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280 GHz Focal Plane Unit Design and Characterization for the SPIDER-2 Suborbital Polarimeter

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Abstract We describe the construction and characterization of the 280 GHz bolometric focal plane units (FPUs) to be deployed on the second flight of the balloon-borne SPIDER instrument. These FPUs are vital to SPIDER’s primary science goal of detecting or placing an upper limit on the amplitude of the primordial gravitational wave signature in the cosmic microwave background (CMB) by constraining the B-mode contamination in the CMB from Galactic dust emission. Each 280 GHz focal plane contains a 16 × 16 grid of corrugated silicon feedhorns coupled to an array of aluminum-manganese transition-edge sensor (TES) bolometers fabricated on 150 mm diameter substrates. In total, the three 280 GHz FPUs contain 1,530 polarization sensitive bolometers (765 spatial pixels) optimized for the low loading environment in flight and read out by time-division SQUID multiplexing. In this paper we describe the mechanical, thermal, and magnetic shielding architecture of the focal planes and present cryogenic measurements which characterize yield and the uniformity of several bolometer parameters. The assembled FPUs have high yields, with one array as high as 95% including defects from wiring and readout. We demonstrate high uniformity in device parameters, finding the median saturation power for each TES array to be ∼3 pW at 300 mK with a less than 6% variation across each array at 1σ. These focal planes will be deployed alongside the 95 and 150 GHz telescopes in the SPIDER-2 instrument, slated to fly from McMurdo Station in Antarctica in December 2018.

Keywords Detector Packaging — Magnetic Shielding — Transition-Edge Sensors — Scientific Ballooning — Cosmic Microwave Background

1 Introduction

The field of observational cosmology is moving at a rapid pace, with recent measurements of the polarization in the cosmic microwave background radiation (CMB) pushing the frontier of our investigative capabilities. The current focus is on the primordial B-mode polarization in the CMB, expected to have been sourced by gravitational waves passing through the surface of last scattering. If this signal exists, it is obscured in observations by foreground signal, in particular that from Galactic dust.

The SPIDER instrument includes six refracting telescopes designed to measure the polarization in the CMB from a ballooning platform. These telescopes are collectively encased within a 1300 L liquid helium cryostat with an approximately 20 day hold time. Over 2300 transition-edge sensors (TES) are cooled to their 300 mK operating temperature by a closed-cycle, helium-3 adsorption refrigerator at the base of each telescope, and read out by time-division multiplexing (TDM) superconducting quantum interference devices (SQUID).

The first SPIDER instrument was launched in January 2015 from McMurdo Station in Antarctica and contained six 95 and 150 GHz slot-antenna-coupled TES arrays fabricated at JPL. During the 16 day flight, SPIDER-1 observed 10% of the sky from an altitude of 36 km and collected 2 TB of raw data, currently under analysis.

The second flight is scheduled for December 2018 and will re-deploy three proven SPIDER-1 telescopes alongside three NIST-fabricated feedhorn-coupled 280 GHz TES focal plane units (FPUs). Otherwise, SPIDER-2 will have effectively the same construction as the first instrument.

The 280 GHz band has high signal-to-noise in Galactic dust foregrounds and complements the 95 and 150 GHz data, enabling improved characterization of the foreground spec-

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The continuum contribution of the atmosphere limits detector sensitivity at frequencies above about 200 GHz for ground-based observations. At float altitudes the atmospheric loading is reduced by several orders of magnitude, enabling background-limited observations of the microwave sky at these frequencies.

In this proceeding we describe the mechanical, thermal, and magnetic shielding architecture of the 280 GHz FPUs and present cryogenic measurements of the yield and bolometer parameters.

2 280 GHz Focal Plane

The unique constraints of ballooning necessitate a compact, modular FPU structure and robust shielding from ambient magnetic fields at float. The 280 GHz FPUs were designed to utilize proven SPIDER-1 packaging and magnetic shielding with the NIST sensor array units, and to fit within the SPIDER-1 optical and cryogenic design such that the telescope could slot in to any port in the cryostat.

Figure 1a shows a fully assembled FPU, including the sensor array assembly centered on the copper faceplate on the sky side of the box. The first two stages of SQUID readout are under the sensors, within several layers of high-permeability and superconducting shielding. The 280 GHz SPIDER FPU packaging complies with the previously mentioned constraints while maintaining the advantages of the low loading environment at 36 km altitude.

2.1 Sensor Array Assembly

The 280 GHz sensor array assembly is composed of a 16 × 16 array of conical, corrugated feeds coupled to a monolithic detector array fabricated on a 150 mm diameter silicon wafer. The detector array stack consists of the silicon backshort, detector array wafer, and feedhorn interface wafer. These components are pressed against the corrugated feedhorn array via a 300 mK copper bracket and BeCu springs. The TES wafer is heat sunk to 300 mK through the silicon backshort, which is gold-plated on the bracket side. The detector stack is thermally connected to the 300 mK bracket via tack-bonded gold ribbons, seen in Figure 1b.

On the detector array, each feedhorn is coupled to two TESs, one for each orthogonal polarization mode. Each bolometer pair, considered one pixel, is rotated 45° from its neighbor for simultaneous coverage of Q and U Stokes parameters. Every TES bolometer island contains two sensors in series with differing superconducting critical temperatures ($T_c$): a 420 mK AlMn sensor for flight operations and a 1.6 K Al sensor for on the ground pixel characterization.

In total, the three SPIDER-2 280 GHz focal planes contain 765 spatial pixels and 1,530 polarization sensitive bolometers. These exhibit an electrical noise equivalent power (NEP) of $2.6 \times 10^{-17}$ W/$\sqrt{\text{Hz}}$, as reported in Hubmayr, et al, 2016. Details of the sensor array and detector stack, including corrugated feeds and pixel fabrication, are well-described in previous publications.

2.2 Sensor Packaging

Thermal connections on and within the focal plane begin with the gold-plated copper faceplate, which is itself thermally coupled to the cold point of the 3He adsorption refrigerator at the base of the telescope via a copper bus and 300 mK ring cradling the FPU. The high heat capacity faceplate and stainless steel mounting blocks between the ring and FPU serve as low-pass filters for any thermal fluctuations from the adsorption refrigerator. The components internal to the FPU are maintained at 300 mK by two gold-plated copper heat straps.
Fig. 1: (a): Fully assembled 280 GHz focal plane unit, ready for installation. The SPIDER-2 FPUs are designed to reuse proven SPIDER-1 packaging and magnetic shielding, hybridized with the NIST sensor array units. Image (a) shows the 300mK gold-plated copper faceplate, with the corrugated feedhorns of the sensor array assembly set in the cutout at the center. The two gold-plated copper heat straps are thermally sunk to the faceplate and bent in to the box, cooling the internal components from the center to minimize trapping magnetic flux. The superconducting NbTi cabling electrically connects the first and second stage multiplexing SQUIDs to the 2K SSA and warm readout electronics. The niobium box serves as both the outer layer of the packaging and as superconducting magnetic shielding. (b): Sensor array unit. Here, the sky side of the detector stack is in to the page. This focal plane is in the middle of assembly, with detector and flexible cabling (“flexi”) bondpads exposed in preparation for bonding. Nb-traced silicon breakout chips are used to mate the flexi cable and detector connections. For bonding, the flexi cables are protected by temporary covers, and the MUX readout PCBs are folded under the faceplate (between the table and faceplate in the photo).

These are contorted such that they can be firmly heat sunk to the copper faceplate on the outside of the box, as shown in Figure 1a, then bent inside to form solid thermal contact with (1) the top of the niobium box and high-µ Cryoperm sheet and (2) the high-µ sleeves and copper backplates of the four multiplexing (MUX) SQUID readout boards. The thermal straps cool these components from the center to avoid trapping magnetic flux within the shielded focal plane box.

The multiplexing SQUID readout is set behind the detector stack, within the shielding niobium and aluminum boxes, and connected to the sensor array unit via superconducting aluminum traced flexible circuits (“flexi cables”). The flexi cables electrically connect the TES bondpads to the silicon shunt resistor and MUX chips. At the detector end, the flexi cables are secured on the underside of the faceplate as in Figure 1a. Niobium-traced silicon breakout chips (NIST-designed and fabricated) are used between the detector wafer and flexi cabling, due to the fact that the 170 µm detector bondpad pitch could not be matched by commercial flexi cables.

The shunt resistor and MUX chips are mounted on Tech-Etch printed circuit boards (PCBs) attached to the flexi cabling. These PCBs connect to superconducting NbTi cables via a breakout board, then exit the FPU through slits in the niobium box (see cabling exiting the FPU in Figure 1a). The NbTi cables connect the FPU to the 2K SQUID series array (SSA) amplifier and warm Multi-Channel Electronics mounted to hermetic flanges on the belly of the cryostat.

2.3 Magnetic Shielding

In ballooning, the magnetic environment of the payload is changing over time in a way that is challenging to measure or model. It is not easily removed in post-processing as part of scan synchronous noise, and will not necessarily integrate down over time. These time-varying
magnetic fields can impact instrument performance by altering the critical temperature of the TES and by adding flux through the SQUIDs, both of which can mimic signal. At the SSA, a nonuniform field can cause destructive interference of the V-φ curves, resulting in weak signal amplification.

The SQUIDs within the SPIDER 280 GHz FPU box are shielded by several layers of high-µ and superconducting metals. These include Amuneal-4K (A4K) high-µ sleeves, superconducting Al and Nb boxes, and high-µ Metglas sheets (Figure 2). This shielding was designed to attenuate ambient magnetic fields at the SQUIDs by $10^7$ and at the detector wafer by $10^4$. These values were deemed sufficient prior to the 2015 flight from laboratory testing and magnetic modeling. With the exception of a Nb backshort found in the SPIDER-1 FPUs that served as both a detector backshort and Nb box top, the 280 GHz magnetic shielding configuration is identical to the SPIDER-1 strategy described in Runyan, et al, 2011. From simulations using the Ansys Maxwell 3D magnetostatic software, we found that we can meet our shielding goals without the Nb backshort.

In this configuration, the TES are outside the superconducting boxes, and thus are not as well shielded as the SQUIDs. This is addressed by placing a high-µ “spittoon” above the focal plane. The SSA is shielded separately from the rest of the components by a high-µ sleeve inside a superconducting Nb case, which is then wrapped in ten layers of Metglas sheeting. Finally, the entire insert is surrounded by two layers of A4K.

Due to the non-uniformity in magnetic coupling across the wafer found in lab testing, dark SQUIDs (SQUIDs with no TES) can provide the magnitude of scan-synchronous pickup within the FPU box, but not an adequate template for noise removal. From analysis of SPIDER-1 flight data, the scan-synchronous noise in the dark SQUID channels is less than 1% and can therefore be entirely accounted for by the estimation of cross-talk. This indicates that the SPIDER-1 design provided sufficient magnetic shielding at the SQUIDs; thus we were motivated to utilize the same architecture for the SPIDER-2 focal planes.

To compare the pickup in the SPIDER-1 and -2 focal planes, we took measurements of
The median critical temperature is 445 mK with a 1.6%, or 7 mK, spread at 1σ. A and B detectors, shown here as top (A) and bottom (B) halves of a pixel square, are orthogonal (§2.1). Note that white detectors are dead (§3), however detector column 8 has since been fixed.

Notably, the pickup is highest when the TESs are in their superconducting state; indicative of parasitic pickup in the wiring that connects the TESs to the SQUIDs, and not due to $T_c$ coupling to the magnetic field. These measurements showed no dark SQUID pickup in either FPU, further indication that the shielding inside the FPU box is sufficient.

In the 280 GHz FPU, this pickup in the wiring is approximately 10 times that of the dark SQUID noise floor; however these tests were done without the high-$\mu$ spittoon and multiple cylindrical shields. From simulation, these shields attenuate ambient fields by $10^3$, therefore we expect the full shielding configuration to mitigate this coupling below the noise floor.

### 3 Focal Plane Characterization and End-to-End Fabrication Yield

Despite the elaborate packaging scheme described in §2, we achieved high yields for the assembled focal planes. Cold, dark characterization of the 280 GHz focal planes was accomplished by taking IV curves at several bath temperatures ($T_{\text{bath}}$) from 461 mK down to 330 mK, the minimum temperature of the test dewar, with a gold-metalized silicon plate covering the feedhorns. Each curve provided a saturation power ($P$) for the detector at that temperature step. From there we obtained the critical temperature, conductivity of the bolometer legs at $T_c$ ($G_c$), and curvature ($\beta$) by fitting to the functional form

$$P = \frac{G_0 T_0}{1 + \beta} \left[ \left( \frac{T_c}{T_0} \right)^{1+\beta} - \left( \frac{T_{\text{bath}}}{T_0} \right)^{1+\beta} \right],$$

and extrapolating to 300 mK, the nominal FPU operating temperature. The thermal conductance of the bolometer legs depends on the temperature like so

$$G(T) = G_0 \left( \frac{T}{T_0} \right)^\beta.$$
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Table 1: End-to-end yield and parameter estimates at 300 mK for the three 280 GHz focal planes, dubbed Y3, Y4, and Y5 depending on which SPIDER-1 FPU their packaging came from. Parameters were measured at various bath temperatures down to 330 mK, then extrapolated to the 300 mK flight operating temperature (§3). The median values are listed and the quoted spread at 1σ is the maximum for any of the FPUs.

Note that $T_0$ is the reference temperature where $G_0 \equiv G(T_0)$ is defined. Normal resistances ($R_n$) were obtained directly from the slope of the load curves at high bias. The resulting parameter estimates for all three 280 GHz SPIDER focal planes are listed in Table 1. Figures 3a and b show the saturation powers and critical temperatures across one 280 GHz FPU.

For all three arrays, the bolometer fabrication yield is $>99\%$. Additional loss of yield is either broken wiring or SQUID amplifier failures, the former being the dominant source. For example, the initial wire bonds could not withstand cold cycling due to differential thermal contraction between the detector array and the silicon breakout chips (Figure 1b). Optimizing bonding parameters appears to have largely solved the problem, but we suspect this remains the largest source of pixel loss.

Based on load curve analysis at 330 mK, the 280 GHz FPUs have conservatively estimated end-to-end yields of 81%, 88%, and 95% (Table 1), including wiring and readout defects. Subsequent fixes and warm continuity tests lead us to believe that these yields have improved further.

4 Conclusion and Status

The three 280 GHz focal planes have been assembled and cryogenically screened. We find bolometer parameters $P_{sat}$, $T_c$, and $G_c$ that match our targets with high yield and uniformity ($<6\%$ spread) across the wafer. We are currently integrating the arrays into the telescopes and flight cryostat for further characterization. This will include polarimetric measurements, full-system optical efficiency and noise measurements, bandpass measurements via Fourier transform spectroscopy, and internal loading estimates. The SPIDER-2 instrument is on track for a December 2018 flight from McMurdo Station in Antarctica.

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