Numerical Analysis on the Performance of a Radiant Cooling Panel with Serpentine-Based Design

Mohammad Hakim Mohd Radzai 1, Chong Tak Yaw 2, Chin Wai Lim 1,* 1, Siaw Paw Koh 2 and Nur Amirani Ahmad 1

1 Department of Mechanical Engineering, University Tenaga Nasional, Jalan IKRAM-UNITEN, Kajang 43000, Selangor, Malaysia; hakimradzai@gmail.com (M.H.M.R.); meya_ahmad14@yahoo.com (N.A.A.)
2 Institute of Sustainable Energy, University Tenaga Nasional, Jalan IKRAM-UNITEN, Kajang 43000, Selangor, Malaysia; chongty@uniten.edu.my (C.T.Y.); johnnykoh@uniten.edu.my (S.P.K.)

Abstract: Radiant cooling systems (RCS) are gaining acceptance as a heating, ventilation, and air conditioning (HVAC) solution for achieving adequate thermal comfort and maintaining acceptable indoor air quality inside buildings. RCS are well known for their energy-saving potential; however, serious condensation problem hinders the growth of this technology. In order to prevent the risk of condensation, the supply water temperature is kept higher than the dew point temperature of the air inside the room. The full potential of the cooling power of a radiant cooling panel is limited. Therefore, this article is on maximizing the cooling capacity of a radiant cooling panel, in terms of flow configuration. Radiant cooling panels (RCP) with different chilled water pipe configurations are designed and compared, side by side with the conventional serpentine flow configuration. The cooling performance of the radiant cooling panels is evaluated by using computational fluid dynamics (CFD) with Ansys Fluent software (Ansys 2020 R2, PA, USA). Under similar flow and operating conditions, the common serpentine flow configuration exhibits the least effective cooling performance, with the highest pressure drop across the pipe. It is concluded that the proposed designs have the potential of improving the overall efficiency of RCP in terms of temperature distribution, cooling capacity, and pressure drop.

Keywords: ansys fluent; cooling capacity; computational fluid dynamics; radiant cooling system; serpentine flow

1. Introduction

The radiant cooling panel (RCP) system has become one of the solutions to improving indoor thermal comfort, due to its low energy consumption compared to all-air systems [1]. RCP provides extensive flexibility in terms of installation and zoning compared to other radiant system types, such as thermally active building systems (TABS) and capillary tube systems, which have tubes embedded in the concrete layer of the building structure [2]. RCP can easily be placed on the ceiling of the building, as well as on the wall column inside a room. Because of the ease of installation, RCP can be used in both new and retrofitted buildings [2].

However, the radiant cooling (RC) system has a condensation problem that hinders the widespread growth of this new technology [3]. The high risk of surface condensation limits the usage of radiant cooling in hot and humid areas [2]. In order to prevent the risk of surface condensation, the supply water temperature should be higher than the dew point temperature of the air inside the room [1]. However, by increasing the supply water temperature, the cooling power of the radiant cooling system will not perform to its full potential [1]. Therefore, research into RCP flow configurations is vital to achieving the maximum cooling potential of the system. S. Oubenmoh et al. [4] conducted a numerical
investigation on the effect of pipe arrangement on RCP performance. The design considerations chosen for the investigation are counterflow, serpentine, and modulated spiral. It is noticeable that the tubing’s geometry influences the average temperature of the core surface where each of the pipe arrangements produces a different maximum core surface temperature. In terms of pressure losses, the modulated spiral configuration exhibits the lowest pressure losses across the pipe, with better thermal homogeneity. The energy consumption of the pump can further be reduced by reducing the inlet water velocities, but more considerations about the design and configuration should be taken into account.

M. Mosa et al. [5] studied two design groups of RCP systems, these being serpentine flow configurations and canopy-to-canopy-flow configurations. The simulation carried out was intended to find the influence of the panel’s aspect ratio on the surface temperature distribution of the RCP. The aspect ratio is calculated as the width over the length of the RCP. A serpentine flow configuration with a higher aspect ratio exhibits greater pressure drop, due to the increasing number of bends over the serpentine tube’s total length. More pumping power is required on a serpentine tube that has higher aspect ratios. However, for the canopy-to-canopy flow configurations, the aspect ratio of the RCP does not influence the cooling performance of the RCP. M. Mosa et al. [5] claim that the compactness in the RCP flow configuration can increase the ratio of cooling capacity per pumping power.

A multi-segmented mini-channel radiant ceiling cooling panel (MCRCP) is a new RCP design, as proposed by Lan Ding et al. [6]. Five simulation scenarios are conducted to analyze the relationship between the MCRCP panel area and its cooling performance. MCRCP panels with aspect ratios of 1.5, 2, 2.5, 3 and 4 are chosen for analysis. Based on the findings of Lan Ding et al. [6], the cooling capacity of the MCRCP increases when the aspect ratio of the panel increases from 1 to 3. Further increasing the aspect ratio of the panel will result in decreasing the cooling capacity; the distance between the radiant panels becomes narrower, hindering the natural convection flow of the inside air [6].

Although the research on RCP flow configurations has made some progress, studies on the dynamic changes of the RCP flow configurations on the cooling performance of the RCP are still limited. There is still a wide range of flow configurations that should be explored and implemented in the RCP system. A more uniform temperature distribution with minimum pressure drop will be likely to have better cooling capacity and efficiency. Therefore, it is expected that different flow configurations will demonstrate different performances in RCP systems.

In this work, a study on the implementation of a different flow configuration on the RCP, other than the conventional serpentine flow configuration, is conducted. The novelty of this study is to evaluate the cooling performance of the conventional serpentine flow configuration, and different designs are compared under similar flow and thermal conditions. Innovative flow patterns for the serpentine-based designs are created by using the constructible approach, where the flow passage is varied in the RCP design, mainly by increasing the flow passages for the chilled water to pass through the RCP. The objective of this research is to analyze the cooling capacity, temperature distribution, and pressure drop of the RCP for each of the designs. The results from this research could provide technical information for the development of new RCP serpentine-based flow configurations.

The paper is categorized as follows. In Section 2, proposed methods are presented. In Section 3, the results and discussions of the proposed RCP are discussed. Section 4 presents the authors’ findings and their conclusions.

2. Methods

This research focuses on the numerical simulation of three types of pipe arrangement configurations of the RCP, by varying the number of inlets and outlets of the serpentine flow configurations.
2.1. Computational Fluid Dynamics (CFD)

This research will be conducted using computational fluid dynamics (CFD) software, known as Ansys Fluent [7,8]. The Ansys Fluent software is used to simulate the fluid flow and investigate the heat transfer characteristics for different pipe arrangement configurations of the RCP. The Ansys Fluent software uses the Navier-Stokes equation in the finite volume method (FVM) to analyze the results of the simulation [9,10].

The steady-state continuity equation, momentum conservation equation, and energy conservation equation are solved by Ansys Fluent with the basic governing equations described below [11,12]:

\[
\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0
\]  

(1)

The continuity equation is shown in Equation (1), where \( u, v \) and \( w \) are the velocity component in \( x, y \) and \( z \) directions, respectively.

\[
\rho \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right) = \rho \frac{\partial u}{\partial x} + \mu \left( \frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} + \frac{\partial^2 u}{\partial z^2} \right)
\]

(2)

\[
\rho \left( \frac{\partial v}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right) = \rho \frac{\partial v}{\partial y} + \mu \left( \frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} + \frac{\partial^2 v}{\partial z^2} \right)
\]

(3)

\[
\rho \left( \frac{\partial w}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right) = \rho \frac{\partial w}{\partial z} + \mu \left( \frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} + \frac{\partial^2 w}{\partial z^2} \right)
\]

(4)

The momentum conservation equation in three dimensions is shown in Equations (2)–(4).

\[
\partial \left( \frac{\rho u T}{\partial x_i} \right) = \frac{\partial}{\partial x_i} \left( k \frac{\partial T}{\partial x_i} \right) + S_T
\]

(5)

The energy conservation equation is shown in Equation (5), where \( T \) is the fluid temperature, \( u_i \) is the velocity in different directions in Cartesian coordinates, \( x_i \) is the Cartesian coordinates, \( k \) is the thermal conductivity, \( C_p \) is the constant-pressure specific heat of fluid, and \( S_T \) is the radiative energy source term [11,13].

The Navier-Stokes equations are then transferred into discretized form for numerical solution. By using the FVM, the general form of the Navier–Stokes equation is integrated over a control volume, and the Gauss divergence theorem is applied. Equation (6) is used as a starting point for computational procedures in the FVM [12]. \( \phi \) is introduced as a general variable for the conservative form of all fluid flow equations [12]:

\[
\int_{cv} \frac{\partial (\rho \phi)}{\partial t} dV + \int_{cv} \text{div}(\rho \phi u) dV = \int_{cv} \text{div}(\Gamma \text{grad} \phi) dV + \int_{cv} S_{\phi} dV
\]

(6)

where \( \frac{\partial (\rho \phi)}{\partial t} \) is the rate of increase of the \( \phi \) fluid element, \( \text{div}(\rho \phi u) \) is the net rate of flow of \( \phi \) out of the fluid element, \( \text{div}(\Gamma \text{grad} \phi) \) is the rate of increase of \( \phi \) due to diffusion, and \( S_{\phi} \) is the rate of increase of \( \phi \) due to the source terms.

The computational domain is divided into a finite number of close-proximity control volumes as discrete elements. They are located in the centroid of the control volume, and are called variables of interest [14]. FVM uses interpolation to integrate the differential forms of the governing equations, to create discretized equations [14]. The general discretized equation for interior nodes in three dimensions is shown in Equation (7) [12]:

\[
a_{\rho} \phi_p = \sum a_{nb} \phi_{nb} + S_u
\]

(7)

where \( \Sigma \) is the diffusion summation of all neighboring nodes \( nb \) and \( a_{nb} \) is for the neighboring coefficients. \( \phi_{nb} \) and \( S_u \) are the values of \( \phi \) at the neighboring nodes and linearized source term, respectively. The equation is mainly used in mesh grid sizing and grid spacing and will drastically affect the quality of the mesh. High-quality mesh can be produced by
using Ansys Fluent, as the meshing runs the Navier–Stokes equation in a form of Fourier series. By using Ansys Fluent for this research, FVM is used for the generation of mesh, to find the cooling characteristics of the RCP.

2.2. Pipe Arrangement Configurations

Figure 1a–c shows the proposed designs of the pipe arrangement configuration that will be used in this research. Figure 1a is the common serpentine flow configuration and is widely used with RCP [1,15–22]. Figure 1a will serve as a benchmark for the other two proposed designs. The design of Figure 1b is known as a counterflow serpentine, with two-inlet-two-outlet flow configurations, while Figure 1c is a serpentine design with three-inlet-three-outlet flow configurations. The relevancy of the number of inlet and outlet flows for an RCP is determined by studying the three proposed designs.

![Flow configurations](image)

**Figure 1.** Flow configurations: (a) one-inlet-one-outlet serpentine, (b) counterflow two-inlet-two-outlet serpentine, (c) three-inlet-three-outlet serpentine configurations. Note: inlet (blue arrow) and outlet (green arrow).

2.3. Mathematical Model

Based on the sketch in Figure 1, chilled water flows from the inlet of the copper tube and goes through the channels of the tube. The top and sides of the radiant panel are insulated, and heat flux is applied from beneath the radiant panel. Therefore, heat exchange occurs between the aluminum plate and the chilled water flowing through the copper tube. The fluid inlet temperature and ambient temperature are kept constant throughout the simulation. The plate area and flow volume are defined as Equations (8) and (9).

\[
A_p = W \times L \tag{8}
\]

\[
V_{flow} = \frac{\pi}{4} D^2 L_{tube} \tag{9}
\]

where \(W\) and \(L\) is the width (m) and length of the plate (m), respectively. \(D\) is the tube diameter (m) and \(L_{tube}\) is the flow path length (m). The governing equations for the model are shown in Equations (1)–(5).

The total heat flux received by the panel is expressed by Equation (10). The total heat flux is the summation of radiation and convection heat transfer between the surrounding temperature and the panel’s surface.

\[
q_{rc}'' = h_r (AUST - T_p) + h_c (T_{amb} - T_p) \tag{10}
\]

where \(h_r\) is the radiation heat transfer coefficient, \(AUST\) is the area-weighted average temperature of all the uncontrolled surfaces in the conditioned space, \(T_p\) is the effective radiant panel surface temperature and \(h_c\) is the convection heat transfer coefficient. The
equation for radiation and convection heat transfer coefficient is expressed as shown in Equations (11) and (12), respectively [23]:

\[
h_r = \frac{\sigma \sum_{j=1}^{n} F_{\varepsilon_{s,j}} (T_{s} - T_{p})}{(AUST - T_{p})}
\]  

(11)

where \( \sigma \) is the Stephan–Boltzmann constant, and \( F_{\varepsilon_{s,j}} \) is the radiation interchange factor between the panel surface temperature and \( j \)th surface temperature, \( T_{j} \).

\[
h_c = \frac{q''_{c}}{(T_{amb} - T_{p})}
\]  

(12)

where \( T_{amb} \) is the ambient air temperature.

However, based on data from the American Society of Heating, Refrigerating and Air-Conditioning Engineers (ASHRAE), the radiation and convective heat fluxes for a suspended RCP may also be estimated by using Equations (13) and (14) [24]:

\[
q''_{r} = 5 \times 10^{-8} \left( (AUST + 273)^4 - (T_{p} + 273)^4 \right)
\]  

(13)

\[
q''_{c} = 2.31 (T_{amb} - T_{p})^{0.31} (T_{amb} - T_{p})
\]  

(14)

For the calculation of heat flux, an initial value of plate temperature, \( T_{p} \), is needed to be assumed and adjusted accordingly throughout the simulation. \( AUST \) is assumed to be the same as \( T_{amb} \) for ease of calculation, normally when the cases and design adhere to ASHRAE 1999a [16,25]. Table 1 below shows the heat transfer characteristics, based on the mean panel temperature determined by Mumma et al. [25].

| Mean Panel Temperature (°C) | Room Temperature/ AUST (°C) | \( q_{radiation} \) \( (\text{W/m}^2) \) | \( q_{convection} \) \( (\text{W/m}^2) \) | \( q_{total} \) \( (\text{W/m}^2) \) | U, Overall Heat Transfer Coeff. \( (\text{W/m}^2) \) |
|-----------------------------|-------------------------------|----------------------------------------|---------------------------------|-------------------------------|----------------------------------|
| 10                          | 22                            | 60                                     | 57                              | 117                           | 9.43                             |
| 13                          | 22                            | 47                                     | 41                              | 88                            | 9.14                             |
| 16                          | 22                            | 35                                     | 25                              | 60                            | 8.75                             |
| 18                          | 22                            | 19                                     | 13                              | 32                            | 8.23                             |
| 10                          | 26                            | 76                                     | 76                              | 151                           | 9.82                             |
| 13                          | 26                            | 63                                     | 60                              | 123                           | 9.65                             |
| 16                          | 26                            | 51                                     | 44                              | 95                            | 9.37                             |
| 18                          | 26                            | 38                                     | 28                              | 66                            | 9.03                             |

### 2.4. Material Properties

The thermophysical properties of the elements found in the RCP are given in Table 2 [26].

| Materials         | Density \( (\text{kg/m}^3) \) | Specific Heat \( (\text{J/kg K}) \) | Thermal Conductivity \( (\text{W/m-K}) \) |
|-------------------|--------------------------------|-------------------------------------|------------------------------------------|
| Water             | 998.2                          | 4182                                | 0.6                                      |
| Copper            | 8978                           | 381                                 | 387.6                                    |
| Aluminum          | 2719                           | 871                                 | 202.4                                    |

### 2.5. Simulation Domain and Boundary Conditions

The boundary conditions are taken from real conditions selected from other research papers. The boundary conditions and domains are obtained from [1,5] and are summarized as shown in Table 3.
Table 3. Summary for boundary conditions and domains.

| Domain          | Boundary                      | Boundary Properties                                      |
|-----------------|-------------------------------|----------------------------------------------------------|
| Tube Walls      | Stationary wall               | • No-slip                                                 |
|                 |                               | • Standard Roughness                                     |
|                 |                               | • Thermal Conditions: Thermally Coupled                   |
|                 |                               | • Heat Generation Rate = 0 W/m³                           |
| Water Inlet     |                               | • Velocity Specification Method: Magnitude, Normal to Boundary |
|                 |                               | • Reference Frame: Absolute                               |
|                 |                               | • Mass Flow Rate = 0.004 kg/s                             |
|                 |                               | • Velocity Magnitude = 0.051 m/s                          |
|                 |                               | • Initial Gauge Pressure = 0 Pa                           |
|                 |                               | • Temperature = 288 K                                    |
| Water Outlet    |                               | • Reference Frame: Absolute                               |
|                 |                               | • Gauge Pressure = 0 Pa                                   |
|                 |                               | • Pressure Profile Multiplier = 1                         |
|                 |                               | • Direction: Normal to Boundary                           |
|                 |                               | • Pressure Specification: Total Pressure                  |
| Top and Sidewalls| Stationary wall               | • No-slip                                                 |
|                 |                               | • Standard Roughness                                     |
|                 |                               | • Thermal Conditions: Heat Flux                           |
|                 |                               | • Heat flux = 0 W/m²                                      |
| Plate Bottom Wall | Stationary wall              | • No-slip                                                 |
|                 |                               | • Standard Roughness                                     |
|                 |                               | • Thermal Conditions: Heat Flux                           |
|                 |                               | • Heat flux = q”rc                                        |

The boundary conditions are as follows:

(a) The mass flow rate at the pipe inlet is 0.004 kg/s, with a temperature of 15 °C.
(b) The top and sides of the radiant panel are insulated.
(c) Heat flux is applied from underneath the radiant panel as it is exposed to room temperature.
(d) The ambient air temperature is assumed to be 24 °C.

There are also some assumptions made to simplify the simulation model. The assumptions are as follows:

(a) The flow inside the copper tube is laminar and incompressible.
(b) The pipe wall thickness is proposed to be a thin layer, as the thermal conductivity of copper is high, and the wall thickness is very small. Therefore, the wall thickness is ignored.
(c) The aluminum panel has perfect contact with the copper tube.

2.6. Operating Conditions and Design Parameters

Based on M. Mosa et al. [5], a radiant panel with a 1.05 aspect ratio gives the highest cooling capacity for the serpentine flow configuration. Therefore, this research uses the given aspect ratio of the radiant panel as a reference. Note that the aspect ratio is the width to length ratio of the RCP.

The values for the aspect ratio ($A_r$), panel area ($A_p$), width ($W$), length ($L$), panel thickness ($t$), inner diameter ($D_o$), and outer diameter ($D_i$) of the copper tube are summa-
rized in Table 4. The values for the operating condition for water inlet temperature ($T_{w,in}$), ambient temperature ($T_{amb}$), water inlet mass flow rate ($\dot{m}_{in}$), and area-weighted average temperature ($AUST$) are also shown in Table 4.

Table 4. Summary of parameters and operating conditions for the simulation.

| Design Parameters | Value               | Operating Conditions             |
|-------------------|---------------------|---------------------------------|
| Aluminum Panel    | $A_r = 1.05$        | Laminar flow                    |
|                   | $A_p = 1.35 \text{ m}^2$ | Incompressible                  |
|                   | $W = 1.19 \text{ m}$ | $T_{w,in} = 15 \text{ °C}$      |
|                   | $L = 1.13 \text{ m}$ | $T_{amb} = 24 \text{ °C}$       |
|                   | $t = 3 \text{ mm}$  | $\dot{m}_{in} = 0.004 \text{ kg/s}$ |
| Copper Tube       | $D_o = 12.5 \text{ mm}$ | $AUST = 24 \text{ °C}$         |
|                   | $D_i = 10 \text{ mm}$|                                 |

2.7. Geometrical Setup of the Proposed Design

The 3D design of the RCP model is created by using Autodesk Fusion 360. This software is used because it makes it easier to construct 3D modeling. The completed 3D design of the RCP is then converted into a .step file. The .step file of the design is then imported into SpaceClaim for flow simulation. The dimension of this model follows the parameters proposed in Table 4. The generated model of a one-inlet-one-outlet serpentine flow configuration, as modeled by using Autodesk Fusion 360 in SpaceClaim, is shown in Figure 2.

![Figure 2. 3D model of the one-inlet-one-outlet serpentine flow configuration.](image)

After generating a 3D model of the RCP, a hexahedral mesh is used for the generation of mesh on the design model in Figure 3. Hexahedral mesh is chosen because it enables a more accurate result on a simple geometry model. Edge sizing is used at the inlet and outlet flow of the fluid domain. The mesh is smoothed by applying 10, 15, 20 and 25 divisions on the mentioned edge for the grid independence test, to ensure that the grid size used is suitable to obtain a converged solution.
The simulation domains and boundary locations are specified after the mesh generation of the model is completed. The fluid domain is the pipe carrying chilled water, and the solid domain is the aluminum plate located below the fluid domain. Heat exchange will occur between the two domains, with the boundary conditions mentioned in Table 3. The fluid domain is set to have the properties of water, and the solid domain has the properties of aluminum.

For the solver execution part, a laminar model is chosen because the calculated Reynolds number is in the laminar region. The energy equation is turned on as heat exchanges occur between the fluid domain and the solid domain. The solver type is pressure-based and steady-state. A steady-state simulation is faster than a transient one and is thus suitable for this study, to save time and computational load.

For the solution part, a Semi-Implicit Method for Pressure Linked Equations (SIMPLE) is used for the pressure-velocity coupling and is set as the solution method. A second-order upwind is chosen for the spatial discretization of the momentum and energy equations, whereas the gradient discretization is based on the least-squares cell, and pressure discretization is set as second-order. The residual tolerance used for continuity, \(x\)-velocity, \(y\)-velocity, \(z\)-velocity, and energy is 0.001, which is the default value from ANSYS Fluent. Hybrid initialization is used, and 1000 iterations are chosen to make sure that the value obtained is accurate and converged.

Finally, the results obtained in the post-processing stage are analyzed and examined for discussion. The results that will be taken for analysis are the temperature contour of the RCP, pressure contour, outlet water temperature, surface temperature of RCP, and pressure drop inside the tube. The cooling capacity and temperature distribution uniformity are determined from the surface temperature of the RCP.

The steps above will be repeated on the other two proposed designs of the RCP flow configurations.

2.8. Validation

In order to validate the CFD simulation, the convergence of the simulation result needs to be obtained. To further validate the CFD simulation, the results obtained are compared with other literatures.

Table 5 shows the calculated results and experimental results of the RCP model obtained under constant indoor environmental factors. These data are taken from W. Jin et al. [27] and Zhang et al. [28], and are compared with the conventional serpentine flow configuration results. Based on the comparisons, the outlet water temperature and the surface panel
temperature of the RCP differ slightly from the simulated conventional serpentine. The percentage errors of the outlet water temperature and surface panel temperature for comparison 1 are 4.30% and 5.33%, respectively, while the percentage errors for comparison 2 are 1.07% and 4.44%, respectively. The error caused is mainly because of the different inlet water temperatures used for the simulation. Nevertheless, the error obtained for both of the comparisons is within 5%, which is acceptable for engineering applications. Therefore, the experimental results from [27,28], as well as the temperature contour obtained, verified the simulation result of the conventional serpentine flow configuration.

Table 5. Comparison with experimental results from other literatures.

|                | Comparison 1 | Percentage Error (%) | Comparison 2 | Percentage Error (%) |
|----------------|--------------|----------------------|--------------|----------------------|
| Inlet          | 15.7 °C      | 15.0 °C              | 14.6 °C      | 15.0 °C              |
| Outlet         | 17.8 °C      | 18.6 °C              | 18.4 °C      | 18.6 °C              |
| Surface        | 21.3 °C      | 22.5 °C              | 21.5 °C      | 22.5 °C              |

3. Results and Discussions

3.1. Grid Independence Test

A grid independence test was carried out on Design 1 (see Figure 1a) to determine the best number of elements required for the simulation. Table 6 shows the grid independence test result, where the number of elements is the variable, and the pressure drop and outlet temperature are taken for analysis.

Table 6. Summary of parameters and operating conditions for the simulation.

| Number of Elements | Outlet Temperature (K) | Pressure Drop (Pascal) |
|--------------------|-------------------------|------------------------|
| 275,090            | 291.9821                | 129.5508               |
| 283,680            | 291.7517                | 129.4224               |
| 313,492            | 291.6585                | 129.3006               |
| 383,305            | 291.6650                | 129.6958               |

In the grid independence test analysis, the operating conditions and parameters were kept constant, while the number of meshing elements was varied. Based on the graph presented in Figures 4 and 5, the outlet temperature became constant as the number of elements increased, while the pressure drop did not seem to be affected greatly by the number of elements, since the result was showing a consistent value with a 0.0337 variance. Therefore, the total number of elements used in the simulation was chosen to be 313,492, since the outlet temperature began to converge into a steady value while there was negligible variation in the pressure drop at this number of elements. Choosing a larger number of elements would only increase the computing time, without any significant impact on the simulation.
3.2. CFD Simulation—ANSYS Fluent

The results obtained from the simulations conducted for all three proposed flow configurations (Figure 1a–c) of the RCP are shown in this section. All the designs simulated have achieved a converged solution, where the solution is unique and, therefore, can be used for further analysis.

3.2.1. Residual Plot

Figures 6–8, below, show the residual plot for all three flow configurations. Here, they are denoted as Design 1, Design 2 and Design 3 for Figure 1a–c, respectively. A residual plot is important in determining the reliability of the simulation result. In the simulations conducted, all the flow configurations have achieved convergence and, therefore, the results of the simulation can be considered accurate, with low residual error.
**Figure 6.** Converged residual plot for Design 1.

**Figure 7.** Converged residual plot for Design 2.

**Figure 8.** Converged residual plot for Design 3.
3.2.2. Temperature Contour and Pressure Contour

Temperature contour and pressure contour for all three flow configurations, obtained from the simulations, are shown in Figures 9 and 10.

![Temperature Contour](image1)

**Figure 9.** Temperature contours for the three different flow configurations.

![Pressure Contour](image2)

**Figure 10.** Pressure contours for the three different flow configurations.

3.3. Analysis and Discussions

The findings obtained from the simulation results are further analyzed in this section. This includes the analysis of the temperature distribution of the RCP with different serpentine flow configurations. In addition, the pressure drop inside the tube is identified based on the simulation. Moreover, important parameters, such as outlet water temperatures, minimum and maximum plate temperature, as well as the average plate temperature, are obtained and analyzed for each of the serpentine flow configurations.
3.3.1. Maximum Outlet Water Temperature and Total Heat Transfer Heat

Based on the results obtained from the simulations, all the maximum temperatures and total heat transfer rate for each serpentine flow configuration are tabulated in Table 6 below, with the water inlet temperature, \( T_{\text{inlet}} = 288 \, \text{K} \).

Based on Table 7, Design 2, with the two-inlet-two-outlet serpentine flow configuration, has the highest maximum rate of heat transfer with the maximum temperature of 289.716 K at outlet 1, and 289.727 K at outlet 2. The temperature difference at outlet 1 is 1.716 K and 1.727 K at outlet 2, with a total heat transfer of 28.111 W. Based on the simulation results, heat is absorbed equally between the two pipes in Design 2, resulting in a lower temperature difference with a higher total heat transfer rate. In the case of Design 3, the middle pipe, which is outlet 2, has only 1130 mm of pipe length, resulting in a lower performance of heat absorption from the panel surface.

### Table 7. Maximum outlet water temperature and total heat transfer rate.

| Design | Maximum Outlet Water Temperature, (K) | Temperature Difference \((T_{\text{inlet}} - T_{\text{outlet}})\), (K) | Total Heat Transfer Rate (W) |
|--------|--------------------------------------|-----------------------------------------------|-----------------------------|
| 1 Outlet 1 | 291.659 | 3.659 | 23.122 |
| 2 Outlet 1 | 289.716 | 1.716 | 28.111 Outlet 2 | 289.727 | 1.727 |
| 3 Outlet 1 | 289.476 | 1.476 | 26.247 Outlet 2 | 288.822 | 0.822 Outlet 3 | 289.417 | 1.417 |

3.3.2. Temperature Distribution

Table 8 shows the minimum and maximum temperatures of the panel’s surface, with its mean panel temperature as calculated by Ansys Fluent.

### Table 8. Radiant cooling panel surface temperature.

| Design | Minimum Panel Temperature (K) | Maximum Panel Temperature (K) | Mean Panel Temperature (K) |
|--------|-------------------------------|-------------------------------|----------------------------|
| 1      | 293.441                       | 298.919                       | 295.533                    |
| 2      | 291.206                       | 295.711                       | 293.368                    |
| 3      | 291.626                       | 296.801                       | 293.943                    |

Based on Table 8, Design 2 has the lowest mean panel temperature, indicating that the temperature distribution in Design 2 has the most uniformity. The common serpentine flow configuration in Design 1 has the highest mean panel temperature, with a 298.919 K maximum panel temperature and a 293.441 K minimum panel temperature. Based on the panel’s surface temperature, shown in Table 8, Design 1 is not as good as the other two designs in respect of obtaining uniform temperature distribution. This is because the water flowing inside the tube becomes hotter with the increasing length, thus decreasing the heat absorption performance, resulting in a higher panel temperature at the area around the water outlet. This phenomenon can be seen from the temperature contour shown in Figure 9, where the edge of the panel surface located at the water outlet has a higher temperature compared to the panel surface located near the water inlet. The concentration of the cold region can be seen clearly at the water inlet, with temperature increases along the pipe length. This temperature contour agrees with Mohamed Mosa et al. [1] showing a similar temperature distribution characteristic.

In the case of Design 2, the cold region is concentrated in the middle part of the RCP. This is because the water inlets are located next to each other, thus creating a very low minimum panel surface temperature at that region. In addition, the temperature contour for Design 2 shows a uniform temperature distribution throughout the panel’s surface.
compared to those of Design 1 and Design 3. There is a smaller greenish region observed in the temperature contours of both Design 1 and Design 3. What is more, the greenish region that represents a relatively low-temperature zone is not evenly spread throughout the panel’s surface in both Design 1 and Design 3.

The middle pipe has no significant impact on the cooling performance of the RCP in Design 3, and the cold region is only concentrated at the right side of the panel’s surface where the water inlet is located. The left side of the panel’s surface, where the water outlets are placed near each other, shows an increase in temperature due to the increase of water temperature being concentrated in that region.

For all three designs, the tubes did not cover the edges of the plate, resulting in a hotter region compared to the rest of the surface [1]. This is because the chilled water has less access to the surface of those regions [1]. To gain more understanding of the temperature distribution of the panel’s surface, a temperature distribution graph is plotted by dividing the plate into three equally spaced segments. The plate is divided into the right-hand side, middle, and left-hand side of the plate in the graphical analysis shown in Figures 11–13.

Figure 11. Temperature distribution of the cooling panel for Design 1.

Figure 12. Temperature distribution of the cooling panel for Design 2.
Figure 13. Temperature distribution of the cooling panel for Design 3.

By comparing the three temperature distribution graphs, it can be seen that Design 1 has the least temperature distribution uniformity where the plotted temperature for each segment is not concentrated in the same location, unlike in Design 2. Although the minimum, maximum, and mean plate temperatures in Design 3 do not differ that much from those of Design 2, the temperature distribution graph shows that the temperature is not distributed uniformly throughout the plate.

In terms of calculation, the temperature difference of Design 2 is the lowest compared to those of Design 1 and Design 3 at 4.505 °C, whereas Design 1 and Design 3 have a temperature difference of 5.478 °C and 5.175 °C, respectively. A smaller temperature difference indicates that the temperature distribution of a particular radiant cooling panel has a higher uniformity because its temperature spread is limited to that specific range. From this finding, Design 2 shows the most promising result for temperature difference calculation, with the most concentrated temperature distribution graph plotted. Therefore, Design 2, with a two-inlet-two-outlet counterflow serpentine configuration, has the highest uniformity in terms of temperature distribution.

3.3.3. Cooling Capacity

The cooling capacity of the RCP for each of the flow configurations was analyzed and calculated by using Equations (13) and (14), as given in Section 2.3. The cooling capacity formula was taken from ASHRAE Transactions [29], where the apparent heat is removed from the room by convection and radiation, which is absorbed by the RCP. Therefore, the cooling capacity, the $Q_{\text{cool}}$ of the RCP can be determined by calculating the total heat flux absorbed through radiation and convection.

$$Q_{\text{cool}} = q''_r + q''_c$$

(15)

Based on Table 9, Design 2 had the highest cooling capacity compared to the other two designs. Design 1, which is the common serpentine flow configuration, had the lowest cooling capacity, which was 19.752 W/m² smaller than that of Design 2. Design 2 is created from a heat exchanger perspective, where counterflow is more efficient than a parallel flow. By implementing this idea into the RCP where the inlet and outlet are located on the same side of the plate, a significant improvement of its cooling performance was achieved. The cooling performance of the RCP agrees with Mohamed Mosa et al. [1], where the counterflow design group is better than the conventional flow design group. Therefore, the arrangement of pipe configuration can have a significant impact on its cooling performance.
In addition, the heat transfer characteristics obtained also agree with Miriel et al. [30], where the heat absorption consists of 2/3 radiation and 1/3 convection of the total heat transfer.

Table 9. Radiant cooling panel cooling capacity:

| Design | Room Temperature/ AUST (K) | Mean Plate Temperature, \( T_p \) (K) | \( q_r^r \) radiation (W/m\(^2\)) | \( q_c^r \) convection (W/m\(^2\)) | Cooling Capacity, \( Q_{cool} \) (W/m\(^2\)) |
|--------|-----------------------------|----------------------------------------|-----------------------------------|-----------------------------|----------------------------------------|
| 1      | 297                         | 295.533                                | 7.630                             | 3.816                       | 11.446                                |
| 2      | 297                         | 293.368                                | 18.684                            | 12.514                      | 31.198                                |
| 3      | 297                         | 293.943                                | 15.772                            | 9.985                       | 25.757                                |

3.3.4. Pressure Drop

Based on the pressure contour shown in Figure 10, the pressure decreases from the upstream to the downstream region for all of the flow configurations. This is due to pressure losses along the flow in the pipe. Pressure drop is closely related to pumping power, meaning that the higher the pressure drop, the larger the pumping power needed to operate the system. Therefore, it is important for us to determine the total pressure drop in order to design a more efficient cooling system for the RCP.

The pressure drop is determined by using the Ansys Fluent post-processing method, and is compared to a manual calculation derived from Bernoulli’s equation for pressure drop [31], as shown in Equation (16):

\[
\frac{P_1}{\rho g} + \frac{v_1^2}{2g} + Z_1 = \frac{P_2}{\rho g} + \frac{v_2^2}{2g} + Z_2 + h_L
\]

where \( P_1 \) is the static pressure at the inlet and \( P_2 \) is the static pressure at the outlet. \( v \) is the fluid velocity, \( Z \) is the elevation, \( h_L \) is the total head loss, \( \rho \) is the fluid density, and \( g \) is the gravitational acceleration. The total head loss, \( h_L \) is determined by using Equation (17):

\[
h_L = \left[ f \frac{L}{D} + \sum k \right] \frac{v^2}{2g}
\]

where \( f \) is the friction factor, \( L \) is the pipe length, \( D \) is the pipe diameter and \( k \) is the loss coefficient value. Because the flow is laminar and the pipe is circular, the formula used to calculate the friction factor is shown in Equation (18) [31]:

\[
f = \frac{64}{Re}
\]

Equation (19) is the simplified Bernoulli’s equation, where the inlet velocity is assumed to be the same as the outlet velocity. The pipe configuration is horizontal, thus the change in potential energy is ignored. The loss coefficient value, \( k \), is determined to be 0.2 because the pipe is circular and has a 180° bend.

Based on Table 10, the theoretical value of pressure drop is almost identical to the simulated value obtained from the simulation. These data prove that the simulated pressure drop can be validated with the governing Bernoulli’s equation, as the results obtained have a very low percentage of error. Comparing the results among all the three designs, Design 1 has the most pressure drop across the pipe. This is because the pipe configuration in Design 1 has a total of six bends in a single, long pipe. Unlike Design 2, the pipe configuration is divided into two pipes, meaning that it has only three bends for each pipe. For Design 3, the middle pipe is a straight pipe without any bends, while the other two pipes have two bends each. Because of this, the middle pipe of Design 3 has the lowest pressure drop. As can be noticed from the pressure contour figure, the pressure drop of the last serpentine is less than the first serpentine, because of the decreasing pressure...
throughout the flow toward the outlet [32,33]. With the increasing number of bends of the pipe, a larger pressure drop across the pipe is observed, as shown in Figure 14. Therefore, the pressure drop across the pipe is highly related to the number of bends and the path length of the fluid.

![Figure 14. Effect of the number of bends on a pipe’s pressure drop.](image)

### Table 10. Simulated and theoretical pressure-drop values.

| Design | Total Loss Coefficient Value, ∑k | Friction Factor, f | Theoretical Pressure Drop (Pascal) | Simulated Pressure Drop (Pascal) | Percentage Error (%) |
|--------|---------------------------------|-------------------|-----------------------------------|---------------------------------|----------------------|
|        | 1                               |                   |                                   |                                 |                      |
| 1      | Pipe 1                          | 1.2               | 0.1429                            | 127.8954                        | 129.4225             | 1.19                |
| 2      | Pipe 1                          | 0.6               | 0.1429                            | 74.3315                         | 74.8937              | 0.76                |
|        | Pipe 2                          | 0.6               | 0.1429                            | 74.3315                         | 75.1780              | 1.14                |
| 3      | Pipe 1                          | 0.4               | 0.1429                            | 56.4650                         | 56.3966              | 0.12                |
|        | Pipe 2                          | 0                 | 0.1429                            | 20.7530                         | 19.8199              | 4.50                |
|        | Pipe 3                          | 0.4               | 0.1429                            | 56.4650                         | 56.7390              | 0.49                |

### 3.4. Analysis and Discussions

The simulation results presented in Section 3.2 show that the cooling capacity of the RCP can be enhanced by manipulating the flow configurations of the RCP. It was found that the proposed flow configurations enhanced the cooling capacity of RCP by 172.57% for Design 2 and 125.03% for Design 3. The percentage increase is compared with the cooling capacity of Design 1, which is the conventional serpentine flow configuration that acts as a benchmark for the two proposed designs. The RCP operates in an enclosed environment, in which the air and all the surfaces are kept at a constant temperature. Chilled water enters the RCP at 15 °C, resulting in a temperature difference between the RCP and the conditioned surrounding. Heat exchanges occur between the RCP and the conditioned surroundings due to the temperature difference. From this natural phenomenon, the heat exchange that occurs is more efficient in Design 2 compared to Design 1 and Design 3, because Design 2 enables a more uniform temperature distribution with the highest cooling capacity.

As shown in Equation (15), the cooling capacity, which is also known as total heat flux, as absorbed by the RCP, is the summation of radiation and convection heat transfer between the surrounding temperature and the panel’s surface. Our results agree with ASHRAE’s definition of radiant systems, where the radiation heat transfer amounts to 60% of the total heat flux absorbed by the RCP. Thus, these results could be applied to develop new RCP flow configurations with maximum cooling potential, because the cooling capacity of RCP is often insufficient in hot and humid climates, due to condensation problems. If proper RCP flow configurations have been discovered, the required cooling capacity can be achieved at a relatively higher chilled water temperature.
Although the proposed flow configurations effectively enhanced the cooling capacity of the RCP, the simulated results are recommended for validation with an experimental investigation. As this research is only for a single panel, appropriate experimentation and testing conditions are necessary to find out the required number of panels that should be integrated and distributed over the ceiling for effective cooling. With a larger number of panels, a higher water flow rate is needed throughout the large radiant surface of the RCP. Therefore, proper experimental research is needed to evaluate the pumping power of these designs to determine their applicability.

4. Conclusions

The aim of this research is to develop CFD numerical simulations in order to test and simulate some basic design aspects of the RCP system. A three-dimensional model of the RCP is used to study the performance of the RCP, in terms of temperature distribution, cooling capacity, and pressure drop across the pipe. The study is conducted on three different pipe configurations, which are one-inlet-one-outlet serpentine, two-inlet-two-outlet counterflow serpentine, and three-inlet-three-outlets serpentine flow configurations. Under the assumed operating conditions, the main findings of this research are as follows:

1. The two-inlet-two-outlet counterflow serpentine flow configuration leads to better temperature uniformity. The best result in terms of low and uniform temperature distribution can be obtained with the counterflow design because it produces the lowest temperature difference of the panel’s surface temperature, with a 4.505 °C temperature difference.

2. The common serpentine flow configuration exhibits the lowest cooling performance, with the highest pressure drop across the pipe. The single pipe of the common serpentine flow configuration has a pressure drop of as much as 129.4225 Pa, which greatly influences the energy efficiency of the pump.

3. Pressure drops can be reduced by reducing the number of bends of the pipe, as the loss coefficient factor increases with every bend. Design 1 has six bends in a single long pipe, resulting in the highest total loss coefficient value, while Design 2 and Design 3 have fewer bends for each pipe. The pressure drop obtained is higher in a pipe that has a higher number of bends.

4. Increasing the number of inlets and outlets of the serpentine flow configuration may not make a significant impact on the cooling performance. However, the arrangement of the pipe configuration, such as the location of the inlets and outlets of the pipe and also the compactness of the pipe arrangement, may influence the cooling performance of the RCP.

It can be concluded that the common serpentine flow configuration does not give the best performance for the RCP. The common serpentine flow configuration exhibits the lowest cooling capacity, with 11.446 W/m². The highest cooling capacity of 31.198 W/m² is obtained from the two-inlet-two-outlet counterflow serpentine flow configuration. Both of the proposed flow configurations outperform the common serpentine flow configuration in terms of cooling capacity, temperature distribution, and pressure drop. Therefore, the cooling performance of RCP can be improved by implementing different flow configurations for the RCP.

Moreover, flow uniformity is a significant factor influencing the efficiency of any thermal system, including an RCP. Applying a different pipe configuration to the RCP will affect the flow uniformity and the temperature distribution uniformity. Therefore, an RCP should be designed with care to provide sufficient indoor cooling with maximum thermal comfort. In order to achieve this, it is recommended that more research be conducted on the design of RCP; there are many other innovative flow configurations that can be implemented in the RCP. The search for the best design of RCP is a never-ending journey as there is no limit to ideas. In the quest for an improved panel architecture, it is important for us to report and document the current results, as they can serve as guidance for future research. Conducting the numerical simulation by using a full version of Ansys Fluent
is suggested, to eliminate any uncertainties from the meshing, so that grid-independent tests can be conducted with a larger number of elements. To further validate the energy efficiency of these three designs, future research plans will include experimental and simulation studies on energy consumption for the pumping power of the three designs.

**Author Contributions:** Conceptualization, C.W.L. and S.P.K.; methodology, M.H.M.R., C.W.L. and N.A.A.; software, M.H.M.R.; validation, M.H.M.R. and C.W.L.; formal analysis, M.H.M.R. and C.W.L.; investigation, M.H.M.R. and C.W.L.; resources, M.H.M.R., C.T.Y. and C.W.L.; data curation, M.H.M.R. and C.W.L.; writing—original draft preparation, M.H.M.R., C.T.Y., C.W.L. and N.A.A.; writing—review and editing, M.H.M.R., C.T.Y., S.P.K. and C.W.L.; visualization, C.W.L., S.P.K. and C.T.Y.; project administration, C.W.L.; funding acquisition, C.W.L. All authors have read and agreed to the published version of the manuscript.

**Funding:** This research was supported by the Fundamental Research Grant Scheme (FRGS) funded by the Ministry of Education, Malaysia (FRGS/1/2018/TK03/UNITEN/02/3).

**Institutional Review Board Statement:** Not applicable.

**Informed Consent Statement:** Not applicable.

**Data Availability Statement:** Not applicable.

**Acknowledgments:** The authors wish to thank the Ministry of Education of Malaysia (MOE) for providing the Fundamental Research Grant Scheme (FRGS) grant research. The authors also acknowledge the contribution of the facilities of Universiti Tenaga Nasional Sdn. Bhd, Malaysia (UNITEN) for providing all out-of-laboratory support.

**Conflicts of Interest:** The authors declare no conflict of interest.

**References**

1. Mosa, M.; Labat, M.; Lorente, S. Constructal design of flow channels for radiant cooling panels. *Int. J. Therm. Sci.* 2019, 145, 106052. [CrossRef]

2. Ning, B.; Chen, Y.; Liu, H.; Zhang, S. Cooling capacity improvement for a radiant ceiling panel with uniform surface temperature distribution. *Build. Environ.* 2016, 102, 64–72. [CrossRef]

3. Imran, M.S.; Baharun, A.; Ibrahim, S.H.; Abidin, W.A.W.Z. Performance of low cost alternative radiant cooling panel in Malaysia. *Int. J. Sustain. Constr. Eng. Technol.* 2017, 8, 29–45.

4. Oubenmoh, S.; Allouhi, A.; Ait Mssad, A.; Saadani, R.; Kouksou, T.; Rahmoune, M.; Bentalb, M. Some particular design considerations for optimum utilization of under floor heating systems. *Case Stud. Therm. Eng.* 2018, 12, 423–432. [CrossRef]

5. Mosa, M.; Labat, M.; Lorente, S. Role of flow architectures on the design of radiant cooling panels, a constructal approach. *Appl. Therm. Eng.* 2019, 150, 1345–1352. [CrossRef]

6. Ding, L.; Katsura, T.; Radwan, A.; Nagano, K. Thermal performance analyses of a new multi-segmented minichannel-based radiant ceiling cooling panel. *Energy Rep.* 2020, 6, 1409–1415. [CrossRef]

7. Harijishnu, R.; Jayakumar, J. Thermal radiation heat transfer in participating media by finite volume discretization using collimated beam incidence. *IOP Conf. Ser. Mater. Sci. Eng.* 2017, 243, 012009. [CrossRef]

8. Komen, E.; Hopman, J.; Frederix, E.; Trias, F.; Verstappen, R. Energy-preserving discretisation for LES/DNS with unstructured collocated grids in OpenFOAM. In Proceedings of the 8th European Congress on Computational Methods in Applied Sciences and Engineering (ECCOMAS 2020), Paris, France, 11–15 January 2021.

9. Laha, D. Computational Aerodynamic Analysis of a Rear Spoiler on a Car in Two Dimensions. Available online: https://www.academia.edu/25504634/Computational_aerodynamic_analysis_of_a_rear_spoiler_on_a_car_in_two_dimensions (accessed on 10 July 2021).

10. Ranade, R.; Hill, C.; Pathak, J. DiscretizationNet: A machine-learning based solver for Navier–Stokes equations using finite volume discretization. *Comput. Methods Appl. Mech. Eng.* 2021, 378, 113722. [CrossRef]

11. Trias, F.X.; Valle, N.; Gorobets, A.; Oliva, A. Symmetry-Preserving Discretization of Navier-Stokes on Unstructured Grids: Collocated vs Staggered. In Proceedings of the 14th WCCM-ECCOMAS Congress 2020, Paris, France, 11–15 January 2021. Available online: http://fxtrias.com/docs/ECCOMAS20_SymPres_Unstruct_STG_paper.pdf (accessed on 10 July 2021).

12. Xie, D.; Wang, Y.; Wang, H.; Mo, S.; Liao, M. Numerical analysis of temperature non-uniformity and cooling capacity for capillary ceiling radiant cooling panel. *Renew. Energy* 2016, 87, 1154–1161. [CrossRef]

13. Versteeg, H.; Malalasekera, W. The finite volume method. In *An Introduction to Computational Fluid Dynamics*; Longman Scientific and Technical: Essex, England, 1995.

14. Dong, J.; Zheng, W.; Song, M.; Chen, Z.; Jiang, Y. Numerical simulation on heating performances of a radiant-convective heating terminal. *J. Build. Eng.* 2021, 39, 102307. [CrossRef]
15. Molina-Aiz, F.; Fatnassi, H.; Boulard, T.; Roy, J.-C.; Valera, D. Comparison of finite element and finite volume methods for simulation of natural ventilation in greenhouses. *Comput. Electron. Agric.* 2010, 72, 69–86. [CrossRef]

16. Yu, G.; Xiong, L.; Du, C.; Chen, H. Simplified model and performance analysis for top insulated metal ceiling radiant cooling panels with serpentine tube arrangement. *Case Stud. Therm. Eng.* 2018, 11, 35–42. [CrossRef]

17. Tye-Gingras, M.; Gosselin, L. Investigation on heat transfer modeling assumptions for radiant panels with serpentine layout. *Energy Build.* 2011, 43, 159–1608. [CrossRef]

18. Ahmad, N.; Lim, C. Simplified model and performance analysis for radiant cooling panel with serpentine tube arrangement and thin insulation layer for moisture control in tropical climate. *J. Phys. Conf. Ser.* 2020, 1630, 012002. [CrossRef]

19. Grinham, J.; Craig, S.; Inger, D.E.; Bechthold, M. Origami microfluidics for radiant cooling with small temperature differences in buildings. *Appl. Energy* 2020, 277, 115610. [CrossRef]

20. Mumma, S.A. Condensation Issues with Radiant Cooling Panels. Available online: http://doas-radiant.psu.edu/IAQ3.pdf (accessed on 15 July 2021).

21. Mosa, M.; Malley-Ernewein, A. Constructal design applications in buildings: Radiant cooling panels and thermochemical energy storage. *Heat Transf.* 2020, 49, 3981–3996. [CrossRef]

22. Laouadi, A. Development of a radiant heating and cooling model for building energy simulation software. *Build. Environ.* 2004, 39, 421–431. [CrossRef]

23. Yin, J.-Y.; Cao, R.-X.; Zhao, L. The Effect of Operational Condition on Performance of Serpentine Chilled Ceiling Panel. *Constr. Conserves Energy* 2008, 6, 13–17.

24. Cholewa, T.; Anasiewicz, R.; Siuta-Olcha, A.; Skwarczynski, M.A. On the heat transfer coefficients between heated/cooled radiant ceiling and room. *Appl. Therm. Eng.* 2017, 117, 76–84. [CrossRef]

25. ASHRAE. *Handbook: HVAC Systems and Equipment*; ASHRAE: Atlanta, GA, USA, 2004.

26. Conroy, C.L.; Mumma, S.A. Ceiling radiant cooling panels as a viable distributed parallel sensible cooling technology integrated with dedicated outdoor air systems/discussion. *ASHRAE Trans.* 2001, 107, 578.

27. Prasanna, P.; Moorthy, P.; Krishnan, A.; Balakrishnan, V. Analysis and optimization of radiant cooling panel with wave-type embedded pipes. *Eng. Trans.* 2020, 68, 47–67.

28. Jin, W.; Ma, J.; Jia, L.; Wang, Z. Dynamic variation of surface temperatures on the radiant ceiling cooling panel based on the different supply water temperature adjustments. *Sustain. Cities Soc.* 2020, 52, 101805. [CrossRef]

29. Zhang, L.; Liu, X.-H.; Jiang, Y. Simplified calculation for cooling/heating capacity, surface temperature distribution of radiant floor. *Energy Build.* 2012, 55, 397–404. [CrossRef]

30. Mumma, S.A. Chilled ceilings in parallel with dedicated outdoor air systems: Addressing the concerns of condensation, capacity, and cost. *ASHRAE Trans.* 2002, 108, 220.

31. Miriel, J.; Serres, L.; Trombe, A. Radiant ceiling panel heating–cooling systems: Experimental and simulated study of the performances, thermal comfort and energy consumptions. *Appl. Therm. Eng.* 2002, 22, 1861–1873. [CrossRef]

32. Cengel, Y.A.; Cimbala, J.M. *Fluid Mechanics: Fundamentals and Applications*, 4th ed.; McGraw-Hill Education: New York, NY, USA, 2018.

33. Yusof, M.; Sainan, K.; Zakaria, I. Pressure Drop Analysis of Serpentine Mini Channel through Local Loss Coefficient Value. *J. Mech. Eng.* 2020, 9, 79–94.