Effect of baffle shape in heat transfer for jet impingement on a solid block

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Abstract. The numerical solution solution is obtained for fluid flow and heat transfer in a confined impinging slot on a solid block with the presence of baffles. In order to consider the effect of baffle shape the rectangular and semi circular baffles are considered and for the effect for Reynolds number the Reynolds number is varied from 100 to 300 with the step of 50. The present study reveals the vital impact of Baffle shape and Reynolds number (Re) on the fluid flow and heat transfer characteristics over a wide range. It is finally added that the presence of baffle improves the Nusselt number. The Nusselt number increases with the increase of Reynolds number. The present study proved that, the primary peak of Nusselt number occurs nearer to the reattachment length. The secondary peak of Nusselt number occurs nearer to the baffle. It is observed that for semi circle baffle the velocity attains maximum one compared to rectangular baffle.

1 Introduction

The jet impingement cooling technique is used in industries like turbine blade cooling, rocket launcher cooling, electrical, metal annealing, and textile drying and electronic equipments in order to confiscate a huge amount of heat. The common types of impingement are confined and unconfined jets. The design and fabrication of confined jets are very complicated compared to unconfined jets. There are many numerical and experimental investigations are conducted to find out the effect of confinement wall on jet impingement. The experimental and numerical predictions are mostly related to turbulent jet impingement to have fast cooling. But in the case of cooling of electronic equipments the area need to be cooled is small and also the velocity. In order to consider the present situation, the laminar regime is considered for the entire computational domain.

Habib et al[1] experimentally investigated that there was increase in average nusselt number with the increase of Reynolds number and block height. Hussein M. Maghrabie et al[2] numerically investigated the heat transfer and pressure drop of in line array of heated obstacles cooled by jet impingement in cross flow. They concluded that there was a significant change in jet impinging cross flow and normal flow of jet impingement. Withada Jedsadaratanachai et al.[3] predicted numerically V baffles can induce a drastic increase in Nusselt number and higher friction.

Yang etal [4] investigated experimental as well as numerically in array of jet impingement on a concave surfaces. The finally concluded that increase in jet diameter hole will not improve the heat transfer significantly. Eiamsa ard[5] et al numerically examined the fluid flow and heat transfer characteristics in the presence of triangular wavy baffles. They concluded that for 30° and 45° baffles endow with more nusselt number compared to 60° baffles. Alsanossi M. Aboghrara [6] examined analysis of solar air heater with jet impingement on corrugated absorber plate. They exposed that corrugated plate is the strong function of heat transfer enhancement. In previous series of studies performed by author [7&8] concentrated more on fluid flow structure behavior of different fluids like air, water and nano fluid jet impingement on a solid block. In the present paper the author revealed a numerical study report on the effect of imposing the different shape of baffles on jet impingement on a solid block. The primary aim of this study is to elucidate the enhancement in heat transfer by incorporating baffles by (i.e. increasing area).

2 Problem Descriptions

A The schematic diagram for single slot jet impingement on a solid block in the presence of baffles are shown in figure 1. The air jet come out from a nozzle of width W, with a uniform velocity (U), impinges normal to a
computational domain which is called impingement plate at a distance of L from the nozzle. The horizontal and vertical velocity are considered as $U$ and $V$ respectively. Since the computational domain is symmetry about vertical or y-axis, it is prudent to model and simulate only right hand of the computational domain thereby saving precious computing power and time.

![Fig. 1(a). Schematic diagram of computational domain with rectangular deflector](image)

The inlet velocity of air ($U$) is calculated by using the formula: $Re = (\rho Ud_h / \mu)$ where $\rho$ denotes the density of the fluid; $D_h$ the hydraulic diameter, and $\mu$-dynamic viscosity of the fluid. The deflectors are provided at the top and bottom walls at equal distance having same area. Dimensionless velocity ($U$&$V$) are determined by $U = u / U_{in}$, $V = v / U_{in}$ where $u$ and $v$ are the dimensional velocities and $U_{in}$ is considered as the reference velocity and equal to unity. The aspect ratio (AR) is defined as length of the domain in Y direction($L_y$) to hydraulic diameter($D_h$)

### 2.1 Boundary Conditions

The numerical study had a uniform wall temperature at the bottom wall and top wall is considered as an adiabatic one. Uniform velocity is considered as inlet and pressure outflow is considered at outlet. Symmetry condition is adopted by considering right hand of the computational domain inorder to reduce the computational time.

![Fig. 1(b). Schematic diagram of computational domain with baffle shaped deflector](image)

### 2.2 Assumptions

The following assumptions were made while solving this problem:

- Flow was taken as laminar i.e. Reynolds number varies from 50 to 300 with the step of $Re=50$ only since the presence of the block induces turbulence at low Reynolds number also.
- Incompressible fluid.
- The case solved here is that of non-conjugate heat transfer i.e. the effect of the conductivity of the slab material is not taken into account.
- The geometry at the outlet is assumed to be so long that the flow eventually gets fully developed.

### 2.3 Commercial Package Used

The continuity equation, momentum equation and energy equation are solved by commercial package ANSYSCFD15.0.

\[
\frac{\partial U}{\partial X} + \frac{\partial V}{\partial Y} = 0 \quad \text{Continuity Equation}
\]

\[
X \text{ Momentum Equation} \\
U \left( \frac{\partial U}{\partial X} \right) + V \frac{\partial U}{\partial Y} = -\frac{\partial P}{\partial X} + \frac{1}{Re} \left( \frac{\partial^2 U}{\partial X^2} + \frac{\partial^2 U}{\partial Y^2} \right)
\]

\[
Y \text{ Momentum Equation} \\
U \left( \frac{\partial V}{\partial X} \right) + V \frac{\partial V}{\partial Y} = -\frac{\partial P}{\partial Y} + \frac{1}{Re} \left( \frac{\partial^2 V}{\partial X^2} + \frac{\partial^2 V}{\partial Y^2} \right)
\]

Energy equation:

\[
U \frac{\partial \theta}{\partial X} + V \frac{\partial \theta}{\partial Y} = \frac{1}{Re Pr} \left( \frac{\partial^2 \theta}{\partial X^2} + \frac{\partial^2 \theta}{\partial Y^2} \right)
\]

The non-dimensional variables considered for this problem are,

$X = Lx / D_h$, $Y = Ly / D_h$, $U = u / U_{in}$, $V = v / U_{in}$ where $u$ and $v$ are the dimensional velocities and $U_{in}$ is considered as the reference velocity and equal to unity. The aspect ratio (AR) is defined as length of the domain in Y direction($L_y$) to hydraulic diameter($D_h$)

\[
Re = (\rho u D_h / \mu)
\]

$Nu = (\rho u D_h / \mu)$ and $Nu = \left( \frac{\partial \theta}{\partial Y} \right)_{Y=0}$

where $\theta_i = \frac{T_{r-T_{c}}}{T_{r-T_{c}}}$ is the non dimensional temperature in the fluid region and $\theta_s = \frac{T_{w-T_{c}}}{T_{w-T_{c}}}$ is the non dimensional temperature in the solid region.

The term $\theta_{s}$ is always 1 for the non-conjugate case, i.e., the effect of the slab thickness is neglected. Here the temperature of the wall is assumed as the temperature of the hot fluid ($T_{w}=T_{h}$).

### 2.4 Validation of the problem

In order to prove our algorithm the validation work has been done against sivasamy et al paper[4]. The centers of the primary and secondary vortexes for aspect ratio 5 for Reynolds number from 100 to 400, the same for Reynolds number 200 and aspect ratio 2 to 5 has been used as the parameters for validation.
Table 1 compares the center of primary vortex in x coordinate produced by present algorithm with sivasamy et al result. Table 1 compares the center of primary vortex in y coordinate produced by present algorithm with sivasamy et al result.

Table 1 Centres for primary vortex in X &Y-coordinate.

| Re  | X         | Y         | X         | Y         |
|-----|-----------|-----------|-----------|-----------|
|     | Present   | Sivasamy  | Present   | Sivasamy  |
| 100 | 1.90      | 1.90      | 1.43      | 1.41      |
| 200 | 3.12      | 3.07      | 1.52      | 1.49      |
| 300 | 4.08      | 4.05      | 1.60      | 1.56      |
| 400 | 4.86      | 4.84      | 1.64      | 1.63      |
| 500 | 5.43      | 5.42      | 1.71      | 1.67      |

The figure 3 shows that, the coefficient of friction in bottom wall is compared with sivasamy et al and it is having good agreement also.

Fig 2. X-velocity contours and streamlines for aspect ratio 5 and Re=100

Table 2 Grid Independence Study

| Grid level | Total No of nodes | Average Nu Number for the bottom wall |
|------------|-------------------|--------------------------------------|
| Grid 1     | 259246            | 8.31                                 |
| Grid 2     | 274584            | 10.5                                 |
| Grid 3     | 289584            | 11.367                               |
| Grid 4     | 304584            | 12.31                                |
| Grid 5     | 319922            | 12.3768                              |
| Grid 6     | 335260            | 12.4092                              |

Table 2 reveals the prediction of nusselt number for same aspect ratio and Reynolds number. From the table it is concluded that there was not a substantial change in Nusselt Number after Grid Number 4. So Grid Number 4 is considered as the grid selection for the particular aspect ratio.

3 Numerical Approach and Procedures

In the present numerical investigation the two dimensional laminar incompressible jet is considered for the present study. The single phase model is adopted in the present simulation for the sake of simplicity. No slip condition and no penetration conditions are applied for the velocity components on the solid wall. The present numerical study concentrates on heat transfer investigation in the presence of solid block with different shape of deflector. The algorithm used for solving is the semi implicit method for pressure linked equation algorithm. The numerical investigations based on finite volume method are simulated by ANSYS FLUENT 15.0 CFD code. The Tecplot 9.0 (postprocessor) is used to interpret the result, for obtaining the contour plots of X velocity, Y velocity and streamline contour for flow field.

4 Results and Discussion

The present author has already done heat transfer investigation in jet impingement on a solid block. Numerical simulations of heat transfer characteristics of
jet impingement in the presence of baffles are presented. In the present investigation rectangular and semi circle (Dimple) shaped baffles are adopted. The influence of rectangular and dimple shaped baffles are presented in terms of velocity of fluid and Nusselt number.

The following figures show the temperature contour for different Reynolds number in the presence of dimple shaped baffles.

**Fig. 5(a).** Temperature contour in the presence of dimple baffle for Re=150 for AR=3

**Fig. 5(b).** Temperature contour in the presence of dimple baffle for Re=250 for AR=3

**Fig. 5(c).** Temperature contour in the presence of dimple baffle for Re=300 for AR=3

From figure 5 it is inferred that the average surface temperature increases with the increase of Reynolds number. When the Reynolds number increase the thermal boundary layer thickness decreases in the wall region.

**Fig. 6(a).** Temperature contour in the presence of rectangular baffle for Re=100 for AR=3

**Fig. 6(b).** Temperature contour in the presence of rectangular baffle for Re=200 for AR=3

**Fig. 6(c).** Temperature contour in the presence of rectangular baffle for Re=300 for AR=3
The figure 6 shows the temperature contour for different Reynolds number in the presence of rectangular baffles. From figure 5 & 6, it is concluded that the vortex induced by the baffle increases mixing and thus enhances the heat transfer. The figure 5 & 6 shows that there was a reduction in thickness of shear layer with the increase of Reynolds number.

The influence of baffle on the heat transfer can be determined by comparing the heat transfer characteristics downstream of the baffle with the presence of the baffle and without the presence of baffle. It is noticed that peak horizontal velocity in the presence of baffle is higher than peak horizontal velocity in the absence of baffle.

The followings table shows the peak Nusselt number with the presence of rectangular baffle & without baffle.

| Re Number | Peak Nu without baffle | Peak Nu with baffle |
|-----------|------------------------|---------------------|
| 100       | 9.881                  | 11.4216             |
| 200       | 14.856                 | 16.2422             |
| 300       | 20.4421                | 21.245              |

Figure 7(a) & 7(b) Local Nusselt number in the downstream location in the presence of rectangular baffle. For Reynolds number 100 nearly 15.59% improvement in Nusselt number and similiar for 9.33% and 3.92% increment in for Re=200 & Re=300 respectively.

The figure 7(a) and 7(b) shows the local nusselt number in the downstream location in the presence of rectangular baffle there are three peak nusselt number are identified in the downstream location. This will improve the average nusselty number in the donstream location. Primary peak occurs at the reattachment of fluid in the computational domain. Seconday peak occurs due to the baffle and tertiary peak occurs due to high inertia force which cause flow separation.

| Reynolds Number | Peak Nusselt number Rectangular | Dimple |
|-----------------|---------------------------------|--------|
| 100             | 7.02                            | 8.16   |
| 150             | 10.52                           | 12.95  |
| 200             | 13.46                           | 14.02  |
| 250             | 19.57                           | 21.25  |
| 300             | 23.70                           | 23.58  |

The table 4 shows that the presence of dimple shaped baffle produces high heat transfer compared to rectangular
baffle. From the table 4 it is confirmed that the percentage of increase in nusselt number decreases with the increase of Reynolds number.

### 5 Conclusion

In the present numerical analysis of heat transfer, while the jet impinges on a block, is investigated using commercial CFD codes. The computed thermal fields in terms of the Nusselt number ranging from 2 to 5 with the Reynolds number ranging from 100 to 300 are analyzed in which the flow comes under the laminar regime.

Due to the presence of baffles, there was increase in nusselt number. The nusselt number profile along the downstream location is plotted for various Reynolds number.

The Nusselt number attains first peak value nearer to the reattachment length and second peak occurs nearer to the baffle and third peak occurs due to the higher Reynolds number.

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