Wall cratering upon high velocity normal dust impact

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Abstract

Dust-wall high speed impacts, triggered by the termination of runaway electrons on plasma facing components, constitute a source of erosion. Normal high velocity mechanical impacts of tungsten dust on bulk tungsten plates are reproduced in a controlled manner by light gas gun shooting systems. Post-mortem surface analysis revealed that three erosion regimes are realized; plastic deformation, bonding and partial disintegration. The large impact statistics allowed the extraction of reliable empirical damage laws in the latter regime, which can be employed for erosion estimates in future reactors.

Keywords: dust in tokamaks, runaway electron termination, mechanical impacts, damage laws, impact bonding

1. Introduction

The presence of dust in fusion devices has important operational and safety implications, especially in future fusion reactors where metallic dust generation could scale up by orders of magnitude compared to existing devices and nuclear licensing requirements impose several restrictions on the total and hot dust inventories. Hence, the understanding and modelling of dust transport, production, re-mobilization, adhesion and survivability have received wide attention within the fusion community. Dust-wall mechanical impacts have emerged as an essential feature of dust dynamics in tokamaks. As a consequence of the curved ion flow and inertial effects, mechanical impacts, that span a very wide range of incident speeds, are unavoidable. They have been revealed to control dust migration in the fusion boundary plasma, to influence dust survivability, to play a pivotal role in the formation of dust accumulation sites and to constitute a source of plasma-facing component (PFC) damage.

Experimental and theoretical investigations mainly focused on the low-to-moderate impact speed range, \( v_{imp} \lesssim 200 \text{ m/s} \), being the most typical for micrometer dust-wall collisions in tokamaks. Studies also targeted the hyper-velocity range, \( v_{imp} \gtrsim 4000 \text{ m/s} \). Mechanical collisions between solid spherical micrometric projectiles and semi-infinite bulk solid targets have been extensively studied in the literature due to their relevance in geological phenomena, space applications, weapons research and technological applications. For refractory metal projectiles and targets (W, Mo), three impact speed ranges can be distinguished that are further separated in multiple regimes. In what follows, we shall present the elementary characteristics of each velocity range and briefly discuss its relevance to dust in tokamaks. It is worth pointing out that each velocity range limit depends not only on the material composition but also on the dust size and dust-wall temperature. Thus, the limits stated below should be deemed as indicative. The discussion is also applicable to oblique impacts, but not to molten projectiles or targets.

In the low-to-moderate velocity range, \( v_{imp} \lesssim 200 \text{ m/s} \), the mechanical impact induces weak plastic deforma-
mations to the projectile and the target. Two regimes can be distinguished; the *sticking regime* of $v_{\text{imp}} \lesssim 5 \text{ m/s}$ where the irreversible adhesive work performed during the near-elastic impact dissipates the incident kinetic energy of the projectile and thus leads to immobilization; the *inelastic rebound regime* $5 \lesssim v_{\text{imp}} \text{[m/s]} \lesssim 200$ where the projectile rebounds with a decreased kinetic energy due to adhesive work, plastic dissipation and frictional losses. This is the most typical range for tokamak dust that has been studied experimentally and theoretically for fusion-relevant materials. In particular, a unified analytical description has been achieved for both regimes by combining an elastic-perfectly plastic adhesive impact mechanics model with a rigid sliding body model.

In the **high velocity range**, $200 \lesssim v_{\text{imp}} \text{[m/s]} \lesssim 4000$, the mechanical impact induces strong plastic deformations, surface melting and partial fragmentation to both the projectile and target. Three regimes can be distinguished; the *deformation regime* of $200 \lesssim v_{\text{imp}} \text{[m/s]} \lesssim 500$ that is accompanied by severe projectile flattening & shallow target crater formation; the *bonding or cold spray regime* of $500 \lesssim v_{\text{imp}} \text{[m/s]} \lesssim 1000$ that is characterized by the adhesion of the projectile on the target, which can be realized by various mechanisms such as localized melting, adiabatic shear instability, viscous interlocking and interfacial amorphization; the *partial disintegration regime* of $1000 \lesssim v_{\text{imp}} \text{[m/s]} \lesssim 4000$ that is accompanied by material splash ejection and partial fragmentation. It has been recently demonstrated that the high velocity range is relevant for the solid dust that is produced by the explosive stopping of runaway electrons on PFCs. In particular, postmortem and *in situ* evidence from FTU (2013 shutdown and 2019 decommissioning), strongly supported by dust transport simulations and dedicated laboratory tests, revealed that the striking of runaway electrons on the equatorial poloidal lid led to the generation of very fast solid dust that generated multiple shallow craters upon impact on the adjacent toroidal limiters.

In the **hyper-velocity range** of $v_{\text{imp}} \gtrsim 4000 \text{ m/s}$, the impact velocity is comparable or larger than the compressive sound speeds of the projectile and the target. This triggers the emergence of strong shockwaves, upon whose sudden compaction extreme transient pressures and temperatures arise at the collision zone. Upon shockwave release, both materials extensively vaporize while the localized energy is large enough to cause atomic excitation and partial ionization. As a result, hyper-velocity impacts are accompanied by strong electrostatic and spectroscopic signals. In addition, the emerging craters feature excavated volumes that by far exceed the volume of the projectile. Finally, very high velocity solid ejecta can be released that cause further damage to the surrounding material. Solid evidence of hypervelocity impacts have been obtained in FTU that concerned the in-situ detection of dust impact ionization by electrostatic probes and the post-mortem observation of craters. However, considering the typical initial dust velocity distributions

![Figure 1: Image of a damaged bulk W target after the high velocity impact of spherical W dust (here $D_d = 51 \mu m$ and $v_{\text{imp}} = 2075 \text{ m/s}$) together with magnified SEM images of two regions that have been included in the statistical crater analysis.](image)

| $D_d$ | $v_{\text{imp}}$ | $\#$ | $D_c$ | $\sigma_D$ | $H_c$ | $\sigma_H$ |
|------|-----------------|------|------|-----------|------|-----------|
| 51   | 660 97          | 15.5 | +7.5 | 3.3       |
| 51   | 984 83          | 15.3 | -13.3| 3.9       |
| 51   | 1565 62         | 17.7 | +14.9| 5.7       |
| 51   | 2075 73         | 20.9 | +32.2| 5.1       |
| 51   | 2506 47         | 22.3 | +40.4| 5.3       |
| 51   | 2561 36         | 22.9 | +38.4| 7.1       |
| 51   | 3128 37         | 24.4 | +50.0| 6.7       |
| 63   | 596 110         | 17.5 | +8.1 | 3.2       |
| 63   | 764 64          | 18.4 | -26.3| 4.2       |
| 63   | 1012 63         | 18.2 | -14.3| 4.6       |
| 63   | 1534 72         | 20.5 | +21.4| 6.4       |
| 63   | 2039 78         | 24.8 | +43.3| 7.3       |
| 63   | 2485 37         | 26.8 | +51.1| 8.1       |
| 63   | 2500 45         | 27.4 | +47.5| 8.2       |
| 63   | 3108 33         | 35.1 | +62.4| 11.3      |
| 63   | 3190 36         | 30.8 | +66.6| 8.2       |
| 76   | 583 99          | 21.3 | +8.7 | 3.2       |
| 76   | 997 48          | 22.2 | -16.8| 5.8       |
| 76   | 1563 57         | 26.6 | +31.6| 10.5      |
| 76   | 2033 35         | 29.4 | +50.1| 7.3       |
| 76   | 2058 43         | 30.7 | +48.1| 5.8       |
| 76   | 2513 25         | 31.4 | +64.4| 8.9       |
| 76   | 2551 36         | 33.4 | +65.3| 10.7      |
| 76   | 3069 27         | 42.3 | +81.9| 14.2      |
| 76   | 3126 20         | 38.4 | +80.1| 9.2       |
and given the generally restricted acceleration lengths, no acceleration mechanisms are currently known that can lead to impact speeds within the hyper-velocity range $\frac{3}{8}$.

3. Experimental

High-sphericity low internal porosity W dust was purchased from “TEKNA Plasma Systems”. The original batch had a nominal size distribution of 45 – 90 µm. From this polydisperse batch, three nearly monodisperse sub-populations were meshed out using a sequence of six sieves with nominal sizes of 80, 75, 71, 63, 56 and 50 µm. The mean W dust diameters are $51(\pm 5)\mu m$, $63(\pm 5)\mu m$ and $76(\pm 5)\mu m$, sizes that are comparable to the most probable sizes of the fast Mo solid dust that was observed in FTU $\frac{13}{14}$. The bulk square W targets have 23 mm length and 4 mm thickness.

High velocity dust-wall impacts are realized by means of a light gas gun system $\frac{13}{14}$. In the two stage configuration, the first stage is a high pressure reservoir connected to the second stage with a fast valve. The second stage (or pump tube) is a cylinder in which the light gas is fed at relatively low pressure and compressed by a free piston following the fast valve operation. The compressed light gas rapidly expands into the launch tube, which has a smaller diameter than the pump tube, simultaneously accelerating a macro-scale projectile (sabot) which features a cavity that is loaded with micron dust. The dust particles are separated from the sabot at the end of the launch tube where the latter impinges on a diaphragm. The dust particles free stream with high speeds into a vacuum chamber that features the target. Impact speeds within the range of $1000 \leq v_{\text{imp}}[m/s] \leq 3500$ have been realized by employing different light gases (nitrogen or hydrogen) and setting different initial pressures for the first (20 – 40 bar) & second (0.8 – 1.2 bar) stage. In the single stage configuration, the high-pressure hydrogen reservoir is directly connected to the launch tube without an intervening piston and the achieved impact speeds lie within $400 \leq v_{\text{imp}}[m/s] \leq 1000$.

The dust speed is measured through the dust transit time between two laser sheets with an uncertainty that is generally less than ±1%. This measurement error primarily originates from the relatively large thickness of the individual beams (1 mm) when compared to the beam spacing (100 mm). The optical method was preferred over the time-of-flight method $\frac{14}{15}$, where the start trigger is provided by the muzzle flash due to sabot ejection from the barrel and the stop trigger is provided by the impact flash, whose precision is $\sim \pm 3\%$. The dust cloud speed should well approximate the dust particle speed, given the nearconstant small cloud width recorded by the laser sheets.

Overall, 34 impact tests were performed with the three dust sub-populations and impact speeds covering the high velocity range. Light gas gun acceleration leads to a large number of W-on-W impacts, since it is not possible to load a small dust number on the sabot cavity. However, the useful crater statistics are rather restricted, since the central spot features many overlapping craters that should be excluded from the analysis (see figure $\frac{11}{13}$) and since the central spot also features few visibly smaller craters due to dust-dust impacts in the proximity of the target (see the finite width of the dust cloud). Naturally, it is possible to minimize the central spot by loading less dust on the sabot, but this compromises the accuracy of the speed measurement and does not necessarily improve statistics. From the 34 shot samples, 25 featured a large enough number of isolated craters suitable for statistical analysis (Table $\frac{17}{19}$).

The shot W-on-W samples were first mapped by means of a Scanning Electron Microscope (SEM), at low magnification and at high resolution, in order to statistically estimate the average value of the crater diameter at different impact conditions. The crater depth was then measured with a precision optical microscope (Leitz Wetzlar Ortolux) of 0.5 µm sensitivity. Moreover, few craters from each shot sample were mapped with a mechanical profiler (KLA-Tencor mod. P-15) in order to verify the precision of the above techniques. The instrumental uncertainty was estimated to be $\pm 3\%$ for the crater depth (optical) and $\pm 15\%$ for the crater diameter (SEM). The average values and standard deviations of the crater dimensions (instrumental and statistical errors) are listed in Table $\frac{19}{19}$.

Since the impact speeds varied from 583 m/s to 3190 m/s, all three impact regimes of the high velocity range were observed for all the three dust sub-populations. Characteristic examples are illustrated in figure $\frac{12}{20}$.
4. Analysis

4.1. Empirical damage laws

Impact damage laws are empirical correlations of the crater diameter and depth as function of the target and projectile material properties, impact speed, impact angle and dust size \[46, 47, 48, 49, 50, 51, 52\]. Systematic hyper-velocity dust impact experiments have led to the formulation of empirical damage laws that are applicable in quite extended impact velocity and dust size ranges \[46, 47, 48\]. Considering their validity for various target-projectile combinations, such general damage laws are of limited accuracy, especially in the high impact velocity range. In what follows, we utilize our extensive W-on-W experimental data to extract empirical damage laws valid within the disintegration regime of the high velocity range, \(1 \lesssim v_{\text{imp}} [\text{km/s}] \lesssim 3.5\).

The experimental crater depth and crater diameter data are fitted in the standard power law form of \(\alpha (D_d)^{\beta} (v_{\text{imp}})^{\gamma}\) with \(\alpha, \beta, \gamma\) the fitting parameters. The two-variable three-parameter non-linear fit led to the empirical damage laws

\[
D_c = 0.0330 (D_d)^{1.005} (v_{\text{imp}})^{0.527},
\]

\[
H_c = 0.0000114 (D_d)^{1.264} (v_{\text{imp}})^{1.282},
\]

with \(D_c, H_c, D_d\) measured in \(\mu m\) and \(v_{\text{imp}}\) in m/s. The mean absolute relative fitting error is 1.87% for the diameter and 6.89% for the depth. The standard errors for the fitting parameters of the diameter are 0.0009 (\(\alpha\)), 0.041 (\(\beta\)), 0.027 (\(\gamma\)). The standard errors for the fitting parameters of the depth are 0.00000803 (\(\alpha\)), 0.098 (\(\beta\)) and 0.072 (\(\gamma\)). The empirical damage laws are plotted in figure 3.

An established general damage law for the crater depth, appropriate for metallic dust exceeding 50 \(\mu m\) and impact speeds within \(2 \lesssim v_{\text{km/s}} \lesssim 12\), reads as \[46, 47, 48\]

\[
H_c = 5.24 \frac{D_d^{10/18}}{h_c^{1/4}} \left(\frac{\rho_d}{\rho_t}\right)^{1/2} \left(\frac{v_{\text{imp}}}{c_t}\right)^{2/3},
\]

where \(H_c\) is the crater depth in cm, \(D_d\) the dust diameter in cm, \(h_t\) the target’s Brinell hardness number, \(\rho_d, \rho_t\) the mass densities of the dust and the target, \(c_t\) the target’s sound speed and \(v_{\text{imp}}\) the dust impact speed. It is evident from the exponents that the general damage law exhibits a weaker dependence on both the dust size and the impact speed than the W-on-W damage law. More specifically, as discerned from figure 4 the general damage law significantly underestimates the crater depth for large dust sizes and high impact speeds. This confirms the importance of target-projectile specific damage laws as those of Eqs. (12).

Finally, empirical damage laws allow for an estimate of the excavated volume provided that an additional assumption is made for the geometry of the crater. As expected from symmetry considerations arising from the high dust sphericity, local target planarity and normal nature of the impact, the crater geometry can be well-approximated by a spherical cap. This has been consistently observed in the laboratory impact tests, see figure 5 and is consistent with
the experimental hyper-velocity literature. Thus, the excavated volume can be approximated by

\[ V_c = \frac{1}{6} \pi H_c \left( \frac{3}{4} D_c^2 + H_c^2 \right). \]  

(4)

Some characteristic examples are provided in figure 6. Note that the excavated material from a single impact crater is much larger than the excavated material from a single unipolar arc crater, but much smaller than the material excavated during an edge-localized mode or major disruption driven melt event.

4.2. Critical velocities

Here, we shall discuss the transition between the high velocity regime for normal W-on-W impacts, which dictates the applicability limits of our empirical damage laws.

The critical erosion velocity \( v_{\text{crit}}^{\text{ero}} \) is the transition velocity between the bonding and disintegration regimes of the high velocity range. An analytical expression has been derived for \( v_{\text{crit}}^{\text{ero}} \) that associates the transition with the impact kinetic energy that triggers melting. A first-order energy balance analysis that contains a number of approximations leads to the expression [31]

\[ v_{\text{crit}}^{\text{ero}} = \left( \frac{20 \epsilon_{\text{th}} I_{\text{melt}}}{\rho_d \sqrt{D_d}} \right)^{2/5}, \]

where \( \epsilon_{\text{th}} = \sqrt{\rho_d k_d c_{pd}} \) is the so-called thermal effusivity with \( k_d \) the thermal conductivity and \( c_{pd} \) the specific isobaric heat capacity, where \( I_{\text{melt}} = T_m - T_0 + \Delta h_f/c_{pd} \) is a melting index with \( T_m = 3695 \) K the melting point of W, \( T_0 = 300 \) K the room temperature and \( \Delta h_f \) the latent heat of fusion. This expression should be expected to underestimate the critical erosion velocity but has exhibited a good agreement with dedicated Sn, Bi, Zn, Ti experiments [31].

The critical bonding velocity \( v_{\text{crit}}^{\text{bond}} \) is the transition velocity between the deformation and bonding regimes of the high velocity range. A closed-form correlation is available for \( v_{\text{crit}}^{\text{bond}} \) that is based on numerical simulations within the assumption that impact bonding can be attributed solely to the onset of adiabatic shear instabilities. Additional numerical modelling and experimental results allowed the consideration of size effects in the original expression [34].

The final expression reads as [33, 34]

\[ v_{\text{crit}}^{\text{bond}} = \left[ 667 - 0.014 \rho_d + 0.08(T_m - T_0) + 10^{-7} \sigma_y \right] (D_d)^{-0.07}, \]

where all quantities are in SI units and \( \sigma_y \) denotes the yield strength. It is worth pointing out that the adopted size exponent is a factor of two less than its traditional value [33] in accordance with state-of-the-art experiments [34].

In figure 7, the two W-on-W critical velocities are plotted as a function of the dust size. The room temperature recommendations of Ref. [59] have been followed for the W thermophysical properties. Velocity-size combinations that have been experimentally verified to lead to impacts that belong to the deformation and bonding regimes have also been included. It is evident that the aforementioned general expressions underestimate both critical velocities.

Figure 5: SEM images of high velocity W-on-W normal dust impacts that fall into the partial disintegration regime for different dust sizes and different impact speeds: (a) \( D_d = 51 \) µm, \( v_{\text{imp}} = 1506 \) m/s, (b) \( D_d = 63 \) µm, \( v_{\text{imp}} = 3190 \) m/s, (c) \( D_d = 76 \) µm, \( v_{\text{imp}} = 2033 \) m/s.

Figure 6: The excavated volume predictions of the empirical damage laws for the crater diameter and the crater depth in the partial disintegration regime under the spherical cap assumption. Excavated volume as a function of the impact speed for various W dust sizes.
5. Summary and future work

The first experimental study of high velocity W dust - W wall impacts, relevant for cascade-like PFC damage caused by runaway electron stopping on PFCs, has been carried out. Controlled normal high velocity impacts have been realized with a light gas dust gun. Surface analysis allowed the identification of three wall damage regimes (plastic deformation, impact bonding, partial disintegration) and the extraction of empirical damage laws for the most harmful disintegration regime. Work in progress focuses on the extraction of empirical damage laws for the most harmful disintegration regime. Work in progress focuses on the extension of the impact data to obtain more accurate damage laws, the determination of the critical velocities that demarcate the damage regimes and the realization of oblique impacts. In the future, we aim to quantify the effect of elevated dust temperatures; a step that is particularly important given that explosive runaway electron termination on PFCs is expected to produce fast hot solid dust. Concerning the potential significance of high velocity solid dust impacts in fusion devices, the following remark is important given that explosive runaway electron termination could have a major extent, it has a localized character and it could be constrained to sacrificial limiters or to replaceable divertor plates in future fusion reactors. On the other hand, secondary high velocity dust impact damage should be considerably less dramatic, but it has a delocalized character and could thus compromise the integrity of plasma facing components including diagnostics.

Acknowledgments

The work has been performed within the framework of the EUROfusion Consortium, funded by the European Union via the Euratom Research and Training Programme (Grant Agreement No 101052200 - EUROfusion). Views and opinions expressed are however those of the authors only and do not necessarily reflect those of the European Union or European Commission. Neither the European Union nor the European Commission can be held responsible for them.

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