ADVANCES OF FLASH LIDAR DEVELOPMENT ONBOARD UAV

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ABSTRACT:

A small cost-low civilian UAV (Unmanned Aerial Vehicle - UAV) platform usually requests that all carried components should be light in weight, small in volume, and efficient in energy. This paper presents the advance of a pre-mature of flash LiDAR system including laser emitting system, associate with the pulsed voltage technology. A complete laser emitting system, including laser diode, conic lens, alignment, divergence angle, etc., has been designed and implemented. The laser emitting system is first simulated and tested using 3D-Tool software, and then manufactured by an industrial company. In addition, a novel power supply topology based on two coupled coils, pulse generator circuit, and a fast switch, is proposed since several 100 V in voltage, 10-100 A in current, several hundred millisecond in pulse width is needed for flash LiDAR system onboard a small low-cost civilian UAV platform, and the traditional power supply had problems in efficiency and bulk. Finally, laser emitting and the power supply are assembled and tested. The size of laser footprint is 4398.031 mm x 4398.031 mm in x and y axes, respectively, when shifting from a flight height of 300 m, which is close to the theoretic size of 4.5 m x 4.5 m. The difference of 102 mm can meet the requirement of flash LiDAR data collection at a flight height of 300 m. Future work on extensive and on-going investigation and investments for a prototype of flash LiDAR system is drawn up as well.

1. INTRODUCTION

The 3D imagery has widely been applied in such as terrain mapping, disaster rescue, helicopter obstacle avoidance, range navigation, urban planning, environmental monitoring, resource exploration, and so on. Many technologies have been developed for automatic acquisition of 3D imagery, such as typical stereo aerial imager onboard airborne. Compared with conventional intensity image captured by optical sensors onboard either airborne or spaceborne, three dimensional (3D) image captured by active sensor offers more information about the target, such as elevation, distance, position and structure (Zhou et al., 2011). Traditionally typical airborne scanning LiDAR (Light detection and Ranging) sensor onboard manned airplane is an active sensor, and is capable of acquiring 3D imagery of an arbitrary object or scene by measuring the time delay of a round trip of a laser pulse directed at points.

In recent years, a UAV platform becomes more and more interesting, since it is capable of quickly reaching the target area and deploying mission to the remote sites at a low-cost. A small low-cost civilian UAV platform often has limitation to its carried components in volume and weight. These limitations have seriously hindered its applicability in practice, since it requires lightness in weight, small in volume, and efficiency in power supply. For this reason, this paper presents an innovative idea about airborne flash LiDAR system. The flash LiDAR systems are analogous to a camera with a flashbulb (flood illumination), but with the flash being provided by laser illumination and the use of a detector with a clock to determine the time it takes for the flash to depart, reflect off of the target, and return (**). By measuring the time of flight of the reflected laser pulse, the sensor can determine a range measurement along with intensity for each pixel in the image. This information set over the range of all detector pixels is referred to as a 3D LiDAR imagery (Zhou et al., 2011; Yang and Zhou, 2011).

2. DESIGNED FLASH LIDAR FLOWCHART

2.1 General Framework

The proposed flash LiDAR onboard a UAV system is depicted in Figure 1. As seen from Figure 1, it consists of five basic functionality modules. They are (1) laser and laser emitting sub-system, (2) APD and laser receiver sub-system, (3) micro-control and processing sub-system, (4) POS subsystem, and (5) LiDAR point cloud post-processing sub-system. The details of each sub-system are presented by Zhou and Yan (2011). A brief review is presented below.

![Figure 1. The flowchart of airborne flash LiDAR system](image)
(1) Laser emitting sub-system: TPGAS2S09H diode laser with wavelength of 905 nm is used for this project. This laser emitting sub-system is to generate high energy of laser. As a initial research, a prototype with 5 × 5 pixels2 3D image array with <5cm ranging accuracy, in real time at rates up to 30 Hz, is designated.

(2) Receiver sub-system: The most common detector used to detect the laser echo pulse is linear mode avalanche photodiode (APD). This research adopted the Geiger-mode APD for receiving photon single. 3D imaging laser radar that uses Geiger-mode APDs is being developed (Aull and Marino, 2005; Daniel, 2003; Johnson, 2003; Aull et al., 2002; Marion et al., 2003). Because of the effect of background light noise, a new system design and new data processing method need to be developed. The details of this sub-system will be reported in the near future.

(3) Micro-Control Sub-system: which is core of system. The microprocessor controls the entire system and enables real-time control laser emitting, receiving, measure of time interval, data sampling, storage, etc. The control device also provides interactive communication with each modules. The control device is, therefore, usually designed as keyboard display unit.

(4) POS subsystem: which is used to provide the attitude of each laser ray and central position of laser emitter for calculation of 3D geodetic coordinates in a given coordination system.

(5) LiDAR point cloud pre-processing: which is used to generate XYZ coordinates on the basis of range, attitude angles of each laser ray and central coordinates of laser emitter.

2.2 Principle of Flash Laser Emitting

A brief description is given in this Section.

If the size of the emitting plan of a diode laser is \( L_v \) and \( L_h \) in length and width, and the divergence angle of the bundle of lasers is \( \theta_v \), and \( \theta_h \) in horizontal and vertical plan, the focal length is \( f_v \), the laser source is located at the focus (see Figure 2).

With geometry in Figure 2, we have
\[
\theta_v = 2 \arctan \left( \frac{L_v}{2f_v} \right) \tag{1}
\]

Generally, \( L_v \ll 2f_v \), so the divergence angle of plan is
\[
\theta_v \approx \frac{L_v}{f_v} \tag{2}
\]

As observed in Eq. 2, the divergence angle is negatively proportional to the focal length.

To make the direction of emitting laser is parallel to the direction of entry laser, the diameter of aperture must be at least
\[
D_d = L_v + 2f_v \tan \left( \frac{\theta_v}{2} \right) \tag{3}
\]

Usually, \( L_v \ll D_d \), the least diameter of aperture is expressed by
\[
D_d \approx 2f_v \tan \left( \frac{\theta_v}{2} \right) \tag{4}
\]

So, the least width of laser plan is
\[
D_{vo} = D_d - L_v \tag{5}
\]

Substitute Eq. 3 into Eq. 5, we have
\[
D_{vo} = 2f_v \tan \left( \frac{\theta_v}{2} \right) \tag{6}
\]

As observed from Eq. 6, the width of laser plan is positively proportional to the focus length.

Combined Eq. 2 and Eq. 6, it is hard to simultaneously meet both conditions, which results in difficulty of designing a reasonable and powerful laser emitter sub-system.

With the same method, we have divergence angle is
\[
\theta_h \approx \frac{L_h}{f_h} \tag{7}
\]

To make laser power in the emitting system in vertical plan, the least diameter of aperture is
\[
D_{h} \approx 2f_h \tan \left( \frac{\theta_h}{2} \right) \tag{8}
\]

The least width of emitter laser in vertical plan is
\[
D_{ho} = 2f_h \tan \left( \frac{\theta_h}{2} \right) \tag{9}
\]

For example, if the emitter plan is 225×400μm, the divergence angle is 10° in planar direction and 25° in vertical direction, and divergence angle is 30-33mrad after alignment.

With the two cases above, this paper select the fast –axis conical lens with a focal length of 7.7 mm, and 7 mm and 9 mm in height and length; select a slow-axis conic lens with a focal length of 13.7 mm, and 13 mm and 15 mm in height and length.

The major purpose of the emitting system is to align the emitting laser. Thus, when selecting an emitting laser system, we have to simultaneously consider both the alignment characteristic and the complexity of optical system in structure, processing and manufacturing. With the computational parameters, the 3D model is simulated using ZEMAX software, as depicted in Figure 3. The simulated 3D model, associated with its size is depicted in Figure 4.
3. COMPONENTS OF LIGHT-EMITTING SYSTEM

3.1 Laser Diode

TPGAS2S09H diode laser with wavelength of 905 nm is used for this project. Diode lasers in the 1300 nm to 1600 nm regime are used in a variety of applications including pumping, range finding, materials processing, and aesthetic medical treatments. In addition to the compact size, efficiency, and low cost advantages of traditional diode lasers, high power semiconductor lasers in the eye-safe regime are becoming widely used in an effort to minimize the unintended impact of potentially hazardous scattered optical radiation from the laser source, the optical delivery system, or the target itself.

The junction plane, light-emitting surface and divergence angle of diode laser are depicted in Figure. As seen, the light-emitting surface is not symmetric with the divergence angle. For each light-emitting element, the length in the direction parallel to junction plane is about 230 μm, and the height in the direction of vertical to junction plane is 1 μm. The divergence angles in direction parallel to and vertical to junction plane is 10° and 25°, respectively. The major functionality of light-emitting optical system of diode laser is aligning and shaping to the bundle of light, which are no-symmetric, big divergence angle and poor quality, to make them meet the requirement of laser, i.e., small divergence angle, alignment and high energetic light.

The final required bundle of laser is: Divergence angles in horizontal and vertical direction is approximately 30~33 mrad, the energy Transmittance laser reach above 95%.

This laser emitting sub-system is to generate high energy of laser. As a initial research, a prototype with $5 \times 5$
pixels square 3D image array with <5 cm ranging accuracy, in real time at rates up to 30 Hz, is designated.

3.2 Power Driving Circuit for Laser Diode

A laser diode accompanying with its power supply is an important part of LiDAR system onboard UAV. The traditional power supply had problems in efficiency and bulk, has been demonstrated that it is not proper for application on a small low-cost civilian UAV platform. How to design a power supply for laser diode to meet the requirement of light, small, and energy efficiency, is a valuable work. In this paper, a novel power supply topology for LiDAR system on board UAV platform is presented. The power supply is composed of two coupled coils, pulse generator circuit, and a fast switch (Zhou and Yang, 2011).

Coffey (2009) though that the power-supply largely impacts the performance of laser-diode for a given specification. Different methods of design and implementation of the laser diode power supply have been proposed by Cui et al. (2011), Zhou et al. (2011), Yang et al. (2011). A novel low power supply for DC-coupled 1.25 Gb/s laser diode driver is suggested by Fu et al. (2006). With the MAX797, driver circuit of the high-power laser diode was proposed by Li and Xu (2008). For a pulsed power modulated for high output power for laser fuze was proposed by Guo et al. (2011). The automatic power control of DC-coupled burst-mode laser diode was presented by Zhang et al. (2009) and Li et al. (2008).

A proposed schematic diagram of the power supply for laser diode is depicted in Figure 7. As seen from Figure 7, Driving power supply is composed of two coupled coils (could be replaced by a pulsed transformer), a thyristor, TTL pulse signal, resister and capacitor [3-6]. With this topology of power supply, the input voltage of the power supply is a +28 V DC voltage from airplane, and the output maximum voltage is 300 V. Before the TTL pulse signal coming, the power supply is in a steady state. During this steady state, capacitor C1 is charged by the input voltage through R1 and L1 to +28V; the thyristor Q1 is turn off because the TTL pulse signal is in a low state; there is no current in L2; and the output voltage is 0. If a pulse signal comes to the gate of the thyristor Q1, Q1 will turn on, and C1 will release its energy through Q1 and L1 rapidly cause the low resistance in the circuit. This situation will generate a high voltage across L2; for L1 and L2 are strongly coupled. The voltage across L2 can be controlled through turns ratio of L2 and L1, here we should choose a turns ratio of 10. Assuming the other parameters of the two inductors are all the same besides the turns ratio, the inductance of L2 will be 100 times larger than L1. The generated pulsed voltage is around 300 V (voltage of L1 plus L2). The voltage generated by L2 is coupled to laser diode D1 through R2 and C2.

In order to meet the requirement of laser diode adopted in LiDAR scanner onboard small UAV platform, the parameters of each circuit elements have been specified and are depicted in Figure 8. Instead of traditionally trying every element parameter one by one, a circuit model by PSpice is set up and simulation experiment is conducted. This method is fast and cost efficiency.

With the above design, experiments and test, a prototype of power supply is produced, as shown in Fig. 9. The prototype of power supply is synchronous with a pulse signal generated by control circuit, and the output voltage and current adjustable to fit laser diode, and the repeat pulse generation is up to 1000 pulses per second.
be bigger than 102 mm due to error, alignment, atmospheric impact, etc. In addition, the divergence angles in horizontal and vertical directions are almost the same.

![Diagram of footprint](image)

**Figure 10.** Simulated footprint at a flying height of 300 m above local average elevation

5. **MANUFACTURES**

The laser emitting device is manufactured by Zhenhong LLC at Dongkuan, Guangdong, China. With the designed parameters by our theoretic analysis, 3D-Tool software is used for the visualization of laser emitting device, as shown in Figure 11. 3D-Tool is a powerful, cost effective tool that has helped customers substantially in the management of manufacturing technologically advanced products. 3D-Tool allows to see, evaluate, measure, cross section.

The emitting laser component consists of basis, front and rear cushion, ring, tube, polysulfone sets, septa, diode laser, as depicted in Figure 11. Terms are as follows.

- a1: base for rear lens
- a2: base for front lens
- a3: isolation piece for separating front and real lens
- a4: flating pieces for stabling the front lens
- a5: screw tube for installing and stabling polysulfone
- a6: polysulfone for stabling laser diode
- b1: cushion (base)
- c1: lock ring for stabling tube
- d1: laser diode
- e1: emitting tube with two lens, front lens and rear lens inside

6. **CONCLUSION**

This paper presents the advances of the flash LiDAR initiative with focus on light-emitting system. The proposed flash LiDAR is imaged with the 3D imaging mode, and the entire scene within the sensor’s field of view (FOV) at a single flash of the laser. This is because the proposed flash LiDAR have many advantages such as inherently insensitive to ambient and stray light, and the glint and clutter outside of the expected range to the target filtered and discarded. Since the flash LiDAR captures a complete image with each flash. This capability makes data pre-processing and post-processing extremely efficient and reliable since range, bearing, and pose algorithms do not have to deal with interpreting data from glint and clutter. Rapid frame rates, waveband filtering, and time gating on the return signal increase glint tolerance even further. Thus these advantages will bring significant improvement for its application in the traditional fields such as topographic mapping and disaster monitoring.

This paper only reports a pre-mature of flash LiDAR technology. Extensive and on-going investigation and investments for a prototypes of flash LiDAR system is under development. The system will be tested in the laboratory and in-flight airplane and helicopter in field tests.

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Figure 11. Hardware components of laser emitting system