A Convenient Approximation of the Projectile Motion with Quadratic Air Resistance

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A convenient approximation of the projectile motion with quadratic air resistance

Marc Landon

Abstract A convenient approximated analytic solution is proposed for the problem of the motion of a body under a resistive force, acting in the magnitude of the squared velocity of the body. This solution is an explicit function of time, that keeps a good behavior both near the initial state and far from the initial state.

To obtain a general analytic solution, we firstly used a reduction principle to be able to manipulate scalar objects, and we analyzed limit behaviors, both near the initial state and far from the initial state. Secondly, we proposed an approximated analytic solution with heuristics based on the built knowledge.

Finally, a robust and stable integration scheme is proposed, based on the obtained analytic solution. We compared the scheme with other standard integration schemes.

Keywords particle dynamics · quadratic drag · projectile motion · integration scheme

1 Introduction

One of the classic problems of physics concerns an object moving through a resistive medium. We focus on two particular laws of the resistive force, either linear or quadratic in the relative velocity in respect with the medium. Both laws are approximation in specific regimes of the motion, as detailed in any textbook on hydrodynamics. Solutions with a linear and a quadratic resistive force have already been obtained. The case of a motion with a linear resistive force is explicitly solvable, by meaning that the exact solution can be expressed by elementary functions of time. The case with a quadratic resistive force is not explicitly solvable. We call ‘elementary functions’ all functions that can be locally expressed with convergent power series.

Our aim is to provide a stable integration scheme of the projectile motion, where ‘stable’ means that the scheme is unconditionally stable in regards with the time increment (also called delta-time or time-step). This scheme could be used

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for real-time applications for which the time-step is unbounded. Also, it could
be used to handle pre-warmed simulations, where simulations start at a non-zero
time. (For example, loading a game level with smoke stacks already fully grown).

1.1 Previous work

Implicit analytic solutions have already been proposed. The solution described in
[1] parameterizes the velocity magnitude with direction angle of the velocity. Even
when the proposed solutions can be expressed by elementary functions, they are
expressed with parametric variables which are built from trajectory parameters.
Therefore, such solutions cannot be used to implement time-based integration
schemes.

Explicit solutions have already been proposed using various approximation
methods. The solutions found by [2] are valid for a nearly flat trajectory, and only
near the initial state or far from the initial state. More general solutions have also
been found in the form of Taylor series near the initial state, by [3]. Lastly, [4]
provides solutions as a ratio of two series expansions near the initial state. However,
time-based integration schemes implemented built from series expansions won’t be
unconditionally stable.

1.2 Equation of motion

From the first Newton law, the time derivative of the motion quantum is equal
to the resultant of the forces applied on the body. We consider a resultant of the
forces that is composed of a constant and uniform force, and a resistive force (also
called drag force) that depends on the velocity. We also consider that the mass of
the body is constant, and we reduce the body to its center of inertia and its mass.
We obtain the following well-known motion equation:

\[
m \frac{d\vec{v}}{dt} = \vec{F}_c + \vec{F}_{\text{drag}}
\]

where \( m \) is the non-null mass of the body, \( \vec{v} \) its velocity, \( \vec{F}_c \) a constant and
uniform force and \( \vec{F}_{\text{drag}} \) the drag force. \( \vec{F}_c \) often includes the gravity action, and
it is replaced by \( \vec{Q} \) such as \( \vec{F}_c = m \vec{Q} \). We obtain the general form of the equation,
where the varying terms are placed on the left-hand side and the constant terms
on the right-hand side:

\[
\frac{d\vec{v}}{dt} - \frac{1}{m} \vec{F}_{\text{drag}} = \vec{Q}
\]  

(1)

1.3 The drag force

The drag force is the resistance of the medium to the body motion. The medium
is considered in a stable equilibrium state, uniform in space and infinitely wide.
In this paper, we will address two laws that describe the drag force: a linear
law and a quadratic law:
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\[
\vec{F}_{\text{linear}} = -k_l (\vec{v} - \vec{v}_{\text{wind}}) \\
\vec{F}_{\text{quadratic}} = -k_q \|\vec{v} - \vec{v}_{\text{wind}}\| (\vec{v} - \vec{v}_{\text{wind}})
\]

where \(\vec{v}_{\text{wind}}\) is the constant and uniform velocity of the medium, \(k_l\) and \(k_q\) are the positive constants, called drag coefficients.

1.4 Analytic solution with a linear drag

In this section, we recall common results when the drag force follows a linear law.

The insertion of linear drag force (2) into (1) gives:

\[
\vec{v}'(t) + \alpha (\vec{v}(t) - \vec{v}_{\text{wind}}) = \vec{Q}
\]

where \(\alpha = \frac{k_l}{m}\) and \(\vec{v}'\) denotes the derivative of \(\vec{v}\) of time. As the wind is constant and uniform, this equation can be rewritten with the relative velocity \(\vec{u} = \vec{v} - \vec{v}_{\text{wind}}\):

\[
\vec{u}' + \alpha \vec{u} = \vec{Q}
\]

given the initial condition \(\vec{u}(t_0) = \vec{u}_0\).

**Solution of the degenerated case** With \(\alpha = 0\), the equation admits a solution \(\vec{u}(t) = \vec{u}_0 + (t - t_0)\vec{Q}\). By removing the drag force, the motion is the well-known motion of an uniformly accelerated body in a Newtonian frame.

**Solution of the general case** With \(\alpha > 0\), the equation admits a solution \(\vec{u}(t) = (\vec{u}_0 - \vec{u}_\infty)e^{-\alpha(t-t_0)} + \vec{u}_\infty\) where \(\vec{u}_\infty = \frac{1}{\alpha}\vec{Q}\). The solution converges toward the terminal velocity \(\vec{u}_\infty\).

As the velocity is a function that can be analytically integrated, the position can be expressed as a function of time. The exact analytic solution can be used in the integration scheme. Consequently, the scheme is exact, and it obviously does not diverge when the time increment becomes large. When \(\alpha\) tends toward zero, the position can be expressed as a Taylor expansion in order to prevent numerical instabilities.

**Scheme (integration scheme with a linear drag)** Given the position \(\vec{x}_n\) and the velocity \(\vec{v}_n\) at the time \(t_n\), the new position and velocity at time \(t_{n+1} = t_n + \delta t\) are:

\[
\vec{v}_{n+1} = \vec{v}_{\text{wind}} + \vec{u}_\infty + (\vec{v}_n - \vec{v}_{\text{wind}} - \vec{u}_\infty)e^{-\alpha \delta t} \\
\vec{x}_{n+1} = \vec{x}_n + (\vec{v}_{\text{wind}} + \vec{u}_\infty)\delta t + (\vec{v}_n - \vec{v}_{\text{wind}} - \vec{u}_\infty)\frac{1 - e^{-\alpha \delta t}}{\alpha}
\]
2 Implicit solution with a quadratic drag

Unlike with the linear drag, an exact analytic solution that can be expressed with elementary functions of time does not exist, to our knowledge so far. We will first deduce some properties of the solution. Secondly, we will express the solution in an implicit form, by meaning that the solution is expressed with the solution itself.

The equation of motion (1), with the quadratic drag force (3), becomes:

\[ \vec{v}' + \alpha \| \vec{v} - \vec{v}_{\text{wind}} \| (\vec{v} - \vec{v}_{\text{wind}}) = \vec{Q} \]

where \( \alpha = \frac{k}{m} \). As the wind is constant and uniform, this equation can be rewritten with the relative velocity \( \vec{u} = \vec{v} - \vec{v}_{\text{wind}} \):

\[ \vec{u}' + \alpha \| \vec{u} \| \vec{u} = \vec{Q} \]  

(5)

given the initial condition \( \vec{u}(t_0) = \vec{u}_0 \).

2.1 Properties

In this section, the properties of the solution of the differential equation (5) will be detailed. This will help to restrict approximations, and to guarantee their behavior.

The motion without drag \( (\alpha = 0) \) admits a solution, already described in section 1.4. So we enforce \( \alpha > 0 \). Firstly, we can restrict the relative velocity \( \vec{u} \) to be in a 2D space. Indeed, the variation of the relative velocity is a weighted sum of the vectors \( \vec{Q} \) and itself. Consequently, the velocity stays in the plane formed by \( \vec{Q} \) and \( \vec{u}_0 \). Moreover, if \( \vec{Q} \) is null, then the velocity will stay in a 1D space. This case is detailed in the section 3. Then, we consider here that \( \vec{Q} \) is not null. Secondly, we assume that the solution exists, is unique, and admits a finite limit on infinite time.

Stationary solution : The equation admits a stationary solution:

\[ \vec{u}(t) = \vec{u}_\infty = \frac{1}{\sqrt{\alpha \| \vec{Q} \|}} \vec{Q} \]  

(6)

We define \( \gamma = \sqrt{\alpha \| \vec{Q} \|} \).

Asymptotic properties of the solution around the initial time : Around the initial state, we can trivially write:

\[ \vec{u}(t_0) = \vec{u}_0 \]

\[ \vec{u}'(t_0) = \vec{Q} - \alpha \| \vec{u}_0 \| \vec{u}_0 \]
Asymptotic properties of the solution on infinite time: The solution converges toward \( \vec{u}_\infty \) on infinite time. Let’s introduce a decomposition of \( \vec{u} \) such as:

\[
\vec{u}(t) = \vec{u}_\infty + f(t)\vec{W}(t) \tag{7}
\]

where \( f \) is a non-null function of time that tends toward zero on infinite time, and \( \vec{W} \) is an unit vector (\( \lVert \vec{W}(t) \rVert = 1 \)). As \( \vec{W} \) is an unit vector, its derivative is perpendicular to \( \vec{W} \). Because the velocity belongs to a 2D-space, we decompose\( \vec{W} \) such as:

\[
\vec{W}(t) = \cos(\theta(t))\vec{n}_\infty + \sin(\theta(t))\vec{r}_\infty \tag{8}
\]

where \( \theta \) is a function of time, \( \vec{n}_\infty \) the unit vector with the same direction as \( \vec{u}_\infty \), and \( \vec{r}_\infty \) an unit vector, perpendicular to \( \vec{n}_\infty \). We inject (7) into (5):

\[
f(t)\vec{W}(t)' + f(t)'\vec{W}(t) + \alpha\vec{u}_\infty + f(t)\vec{W}(t)\lVert \vec{u}_\infty + f(t)\vec{W}(t) \rVert = \vec{Q}
\]

We assume that \( \vec{Q} \) is an unit vector, \( \vec{n}_\infty \) the unit vector with the same direction as \( \vec{u}_\infty \), and \( \vec{r}_\infty \) an unit vector, perpendicular to \( \vec{n}_\infty \). We inject (7) into (5):

\[
f(t)\vec{W}(t)' + f(t)'\vec{W}(t) + \alpha\vec{u}_\infty + \vec{u}_\infty = 0
\]

We project on \( \vec{W}' \) (\( \vec{W} \) terms vanish), and we factorize by \( f \):

\[
\left( \vec{W}'(t) \cdot \vec{W}'(t) \right) + \alpha \left( \vec{W}(t) \cdot \vec{u}_\infty \right) \left( \vec{u}_\infty \cdot \vec{W}'(t) \right) = 0
\]

We are using (8), and by excluding the equation \( \theta'(t) = 0 \), we obtain:

\[
\theta'(t) + \frac{1}{2} \alpha\lVert \vec{u}_\infty \rVert^2 \sin(2\theta(t)) = 0
\]

This ordinary differential equation admits a limit on infinite time, and 2 stationary points (within \([0, \pi]\)). However, only the stationary point \( \theta = 0 \) is stable. Finally, we can conclude that, either it stays on the unstable stationary point:

\[
\vec{u}_\infty \cdot \vec{W}'(t) = 0 \quad (\vec{u}_0 \text{ and } \vec{Q} \text{ are collinear})
\]

either it converges toward the stable stationary point:

\[
\lim_{t \to \infty} \left( \vec{W}(t) \cdot \vec{Q} \right) = 0 \tag{9}
\]

2.2 Implicit form of the solution

An implicit solution can be found by using the reduction principle for certain coupled systems of ordinary differential equations introduced by [4]. Let \( P \) denote the scalar function \( P : t \to \alpha \lVert \vec{u}(t) \rVert \). The equation (5) can be written in the
canonized form:

\[
\vec{u}' + P\vec{u} = \vec{Q} \tag{10}
\]

The solution of (10) can be written as:

\[
\vec{u}(t) = \vec{u}(t_0) + \vec{Q} \left( e^{- \int_{t_0}^{t} P(r) dr} \right) \int_{t_0}^{t} e^{\int_{s}^{t} P(r) ds} ds dr \tag{11}
\]

Proof The first step is to find the integral solution of the homogeneous equation \( \vec{u}' + P\vec{u} = 0 \). The second step is to write the solution of the origin equation as the
product of the homogeneous solution and an unknown function. The last step is to remind the solution exists and is unique.
Let \( \varphi: t \to e^{\int_{t_0}^t P(r) dr} \) denote a scalar function of time. The dependency on the variable \( t_0 \) is implicit. Let \( \phi \) denote the integral of \( \varphi: \phi: t \to \int_{t_0}^t \varphi(r) dr \). Then the solution (11) can be expressed as:

\[
\vec{u}(t) = \vec{u}_0 + \frac{\vec{Q} \phi(t)}{\phi'(t)}
\]

**Differential equation on \( \phi \):** The scalar function \( \phi \) is a solution of the second-order differential equation:

\[
\phi''(t) = \alpha \|\vec{u}_0 + \phi(t)\vec{Q}\|
\]

**Proof** By injecting (12) into (5), or by following the deduction detailed in [4]. \( \square \)

**Asymptotic properties of \( \phi \) at the initial time:** From the asymptotic properties of \( \vec{u} \) at the initial time and the equation (13), we can trivially determine the derivatives at \( t_0 \).

\[
\begin{align*}
\phi(t_0) &= 0 \\
\phi'(t_0) &= 1 \\
\phi''(t_0) &= \alpha \|\vec{u}_0\| \\
\phi'''(t_0) &= \alpha \frac{\vec{u}_0 \cdot \vec{Q}}{\|\vec{u}_0\|}
\end{align*}
\]

**Asymptotic properties of \( \phi \) on infinite time:** Given the asymptotic properties of \( \vec{u} \) on infinite time and the expression of \( \vec{u} \) in (12), we can deduce that

\[
\phi(t = +\infty) \simeq ae^{\gamma t} - \frac{\vec{u}_0 \cdot \vec{Q}}{\|\vec{Q}\|^2} + o(1)
\]

with \( \gamma = \sqrt{\alpha \|\vec{Q}\|} \) and \( a \) a constant.

**Proof** We equalize (12) and (7):

\[
\vec{u}(t) = \vec{u}_0 + \frac{\phi(t)\vec{Q}}{\phi'(t)} = \vec{u}_\infty + f(t)\vec{W}(t)
\]

We project on \( \vec{Q} \):

\[
\vec{u}_0 \cdot \vec{Q} + \phi(t)\|\vec{Q}\|^2 = \phi'(t) \left( \frac{\|\vec{Q}\|^2}{\gamma} + f(t)\vec{W}(t) \cdot \vec{Q} \right)
\]

We focus on the behavior on infinite time. Let \( \phi_\infty \) denote the asymptotic function of \( \phi \) at infinite time. Using (9), we obtain the following differential equation:

\[
\frac{\vec{u}_0 \cdot \vec{Q}}{\|\vec{Q}\|^2} + \phi_\infty(t) = \frac{1}{\gamma} \phi'_\infty(t)
\]

Thus:

\[
\phi_\infty(t) = ae^{\gamma t} - \frac{\vec{u}_0 \cdot \vec{Q}}{\|\vec{Q}\|^2}
\]

\( \square \)
3 Analytic solution with a quadratic drag in one dimension

In one dimension, the motion under a quadratic drag admits an analytically solution. This also matches with the multi-dimensional motion where \( \vec{u}_0 \) and \( \vec{Q} \) are co-linear, or when \( \vec{u}_0 \) or \( \vec{Q} \) is null. The scalar variant of the equation (5) is:

\[ u' + \alpha |u|u = Q \tag{19} \]

given the initial condition \( u(t_0) = u_0 \). We assume \( \alpha > 0 \). Without loss of generality, we can enforce \( Q \geq 0 \). Indeed, as \( u^* = -u \) is solution of the equation \( u'^* + \alpha |u^*|u^* = -Q \), we can choose \( u^* \) as the new unknown velocity. This is equivalent to inverting the space axis.

First-all, the equation (19) admits a stationary solution:

\[ u(t) = u_\infty = \sqrt{\frac{Q}{\alpha}} \]

The analysis of the equation (19) is not detailed here, but we remind important results: the equation admits a unique solution (given an initial condition), either a solution is the stationary solution, either the solution never reaches or crosses \( u_\infty \). Solutions are monotone and bounded, and they always converge to \( u_\infty \).

For \( Q = 0 \), we have the homogeneous equation; the related motion’s regime is called the homogeneous regime. For \( Q > 0 \), as \( u \) tends towards \( u_\infty \) which is greater than zero, and by the limit definition, we obtain the following statement:

\[ \exists t_p > t_0 \ \forall t > t_p, u(t) > 0 \]

From this, we deduce that \( u \) will become and stay positive after a certain time. If \( u_0 \) is negative, knowing that the solution is monotone, then \( u \) will stay negative within a limited time, and will become positive. We call down-to-up regime the time interval in which \( u \) is negative, and terminal regime the time interval in which \( u \) is positive.

3.1 Down-to-up regime

We have \( u_0 < 0 \) (and \( Q > 0 \)). Let \( t_p \) such as \( u(t_p) = 0 \). Within \([t_0, t_p]\), the motion equation becomes:

\[ u' - \alpha u^2 = Q \]

The solution is:

\[ u(t) = \tan(\gamma(t - t_p)) \]

for \( t \in [t_0, t_p] \), where \( \gamma = \sqrt{\alpha Q} \) and

\[ t_p = t_0 + \frac{-1}{\gamma} \text{arctan}(\frac{u_0}{\sqrt{Q}}) \]

From this solution, we did not find an analytic expression of \( \phi \).
3.2 Terminal regime

We have $u_0 \geq 0$ (and $Q > 0$). The motion equation becomes:

$$u' + \alpha u^2 = Q$$

The solution is:

$$u(t) = \frac{1}{\frac{\alpha}{u_0} + \gamma} \frac{1}{\frac{2u_\infty}{2u_\infty}} e^{-2\gamma(t-t_0)} + \frac{1}{2u_\infty} - u_\infty$$

where $\gamma = \sqrt{\alpha Q}$. From this solution, we obtain an analytic expression of $\phi$:

$$\phi(t) = \frac{1}{\gamma^2} \left( \frac{\alpha u_0 + \gamma}{2} e^{\gamma t} + \frac{\alpha u_0 - \gamma}{2} e^{-\gamma t} - u_0 \right)$$

3.3 Homogeneous regime

When $Q = 0$, the solution is:

$$u(t) = \frac{u_0}{1 + u_0 \alpha (t - t_0)}$$

From this solution, we obtain an analytic expression of $\phi$:

$$\phi(t) = t + \alpha u_0 \frac{t^2}{2}$$

This solution fits with the limit of the terminal solution when $Q \to 0$. The proof can be done with a Taylor expansion from the expressions of $u$ and $\phi$ in the terminal regime.

3.4 Degenerated regime

When $\alpha = 0$, the solution is $u(t) = Q(t - t_0) + u_0$ and $\phi(t) = t$. Like the homogeneous regime, the degenerated regime is a limit of the terminal regime when $\alpha \to 0$. We can obtain a truncated Taylor expansion of $\phi$ from its expression in the terminal regime:

$$\phi(t) = t + \alpha u_0 \frac{t^2}{2} + \alpha Q \frac{t^3}{3!} + \alpha^2 Q u_0 \frac{t^4}{4!} + \alpha^2 Q^2 \frac{t^5}{5!} + ...$$

4 Approximation of the motion of a point under a quadratic drag

Given that the velocity can be expressed with the function $\phi$ thanks to the equation (12), we will propose an approximation of $\phi$. One of our main outcome is to write $\phi$ as a sum of exponential terms of time. Moreover, the approximation can match with the exact solution when the motion is in the one-dimensional terminal-regime.

We can have up to 7 parameters to build the expression of $\phi$ since we gathered 7 relations: 5 relations near the initial state with the equations (14), (15), (16) and
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(17), plus 2 relations far from the initial state with the equation (18). The general form is proposed as:

$$\phi_G(t) = a_0 + a_1 e^{\gamma_1 t} + a_2 e^{\gamma_2 t} + a_3 e^{\gamma_3 t}$$

(20)

With this form, the previous properties of $\phi$ can be more detailed:

– Equations (14) and (15) are essential as they are deducted from the definition of $\phi$. It leads to the following system:

$$a_0 + a_1 + a_2 + a_3 = 0$$
$$\gamma_1 a_1 + \gamma_2 a_2 + \gamma_3 a_3 = 1$$

(21)

– Equation (16) allows that the velocity obeys to the equation of motion at the initial state. It leads to:

$$\gamma_1^2 a_1 + \gamma_2^2 a_2 + \gamma_3^2 a_3 = \alpha \|\vec{w}_0\|$$

(22)

– Equation (17) allows even more accurate motion near the initial state, but the coefficients will be more sensitive to the initial state. It leads to:

$$\gamma_1^3 a_1 + \gamma_2^3 a_2 + \gamma_3^3 a_3 = \alpha \frac{\vec{a}_0 \cdot \vec{Q}}{\|\vec{Q}\|}$$

(23)

– Equation (18) is in fact composed of 2 relations: the exponential coefficient value allows that the velocity converges to the terminal velocity far from the initial state, while the constant coefficient value impacts the way the velocity converges to the terminal velocity. It leads to:

$$\gamma_1 = \sqrt{\alpha \|\vec{Q}\|}$$
$$\gamma_2 < 0$$
$$\gamma_3 < 0$$
$$a_0 = -\frac{\vec{a}_0 \cdot \vec{Q}}{\|\vec{Q}\|^2}$$

(24)

(25)

4.1 Identification of the parameters

Without loss of generality, we fix $\vec{Q} = -qe_y^e$ and $\vec{w}_0 = w_0 (\sin(\theta_0)e_x^e - \cos(\theta_0)e_y^e)$ where $e_x^e$ and $e_y^e$ are 2 perpendicular unit vectors and $q$ is a positive scalar.

There is multiple ways to identity the parameters of $\phi_G$. We have enough expressions to directly solve the non-linear system formed by equations from (21) to (25), but this strategy leads to spurious results as it tries to equalize an approximated form with asymptotic constrains. Therefore, we used error-minimization techniques.

To do so, we used a curve-fitting method to fit $\phi_G$ with reference data obtained from a numeric integration (that we call $\phi_{ref}$). Equality-constrains are implemented using Lagrange-multipliers technique, while inequality-constrains are implemented using interior penalization technique. We also choose $\gamma_3 < \gamma_2$ to distinguish both parameters.
From the obtained parameters values, as shown in figures 1 and 2, we can either create a 3-dimensional tabulation, or fit the data with functions. In this paper, we decided to find functions that can approximate the values of $\gamma_2$ and $\gamma_3$, and to deduct other parameters from the equations. To build the functions, we did some observations: firstly, the exponential coefficients $\gamma_2$ and $\gamma_3$ are close to be proportional to $\gamma_1$. Secondly, with help of the knowledge of the one-dimensional terminal-regime (in section 3.2), we can identify one of the two exponential coefficients to $-\gamma_1$. Thirdly, the exponential coefficients must converge to a finite value when $u_0$ tends toward infinity. We finally end with the following empirical
expressions of the exponential coefficients:

\[ c_{\theta} = 1 - \cos(\theta_0) \]

\[ \gamma_2 = -\sqrt{\alpha q} \left( 0.81 - 0.27 c_{\theta} + 0.66 c_{\theta} e^{-u_0} + 0.68 c_{\theta} u_0 e^{-6u_0^2} + 0.14 c_{\theta}^2 (1 - e^{-u_0}) \right) \]

\[ \gamma_3 = -\sqrt{\alpha q} \left( 1 - 0.27 c_{\theta} + 0.72 c_{\theta} e^{-u_0} + 0.30 c_{\theta} u_0 e^{-6u_0^2} + 0.12 c_{\theta}^2 (1 - e^{-u_0}) \right) \] (26)

Then, other parameters \((a_1, a_2, a_3)\) can be deducted and computed from the linear-system:

\[
\begin{bmatrix}
1 & 1 & 1 \\
\gamma_1 & \gamma_2 & \gamma_3 \\
\gamma_1^2 & \gamma_2^2 & \gamma_3^2
\end{bmatrix}
\begin{bmatrix}
a_1 \\
a_2 \\
a_3
\end{bmatrix}
= \begin{bmatrix}
-a_0 \\
1 \\
a_0 u_0
\end{bmatrix}
\] (27)

We can recognize here the Vandermonde matrix, which is inversible when \(\gamma_1, \gamma_2\) and \(\gamma_3\) have distinct values. The function \(\phi_G\), approximation of \(\phi\), is now fully determined. From the implicit solution (12), the approximated velocity is:

\[
\vec{u}_G(t) = \frac{\vec{u}_0 + \vec{Q} \phi_G(t)}{\phi_G(t)} = \vec{u}_0 + \vec{Q} (a_0 + a_1 e^{\gamma_1 t} + a_2 e^{\gamma_2 t} + a_3 e^{\gamma_3 t})
\]

(28)

The figure 3 shows the velocity path during the motion, from different initial velocity. Hopefully, the velocity converges towards the same value, and we remark also that the velocity converges with a certain tangent. We proved above that this tangent is perpendicular with the constant right-hand side vector \(\vec{Q}\) (so \(\vec{e}_y\) here).

While we obtained good approximation at both asymptotic limits, the in-between approximation is less accurate. The inaccuracy is getting stronger when the motion spends more time in the down-to-up regime (so when \(\cos(\theta_0)\) is closer to \(-1)\).

Fig. 3 The velocity \(\vec{u}\) with \(\theta_0 = 0.05, 1.53, 2.55\), \(u_0 = 1\), \(\alpha = 1\) and \(q = 1\). The solid line shows our approximation, the dashed line shows the reference. Each point represents a time increment of 0.2 seconds.
4.2 Degenerated motions

The present method will fail when $\lambda \to 0$. This scenario can happen when either $\alpha \to 0$, either $\vec{Q} \to \vec{0}$. In the first case, the terminal velocity diverges. In the second case, the motion tends to the one-dimensional motion, where the velocity path does no longer tends toward the terminal velocity with the same tangent. In both limit cases, the function $\phi$ folds into a polynomial in $t$. We can find a truncated Taylor expansion on $\phi$, with help of the first derivatives of $\phi$ from equations (14) to (17):

$$\phi(t) = t + \alpha \| \vec{u}_0 \| \frac{t^2}{2} + \alpha \frac{\vec{u}_0 \cdot \vec{Q} \ t^3}{3!} + ...$$

4.3 Approximation of the position

Whereas the velocity has an analytic expression, its integral (the position) cannot be expressed by elementary functions of time. A first approach is to use numeric integral schemes, that will still benefit from our stable approximation of the velocity. Fortunately, as shown in the figure 4, our approximation is still relevant in terms of accuracy, combined with multi-step integral schemes. We can illustrate such schemes with the following mixed scheme:

**Scheme (mixed integration scheme with a quadratic drag (Ap-E2))** Given the position $\vec{x}_n$ and the velocity $\vec{v}_n$ at the time $t_n$, the new position and velocity at time $t_{n+1} = t_n + \delta t$ are:

$$\vec{u}_G = ...$$

(as defined by (28))

$$\vec{v}_{n+1} = \vec{v}_{\text{wind}} + \vec{u}_G$$

$$\vec{x}_{n+1} = \vec{x}_n + (\vec{v}_n + \vec{v}_{n+1}) \frac{\delta t}{2}$$

Our reference integration scheme is using a 4th order Runge-Kutta integration for the velocity, and a 2nd order Euler integration for the position:

**Scheme (reference integration scheme with a quadratic drag (RK4-E2))** Given the position $\vec{x}_n$ and the velocity $\vec{v}_n$ at the time $t_n$, the new position and velocity at time $t_{n+1} = t_n + \delta t$ are:

$$\vec{u}_0 = \vec{v}_n - \vec{v}_{\text{wind}}$$

$$\delta \vec{u}_1 = \left( \vec{Q} - \alpha \| \vec{u}_0 \| \vec{u}_0 \right) \delta t$$

$$\delta \vec{u}_2 = \left( \vec{Q} - \alpha \| \vec{u}_0 + 0.5\delta \vec{u}_1 \| (\vec{u}_0 + 0.5\delta \vec{u}_1) \right) \delta t$$

$$\delta \vec{u}_3 = \left( \vec{Q} - \alpha \| \vec{u}_0 + 0.5\delta \vec{u}_2 \| (\vec{u}_0 + 0.5\delta \vec{u}_2) \right) \delta t$$

$$\delta \vec{u}_4 = \left( \vec{Q} - \alpha \| \vec{u}_0 + \delta \vec{u}_3 \| (\vec{u}_0 + \delta \vec{u}_3) \right) \delta t$$

$$\vec{v}_{n+1} = \vec{v}_{\text{wind}} + \left( \delta \vec{u}_1 + 2\delta \vec{u}_2 + 2\delta \vec{u}_3 + \delta \vec{u}_4 \right) \frac{\delta t}{6}$$

$$\vec{x}_{n+1} = \vec{x}_n + (\vec{v}_n + \vec{v}_{n+1}) \frac{\delta t}{2}$$
One of our goals is to provide a stable integration, and to minimize the drift of the position when the time increment becomes large. To do so, we will approximate again the velocity to be able to have an analytical expression of the position. The expression (28) is not analytically integrable because of the denominator form. The denominator is a sum of three exponential terms, in which one becomes dominant compared to the two others. Then, it is approximated by:

$$\frac{1}{\hat{q}_G(t)} = \frac{1}{a_1 \gamma_1 e^{\gamma_1 t} + a_2 \gamma_2 e^{\gamma_2 t} + a_3 \gamma_3 e^{\gamma_3 t}} = \frac{1}{a_1 \gamma_1 e^{\gamma_1 t} \left( 1 + \frac{a_2 \gamma_2 e^{\gamma_2 t} + a_3 \gamma_3 e^{\gamma_3 t}}{a_1 \gamma_1 e^{\gamma_1 t}} \right)}$$

With this approximation, the integral of $\hat{u}_G$ can be analytically expressed. Finally, the proposed integration scheme is the following:

**Scheme (integration scheme with a quadratic drag (Ap-VP))**

Given the position $\vec{x}_n$ and the velocity $\vec{v}_n$ at the time $t_n$, the new position and velocity at time $t_{n+1} = t_n + \delta t$ are:

$$\vec{v}_{n+1} = \vec{v}_{wind} + \vec{a} + \vec{Q}(a_0 + a_1 \gamma_1 t + a_2 \gamma_2 e^{\gamma_2 t} + a_3 \gamma_3 e^{\gamma_3 t})$$

$$\vec{x}_{n+1} = \vec{x}_n + \vec{v}_{wind} \delta t + c_u \vec{a} + c_Q \vec{Q}$$

with $\vec{a} = \vec{v} - \vec{v}_{wind}$, $p = \frac{\vec{a} \cdot \vec{Q}}{||\vec{Q}||^2}$, $q = \alpha ||\vec{a}||$, $c_q = 1 - \frac{\vec{u} \cdot \vec{Q}}{||\vec{Q}||^2}$ and

$$\gamma_1 = \sqrt{\alpha \|\vec{Q}\|}$$

$$\gamma_2 = -\sqrt{\alpha \left( 0.81 - 0.27 c_0 + 0.63 c_0 e^{-||\vec{a}||^2} + 0.68 c_0 ||\vec{a}|| e^{-||\vec{a}||^2} + 0.14 c_0 (1 - e^{-||\vec{a}||^2}) \right)}$$

$$\gamma_3 = -\sqrt{\alpha \left( 1 - 0.27 c_0 + 0.72 c_0 e^{-||\vec{a}||^2} + 0.30 c_0 ||\vec{a}|| e^{-||\vec{a}||^2} + 0.12 c_0 (1 - e^{-||\vec{a}||^2}) \right)}$$

$$a_0 = -\frac{\vec{u} \cdot \vec{Q}}{||\vec{Q}||^2}$$

$$a_1 = \frac{-\gamma_2 \gamma_3 p + (\gamma_2 + \gamma_3) - q}{(\gamma_3 - \gamma_1)(\gamma_1 - \gamma_2)}$$

$$a_2 = \frac{-\gamma_3 \gamma_1 p + (\gamma_2 + \gamma_3) - q}{(\gamma_1 - \gamma_2)(\gamma_2 - \gamma_3)}$$

$$a_3 = \frac{-\gamma_1 \gamma_2 p + (\gamma_1 + \gamma_2) - q}{(\gamma_2 - \gamma_3)(\gamma_3 - \gamma_1)}$$

$$c_u = \frac{1}{a_1 \gamma_1} e^{-\gamma_1 \delta t} - 1 + \frac{-a_2 \gamma_2}{a_1 \gamma_1} e^{(\gamma_2 - 2\gamma_1) \delta t} - 1 + \frac{-a_3 \gamma_3}{a_1 \gamma_1} e^{(\gamma_3 - 2\gamma_1) \delta t} - 1$$

$$c_Q = \frac{1}{\gamma_1} \delta t - pc_u + \frac{a_2 (1 - a_1 \gamma_2)}{a_1 \gamma_1} e^{(\gamma_2 - \gamma_1) \delta t} - 1 + \frac{a_3 (1 - a_1 \gamma_3)}{a_1 \gamma_1} e^{(\gamma_3 - \gamma_1) \delta t} - 1 + \frac{-a_2 \gamma_2}{a_1 \gamma_1} e^{(\gamma_2 - \gamma_1) \delta t} - 1 + \frac{-a_3 \gamma_3}{a_1 \gamma_1} e^{(\gamma_3 - \gamma_1) \delta t} - 1$$

We compare the different schemes by comparing the given position of the body after 10 seconds, after proceeding by successive time-steps with a fixed delta-time.
As shown in the figure 4, our scheme (Ap-VP) is able to give the final position in one step (delta-time is 10s), with roughly the same precision than the precision obtained by proceeding 15 time-steps with the reference scheme (RK4-E2).

![Graph showing relative error vs delta time]

**Fig. 4** The relative difference of the position of the projectile at t=10s, with $\theta_0 = 2.47$, $u_0 = 0.95$, $\alpha = 1$ and $q = 1$.

Even though the proposed scheme (Ap-VP) gives a good accuracy regardless the number of integration steps, it has a heavier computational complexity. Usage of tabulations may be a way to lower the computational complexity.

5 Conclusions

From a detailed analysis of the equation limits, both near the initial state and far from the initial state, we were able to build a good knowledge of the behavior of the motion. We preferred to keep a good balance between the analysis depth of the initial state and terminal state, rather than going for more precision near the initial state. Furthermore, it was demonstrated that the velocity converges toward the terminal velocity with a particular tangent.

Using the reduction principle was a key element towards the ability to work with a single auxiliary variable, so we successfully built an approximation of this auxiliary variable, that leads to deduct an approximated velocity. Several explorations of the present work are now possible, with different forms of the auxiliary variable and different ways to compute its coefficients. Also, as the approximation form is similar to the analytic form in the one-dimensional terminal regime (when the velocity goes in the same direction as the terminal velocity), the obtained solution is less accurate with starting regimes.

To complete the integration scheme, we also proposed an approximation of the position, that sadly introduced another layer of approximation. Even if the proposed integration scheme is unconditionally stable, its complexity is significantly
higher than standard low-order integration schemes. Some compromises between
the accuracy and the complexity may be subject to further work.

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Authors’ contributions

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Competing interests

The authors declare that they have no competing interests.

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