efficiency. Jadee et al. [13] studied the free vibration characteristics of isotropic plates by using FEM and experimental design. Korgesaar et al. [14] carried out a research on the response of the ice-strengthened grillage frames under a certain pressure range. He et al. [15] combined the theory of two-dimensional plate energy flow and FEM, studied the vibration transmission characteristics of coupled plates under medium–low frequency excitation. Lu [16] used frequency spectrum analysis methods to analyze the vibration characteristics of the ship grillage and optimized the dynamic characteristics of the grillage. Lin et al. [17] separated the mode shapes and frequencies that contribute the most to the acoustic radiation peaks, and optimized structural parameters.

Many engineering structures are made of composite materials, such as floating rafts and equipment seats. The application of composite materials is an effective method to reduce vibration and noise. Compared with traditional materials, composite materials have many new advantages and broad application prospects. Zhang et al. [18] systematically reviewed the research progress on the typical mechanical structures of functionally graded carbon nanotubes, and discussed the current research prospects. Many scholars are engaged in the study of the vibration characteristics of the typical structure of composite materials, such as graphene-reinforced metal matrix composites [19], functionally graded carbon nanotubes [20], and so on. Li et al. [21–24] analyzed the free vibration of different laminated composite shallow shells subjected to different boundary conditions.

Meng [25] simplified the progress of solving the vibration response by establishing a vibration isolation model for the engine in operation. Liu et al. [26] used a simplified mass method to convert the acceleration load of ship equipment into force load to simplify the calculation model. Zhang et al. [27] introduced theories of hull-propeller-rudder coupling to simplify the interaction between the propeller and the shaft. Polic et al. [28] used the energy difference between the propeller and the shaft to study the torsional vibration response of the propulsion shafting system working in snow and ice. Li et al. [29] simplified the fluid-propeller-shaft and flow-structure coupling dynamic model to a coupling model of the propeller in a nonuniform flow, and discussed the influence factors of the propeller wet mode. Maljaars et al. [30] simplified the interaction between the propulsion shaft and the propellers, and explored the mechanical properties of the propeller under cavitation and noncavitation conditions. Li et al. [31] researched the free vibration characteristics of medium-thickness functionally graded plates with general boundary conditions based on the Fourier–Ritz method. Tang et al. [32] carried out an experimental research on the vibration characteristics of composite-reinforced sheet-beam structures.

It can be seen from the reviewed researches that many researches have focused on the free and forced vibration characteristics of beams, plates, grillages, cabinets, and so on. Those studies have revealed the mechanism of vibration generation and the influence of simple structures. However, it is far from enough to study the vibration mechanism of simple structures. The frequency difference between structures and equipment should also be considered to prevent resonance. When resonance occurs, there are generally two treatment methods: one is to change the equipment, and the other is to change the structure. It is commonly known that the cost of changing the equipment selection is high, while adopting a proper structural form can avoid resonance and has a good economic effect.

However, from the existing research literature, it can be found that most of the results are applicable to basic research, for the reason that the research object does not meet the engineering application conditions. Especially for ships, their structural form is quite different from ordinary structural forms. However, there are few research results on the vibration characteristics of ship structures, making it difficult to provide sufficient support for ship design.

In view of the above problems and background, the vibration characteristics of typical ship structures are studied based on the FEM, and the influence of structural form and parameters on vibration characteristics of typical ship structures was investigated, as shown in the Graphical abstract. This study sees multicanbin structure as the benchmark structure model. The influence of structural form on vibration characteristics of target deck and other cabin decks in multicabin structure was analyzed. The effects of different structural parameters on vibration characteristics are investigated, and the vibration protection method and process of ship structure are introduced. The results of this study can provide method basis and data reference for relevant research in future.

### 2 Theoretical formula

The mode superposition method is a classic finite element calculation theory method, which can solve most dynamics problems [33,34]. In this method, the motion equation in
the physical coordinate system is transformed into a modal coordinate system, and decouples the equation through the modal frequency and vibration mode information, thereby transforming the motion equation in the physical coordinate system into the modal coordinate system. Then, the structural response in the original physical coordinate system can be obtained by solving the modal coordinate responses and combining them. The main theoretical formulas of modal superposition method are as presented follows, which mainly provide theoretical support for subsequent numerical calculations.

2.1 Solving the coordinate transformation matrix

Solve the coordinate transformation matrix: 

\[ \Phi^T = M_p \Phi^{-1} M^{-1}, \]  

\[ K \Phi = M \Phi M_p^{-1} K_p, \]  

Separate the coordinate transformation matrix \( \Phi \) according to columns: 

\[ \Phi = [\phi_1, \phi_2, ... \phi_n]. \]  

Each element in \( \Phi \) is a column vector, which can be obtained by multiplication of partitioning matrix:

\[ [K \phi_1, K \phi_2, ..., K \phi_n] = [\lambda_1 M \phi_1, \lambda_2 M \phi_2, ..., \lambda_n M \phi_n]. \]  

The corresponding elements in two matrixes are equal one by one. Therefore, the decoupling problem of the vibration differential equation is transformed into the problem of solving the generalized eigenvalue. The eigen-equation is shown as follows:

\[ |K - \lambda M| = 0. \]  

The polynomial is solved to obtain the eigenvalues \( \lambda \), and corresponding eigenvectors \( \phi_i \), which are sorted in order from small to large, and the following formula is used to calculate the natural frequency \( \omega_i \) of each order and its corresponding principal mode \( \phi_i \):

\[ \omega_i = \sqrt[2]{\lambda_i}. \]  

The mode shape matrix \( \Phi \) could be named coordinate transformation matrix, can be determined by solving the mode shape \( \phi_i \).

2.2 Solving the decoupling equation

When all coefficient matrices of vibration equations can be diagonalized, their component forms can be written out, and both ends of the equal sign can be divided by \( m_i \):

Figure 1: Diagram of cabin structure: (a) deck structure and (b) bulkhead structure.
\[ \ddot{q}_i + \frac{c_i}{m_i} \dot{q}_i + \frac{k_i}{m_i} q_i = \frac{f_i}{m_i}, \]

where \( \frac{k_i}{m_i} = \omega_{ni}^2 \frac{c_i}{m_i} = 2\xi \omega_{ni} \), then \( \xi_i = \frac{c_i}{2m_i\omega_{ni}} = \frac{c_i}{2\sqrt{m_i k_i}} \). When the vibration equation is decoupled by the regular mode shape, it can be written as follow:

\[ \ddot{q}_{Ni} + 2\xi \omega_{ni} \dot{q}_{Ni} + \omega_{ni}^2 q_{Ni} = f_{Ni}. \]

Then the calculation method of equation is

\[
q_{Ni}(t) = e^{-\xi \omega_{ni} t} \left[ q_{Ni}(0) \cos \omega_{ni} t + \frac{\dot{q}_{Ni}(0) + \xi \omega_{ni} q_{Ni}(0)}{\omega_{ni}} \sin \omega_{ni} t \right] + \frac{1}{\omega_{ni}} \int_0^t f_{Ni}(\tau) e^{-\xi \omega_{ni}(t-\tau)} \sin \omega_{ni}(t-\tau) d\tau,
\]

where \( \omega_{ni} = \sqrt{1 - \xi_i^2 \omega_{ni}} \). However, the initial conditions are not contained in the physical coordinate system, so it is

**Figure 2:** Modal results of multicompartent truss: (a) 5.2 Hz, (b) 8.3 Hz, (c) 12.5 Hz, (d) 13.6 Hz, (e) 15.4 Hz, (f) 20.4 Hz, (g) 20.9 Hz, (h) 23.7 Hz, (i) 26.5 Hz, (j) 27.5 Hz, (k) 30.2 Hz, and (l) 31.4 Hz.
necessary to use it to solve the system response in modal coordinate system, and directly use $x = \Phi_N q_N$ for transformation relation to realize the transformation between them, but the calculation of $\Phi_N^T$ is often a huge amount of calculation, and the cutting back error will be generated, so the transformation relation is transformed as follows:

$$M_N^{-1}\Phi_N^T M x = M_N^{-1}\Phi_N^T M \Phi_N q_N.$$  (10)

With regard to $M_N^T = E$ and $\Phi_N^T M \Phi_N = M_B$, the above equation can be deformed into $q_N = \Phi_N^T M x$, so it could complete the conversion between that and plug it into the equation to solve the equation. So far, the solution according to the modal coordinate system has been completed and the modal coordinates are converted into physical coordinates by using the formula $x = \Phi_N q_N$.

### 3 Influence of structural forms on vibration characteristics of each cabin structure

In this study, the cabin structure model adopts a three-compartment model structure (shown in Figure 1). Bulkhead and deck plate thickness $t = 5$ mm. The deck is provided with beams with a spacing of $1.2$ mm and longitudinal bones with a spacing of $0.5$ m. The cross-section size of the beam is $61 \times 25 \times 91 \times 25$. The section size of longitudinal bone is $L100 \text{ mm} \times 63 \text{ mm} \times 7 \text{ mm}$. The spacing of bulkhead stiffeners is consistent with the spacing of bone materials on the deck, and the section size of stiffeners is $L100 \text{ mm} \times 63 \text{ mm} \times 7 \text{ mm}$. The material parameters are defined as $E = 2.1 \times 10^{11} \text{ N} \cdot \text{m}^{-2}$, $\mu = 0.3$, $\rho = 7,850 \text{ kg} \cdot \text{m}^{-3}$. In this study, the calculation is based on classical FEM, the plate adopts two-dimensional shell element, and the bone structure adopts one-dimensional beam element. The grid size is $0.02$ m, and the boundary conditions of the cabin are set as two-way simple supports. The total excitation force is $1$ N through the vertical direction, and the positions are shown in Figure 1 (red circle).

The vibration characteristics of C cabin deck with beams, longitudinal bones, or additional pillars are not considered in the calculation and analysis, and the influence of structure form on the vibration characteristics of the structure is compared with the reference structure. Two additional pillars are added inside cabin C, which are located on the beam. Unless otherwise stated, the influence of additional pillars is not considered in initial analysis.

#### 3.1 Influence of aggregate on structural vibration characteristics of each cabin

The overall vibration modes of C deck are mainly considered, and the first 12 order modal results are shown in Figure 2.
When there is no beam or longitudinal bone on C deck, the first natural frequency of C deck is 5.2 Hz, but when there is beam and longitudinal bone on C deck, the first natural frequency of C deck is 46.6 Hz. So, the beam and longitudinal bone have a great influence on natural frequency of the structure, which is because the structural stiffness of the beam and longitudinal bone increases greatly. In addition, the profile has a great influence on the mode of the structure.

### 3.2 Influence of aggregate on vibration transfer characteristics of each cabin structure

When there is no beam and longitudinal bone in the cabin C deck, the overall vibration response of each cabin deck is calculated. The calculated results are shown in Figures 3–5. It can be seen from the figures that the presence or absence of beam and longitudinal structure of
Cabin C deck has a great influence on the vibration transfer characteristics about deck of cabin C; however, that has less influence on the structural vibration transmission characteristics of cabins A and B.
3.3 Influence of pillars on structural vibration characteristics of each cabin

Adding two additional pillars to the cabin C, which are located on the beam, as shown in Figure 1. The influence of the pillars on the vibration characteristics of the multi-cabin structure is analyzed.

The overall vibration modes of C deck are mainly considered, and the results of the first to ninth modes are shown in Figure 6. Comparison of the first to ninth modes natural frequencies of the infrastructure with or without pillars is shown in Figure 7. Pillar can be seen that set on the structure of the third, six, nine order natural frequency of a smaller effect, has certain influence on other order natural frequency, this is due to the third, six, nine order modal mainly for stiffened plate or plate vibration, and the pillar position is located at cross beam, less effect on the modal, and had certain influence on other vibration modal. The presence of aggregate on the deck of cabin C basically has no effect on the deck modes of cabins A and B. Some modal results are shown in Figures 8 and 9.

3.4 Influence of pillar on vibration transmission characteristics of each cabin structure

When the deck of cabin C is equipped with pillar, the overall vibration response of each cabin deck is calculated. The calculated results are shown in Figures 10–12. The
pillar or nonpillar structure of cabin C deck has a great influence on the deck of cabin C, but that has a very small impact on the vibration transmission characteristics of the structures of cabins A and B.

The root means square value of vibration acceleration response of excitation points with or no pillar was analyzed. The calculated results are shown in Figure 13. The presence of no pillar in cabin C has little effect on transmission functions of the origin of deck excitation input points in cabin A. The vibration acceleration transfer functions at the input point have a large peak value at 65, 135, 149, and 192 Hz.

From the analysis results of free vibration characteristics and vibration transfer characteristics, it can be seen that the aggregate or pillar in C cabin deck has a great influence on vibration characteristics of C cabin deck itself, but basically has no influence on vibration characteristics of other compartments. Therefore, there is no need to consider the influence of structural changes on the surrounding cabins in the vibration optimization design of the target cabin structure.

3.5 Influence of plate thickness on structural vibration characteristics

In order to study the influence of deck thickness on structural vibration characteristics, $T$ is set on deck of cabin $t_1 = 7 \text{ mm}$, $t_2 = 9 \text{ mm}$ two kinds of thickness, and the structural reference thickness $t_0 = 5 \text{ mm}$ for comparative study. Relative to the reference structure, the deck thickness is $t_1 = 7 \text{ mm}$, the deck mass of cabin C increases by about 25%; deck thickness $t_2 = 9 \text{ mm}$, the deck mass of cabin C increases by about 51%. The overall variation trend of the calculated results of the overall natural frequency of cabin C deck under different deck thicknesses is shown in Figure 14. The change of deck thickness has little influence on the first order and second-order natural frequencies of the whole deck, and the increase of plate thickness has a decrease on first-order natural frequencies. This is because the stiffness of the first-order mode is mainly provided by the aggregate, and the increase of plate thickness has a smaller effect on increase of modal mass than that of modal stiffness. As the plate thickness...
increases, the natural frequency of the same order increases. This is because the mode shapes of these modes are dominated by the stiffened plate or lattice vibration.

Under different deck thicknesses, the mean square velocities of C deck are shown in Figure 15.

In 10–86 Hz frequency band, the deck thickness changed from 5 mm to 7 mm, 9 mm has little influence on the overall structural vibration. In the 86–300 Hz frequency band, the deck thickness is changed from 5 mm to 7 mm, 9 mm, and the structural vibration response is reduced. This is because the structural vibration response before 86 Hz is dominated by integral vibration, whereas after 86 Hz, it is dominated by stiffened plate or plate lattice vibration. For 109 Hz peak, the response peak decreases by 15.9 dB when the deck thickness changes from 5 to 7 mm. When the thickness changed to 9 mm, the response peak decreased by 19.7 dB.

### 4 Process of vibration protection of ship structure

First, the information of the main vibration sources on the ship was sorted out, then it uses this information to determine the main excitation frequencies to be considered for each vibration source. Then the range of resonance risk caused by various vibration sources is determined. The first-order natural frequency of the structure is analyzed according to the total and local vibration of the ship. The local vibration is studied according to the three levels of plate frame vibration, stiffened plate vibration, and plate lattice vibration. Determine the frequency range of resonance risk that needs to be considered, according to the frequency range, the vibration protection design is carried out for each structure with resonance risk. The flow chart of vibration protection of ship structure is shown in Figure 16.

### 5 Conclusion

In this study, the influence of the structure form on the vibration mode and vibration transfer characteristics of the target structure and other cabin structures in the multicabin structure is investigated by changing the structure form of the target cabin, and the vibration protection flow of ship structure was given. The conclusion of this study is as follows:

1. The change of the structure form of ship’s target cabin has a great impact on vibration characteristics of target cabin structure, and a small impact of other cabin structures. Thus, it is not necessary to consider the influence of target structure on other structures when designing the ship vibration protection.
2. The change of deck plate thickness has a great influence on the frequency band which is dominated by the vibration response of stiffened plate and grillage, especially the resonance peak of the stiffened plate or grillage is significantly improved, because the change of deck plate thickness avoids the resonance phenomenon of the original natural frequency.

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