RESEARCH PAPER

Determination of LME sensitivity of zinc-coated steels based on the programmable deformation cracking test

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Abstract
The current work presents a new test method to evaluate liquid metal embrittlement (LME) susceptibility of zinc-coated steels in arc processes under application-oriented conditions. The procedure is based on the programmable deformation cracking test (PVR test). The PVR test is a variation of a controlled tensile test for hot cracking investigations in arc welding processes. Two dual-phase steels (DP600, DP980) and five transformation-induced plasticity steels (TRIP690, TRIP700, TRIP1100, TRIP1200) were used. The investigations showed that comparable thermo-mechanical loading conditions can be guaranteed for materials of different sheet thicknesses in the PVR test through a targeted adjustment of the heat input per unit length of weld. Furthermore, it was shown that the critical deformation rate \( v_{cr} \) (used for assessing hot cracking susceptibility) may also be used to assess the LME susceptibility of a particular steel. Furthermore, another LME susceptibility parameter, the relative reduction in load-bearing ability \( \Delta \Sigma \) could be derived, which may be used to understand how LME cracking affects materials’ mechanical and fracture properties.

Keywords PVR test · Hot cracking · Liquid metal embrittlement · Liquid metal-assisted cracking · LME · LMAC · Arc welding · Zinc coating · AHSS steel · LME test method

1 Introduction
Deterioration of solid metallic materials by liquid metals has been known by various terms. In the literature this phenomenon is often referred to as liquid metal embrittlement (LME) or liquid metal-assisted cracking (LMAC) [1]. However it is called, during LME, solid metal exposed to tensile stresses is deteriorated by a liquid metal, which leads to brittle intergranular cracking [2–4]. In the literature, there are four requirements for LME to occur, which are as follows: tensile stress (either residual or externally applied), a solid metal, a liquid metal in contact with the solid metal, and the environmental conditions (e.g., exposure time, temperature) [5–17]. Such conditions can be found in various technical areas of metal-processing industry, e.g., in hot-dip galvanizing, resistance spot welding, arc welding, and arc brazing. Central questions from material manufacturers and users concern the susceptibility of a material to LME under manufacturing conditions and how critical the LME cracks are in the subsequent product life cycle.

In the literature, there exist different test procedures to evaluate and investigate the LME behavior of materials, whereby a distinction must be made between process-dependent and process-independent procedures. Process-dependent methods aim to generate the real, complex conditions of the manufacturing process as precisely as possible, while process-independent methods aim to generate individual influencing quantities as precisely as possible. Thermo-mechanical testing (Gleeble) is an example of a process-independent method for the LME investigation.
[12]. In the area of hot-dip galvanizing, Mendala [18] developed a susceptibility test where a sample loaded in tension is exposed in a liquid zinc bath. In the area of resistance spot welding, there are varieties of different test setups with different objectives, such as the targeted, provoked generation of cracks to investigate the mechanism of LME cracks or the focus on assessing the LME susceptibility of different materials.

Classifications of the susceptibility to LME cracking, such as the one from Bhattacharya [4], were created on the basis of Gleeble and resistance spot welding testing. However, it is unknown whether this insight may be transferred to arc welding and brazing applications as the stress-state, and the heating and cooling rates of the welded material differs in arc and resistance spot welding processes. In the area of arc welding and brazing, limited methods have been investigated to understand the LME behavior of zinc-coated automotive steels. Biro et al. [19] developed a method to invoke LME during arc welding and brazing using a self-restrained sample to load the tested sample, which was able to differentiate the LME susceptibility of various materials. Likewise, Bruscato [20] also used a self-restrained sample to evaluate the LME susceptibility of austenitic steels. These test setups enable the generation of LME cracks and the definition of LME-critical conditions, but the options for targeted variation of the influencing quantities remain limited. Razmipoosh et al. [21] used a method similar to the externally stress coupons during laser beam welding to provoke LME. Driessen et al. [22] have derived a procedure based on the programmable deformation cracking test (german: Programmiert-Verformungs-Riss-Test or PVR for short), standardized in [23–25]). As the PVR externally loads samples while they are arc welded, it should allow testing LME susceptibility under arc welding conditions; however, currently no studies have done this.

To test materials’ hot cracking susceptibility, there are numerous methodologies. These include Gleeble testing [24, 25], testing of self-restrained samples [19, 20, 26], and testing of externally stressed flat tensile tests (PVR test [24, 25]). As discussed previously, the Gleeble [12] and the self-restrained [20] tests have also shown the capable of measuring LME susceptibility. Since the test specimen in the PVR test is subjected to a combined load of temperature, stress, and strain, it can be assumed that this test is also suitable for investigating the LME susceptibility. In analogy to the determination of the critical deformation rate for hot cracks, it stands to reason that there also exists a critical deformation rate for LME crack formation. However, ISO/TR 17641–3 [24] only recommends a certain specimen geometry with a specimen thickness of 10mm and Merkblatt DVS 1004–2 [25] only shows an example for a sheet metal thickness of 2mm. The sheet metal thicknesses in the automotive industry typically vary in the range of 0.8 – 3mm. The specific thickness used depends on the required strength of the component. Therefore, not all materials are available in the same sheet metal thickness for a material comparison. One example can be found in lightweight construction concepts, in which weight savings are achieved through substitution with materials of higher strength. If, in the PVR test, the same welding parameters are used for different sheet thicknesses, different thermo-mechanical conditions result. Therefore, a direct comparison of the materials with regard to their susceptibility to cracking (hot cracking and LME cracking) is not possible. Therefore, it is necessary to adjust the welding parameters depending on the sheet thickness. The current study will resolve these issues. Specifically, it will:

- Investigate the potential of the PVR test to assess the LME behavior of zinc-coated automotive steels (transferability of the assessment quantity \(v_{cr}\) (critical deformation rate), deriving further assessment options).
- Determine the thermo-mechanical conditions that will result in LME as a function of the welding parameters.
- Compare the LME susceptibility of various zinc-coated automotive steels with differing sheet thicknesses.

## 2 Materials and methods

Seven hot dip zinc-coated steels of different strengths were selected for the investigations, including two dual-phase steels (DP) and five transformation-induced plasticity steels (TRIP). Table 1 shows the chemical composition of the base materials. The materials were available in different sheet thicknesses \(t_{material}\) ranging from 1.20 to 1.75mm.

| Material | \(t_{material}\) (mm) | C  | Mn  | Si  | Al  | Cr  | Mo  | Fe  |
|----------|----------------------|----|-----|-----|-----|-----|-----|-----|
| DP600    | 1.22                 | 0.09| 1.7 | 0.2 | 0.03| 0.3 | -   | Bal |
| DP980    | 1.20                 | 0.10| 2.1 | 0.3 | 0.05| 0.2 | 0.3 | Bal |
| TRIP690  | 1.20                 | 0.20| 1.7 | 0.4 | 1.31| -   | -   | Bal |
| TRIP700-I| 1.27                 | 0.21| 1.7 | 0.3 | 0.96| 0.1 | -   | Bal |
| TRIP700-II| 1.75                | 0.20| 1.8 | 0.3 | 0.96| 0.1 | -   | Bal |
| TRIP1100 | 1.60                 | 0.20| 2.2 | 1.6 | -   | -   | -   | Bal |
| TRIP1200 | 1.60                 | 0.22| 2.3 | 1.7 | -   | 0.1 | -   | Bal |

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Figure 1 shows the used specimen geometry. It is based on the standard test specimens according to [24, 25]. The rolling direction is along the 453 mm length of the sample.

All samples were tested in as-received (hot dip zinc coated) as well as the uncoated (reference sample stripped of the hot-dip coating) conditions to understand the material cracking performance in the zinc-coated (LME susceptible) and uncoated (Non-LME susceptible) conditions. Both the tests were carried out using the same test conditions. The reference samples were produced by stripping the zinc coating from the as-received material by soaking the materials in a 20% hydrochloric acid solution mixed with 3.5 g/l hexamethylenetetramine, until bubbling stopped [27]. Afterwards, the samples were rinsed in a bath of distilled water and cleaned with ethanol. Before welding, all stripped samples were exposed to the atmosphere for at least 24 h, to allow any hydrogen absorbed in the stripping process to volatilize. This excludes any influence of the hydrogen on crack formation. The aging time was based on internal investigations, in which after exposure times of different lengths, hydrogen measurements were done using the hot gas extraction method. After an exposure time of 24 h, no more diffusible hydrogen was measured in the hot gas extraction in any of the stripped samples. After stripping, the uncoated samples were manually ground using 600 grit sandpaper. All samples were cleaned with ethanol previous to the welding.

LME tests carried out using the PVR test, which is a standardized method to assess the susceptibility to hot cracking [24, 25]. In the PVR test, the superposition of a thermal and a mechanical stress takes place. Figure 2 shows an extended schematic description of the PVR test procedure and the expected occurrence of cracking.

The test specimen has a constant cross section over the entire test area. The sample is subjected to a linearly increasing deformation rate \( v_{PVR} \), which results in an increasing tensile stress throughout the test. The welding process, which starts at the same time as the tensile process, uses a constant welding speed, \( v_w \), so as the weld progresses, the deformation rate, \( v_{PVR} \), of the local area exposed to the welding heat source will increase. As the deformation rate increases, the tensile stress of the specimen increases, leading to cracking once a critical stress (or deformation rate) has been achieved. After reaching a programmed maximum deformation rate \( v_{PVR,max} \), the test ends and the sample is allowed to cool down without load. Once the sample cools, the location of the first cracks are determined along with the critical longitudinal deformation rate \( v_{cr} \) required to initiate cracking, as shown in Eq. (1)

\[
v_{cr} = \frac{a \cdot L}{v_w}
\]
where $a$ is defined as tensile acceleration and $L$ is the mean value of the position of the first three cracks. Afterwards, the critical longitudinal deformation rate $v_{cr}$ can be used as a quantitative measurement of the cracking susceptibility. The smaller the critical longitudinal deformation rate, the greater the susceptibility of the material to cracking.

Generally, the hot cracking phenomena can be divided into solidification cracking (SC), liquation cracking (LC), and ductility dip cracking (DDC). In the present investigation, only solidification cracking and LME cracking occurred. During the analysis, the uncoated sample provided the basis for differentiating between the observed cracking phenomena (solidification cracking and LME cracking).

Cracking that also occurred on the uncoated material was defined as solidification cracking as LME cracking requires the presence of zinc. Conversely, cracking that only occurred on the hot dip zinc-coated samples could only be LME cracking. By inserting the position of the first solidification cracks ($L_{1,SC}$) or the first LME cracks ($L_{1,LME}$) in Eq. (1), the respective critical deformation rate of the solidification cracking ($v_{cr,SC}$) or the LME cracking ($v_{cr,LME}$) was obtained.

The position of the first three cracks (solidification cracks and LME cracks) was determined after the test by means of the dye penetrant test which was carried out in accordance with ISO 3452–1 [28] and microscopic surface examinations at 25 x magnification.

The welding heat source used for all tests was a tungsten inert gas welding arc (TIG arc). Argon was used as shielding gas with a volume flow of 20l/min. The welding parameters were recorded with a WeldAnalyst from the company HKS Prozesstechnik GmbH.

The tests were divided into 5 test series (see Table 2). In test series 1, all specimens were tested in the hot dip zinc-coated state with the same set of parameters. To understand the heat input applied to the material, the applied heat input per unit length of weld $E$ is calculated using Eq. (2)

$$ E = \frac{U_w \cdot I_w}{v_w} $$

where $U_w$ is defined as welding voltage, $I_w$ as welding current, and $v_w$ as welding speed. As all materials differed in sheet thicknesses, the heat input per unit length of weld was adjusted in test series 2 to create a comparable thermomechanical load condition for all test specimens. This was based on the approach of Biro et al. [19]. This resulted in developing a thickness normalized heat input per unit length

| Test series | Material | State | $t_{material}$ (mm) | $U_w$ (V) | $I_w$ (A) | $v_w$ (cm/min) | $L_0$ (mm) | $v_{PPR,max}$ (mm) | $E$ ($kJ/mm$) | $E_{norm}$ ($kJ/mm^{-1}$) |
|-------------|----------|-------|---------------------|-----------|-----------|----------------|-----------|---------------------|-------------|-----------------------|
| 1           | DP600 GI |       | 1.22                | 9.0       | 52        | 20             | 200       | 0.140               | 0.115       | In test series 1, the heat input per unit length of weld was not normalized |
|             | DP980 GI |       | 1.20                | 9.0       | 52        | 20             | 200       | 0.140               | 0.117       |                       |
|             | TRIP690 GI |     | 1.20                | 9.0       | 52        | 20             | 200       | 0.140               | 0.117       |                       |
|             | TRIP700-I GI | | 1.27                | 9.0       | 52        | 20             | 200       | 0.140               | 0.111       |                       |
|             | TRIP700-II GI |  | 1.75                | 9.0       | 52        | 20             | 200       | 0.140               | 0.080       |                       |
|             | TRIP1100 GI |       | 1.60                | 9.0       | 52        | 20             | 200       | 0.140               | 0.088       |                       |
|             | TRIP1200 GI |      | 1.60                | 9.0       | 52        | 20             | 200       | 0.140               | 0.088       |                       |
| 2           | TRIP700-I GI |     | 1.27                | 9.0       | 55        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.149       |
|             | TRIP700-I Uncoated | | 1.27                | 9.0       | 55        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.149       |
|             | TRIP700-II Uncoated | | 1.75                | 9.0       | 76        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.205       |
|             | TRIP700-II Uncoated | | 1.75                | 13.0      | 53        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.205       |
| 3           | DP600 GI |       | 1.22                | 9.0       | 53        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.205       |
|             | TRIP700-I GI |     | 1.27                | 9.0       | 55        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.143       |
|             | TRIP700-II GI | | 1.75                | 9.5       | 72        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.149       |
|             | TRIP1200 GI |      | 1.60                | 10.0      | 62        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.205       |
|             | TRIP1200 GI |      | 1.60                | 10.0      | 62        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.187       |
|             | TRIP1200 GI |      | 1.60                | 10.0      | 62        | 20             | 200       | $E_{norm}$           | $Q_{base}$  | 0.117                 | 0.187       |

5 All Uncoated same parameter settings as test series 4
of weld, $Q_{\text{base}}$, was calculated by dividing $E_{\text{base}}$ by $t_{\text{base}}$, as in Eq. 3. The values for $E_{\text{base}}$ and $t_{\text{base}}$ resulted from test series 1, where a good fusion zone was achieved for the material TRIP690 with a sheet thickness of 1.20mm and the heat input per unit length of weld of 0.140kJ/mm$^{-1}$. Therefore, the resulting value for $Q_{\text{base}}$ is a constant for normalizing the heat input per unit length of weld.

$$Q_{\text{base}} = \frac{E_{\text{base}}}{t_{\text{base}}} \quad (3)$$

Afterwards, $Q_{\text{base}}$ is calculated, the heat input per unit length for the coated material is determined by multiplying $Q_{\text{base}}$ by its respective sheet thickness $t_{\text{material}}$ (see Table 1). This results in the heat input per unit length of weld normalized for the respective sheet thickness $E_{\text{norm}}$, which is required for the comparison (see Eq. (4)).

$$E_{\text{norm}} = Q_{\text{base}} \cdot t_{\text{material}} \quad (4)$$

Table 2 shows the values resulting from the use of Eq. (4). According to Eq. (2), the heat input per unit length of weld is defined using the three parameters of welding voltage, welding current, and welding speed. This can be used to set the necessary normalized heat input per unit length of weld. To ensure a constant relative movement between the heat source and the specimen, the welding speed was kept constant at $v_w = 20cm/min$ for all tests. It is still possible to control the heat input per unit length of weld by changing the welding voltage and the welding current. Both variants were tested in test series 2.

In addition to the evaluation of the critical deformation rate $v_{cr}$, the test method offers an assessment of how LME cracking affects fracture behavior by comparing the force–time curves of LME-affected and non-LME-affected samples. For this evaluation to be valid test specimens must fail by fracture before the end of the test. To ensure this, the maximum deformation rate was increased to $v_{PVR,\text{max}} = 100mm/min$ in test series 3.

In test series 4, the arc voltage was adjusted to $U_w = 10V$ in favor of a more stable arc, and all test materials were tested under comparable thermo-mechanical loads.

In test series 5, the uncoated reference welding was carried out for each base material under test conditions otherwise identical to test series 4. Due to the limited scope of the coated materials tested, statistical validation and thus the consideration of the reproducibility of the results could only be realized for the fracture behavior of the DP600-GI, TRIP700-I-GI and TRIP700-II-GI.

Table 2 shows the test parameters for test series 1–5 in detail.

In addition to evaluating the critical deformation rate, the fracture behavior of the test specimens was evaluated. For this purpose, the recorded force–time curves were converted into engineering stress-time curves. As the change in cross-section in the PVR test was not recorded by measurement, the stresses plotted over time are to be understood as engineering stress, which can be calculated according to Eq. (5) by including the starting cross-section $A_0$ which is a function of the respective sheet thickness $t_{\text{material}}$.

$$\sigma(t) = \frac{F(t)}{A_0(t_{\text{material}})} = \frac{F(t)}{t_{\text{material}} \cdot 40mm} \quad (5)$$

The assessment of the fracture behavior was used in analogy to the LME Gleeble test evaluation to assess the LME behavior after Béal [12]. By integrating the engineering stress over time, the load-bearing ability of the sample calculated for the states uncoated ($\Sigma_{\text{uncoated}}$) and coated ($\Sigma_{\text{coated}}$) according to Eq. (6). The integration limits for the load-bearing ability extends from the start of welding to fracture, defined as $t_{\text{fracture}}$.

$$\Sigma_{\text{uncoated/coated}} = \int_{0}^{t_{\text{fracture}}} \sigma_{\text{uncoated/coated}}(t)dt \quad (6)$$

Subsequently, the relationship between load-bearing ability of the coated and uncoated materials was calculated, and the relative reduction of the load-bearing ability $\Delta \Sigma$ was determined according to Eq. (7).

$$\Delta \Sigma = \frac{\Sigma_{\text{uncoated}} - \Sigma_{\text{coated}}}{\Sigma_{\text{uncoated}}} \cdot 100\% \quad (7)$$

Finally, based on $\Delta \Sigma$ value, the ability of LME to degrade the fracture behavior of a particular material may be evaluated.

## 3 Results and discussion

### 3.1 Cracking phenomena of the test materials in the PVR test

PVR testing of the materials was carried out. Solidification cracks were detected in all TRIP steels (TRIP690, TRIP700-I, TRIP700-II, TRIP1100 and TRIP1200). The DP steels (DP600 and DP980) showed no solidification cracking. Figure 3a and Fig. 3b show an example of the solidification cracking that occurred at the investigated TRIP steels. It may be confirmed that these cracks were indeed solidification cracks and not LME as cracks were seen in the fusion zone of material that were both in the hot dip zinc coated as well in the uncoated condition.

The occurrence of solidification cracking was used to show that the normalization of the heat input per unit length of weld allows thermo-mechanical load conditions to be compared for all sheet thicknesses (see Sect. 3.3). To show
the usability of this method for various materials, it was sufficient to use the TRIP700-I and TRIP700-II. Therefore, the susceptibility to solidification cracking for the materials TRIP690, TRIP1100, and TRIP1200 was not investigated more in detail.

LME cracking, in the area of the heat affected zone (HAZ) near the fusion zone, only occurred in TRIP1100-GI and TRIP1200-GI. As stated previously, it was determined that these cracks were caused by LME as cracking only occurred in materials that were in the hot-dip zinc-coated condition.

It was observed that the TRIP1100-GI tends to form frequent and shallow LME cracks. Conversely, the LME cracks in the TRIP1200-GI were much deeper; however, fewer cracks were seen than in the tests on the TRIP1100-GI. Figure 4a and Fig. 4b show that LME cracking occurred in the TRIP1100-GI and Fig. 4c and Fig. 4d the LME cracking in the TRIP1200-GI. Compared with the investigations by Wintjes et al. [29], the PVR test showed comparable results with regard to the tendency for LME cracking occurrence. In the resistance spot welding test, the TRIP690-GI and DP980-GI showed no or only a few cracks, while the TRIP1100-GI and TRIP1200-GI showed a high level of LME cracking occurrence [29]. In all cases, the LME cracks were perpendicular to the direction of the external tensile load and the direction of welding. The reason for this is the generated tensile stress in the welding direction due to the linearly increasing deformation rate in the PVR test. This agrees the

![Fig. 3 Solidification cracking in the fusion zone at TRIP700-GI-1.75-E.norm (I)-v100. (a) Cracking indications in the dye penetrant test. (b) Microscope surface image of the first solidification cracking](image1)

![Fig. 4 LME cracking detection using the dye penetrant test. (a–b) First LME cracking in the HAZ of TRIP1100-GI-1.60-E.norm (I)_v100. (c–d) First LME cracking in the HAZ of TRIP1200-GI-1.60-E.norm (I)_v100. (a+c) Overview picture of the dye penetrant test. (b+d) Microscope surface image of the first LME cracking. (d) Cracks go through the entire thickness of the sheet metal and are also open at the bottom (see Fig. 10b)](image2)
observations where LME cracks were created using an externally applied force [12, 30].

3.2 Evaluation of the thermo-mechanical system when comparing different sheet thicknesses in the PVR test

In test series 1, all materials were tested in the hot dip zinc-coated condition using the same welding parameters. As all the materials have various thicknesses, the force–time curves resulting from these experiments could not be directly compared. Figure 5a shows that the test material TRIP700-GI-1.75 ($t = 1.75\text{mm}$), which has a minimum tensile strength of $R_{m,\text{min}} = 690\text{MPa}$ at room temperature, showed higher maximum forces in the test than the material DP980-GI-1.20 ($t = 1.20\text{mm}$), which has a minimum tensile strength of $R_{m,\text{min}} = 980\text{MPa}$ at room temperature. In addition, the material TRIP690-GI-1.20 ($t = 1.20\text{mm}$) and TRIP700-GI-1.27 ($t = 1.27\text{mm}$), with similar chemical composition, had a lower force response throughout the test than the TRIP700-GI-1.75. This behavior can initially be explained by the differences in sheet thickness. However, before a comparison can take place, the force must be converted into the engineering stress using Eq. (5) to eliminate the influence of the sheet thickness. Figure 5b shows the resulting engineering stress-time diagram.

Calculating and ordering the materials with respect to the failure stress matched correlated with the expected strength order when accounting for the material strength values at ambient temperature. However, the engineering stress-time curves of the TRIP700-GI-1.27 and TRIP700-GI-1.75, despite having the same minimum tensile strength at room temperature and almost identical chemical composition, still had a large variation in failure stress under thermal influence (see Table 3, $R_{m,T}$). This difference in failure stress was due to the identical welding parameters used for series 1, which resulted in the heat input per unit length of weld being identical for all materials (see Table 2). All materials

![Fig. 5](image)

**Fig. 5** Evaluation of the thermo-mechanical load on the test specimens over the test period in the PVR test. Interpretation of results without and with consideration of the sheet thickness differences. (a) Representation in the form of force–time curve. (b) Representation in the form of engineering stress-time curve

| Material       | $t_{\text{material}}$ (mm) | $R_{m,\text{min},RT}$ (N/mm²) | $R_{m,T}$ (N/mm²) | $\Delta R_{m}$ (%) | $R_{m,T,E,norm}$ (N/mm²) | $\Delta R_{m,E,norm}$ (%) |
|----------------|---------------------------|-------------------------------|-------------------|-------------------|---------------------------|---------------------------|
| DP600          | 1.22                      | 590*                          | 361               | 39                | 370                       | 37                        |
| TRIP690        | 1.20                      | 690*                          | 444               | 36                | 431                       | 38                        |
| TRIP700-I      | 1.27                      | 690*                          | 416               | 40                | 408                       | 41                        |
| TRIP700-II     | 1.75                      | 690*                          | 498               | 28                | 436                       | 37                        |
| DP980          | 1.20                      | 980*                          | 586               | 40                | 578                       | 41                        |
| TRIP1100       | 1.60                      | 1080**                        | 711               | 34                | 594                       | 45                        |
| TRIP1200       | 1.60                      | 1180**                        | 853               | 28                | 753                       | 36                        |

*Values according to [31]; ** Estimated values according to [31]
were welded with a welding voltage of \( U_w = 9V \), a welding current of \( I_w = 52A \), and a welding speed of \( v_w = 20cm/min \). However, with increasing sheet thickness, the volume to be heated and the thermal inertia of the welded material increases. Accordingly, as the heat input per unit length of weld was \( E = 0.140kJ/mm \), regardless of specimen thickness, the increased temperature field of the thicker specimens (TRIP700-GI-1.75) were narrower and experiences lower peak temperatures than the thinner specimens (TRIP700-GI-1.27). Under these test conditions, a comparison of the materials is not possible. For materials to be compared, it must be ensured that a comparable thermo-mechanical state is similar for all test specimens in the test. The fact that this was not true for test series 1 was confirmed when the susceptibility to solidification cracking for the various materials were compared (see Sect. 3.3). Table 3 compares the minimum values of the tensile strength of the base materials at room temperature \( R_{m,\text{min},RT} \) and under the thermal influence in the PVR test \( R_{m,T} \) (without normalization of the heat input per unit length of weld) and \( R_{m,T,E,\text{norm}} \) (with normalization of the heat input per unit length of weld). The values for \( R_{m,T} \) correspond to the maxima of the engineering stress-time curves from Fig. 5b, and the values for \( R_{m,T,E,\text{norm}} \) correspond to the maxima of the engineering stress-time curves from Fig. 8b (see Sect. 3.6). For a comparison, the relative reduction in tensile strength due to the thermal load \( \Delta R_m \) can be calculated for both using Eq. (8).

\[
\Delta R_m = \frac{R_{m,\text{min},RT} - R_{m,T,E,\text{norm}}}{R_{m,\text{min},RT}} \times 100\%
\]

When considering the TRIP700-GI in the two different sheet thicknesses \( t = 1.27mm \) and \( t = 1.75mm \), the comparison shows the influence of the heat input per unit length of weld related to the sheet thickness. The difference between the PVR measured tensile strengths of the two materials without adjusting the heat input per unit length of weld is 12%. After normalizing the heat input per unit length of weld (see test series 4 and 5), the relative reduction in tensile strength due to the thermal load is more comparable. That reduced the difference between the PVR-measured strengths of the two TRIP700-GI materials to 4 %, as the thermo-mechanical conditions are similar by accounting for thickness when selecting welding parameters. Under these adjusted test conditions, an assessment of the thermo-mechanical system can be made.

### 3.3 Effect of the normalization of the heat input per unit length of weld on the results in the PVR test

By normalizing the heat input per unit length of weld according to Eq. (4), test specimens of different sheet thicknesses experience a comparable thermo-mechanical load in the PVR test. Figure 6a shows a comparison of the engineering stress-time curves of the TRIP700 steels of different sheet thicknesses examined in test series 1 and test series 2 before and after the normalization of the heat input per unit length of weld.

According to Eq. (2), the heat input per unit length of weld can be adjusted using the three parameters of welding voltage, welding current, and welding speed. To ensure that the heat source position with respect to the sample was constant for all specimen (as the crosshead speed is changing through the test), welding speed was constant for all tests, and heat input was adjusted by changing the welding current and voltage. Figure 6b shows that both adaptations initially provided almost identical thermo-mechanical material behavior with regard to the engineering stress-time curves. However, it was seen that when welding voltage was increased, the energy density of the arc reduced sufficiently that the base material did not melt, as can be seen in Fig. 6d. As the base material was not molten, solidification cracking (SC) did not occur. In contrast, the increasing welding current increased heat input while maintaining energy density, so the maintaining base material being melting, as can be seen in Fig. 6c, and solidification cracking appeared in the fusion zone. Figure 6e shows the comparison of the critical deformation rates for the onset of solidification cracking for the samples from test series 1 (same heat input per unit length of weld) and the samples from test series 2 (normalized heat input per unit length of weld). It shows that normalizing heat input per unit length of weld with respect to the material thickness was able to reproduce the thermo-mechanical and solidification cracking behavior across samples of various thicknesses. It was also seen that surface condition of the test specimens (uncoated/coated) did not affect the occurrence of solidification cracking. From these results, the subsequent test series was carried out by adjusting the welding current \( I_w \) to ensure that heat input per unit length of weld was normalized to sheet thickness.

### 3.4 Effect of the maximum deformation rate on the thermo-mechanical behavior in the PVR test

To evaluate how LME cracking affected fracture behavior of the base materials, all tests needed to continue until fracture. Therefore, in test series 3 the necessary maximum deformation rate was investigated. The determined engineering stress-time curves were compared with those from test series 2. Since the materials with lower tensile strength have a higher fracture elongation compared to materials with higher tensile strength, the maximum deformation rate \( \nu_{\text{PVR,max}} \) must be specified using the lower strength materials with higher fracture elongation. For example the DP600 did not fail.
at a maximum deformation rate of $v_{PVR,\text{max}} = 50\text{mm/min}$, but the DP980 did (see Fig. 5). Table 4 shows an example of the fracture elongation for three base materials at room temperature.

By increasing the maximum deformation rate to $v_{PVR,\text{max}} = 100\text{mm/min}$, specimen failure could be provoked for the base materials DP600, TRIP700-I, and TRIP700-II, as shown in Fig. 7. As a result, a maximum deformation rate of $v_{PVR,\text{max}} = 100\text{mm/min}$ was defined for test series 4 and 5. This value can be used as a starting value for future investigations of other materials if they have room temperature fracture elongations $\leq 24$ and are exposed to a comparable heat input.

### 3.5 Assessment of the LME susceptibility of the test materials using $v_{cr}$

Assessment material LME susceptibility was carried out in test series 4, analogous to the evaluation method of the hot cracking, by determining the critical deformation rate $v_{cr}$ according to ISO 17641–3 [24] and Merkblatt DVS 1004–2 [25] (see Sect. 2). Only the TRIP1100-GI and TRIP1200-GI
showed LME cracking in the HAZ after the PVR test of the specimens in the hot dip zinc-coated state (see Sect. 3.1). The critical deformation rates for LME of the two materials were calculated according to Eq. (1). TRIP1100-GI-1.60 showed a critical deformation rate of \( v_{cr,\text{LMEc}} = 20 \text{mm/min} \) and TRIP1200-GI-1.60 showed a critical deformation rate of \( v_{cr,\text{LMEc}} = 18 \text{mm/min} \). Similar to the assessment of hot cracking susceptibility, this shows that the LME susceptibility of both materials are similar. Due to the relatively small difference in values of \( v_{cr,\text{LMEc}} \) of the two test materials, a definitive statement can only be made after the values have been statistically verified.

### 3.6 Assessment of the effect of the LME cracking on the fracture behavior of the test material using \( \Delta \Sigma \)

The evaluation of the fracture behavior of a material by means of the PVR test is a new method that has not been used before. To do this, the engineering stress-time curves for the uncoated and coated surface conditions were determined, so that the relative reduction of the load-bearing ability, \( \Delta \Sigma \), due to LME could be calculated. This was done in the test series 4 and 5. By choosing the maximum deformation rate \( v_{PVR,max} = 100 \text{mm/min} \), the necessary specimen fracture was provoked for all specimens before the end of the test, which is shown by a sudden drop in engineering stress-time curves. Figure 8a shows the engineering stress-time curves of the uncoated reference specimens and Fig. 8b that of the hot dip zinc-coated specimens. It can already be seen that the materials TRIP1100 and TRIP1200 showed an earlier failure due to fracture in the hot dip zinc-coated state than in the uncoated state. While the TRIP1200 breaks shortly after reaching the maximum stress, the TRIP1000 can withstand some plastic deformation before it also fails due to fracture. The behavior is explained by the fact that the extent of the cracks (width, depth) is significantly greater for the TRIP1200 than the TRIP1100 (see Fig. 4b, 4d and Fig. 10b). Larger cracks will impart greater weakening of the load-bearing cross-section and thus further reduced the load-bearing ability of a material. This result coincides with the investigations by DiGiovanni et al. [32] in which was shown that greater crack length have a more significant influence on the failure behavior.

All other test materials showed almost identical engineering stress-time curves when tested in both the coated and uncoated surface conditions. Without cracking in the HAZ, the base material could undergo plastic deformation for a longer period until the specimen finally broke after reaching the material- and temperature-dependent elongation after fracture. Figure 9a shows an example of the engineering stress-time curves for the material TRIP690 in the states uncoated and coated. The failure due to fracture being the same for both states. For this base material, the coating has no effect on the fracture behavior under the present conditions.
test conditions in the PVR test. Figure 9b shows an example of the engineering stress-time curves for the material TRIP1200. In the presence of zinc, failure occurred much earlier than in the uncoated state.

For a quantitative assessment, in analogy to the Gleeble test according to Béal [12], the load-bearing ability of the material up to fracture \( \Sigma_{\text{uncoated}} \) (uncoated state) and \( \Sigma_{\text{coated}} \) (coated state) was calculated according to Eq. (6) for all test materials. Then the relative reduction of the load-bearing ability up to fracture \( \Delta \Sigma \) was calculated according to Eq. (7) for all tested materials. Figure 10a shows the values for the comparative evaluation of the material with regard to the effect of the LME cracks on their fracture behavior. The literature regarding the LME occurrence and the evaluation under application-oriented conditions for arc welding processes is still sparse. There is no clear indication of what is the typical noise in this data, so it cannot be determined what reduction in load-bearing ability indicates that LME affected mechanical properties. To evaluate this question, DP600, TRIP700-I, and TRIP700-II were tested. Three test specimens in both surface conditions (coated and uncoated) were welded, and the standard deviation of the values for

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**Fig. 8** Comparison of the engineering stress-time curves of the test materials. (a) Test specimen in the uncoated state. (b) Test specimen in the hot dip zinc coated initial state

**Fig. 9** Superposition of the engineering stress-time curves of a material in hot dip zinc coated and uncoated state. (a) Typical engineering stress-time curves of a material without LME occurrence with the same load-bearing ability up to fracture using the example of the material TRIP690. (b) Typical engineering stress-time curves of a material with LME occurrence with significantly reduced load-bearing ability in the hot dip zinc coated state until fracture using the example of the TRIP1200
ΔΣ was determined and recorded as a scatter bar, as may be seen in the data for these materials in Fig. 10a. Accordingly, there was good reproducibility and a low scatter of the data in the PVR test of ±5%. From the analysis, a scatter range of ±10% was defined. If the relative reduction in load-bearing ability is upper 10%, one can consider that LME affected mechanical properties. This value agrees with Béal [12] who measured similar variation when measuring fracture strength of LME affected specimens using a Gleeble.

Using the variation noted above, it was seen that LME did not affect the load-bearing ability of TRIP690 and DP980 (see Fig. 10a). However, the test materials TRIP1100 had a ΔΣ_{TRIP1100} = 30% and TRIP1200 with ΔΣ_{TRIP1200} = 51%, which are above the +10% limit of the scatter range and show a clear influence of the LME cracking on the fracture behavior.

This consideration makes it possible to assess the fracture behavior of a base material under the influence of LME depending on its coating condition under a thermal–mechanical load. The higher the relative reduction of the load-bearing ability up to fracture, the greater the effect of the LME cracking on the thermal–mechanical fracture behavior of the base material.

4 Summary

It was shown that the LME sensitivity (susceptibility and fracture behavior) of materials in arc processes can be examined using the PVR test. The experimental setup enables the generation and control of a critical temperature range by heating using a TIG arc and the external application of a critical mechanical stress by linearly increasing deformation rate in the welding direction. A comprehensive assessment of the LME sensitivity can be made using a small number of samples in the coated and uncoated conditions.

The critical deformation rate \( v_{cr,LME} \) indicates the LME susceptibility of a material, and it may be used to compare the relative susceptibilities of multiple materials. The lower the critical deformation rate, the more susceptible is the test material under otherwise identical test conditions. It could be shown that the TRIP1100 and the TRIP1200 (in their present condition) have a similar critical deformation rate. Due to the relatively small difference in values of \( v_{cr,LME} \) of the two test materials, a definitive statement can only be made after the values have been statistically verified.

The evaluation of the fracture behavior of a material by means of the PVR test is a new method that has not been used before. The relative reduction of the load-bearing ability, ΔΣ, enables an assessment of the fracture behavior of the base material depending on its coating condition. The higher the relative reduction of the load-bearing ability up to fracture, the greater is the effect of the LME cracking on the thermal–mechanical fracture behavior of the base material. It could be shown that the effect of the LME cracking is significantly greater for the test material TRIP1200-GI (in its present condition) and leads to a greater reduction in deformability and load-bearing ability in comparison to the test material TRIP1100-GI (in its present condition).

The investigations show that a statement about the LME susceptibility of a material alone is not sufficient to be able to characterize its LME behavior. The effects of the cracks should be included in the evaluation.
The maximum deformation rate of \( v_{PVR,max} = 100 \text{mm/min} \) which was necessary to promote specimen fracture can be regarded as a guide value for materials with a fracture elongation \( \leq 24 \) if they are exposed to comparable thermal conditions.

An advantage offered by the testing method is the possibility of a simultaneously evaluation of the susceptibility of the test materials to hot cracking.

It was possible to show that materials of different thicknesses may be compared by designing the welding conditions to induce comparable thermo-mechanical conditions in the materials being compared. This may be done by normalizing the heat input per unit length of weld with respect to material thickness. An adjustment of the welding current is recommended as the manipulating quantity. This was proven by the evaluation of the susceptibility to solidification cracking as well as by the temperature-dependent tensile strength, which were adjusted to a comparable level.

5 Future work

For the further validation of the test method, investigations on other materials are planned, for which an assessment of the LME behavior is available from the literature based on Gleeble testing and resistance spot welding tests.

To obtain a better understanding of how the results from the presented arc-based testing method correlate with LME results from resistance spot welding tests an in-depth comparison between the present data with that from Wintjes et al. [29] is planned in follow-up examinations.

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Data availability Contact the corresponding author.

Code availability Not applicable.

Declarations

Conflict of interest Not applicable.

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