The Planck-LFI flight model composite waveguides

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Abstract

The Low Frequency Instrument on board the PLANCK satellite is designed to give the most accurate map ever of the CMB anisotropy of the whole sky over a broad frequency band spanning 27 to 77 GHz. It is made of an array of 22 pseudo-correlation radiometers, composed of 11 actively cooled (20 K) Front End Modules (FEMs), and 11 Back End Modules (BEMs) at 300K, each FEM and BEM set comprising two radiometers. The connection between the two parts is made with rectangular Wave Guides (WGs). Considerations of very different nature (thermal, electromagnetic and mechanical), imposed stringent requirements on the WGs characteristics and drove their design. From the thermal point of view, the WG should guarantee good insulation between the FEM and the BEM sections to avoid overloading the cryocooler. On the other hand it is essential that the signals do not undergo excessive attenuation through the WG. Finally, given the different positions of the FEM modules behind the focal surface and the mechanical constraints given by the surrounding structures, different mechanical designs were necessary. A composite configuration of Stainless Steel and Copper was selected to satisfy all the requirements described. Given the complex shape and the considerable length (about 1.5-2 m) of the LFI WGs, manufacturing and testing the WGs was a challenge. This work deals with the development of the
LFI WGs, including the choice of the final configuration and of the fabrication process. It also describes the testing procedure adopted to fully characterize these components from the electromagnetic point of view and the space qualification process they underwent. The Scattering parameters of the WGs were obtained in a one port configuration, minimizing instrumental errors. The space qualification process required manufacturing ad-hoc facilities to support the WGs during vibration tests and to allow electromagnetic tests without removing them. Results obtained during the test campaign are reported and compared with the stringent requirements. The performance of the LFI WGs is in line with requirements, and the WGs were successfully space qualified.

1 Introduction

The PLANCK satellite, ESA’s third generation space mission devoted to the study of the Cosmic Microwave Background (CMB), is designed to produce a map of the CMB anisotropy over the whole sky, with an unprecedented combination of angular resolution (4'-30') and sensitivity ($\Delta T/T \simeq 10^{-6}$), for a wide range of frequencies (27-850 GHz). Two complementary instruments, LFI (Low Frequency Instrument) operating in 3 20% wide frequency bands, centered at 30, 44 and 70GHz (Bersanelli et al. [2009]) and HFI (High Frequency Instrument in the 100-850 GHz range) (Puget et al. [1998]), have been recently integrated together and share the focal plane of a Gregorian off-axis aplanatic optimized telescope (Villa et al. 2002).

LFI consists of an array of 22 wide-band pseudo-correlation radiometers (Cuttaia et al. [2004]), composed of a Front End Unit (FEU) working at 20 K (Davis et al. [2009]; Varis et al. [2009]), where the signal is collected and amplified, and a Back End Unit (BEU) (Artal et al. [2009]; Varis et al. [2009]), working at 300K, where the signal is further amplified and then detected. The LFI WGs are necessary in order to connect the two parts of the radiometers.

The FEU is composed of 11 dual profile corrugated Feed Horns (FHs) (Villa et al. [2002]), placed in the outermost region of the PLANCK focal plane, each one connected to an Ortho Mode Transducer (OMT) (D’Arcangelo et al. [2009]) that splits the incoming radiation in two orthogonal, linearly polarized signals, each feeding two independent Radiometric Chain Assemblies (RCA). A Front End Module (FEM) is mounted behind each OMT; here the signal coming from the sky is continuously compared with a reference black body signal at 4 K provided by a pyramidal horn facing a microwave load (Valenziano et al. [2009]). Every FEM has 4 outputs, two for each RCA. Thus the 44 LFI WGs, arranged in groups of 4, connect the 20 K FEU and the 300 K BEU, covering a distance between about 1.5 and 2 m. A schematic view of an LFI radiometer is given in figure
The goal of this paper is to report the design, fabrication and testing activity of the WG: the results obtained during the tests are thus compared with the specification requirements. The paper is organized as follows: in sections 2 and 3 the design and fabrication technique of the WGs are described. In section 4 the electromagnetic tests made on the WGs and the measurement techniques are presented and discussed. In section 5 the results of the Flight Model (FM) test campaign are shown, while section 6 deals with vibration tests. Section 7 provides a summary and a few concluding remarks.

Figure 1: Scheme of the PLANCK-LFI radiometers.

2 Design

Metallic waveguides are commonly used in microwave technologies as transmission lines. They consist of metal hollow pipes usually with rectangular cross section, guiding electromagnetic waves. The most commonly used are rectangular in section with standard dimensions. For an exhaustive treatment of waveguides see Marcuvitz (1993) and Cronin (1995). For LFI it was decided to use standard rectangular waveguides to reduce costs and manufacturing criticalities. Their unimodal propagation simplifies matching with the MMIC (Millimeter Wave Integrated Circuits) of the FEM and control of discontinuities. Losses are higher than in oversized waveguide but they are still within the performance budget (Bersanelli, 2009). The complexity of LFI translated into the waveguides that required extensive prototyping activity and development strategy for design, manufacturing,
measurements and space qualification. As usual for space components, LFI WGs must exhibit low mass and compatibility with payload vibration at launch, but their characteristics are also strongly driven by the role they play in the radiometer structure, since they interconnect two parts of the instrument operating at very different temperatures. Thus it is essential for them to have a very low thermal conductivity in order to decouple the FEM at 20 K and the BEM at 300 K. On the other hand, they also must show good electromagnetic performance, i.e. acceptable Insertion Loss (IL) and high Return Loss (RL), in order to allow the radiometric signal to reach the last stage of detection without excessive attenuation. Finally, they require a complex geometric routing, since each of them must reach a different location in the focal surface, while allowing integration of HFI in the LFI frame and of both instruments on the satellite (figure 2). Thus the final configuration chosen for the WGs design is a tradeoff between thermal, electromagnetic and mechanical aspects; the first consequence is that the WGs were built using different materials. They are composed of two separate sections (connected by non standard flanges), one made of Stainless Steel (SS) and the other made of Copper (Cu), both of standard rectangular size. SS sections have the same length (700mm) at all frequencies, and a very strong thermal gradient along their path. The inner part is partially gold plated to minimize losses. The Cu section is the one that reaches the FEU, and it accommodates all the bends and twists necessary.

Figure 2: Schematic view of the PLANCK LFI/HFI
to reach different positions of the FEU and to take into account the surrounding structures that impose complex geometric routing. They will work at a constant temperature of 20 K. Copper was selected because of its malleability that minimizes the risk of mechanical damage. It has also a good electrical conductivity at cryogenic temperature and can be easily electroformed. The length and shape of these WG sections is variable between about 800 and 1200 mm, depending on the different positions reached in the FEU. Figure 3 shows 4 Cu WGs at 70 GHz. At the beginning of the waveguide development, it was foreseen to avoid any interconnection flange between the different waveguide sections as in the case of WMAP (Jarosik et al., [2003]). However, the extremely complex waveguide path and the 2 m length did not allow building the waveguide in a single piece.

2.1 Electromagnetic design

The copper section bends and twists were designed optimizing the return loss using analytical formulae in Johnson, [1991]. Then the analytical approach was verified with plane finite element software (HFFS1) for the typical case of H– and E– bends both isolated and cascaded. For E–plane bends the analytic form used for the reflection coefficient is

$$|\rho| = \frac{b^2}{24R^2} \sin \left( \frac{2\pi \ell}{\lambda_g} \right) - A \cdot \cos \left( \frac{2\pi \ell}{\lambda_g} \right), \quad \text{for } \frac{R}{b} > 2$$

(1)

where \( R \) is the radius of curvature of the bend, \( \ell \) is the length of the bend along its path, \( \lambda_g \) is the wavelength inside the waveguide, \( b \) is the short side size of the waveguide. The parameter \( A \) is calculated as follows:

$$A = \left( \frac{96}{\pi^4} \right) \left( \frac{2b}{\lambda_g} \right) \sum_{m=1,3}^{\infty} m^{-5} \left[ \left( \frac{2b}{m\lambda_g} \right)^2 \right]^{-1/2}$$

(2)

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The best match is obtained if the length of the bend satisfies the following condition:

\[ \ell = R \cdot \theta = \frac{\lambda_g}{2\pi} \tan^{-1} A \quad (3) \]

In addition the length \( \ell \) can be increased in steps of \( \lambda_g/2 \) in order to arrive at an allowed value of \( R/b \). Although this is a resonant condition, the simulations and the measurements on all the LFI waveguides showed that these analytical formulas can be used to design complex waveguides. Similarly for H–plane bends the formulas become

\[ \rho = \frac{\lambda_g^2}{32\pi^2 R^2} \left| \sin \left( \frac{2\pi \ell}{\lambda_g} \right) - \frac{128}{\pi^2} \frac{a}{\lambda_g} B \cdot \cos \left( \frac{2\pi \ell}{\lambda_g} \right) \right|, \quad \text{for } \frac{R}{a} > 2 \quad (4) \]

\[ B = \sum_{2, A} m^2 (m^2 - 1)^{-3} \left[ (m^2 - 1) - \left( \frac{2a}{\lambda_g} \right)^2 \right]^{-1/2} \quad (5) \]

\[ \ell = R \cdot \theta = \frac{\lambda_g}{2\pi} \tan^{-1} \left[ \frac{128}{\pi^2} \frac{a}{\lambda_g} B \right] \quad (6) \]

where \( a \) is the long side size of the waveguide.

The length of the twist was chosen so that

\[ \ell = (2n) 0.25\lambda_t \quad (7) \]

where \( n \) is an integer and \( \lambda_t \) is the wavelength inside the twist.

To improve the return loss of the copper sections, a straight section at least one wavelength long has been interposed between subsequent discontinuities. This was done in order to reestablish the \( TE_{10} \) fundamental propagation mode in case of unwanted spurious mode excitation and prevent tunneling of higher order modes through discontinuities. During the design phase it was also established to insert a maximum of one twist for each waveguide and to use a long radius of curvature where possible, again with the goal of improving return loss. Manufacturing tolerances were studied based on Alison, ([1972]) and set after a prototype activity on WR10 (W–band) waveguides. This included also the verification of the electromagnetic properties and methods to guarantee a return loss better than 30dB over the entire bandwidth and an attenuation as good as straight waveguides.

2.2 Thermal design

The thermal characteristics of the WGs played an important role in the final design configuration: in fact, since the LFI waveguides connect the warm, \( \sim 300 \text{ K} \) unit to the cold front end of the instrument, along their routing they have been connected
mechanically and thermally to the three thermal shields (V-grooves) of Planck, passively thermalised at $\sim 150$ K, $\sim 100$ K and $\sim 50$ K. Therefore heat transfer and electrical conduction played a key role in the waveguide design, with the goal of keeping the heat load to the 20 K stage at the level of $\sim 100$ mW and the insertion loss at a level of a few dBs. The composite design described in detail in this section has been therefore devised based on an analytical heat transfer model. To estimate the heat transferred by conduction and, therefore, the temperature profile along the waveguide, we have formalised the problem by making simplifying assumptions, i.e.:

- the thermal link between waveguides and thermal shields is assumed to be perfect, which implies that all the heat coming from higher temperature regions will be transferred to the V-groove;
- every V-groove can be considered as an infinite thermal sink;
- no heat will be dissipated by the waveguides between two V-grooves.

Under these assumptions the heat load ($\dot{Q}$) on a given thermal link set at temperature $T_{\text{low}}$ is constant, and determined uniquely by the waveguide characteristics and by the temperatures at the extremes ($T_{\text{low}}$, $T_{\text{high}}$), i.e.:

$$\dot{Q}_{\text{cond}} = \frac{A}{l} \int_{T_{\text{low}}}^{T_{\text{high}}} k(T) dT$$  \hfill (8)

where $A$ is the waveguide wall section, $l$ represents the distance between thermal interfaces at $T_{\text{low}}$ and $T_{\text{high}}$ and $k(T)$ is the thermal conductivity function. From equation (8) it is possible to derive the temperature profile along the waveguide, considering that at any point $\dot{Q}_{\text{cond}}$ is constant and, therefore

$$\dot{Q}_{\text{cond}} = \frac{A}{x} \int_{T_{\text{low}}}^{T(x)} k(T) dT$$  \hfill (9)

for each value of $x$. From a computational point of view the easiest way to calculate $T(x)$ has been to generate a list of temperature values in the range $[T_{\text{low}}, T_{\text{high}}]$ and then calculate the corresponding values of $x$ from the above equation. In Fig 4 we show an example of temperature profile calculated using Eq. 8. The last question considered during the design phase was the necessity for safely reaching vacuum conditions in space, since LFI waveguides may create problem in evacuating the LFI frontend and backend modules during the satellite launch. In radiative heat transfer inside waveguides is negligible compared to conduction because (i) the inner waveguide surface is far from being specular to IR radiation and (ii) twists and bends favour absorption of IR photons.

2Radiative heat transfer inside waveguides is negligible compared to conduction because (i) the inner waveguide surface is far from being specular to IR radiation and (ii) twists and bends favour absorption of IR photons.
order to avoid damages to the radiometers, the WG flanges include venting holes to aid the flow of air. Holes in the waveguide wall behave as a third port, producing mainly two effects: they create discontinuities (generating reflections inside the waveguide) and they radiate electromagnetic power into space. Reflections can be limited by an appropriate location of the hole where perturbation to wall currents is minimum, while radiation can be controlled by keeping the hole diameter below cutoff for the fundamental circular waveguide, i.e.:

\[
  d < d_{\text{max}} = \frac{1.841}{\pi \sqrt{\epsilon_0 \mu_0 \nu_{hf}}} \]

(10)

where \( \nu_{hf} \) represents the highest frequency propagating inside the WG, 77 GHz for the LFI instrument, which corresponds to \( d_{\text{max}} = 2.3 \) mm. Hence, holes dimension must come out as a trade-off between fluid conductance and electromagnetic requirements. The resulting optimized drawing foresees a single hole per waveguide, directly on the flange, with the hole axis parallel to the Hplane. The hole diameters is 1 mm: this value looks large enough for the venting purpose and safe from the electromagnetic point of view. The length of the hole is 6.95 mm. Simulations, performed at the highest in band frequency, demonstrated that the holes do not impact the electromagnetic performance of the WG and that the power radiated through the hole is well below the measurable level.
3 Fabrication technique

The WG section between the Cu WG and the BEU is made, for the first 300 mm, of stainless steel, whose poor thermal conductivity limits the thermal input on the 20 K stage. Since SS is also a poor electrical conductor, the subsequent 400 mm of the WG are gold plated internally, in order to minimize signal losses, while preserving a low thermal conductivity. Thermal decoupling is a critical issue for LFI, so each WG is connected to three thermal shields (V-grooves) to enhance control of the heat load on the FEU. In addition, a paint treatment with Aeroglaze Z306, a high emissivity (0.94-0.97) and low outgassing black polyurethane, was used for the SS section. This paint provides the thermo-optical properties required to dissipate as much as possible into space. The emissivity of Aeroglaze Z306 increases with temperature, thus further preserving the coldest part of the satellite from radiative coupling. The SS sections are straight rectangular WGs with the same length (700 mm) at all frequencies. An image of four 30 GHz SS WGs is shown in figure 5. A schematic reproduction of the inner part of the two WG sections is shown in figure 6 while the physical characteristics for the selected WG’s design are reported in table 1.

![Stainless Steel 30 GHz WG](image)

Figure 5: Stainless Steel 30 GHz WG

![Schematic reproduction of the inner part of the two WG sections](image)
Figure 6: Schematic view of the inner part of the LFI’s WG

The two sections of the WGs are connected with a custom flange. The connection between the Cu section and the FEU requires a non standard flange at all frequencies, while the connection between the SS section and the BEU requires non standard flanges only at 44 and 70 GHz. The two separate WG sections show different criticalities, since they are made of different materials and have different shape and length. The main difficulty in manufacturing the SS section consists in the partial gold plating of its inner part, especially at higher frequencies. For the Cu section, instead, the main problem is to make the bends and twists requested by the design. Two different manufacturing processes were selected for the SS and the Cu sections. The SS parts were directly built in a standard way and the gold deposition was made by placing the anode wire inside the WG, ensuring that it did not hit the internal walls, and providing the necessary flux of electrolyte inside the WG. The part that was not to be plated was masked before the deposition. The Cu section were instead built using the electroforming technique, even if it is a very complicated process. During the design phase, the contribution of discontinuities such as E and H plane bends and twists to losses was estimated analytically. Calculations showed that any possible distortion of the WG size has a larger impact on the RL of the component than the bends and twists themselves (LFI scientific team, 2003). Also the routing of these sections was considered and optimized during the design phase: the number of twists was minimized because of the criticality in manufacturing the mandrels, and the number of different bends

| Central | Internal | Thickness |
|---------|----------|-----------|
| Frequency | section | Cu | SS | Au |
| [GHz] | [mm] | [mm] | [mm] | [µm] |
| 30 | 7.112x3.556 | 0.4 | 0.254 | 2.0 |
| 44 | 5.690x2.840 | 0.4 | 0.254 | 2.0 |
| 70 | 3.099x1.549 | 0.4 | 0.254 | 2.0 |

Table 1: Mechanical properties of the LFI’s WGs
in order to simplify mandrel integration before electroforming. Other criteria were the minimization of WG length and the use of straight sections wherever possible in order to minimize RF losses. The tolerances declared by the manufacturer are reported in tables 2 (Cu section) and 3 (SS section).

|                  | 30GHz | 44GHz | 70GHz |
|------------------|-------|-------|-------|
| Thickness [mm]    | 0.1   | 0.1   | 0.1   |
| straight part length [mm] | 0.127 | 0.127 | 0.05  |
| bend radius [mm]  | 0.127 | 0.127 | 0.05  |
| bend angle [deg]  | 0.1   | 0.1   | 0.1   |
| section dimensions [µm] | 38.1  | 25.4  | 12.7  |

Table 2: *Mechanical tolerances defined by the manufacturer for the LFI’s Copper WGs*

|                  | 30GHz | 44GHz | 70GHz |
|------------------|-------|-------|-------|
| Gold thickness [µm] | 0.25  | 0.25  | 0.25  |
| SS total length [mm] | 0.127 | 0.127 | 0.127 |
| Gold plated length [mm] | 6.35  | 6.35  | 6.35  |

Table 3: *Mechanical tolerances defined by the manufacturer for the LFI’s Stainless Steel WGs*

Even using consolidated manufacturing techniques it was not possible to make the Cu section in a single piece at higher frequencies. In fact, given the small dimensions and the complex routing of the WGs, etching the mandrel was extremely slow, about 50 mm/day. Thus, to optimize the formation process while preserving the quality of final product, three separate pieces were made at 70 GHz and joined together using the copper joint technique. Finally the WGs were washed repeatedly with isopropyl alcohol in order to remove any residuals inside them. The importance of this last step was soon discovered, since in a few occasions a single wash was not enough to remove all residuals, but this was apparent only after the electromagnetic tests.

4 **EM testing**

The LFI WGs must satisfy many requirements of different nature. Despite their complex geometrical shape, they must be able to withstand the mechanical stress of launch and for this reason they have been submitted to vibration tests. Moreover, their main scope is the thermal decoupling of the FEM and BEM without
excessive attenuation of the radiometric signal; thus the scattering parameters of each WG and the isolation within each set of 4 channels have been measured. The main requirements for the electromagnetic properties of the WGs (Guzzi and Villa [2003]) are reported in table 4.

| $\nu$ band [GHz] | IL@20K | RL | Isolation |
|------------------|--------|----|-----------|
| 30GHz            | 27-33  | < 2.5dB | > 25dB | > 30dB |
| 44GHz            | 39.6-48.4 | < 3dB | > 25dB | > 30dB |
| 70GHz            | 63-77  | < 5dB | > 25dB | > 30dB |

Table 4: Electromagnetic requirements for the LFI’s WG at operational temperature

The detailed electromagnetic characterization of the LFI’s WGs was a challenging task. One of the first constraints is related to the length of the WGs; ideally the best solution for measuring the scattering parameters would be working in a two port reflectometer configuration. But when measuring very large components such as the LFI’s WGs, the path difference between the calibration standards and the Device Under Test (DUT) is very large, and this requires very large cable movements. Since the measurements were performed using an AB-Millimetre Vector Network Analyzer (VNA) which works without an external reference channel, excessive cable movements degrade the quality of measurements. For this reason, all tests performed on the WGs were made in a one port reflectometer configuration (figure 7).

Figure 7: Experimental set up for electromagnetic tests.
Another undesired effect of the large path difference between DUT and calibration standards, is the measurement error arising when the frequency stability of the master oscillator is inadequate (or if the VNA operates in swept frequency mode). A way to avoid this problem was found performing measurements with different cable lengths and then combining the results of two different measurements (Simonetto et al. [2006]). Once these main difficulties were solved, the measurement technique was standard. All tests were performed using the variable short, variable load and fixed short calibration. After calibration, the WGs were terminated with a matched load to determine the reflection coefficient and with a short to evaluate the transmission coefficient.

Working in a one port configuration, the quality of data can be degraded if return loss and insertion loss are comparable. In fact, terminating a DUT with an ideal short in a one-port configuration, the measured reflected signal \( b \) at port 1 is

\[
b = S_{11} - \frac{S_{12} \cdot S_{21}}{1 + S_{22}}.
\]

(11)

The measured data are therefore a combination of IL and RL, and of course the contribution of RL cannot be neglected unless the DUT is low loss and very well matched. Since the return loss was measured, those data were used to correct the measurements by direct substitution, as described in (D’Arcangelo et al. [2006]). This practice is far from optimum in terms of error propagation, therefore the results were compared with those obtained with the more accurate (and time consuming) approach of taking the ratio of coefficients in the error matrices extracted from full calibrations with and without DUT. The two methods are consistent within the error bars. A comparison between raw and corrected data is shown in fig. 8. The FEM flanges at all frequencies and the BEM flanges at 44 and 70 GHz required adapters. In order to evaluate losses and reflections of the WGs, it is necessary to evaluate the same quantities for the adapters too, thus they were tested with the calibration technique described previously. Since the losses of the adapters were extremely low and comparable with the measurement error (about 0.1 dB), a very precise determination of this value within the bandwidth was not possible, but anyway results can be considered as an upper limit. Adapters have an impact also on the measured return loss of the WG: a way to remove or minimize this unwanted contribution is using time domain digital filtering, known as time gating in analog terms. The Fourier transform (FT) of the frequency sweep allows in fact a quick identification of the position of the main reflections and their removal if not caused by the WG itself. The ideal condition for this operation would be a very good space resolution, in order to determine with high precision the exact position of the reflection points. The limiting factors are measurement bandwidth and waveguide dispersion. However, it was possible to guarantee a satisfying resolution (about 5-8 mm depending on the LFI frequency
channel). Quite soon, already during the qualification model test campaign, it became apparent that testing the two different WG sections independently would be highly beneficial. On the one hand, this allows a detailed knowledge of the behaviour of each section of the WG, and on the other it helps in determining the location of faults. Thus, after being tested as single units, the Cu and SS WG are integrated and tested again as a single piece in order to guarantee that a good match between the two sections was obtained. Electromagnetic tests turned out to be a very useful tool for spotting problems in the WGs. Typically, the problems were residuals of the aluminum mandrel that was not completely dissolved, and in one case a chemical deposit left inside the SS section after Au deposition.

4.1 Transmission Coefficient

The transmission coefficient of the WGs was determined by subtracting the contribution of the flange adapters, even though it was usually much smaller than the WG loss. At 30 GHz only one adapter was used, since the BEM-side flange was compatible with the standard UG 599 ones. The average amplitude of the transmission coefficient of the 4 WGs belonging to the same bunch is shown in fig. 9 for the Cu and SS sections in the 3 LFI frequency bands. The frequency is normalized to the central one. In this plot, data are shown only on the nominal frequency band, but measurements were taken for all WG over the largest possible frequency band, typically 26.5-40 GHz, 33-50 GHz and 60-80 GHz. These results
are quite representative of the general behaviour of all the LFI’s WGs. Usually, the measured losses of WGs are quite similar (differences of fractions of dB) within the same bunch, the difference being mainly due to the different length of the Cu section (a few centimeters). No requirements having been defined for the amplitude of the transmission coefficient at room temperature, a partial indication of the correct behaviour is given by the comparison with simulated losses. The result can only be considered partial, since simulations were made considering a standard straight rectangular WG, made of gold for 400 mm, SS for 300 mm, and Cu for the rest of its length. Bends and twists were considered as a straight sections with equivalent length with the pure $TE_{10}$ mode propagating inside. Ideal conductivity was used and the roughness of the internal walls was neglected. The data usually showed that the difference between simulations and measurements is mainly caused by the Cu sections. Typical behaviour is reported in figures 10 and 11 where the losses of the two sections of a 70GHz WG (where the difference is maximum) are compared with simulations. Figures 10 and 11 clearly show that the maximum difference between simulations and measurements is smaller than 0.5 dB for the SS section, while the discrepancy is greater for the Cu section, even more than 1.5 dB. This should not be surprising since simulations were made assuming a perfect straight waveguide with ideal conductivity. While return loss simulation techniques of lossless waveguide discontinuities are well established, dedicated simulations to evaluate the insertion loss were not addressed because of the intrinsic difficulty to simulate long (in terms of wavelength) dissipative roughness structures. Moreover the simulation would provide an approximate results.
Figure 10: Amplitude of the transmission coefficient of 4 70GHz LFI’s SS WGs, compared with the simulated losses of a straight rectangular guide.

Figure 11: Amplitude of the transmission coefficient of 4 70GHz LFI’s Cu WGs, compared with the simulated losses of a straight rectangular guide.

anyway, since the ohmic losses depend critically on the quality of the inner surface that could only be evaluated with prototyping. It was then decided to evaluate the effective attenuation coefficient directly from measurements. The analytical simu-
lation where used to demonstrated that mainly the $TE_{10}$ mode propagates inside the copper section since the data and the simulations show the same behaviour. While the pictures at 30 and 44 GHz contain basically the raw data (only the adapter’s contribution having been removed from data), at 70 GHz the data were also corrected for the contribution of reflections, as described above, since they are not negligible in this case (figure 5). Even if specifications were given only at operational temperature, a few considerations can be contemplated. All 30 GHz WGs meet requirements already at room temperature, and losses are expected to decrease at operational temperature. At 44 GHz, a few WGs meet (at room temperature) the insertion loss requirement of 3 dB (at cryogenic temperature), on a limited portion of the bandwidth, while a few others are very near this value. At 70 GHz, instead, results are far from the requirement value of 5 dB. A rough extrapolation of losses to operational temperature was made considering that, for a straight rectangular WG, the attenuation $A$, measured in dB, is

$$A_{dB} = -20 \cdot \alpha \cdot l \cdot \log(e)$$

(12)

where $l$ is the WG’s length while the attenuation coefficient $\alpha$ is given by

$$\alpha = \frac{2}{a} \left( \frac{\pi \epsilon_0 \nu}{\sigma} \right)^{1/2} \left( 1 - \frac{c^2}{4a^2 \nu^2} \right)^{1/2} \left( 1 + \frac{c^2}{4a^2 \nu^2} \right)$$

(13)

where $\nu$ is the frequency, $\sigma=1/\rho$ is the conductivity of the material, $\epsilon_0$ is the vacuum dielectric constant, $a$ is the broad side of the WG and $c$ is the speed of light in vacuum.

Data at room temperature were fitted using this formula, thus finding an effective average resistivity of the material at room temperature. This value is compared with the theoretical one to determine the deviation from ideal case. Performance at operational temperature is then obtained considering the WG as if it were composed of small sections at different constant temperature and assuming the resistivity obtained from the fit to scale as function of temperature like the ideal value.

Results are shown in figure 12 and they are representative of the general behaviour found for the 70 GHz WGs. Extrapolated losses thus match the requirements in this band too.

A statistical analysis was also performed on all the LFI WGs, making a histogram of the ratio between measured and simulated attenuation coefficient. The mean value of the frequency-dependent attenuation coefficient was considered for the analysis. The results point out the general behaviour of the WGs, which seems very similar within each band, as shown in pictures 13, 14, 15.
Figure 12: Transmission coefficient measured at $T$ room, scaled at operational temperature.

Figure 13: Measured-to-simulated attenuation coefficient ratio of the Stainless Steel WGs

The SS WGs have attenuation coefficients close to the simulated one, and this is expected since they are all straight rectangular waveguides. On the other hand for the Cu WGs the value $\alpha/\alpha_0$ is centered around 1.4. Of course, simulations do not take into account twists and bends, but even an incorrect assumption of the conductivity of the material (assumed ideal, with no account for surface roughness) could explain the difference. Also, the spread of the results for Cu is broader than for SS section and this could point to a less reproducible manufacturing process, but the impact of twists and bends should be assessed thoroughly before drawing such a conclusion. The integrated WGs have of course an intermediate behaviour.
and this is an indication that the integration process does not introduce anomalies. The distributions of figures 13, 14, 15 were fitted with a gaussian function: the results of the fit are listed in table 5.
4.2 Reflection coefficient

Representative results for the 70 GHz and the 44 GHz frequency bands are shown in figures 16, 17, 18 and 19 (results at 30 GHz are extremely similar to those at 44 GHz). Since losses at 30 and 44 GHz are similar and not too high, the return loss measured at the FEM and BEM side is similar; in fact the signal reflected at any discontinuity inside the waveguide is not attenuated significantly on the way back to the injection port. At 70 GHz instead, where WG losses are definitely greater, the difference between the FEM and BEM side is clearly visible. In fact, the main sources of reflection are generally in the Cu section. In particular, the regions with the narrowest bends are the most reflective ones. This can be seen when correcting reflection data using time domain filtering (time–gating): the FT of a frequency sweep gives a sequence of peaks as a function of time, representing reflections from different parts of the device under test. This clearly points out that main reflection peaks are located in the CU section of the WGs, as expected.

![Figure 16](image)

Figure 16: Amplitude of the reflection coefficient at 44 GHz, signal inserted from the BEM flange. The two vertical lines represent LFI’s frequency band, while the horizontal one represents the room T requirement.

| WG                  | $\alpha/\alpha_0$ | $\sigma$ |
|---------------------|--------------------|----------|
| Stainless Steel section | 0.99              | 0.064    |
| Copper Section       | 1.4                | 0.13     |
| Integrated WG        | 1.15               | 0.086    |

Table 5: Results of the gaussian fit for the attenuation coefficient.
Figure 17: Amplitude of the reflection coefficient at 44 GHz, signal inserted from the FEM flange. The two vertical lines represent LFI’s frequency band, while the horizontal one represents the room T requirement.

Figure 18: Amplitude of the reflection coefficient at 70 GHz, signal inserted from the BEM flange. The two vertical lines represent LFI’s frequency band, while the horizontal one represents the room T requirement.

The FT is also used to filter (i.e. zeroing out) the unwanted contribution of adapters and transforming back to frequency allows correction of the data. Results show that requirements are met with some margin at 30 and 44 GHz, while at 70GHz the amplitude of the reflection coefficient is around the requirement limit of -25dB for all WG, especially at the FEM side.
Figure 19: *Amplitude of the reflection coefficient at 70 GHz, signal inserted from the FEM flange. The two vertical lines represent LFI’s frequency band, while the horizontal one represents the room T requirement.*

### 4.3 Isolation

The last step in the electromagnetic characterization of the LFI’s WGs is the evaluation of isolation between the 4 WGs belonging to the same bunch. The measurement is made in a two ports configuration with only a simple response correction calibration, basically because the expected measured level is very low and an overly accurate measurement is not necessary. The requirement at room temperature is -30 dB, and results obtained in the 70 GHz band are reported in figure [20](#). They are representative also for the other bands. That figure clearly points out that the measured isolation is very high, dominated by instrumental noise and well below the requirement. This is true for all LFI’s flight WGs, and is expected to hold true also at cryogenic temperature.
Figure 20: 70 GHz isolation, the two vertical lines represent the LFI’s frequency band, while the horizontal one represents the T room requirement.

5 Vibration tests

The LFI WGs must be vibrated at space qualification level but, given their shape and physical dimensions, it was not possible to vibrate the Cu and SS section while integrated. The two sections have been tested separately: each WG has two flanges which, during operations, are submitted to two different dynamic environments. Since the experimental set up did not allow simultaneous excitation of random vibrations at two different locations, each WG was vibrated with an input random level (along the three axis) corresponding to the envelope of those relevant to the two flanges.

Vibration tests were performed with a shaker (figure 21) controlled by a computer, which generates the motion with the requested characteristic. The accelerometers, placed along the WG and its flanges, were used for feedback. The accelerometers’ recorded signals were amplified, digitized and stored for processing.

It was necessary to manufacture two different fixtures (one for the SS WGs, figure 22, and one for the Cu WGs, weighting 50Kg, figure 23, and some specific flange supports for this test. The supports have a double use. On the one hand they were used for solid connection with the vibrating structure, and on the other hand they allow checking the resiliency of flange joints to vibration. They consist of aluminum blocks with four RF channels each, thus the two blocks and the WG form a single unit of four RF channels.

The test started with a search for resonances in the range 5-2000Hz once the WG was firmly mounted on the shaker. The following conditions had to be satisfied for a successful vibration test: after each random vibration on every axis, the WG
should not show visible degradation and all the screws holding it to the fixture should be tight to specified torque. Moreover, the mechanical resonance frequencies before and after vibration should be shifted by less than 5%, while the variation in acceleration should be smaller than 10%.

Finally, electromagnetic tests were repeated to certify success in vibration test. The reflection coefficient of the WG was controlled before and after vibration, because of its sensitivity to mechanical imperfections. These measurements were repeated inserting the signal from both sides, to guarantee that the electromagnetic
effect of any damage was not hidden by the attenuation of the WG. In the very few cases where differences were seen, they were found in the coupling between the aluminium blocks and the adapters used to connect to the microwave circuit. An example of the comparison between the reflection coefficient measured before and after vibrations is reported in figure 24. Moreover also the Fourier Transform of

![Figure 24: Reflection coefficient of a Cu 70 GHz WG measured before and after the vibration tests](image_url)

the signal measured before and after the vibration was calculated and compared. In this way any possible variation in the reflections peaks caused by the vibration could be detected.

12 SS and 12 Cu WGs (i.e. 1 bunch of 4 WGs for each frequency) have been vibrated: the very first tests, performed at low energy, suggested a structural weakness of the WGs. Thus, all the flanges were reinforced, and all the WGs were retested afterwards, to guarantee that reworking did not impact on performance. After these steps, the 12 SS and 12 Cu WGs were successfully vibrated. Only Flight Spare (FS) components were vibrated, in order to avoid excessive stress on the flight components.

## 6 Summary

Six prototypes, 5 Qualification Model, 22 FM and 22 FS WGs were built and extensively tested. They include a straight SS section, partially gold plated, and a Cu section with twists and bends along. Because of their peculiar geometrical shape and physical dimensions, it was necessary to optimize the measurements technique
in order to obtain a precise characterization. The manufacturing technique was also tested and optimized during the qualification phase. Electromagnetic requirements are usually met. A mismatch between simulated and measured losses at room temperature was observed, increasing with frequency. Data collected over the two separate sections clearly pointed out that the difference is basically due to the Cu part, and it should be accounted on the oversimplified simulations neglecting twists and bends. Finally, all FM WGs were successfully vibrated to space qualification level, although flanges needed reinforcement.

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