The Great Inequality and the Dynamical Disintegration of the Outer Solar System

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\textbf{Abstract}

Using an ensemble of $N$-body simulations, this paper considers the fate of the outer gas giants (Jupiter, Saturn, Uranus, and Neptune) after the Sun leaves the main sequence and completes its stellar evolution. Due to solar mass loss—which is expected to remove roughly half of the star’s mass—the orbits of the giant planets expand. This adiabatic process maintains the orbital period ratios, but the mutual interactions between planets and the width of mean-motion resonances (MMR) increase, leading to the capture of Jupiter and Saturn into a stable 5:2 resonant configuration. The expanded orbits, coupled with the large-amplitude librations of the critical MMR angle, make the system more susceptible to perturbations from stellar flyby interactions. Accordingly, within about 30 Gyr, stellar encounters perturb the planets onto the chaotic subdomain of the 5:2 resonance, triggering a large-scale instability, which culminates in the ejections of all but one planet over the subsequent \(~10\) Gyr. After an additional \(~50\) Gyr, a close stellar encounter (with a perihelion distance less than \(~200\) au) liberates the final planet. Through this sequence of events, the characteristic timescale over which the solar system will be completely dissolved is roughly 100 Gyr. Our analysis thus indicates that the expected dynamical lifetime of the solar system is much longer than the current age of the universe, but is significantly shorter than previous estimates.

\textit{Unified Astronomy Thesaurus concepts: Dynamical evolution (421); Solar system (1528)}

1. Introduction

Understanding the long-term dynamical stability of the solar system constitutes one of the oldest pursuits of astrophysics, tracing back to Newton himself, who speculated that mutual interactions between planets would eventually drive the system unstable (Laskar 1996, 2012). Laplace (1799–1825) and Lagrange (1776) successfully challenged this perception by approximating the mutual interactions as perturbations, showing that, to leading order, the long-term evolution of all known solar system planets could be described via cyclic secular variations, thus analytically demonstrating the indefinite stability of the solar system. However, subsequent analyses by Gauss (1809) and Le Verrier (1856) showed that these approximations break down over sufficiently long time intervals, so that more complicated solutions are required. Poincaré (1892) formalized this insight by proving that the full three-body problem could not be solved in closed form—a barrier that has only recently been overcome with the advent of modern computing and $N$-body integration methods.

Unfortunately, even the most precise $N$-body simulations are only able to produce time-limited prognosis for the evolution of the solar system. Due to the chaotic nature of the planetary orbits, deterministic forecasting is impossible over sufficiently long timescales. In particular, the Lyapunov time for the inner terrestrial planets (Mercury, Venus, Earth, and Mars) is of order \(~5\) Myr (Laskar 1989; Sussman & Wisdom 1992), while the outer Jovian planets (Jupiter, Saturn, Uranus, and Neptune) appear chaotic with a Lyapunov time of order \(~10\) Myr\textsuperscript{5} (Laskar 1989; Murray & Holman 1999; Guzzo 2005, 2006). Predictions on timescales significantly longer than these benchmark values are only meaningful in a statistical sense, recasting the question of the solar system’s long-term fate as a probabilistic one.

Over the course of the last three decades, the question of whether or not the orbits of the solar system’s eight planets can remain immutable has come into sharp focus, with state-of-the-art simulations demonstrating that Mercury has a \(~1\%\) chance of becoming unstable within the remaining main-sequence (MS) lifetime of the Sun (Laskar 1994, 2008; Batygin & Laughlin 2008; Laskar & Gastineau 2009; Zeebe 2015). The mechanism for the onset of Mercury’s instability is well understood: by virtue of locking into a linear secular resonance with the $g_5$ mode of the solar system’s secular solution, Mercury’s eccentricity can attain near-unity values, resulting in a collision with the Sun, or even Venus. Intriguingly, general relativistic effects factor into this estimate, with ancillary apsidal precession providing a stabilizing influence on Mercury’s orbit (Lithwick & Wu 2011; Boué et al. 2012; Batygin et al. 2015). Within the context of this narrative, however, the remaining planets appear unaffected and are currently expected to remain stable for a lower limit of \(~10^{18}\) years, when diffusion arising from the overlapping mean-motion resonances (MMR) of Jupiter and Saturn are expected to decouple Uranus (Murray & Holman 1999).

Although the estimate of Murray & Holman (1999) addresses the intrinsic stability of the solar system, on sufficiently long timescales, extrinsic effects come into play. For example, stellar evolution, which is generally not considered in orbital stability studies, represents an important additional aspect of the problem. In particular, the Sun will undergo significant mass loss over the next \(~7\) Gyr, reducing the mass by roughly half, down to \(~0.54\) $M_\odot$ (Sackmann et al. 1993). Over this time span, it is probable that Mercury, Venus, and Earth will be engulfed by the Sun, as its radius expands.
during the red-giant branch (RGB) phase of evolution (Rybicki 
& Denis 2001; Schröder & Smith 2008). This epoch thus marks 
the end of the three innermost planets. Although Mars, Jupiter, 
Saturn, Neptune, and Uranus will survive this phase of stellar 
evolution (Veras & Wyatt 2012), they will experience a \( \sim 1.85 \) 
increase in their semimajor axes (Jeans 1924; Veras et al. 2011; 
Adams et al. 2013; Veras 2016). With their newly expanded 
orbits, the remaining planets are expected to remain stable for a 
minimum of 10 Gyr (Duncan & Lissauer 1998). However, the 
details of orbital evolution that unfolds on much longer 
timescales are less well characterized.

A distinct form of external forcing upon the solar system 
strikes from stellar encounters. As the solar system traces 
through its Galactic orbit it will experience perturbations from 
passing stars, which will act to excite the orbits. If the solar 
system maintained its current orbital configuration, passing 
stars would liberate the planets over timescales of the order of 
\( 10^{14} \) yr (Dyson 1979; Adams & Laughlin 1997). However, the 
interaction cross-section for these gravitational encounters 
scale with the semimajor axis of the planet (Li \& Adams 2015). 
As a result, by accounting for stellar mass loss and the inflation 
of the outer planet orbits, these encounters will become more 
influential. Given enough time, some of these flybys will come 
close enough to disassociate—or destabilize—the remaining 
planets. This paper examines the timescales and mechanisms 
that bring about the demise of the solar system.

This paper is organized as follows. In Section 2, we discuss 
the input parameters of our numerical study, including methods 
for simulating the evolution of the Sun itself and dynamical 
perturbations from stellar flybys. We then provide the results of 
our study and discuss the mechanisms for planetary disassociation 
in Section 3. The paper concludes in Section 4 with a 
summary of our results, a discussion of the significance of these 
findings, and the implications for free-floating planets.

2. Numerical Methods and System Parameters

2.1. The N-Body Simulation

The timescales of interest for this study greatly exceed the 
current age of the solar system. To carry out this simulation, we 
used the IAS15 high-order integrator (Rein \& Spiegel 2015) as 
implemented in the Rebound (Rein \& Liu 2012) software 
package. This 15th order integrator uses a Gau–Radau 
algorithm to numerically solve the equations of motion. 
Despite the high order of this calculation, any use of finite 
timesteps introduces error, which can be characterized by 
deviations from the initial system energy \( \Delta E/E \). In an 
idealized system, IAS15 is capable of achieving \( 10^{-28} \) 
precision, but testing of the outer solar system produces a 
more realistic \( \Delta E/E \) of \( \sim 10^{-19} \). However, this metric 
assumes that the system energy is conserved throughout the simulation 
and the current study considers the effects of extrinsic factors 
that stochastically modify the energy of the solar system itself.\(^6\)

Over sufficient time, the accumulated energy contributions 
from stellar flybys will dominate the \( \Delta E/E \) metric. Without a 
meaningful measure of the systematic integration error, we rely 
on adaptive timesteps (see Quinn et al. 1997), which 
automatically reduce the step size when the expected error 
exceeds machine precision (\( \sim 10^{-15} \)). Doing so, we acknowledge 
our inability to provide an accurate measure of this error, 
but expect \( \Delta E/E \) to be of the order \( \sim 10^{-16} \), the adaptive 
timestep limit.

2.2. The Aging Solar System

Since this study is focused on the long-term effects of the 
outer solar system, we only include Jupiter, Saturn, Uranus, 
and Neptune in our simulations. Although Mars is likely to 
survive the red-giant phase of the Sun’s evolution (Veras 2016), 
its dynamical contribution is negligible and can be ignored in 
order to reduce computational costs. By focusing on the outer 
giant planets, we are only limited by the orbital resolution of Jupiter. 
Furthermore, the orbital period of Jupiter changes as the Sun loses mass, 
increasing the orbital period by a factor of \( \sim 3.4 \) over the lifetime of the Sun. In order to take full 
advantage of this increased period, we break our simulation 
into two parts. The first part (Phase I) includes all of the stellar 
evolution, starting from the present epoch when the Sun is on 
the main sequence, continuing through the red-giant and mass-loss 
phases, and ending as the star becomes a white dwarf. The 
second part (Phase II) includes all of the subsequent temporal 
evolution, when the Sun has a fixed stellar mass and remains as 
a quiescent white dwarf.

The Phase I epoch extends from today to the epoch of final 
mass loss experienced at the end of the Sun’s planetary nebula 
phase. During this phase, \( 46\% \) of the current solar mass will be 
ejected via stellar winds. To ensure that we appropriately 
account for the orbital effects due to this mass loss, we 
implement the Modules for Experiments in Stellar Astrophysics 
(MESA; Paxton et al. 2019) solar mass evolution model 
with zero rotational velocity into our simulation. At each 
timestep (\( \Delta t = 216.63 \) days, corresponding to one-twentieth of 
the orbital period of Jupiter) we update the mass of the Sun to 
reflect the value suggested by this model. To ensure our mass 
loss is smooth, we linearly interpolate the stellar mass between 
the output profiles provided by the MESA simulations. 
Throughout most of the Phase I interval, the Sun retains a 
majority of its current mass and the orbits of the giant planets 
remain static. In the final \( \sim 1 \) Myr of this period, however, the 
Sun loses \( 0.41 M_\odot \), and the semimajor axes \( (a) \) of the giant 
planets expand accordingly. As long as solar mass loss occurs 
over many planetary orbits—which is expected to be the case 
—the orbits will experience adiabatic expansion (for a more in 
depth discussion of the adiabatic limit, see Veras et al. 2011).

The orbital expansion associated with solar mass loss 
can be understood qualitatively as follows. A well-known 
result of classical perturbation theory (see, e.g., Lichtenberg 
& Lieberman 1983) is that an oscillator, subjected to slow 
parametric changes, will preserve the ratio of its energy \( (E) \) 
to its frequency \( (\omega) \). By analogy, the adiabatic invariant associated 
with a Keplerian orbit \( (J) \) has the form

\[
J = \left| \frac{E}{n} \right| \approx \frac{GM_\odot m/2a}{\sqrt{GM_\odot/a^3}} \propto \sqrt{GM_\odot a} \approx \text{constant},
\]

which also corresponds to the first Poincaré action (see 
Morbidelli 2002). Therefore, reduction in stellar mass \( (M_\odot) \) 
will inflate the semimajor axis according to \( a \sim 1/M_\odot \). To 
ensure this orbital expansion remains within the adiabatic limit, 
we have repeated this simulation, artificially slowing down the

\(^6\) We find a slow increase in energy \( \Delta E/E \sim 10^{-18} \) after a typical flyby 
interaction. This value is consistent with the finding of Li \& Adams (2015), 
who did a similar flyby injection test. It is important to note that rare close-
encounter flybys have the ability to produce larger changes in the system 
energy.
mass-loss rate (and hence orbital expansion) by a factor of 100, and find that the orbital parameters remain consistent with results obtained from the expected real-time expansion.

After the epoch of solar mass loss concludes, and the Sun remains as a 0.54$M_{\odot}$ white dwarf, we enter the second phase of our simulations. We now adjust the timestep to become one-twentieth of the increased orbital period of Jupiter ($P_{\text{orb}} = 40.6$ yr, $\Delta t = 740.56$ days) at its new expanded orbit ($a = 9.62$ au) and lower stellar mass. This timestep increase allows for consistent orbital resolution and speeds up the simulations by a factor of 3.4.

2.3. Stellar Flyby Encounters

Throughout the simulations, we introduce stellar flybys which perturb the system, at a rate consistent with the current galactic environment of the solar system. To simulate these flybys, we draw a 10,000 au sphere around the solar system and introduce incoming stars as described below. Only flybys within this sphere are resolved in our calculation. The expected rate ($\Gamma$) for the solar system to encounter passing stars can be written in the form

$$\Gamma = \langle n_s \rangle \pi B^2 \langle v \rangle,$$

where $\langle n_s \rangle$ is the local stellar number density (0.14 pc$^{-3}$; Bovy et al. 2012; McKeel et al. 2015), $B$ is the boundary of the interaction region (10,000 au), and $\langle v \rangle$ is the expected local stellar velocity dispersion ($\sim$40 km s$^{-1}$; Binney & Tremaine 2008). With these parameter values, the expected encounter rate $\Gamma$ is about $4.2 \times 10^{-8}$ stars yr$^{-1}$. In other words, we expect a star to enter our sphere every 23 Myr. It is important to note that we use a static local galactic environment, which may change over the periods considered in the present study. Further discussion of this issue is provided in Section 3.5.2.

To simulate these flyby encounters we follow a procedure similar to that of Heisler & Tremaine (1986). We first randomly select stellar masses from the initial mass function (using the form advocated by Kroupa 2001) within a mass range of 0.08–1$M_{\odot}$ and assign a relative velocity $v_{\text{inf}}$ drawn from a Maxwell–Boltzmann distribution with a scale parameter $\langle v \rangle$. At each timestep, a new star is selected and the expected probability of the encounter is calculated ($\Gamma \Delta t$; exchanging $v_{\text{inf}}$ for $\langle v \rangle$) and an independently drawn random value $\mathcal{R}[0, 1]$ indicates whether the star will enter the sphere or not. If the star is permitted to dynamically engage with the solar system, the inclination for the flyby is drawn from a distribution of $\sin(a \mathcal{R}[0, 1])$. The angular orbital elements $\Omega$ and $\omega$ are uniformly drawn from a distribution of $\mathcal{R}[-\pi/2, \pi/2]$. The impact parameter $b$ of the flyby dictates the distance of closest approach ($r_p$) and is drawn from a distribution of $10,000 \sqrt{\mathcal{R}[0, 1]}$ au, as needed for a uniform sampling of the cross-sectional area. For a hyperbolic orbit, the perihelion $r_p$ is related to the impact parameter according to $r_p = \sqrt{a^2 + b^2} - |a|$, where $a$ is the semimajor axis of the encounter. In other words, the relationship between $r_p$ and impact parameter $b$ takes the form

$$b^2 = r_p^2 \left(1 + \frac{2GM + \langle M' \rangle}{r_p \langle v \rangle^2} \right) \approx r_p^2,$$

where $G$ is the gravitational constant, $M$ is the solar mass, and $\langle M' \rangle$ is the expected mass of the flyby star. The gravitational focusing term represents a small correction (only ~1% for $r_p = 100$ au) which is negligible for nearly all the encounters considered in this study and can be ignored.

To show that our method of randomly selecting stellar encounters produces the correct distribution, we compare our simulations to the expected time $\tau$ at which a given minimum distance of closest approach ($R_{\min}$) is achieved. The quantity $R_{\min}$ is thus the minimum value of the perihelion $r_p$ experience by the solar system as a function of time. Using the simplification of Equation (3), along with the relation $\tau = 1/\Gamma$, we can calculate $R_{\min}$ as a function of $\tau$:

$$R_{\min}(\tau) = \left(\langle n_s \rangle \pi \langle v \rangle \tau^2 \right)^{1/2}.$$

Figure 1 presents the results of this comparison and shows that our sampling procedure replicates the expected time required to reach a given minimum distance of closest approach $R_{\min}$.

Although we insert stellar flybys throughout the full integration, we find that they have little effect on the solar system during Phase I. The tightly packed planets are effectively immune to these distant perturbations during the Sun’s (relatively short) remaining lifetime as a main-sequence star. For example, the cross-section for changing the orbit of Neptune enough to create significant disruption (specifically, so that it crosses the orbit of Uranus) is of the order of 1000 au$^2$ (Laughlin & Adams 2000), which corresponds to a distance of closest approach $R_{\min} \sim 20$ au. Figure 1 shows that this value of $R_{\min}$ is not achieved until a time $\tau \geq 10^{12}$ yr, well beyond the time span of Phase I. Although more distant encounters are expected over the remainder of the Sun’s main-sequence life, these perturbations will produce only small modifications to Neptune’s eccentricity ($\Delta e \sim 0.01$). Moreover, these minor

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5 We select an upper bound of 1$M_{\odot}$ as a conservative estimate of the effects of possible stellar encounters. In practice, we found invoking this bound had little effect on the overall outcome of our simulations. Our results thus provide an upper limit, in that the inclusion of more massive stars would reduce the time needed for planetary disassociation.
perturbations will not have a significant impact on the inner gas giants. As result, the remainder of the paper focuses on the second part (Phase II) of our simulations.

3. Results

We have carried out 10 simulations of the outer solar system’s long-term evolution, with each run spanning $10^{12}$ years. While this ensemble of simulations does not constitute a large statistical sample, we find similar results in each case, indicating that the dynamical picture attained here is representative. In this section we discuss the findings of this study. In addition to determining the timescales for planetary ejection, we also elucidate the mechanisms that cause the solar system to dissolve.

3.1. Ejected Planets

After the Sun has completed its stellar evolution, including mass loss, the solar system will remain stable with the remaining planets orbiting with semimajor axes 1.85 times larger than their current values. As shown in Figure 2, the eccentricities of all the planets remain low ($e \lesssim 0.2$) during the first 10 Gyr of the Phase II era. This finding is consistent with the results of Duncan & Lissauer (1998), who found no significant eccentricity growth during the initial onset of the Sun’s white dwarf phase. However, our results begin to deviate beyond this 10 Gyr timescale. As depicted in Figure 1, the occurrence of a stellar flyby with a perihelion less than 500 au is likely within a 10 Gyr period, and such an encounter provides a significant perturbation to the system. Figure 2 indicates that these stellar encounters can significantly increase the eccentricity of the planets, and even lead to the complete disassociation for many of the planets before the 45 Gyr benchmark for stability reported by Duncan & Lissauer (1998).

In all 10 of our simulations, the four gas giants are ejected from the solar system within $10^{12}$ yr, following the end of solar mass loss. Figure 3 presents the times at which each planet was removed from each of the simulations. The overall average ejection time is roughly 65 Gyr after mass loss (72 Gyr from today). This timescale is far shorter than the $10^{18}$ yr lower bound predicted by Murray & Holman (1999) for internal instability, and shorter than the timescale of $10^{12}$ yr predicted for external perturbations with a compact configuration (Dyson 1979; Adams & Laughlin 1997). Moreover, if we only consider the first planet ejected, we find an average ejection time of about 30 Gyr. In contrast, the last planet is ejected (on average) at a time of $\sim 100$ Gyr. There is no definitive order in which the planets get removed, but typically the ice giants are removed first, with Uranus’s ejection followed by Neptune.

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8 Duncan & Lissauer (1998) acknowledged that calculations beyond 10 Gyr would require a more careful accounting of external stellar encounters, as carried out in this present study.
Saturn, and Jupiter, respectively. Usually the first three planets are all expelled within 5 Gyr of the first ejection. The remaining planet will then linger for an additional 50 Gyr before being removed from the system (see Section 3.4 for further discussion of this final planet). By accounting for the expanded Phase II planetary orbits and external perturbations from stellar flybys, we thus find a significantly reduced expected lifetime for planets to remain bound to the Sun. In the following sections, we discuss the mechanisms that drive this dissolution of the solar system.

### 3.2. Baseline Stability Considerations

Arguably the simplest way to characterize the stability of a planetary system is to require sufficient separation in units of mutual Hill radii (Chambers et al. 1996). This condition is often written in the form

$$a_2 > a_1 + AR_H, \quad (5)$$

where $a_k$ are the semimajor axes of adjacent planets and $A$ is a dimensionless constant that depends on the specific architecture of the system. Typically, one requires $A \gtrsim 10$ for the stability of multiplanet systems (Pu & Wu 2015), whereas smaller values are applicable for two-planet systems (Petit et al. 2018). The mutual Hill radius $R_H$ is given by

$$R_H = \left( \frac{a_1 + a_2}{2} \right) \left( \frac{m_1 + m_2}{3M_*} \right)^{1/3}, \quad (6)$$

where the $m_k$ denote the masses of adjacent planets and $M_*$ is the mass of the Sun. As already discussed above, $M_*$ varies over the history of the solar system, from $M_* = 1 M_\odot$ at the present epoch down to $M_* = 0.54 M_\odot$ after mass loss. For a given pair of adjacent planets, we can thus define a dimensionless parameter $\Delta$ that provides a measure of system stability, i.e.,

$$\Delta \equiv \frac{2(a_2 - a_1)}{a_2 + a_1} \left( \frac{3M_*}{m_2 + m_1} \right)^{1/3}. \quad (7)$$

During the epoch of mass loss, we expect the system to remain in the adiabatic regime so that $aM_* \approx constant$ as the Sun loses mass. As a result, the leading coefficient is invariant and the stability parameter scales according to $\Delta \sim M_*^{1/3}$. As the stellar mass decreases, the stability parameter also decreases, and the system becomes more unstable.

Applying the expected changes in stellar mass over the course of the Sun’s lifetime, one finds that the stability factor ($\Delta$) is reduced from today’s value of 8 to 6 for the Jupiter/Saturn orbital spacing. Likewise, $\Delta$ is reduced from 14 to 11 for both the Saturn/Uranus and the Uranus/Neptune stability pairing.

### 3.3. Jupiter and Saturn Resonance

Although the above discussion indicates that planet–planet interactions are expected to grow stronger due to solar mass loss, the actual source of large-scale instability remains to be identified. Remarkably, our simulations suggest that Jupiter and Saturn’s 5:2 near-commensurability may provide the mechanism that triggers this large-scale instability.

To make this argument, we consider the resonance angle $\phi = 5\lambda_{Saturn} - 2\lambda_{Jupiter} - 3\varpi_{Saturn}$, where $\lambda$ is the mean longitude and $\varpi$ is the longitude of pericentre for the respective orbits. Slow circulation of this angle indicates planets are near (but not in) MMR. In other words, if the planets are not in MMR, the conjunction position will continuously move along the orbit. This behavior is characteristic of Jupiter and Saturn’s present-day configuration. In the top panel of Figure 4, we show the circulation of the Jupiter/Saturn resonant angle as seen today. The period of this circulation is about 900 yr and is directly associated with the modulation in semimajor axes known as the great inequality (see Section 4.1 for further discussion).

As the Sun loses mass, the semimajor axis ratio is conserved, but the width of the 5:2 MMR expands as $\sim M_*^{-1/2}$ (Henrard et al. 1986). This growth leads to the adiabatic capture of Jupiter and Saturn into the 5:2 MMR. In such a capture, the resonant angle will transition from circulation to libration. Our simulations indicate that this transition occurs during the final $\sim 1$ Myr of mass loss (see the during mass-loss panel of Figure 4). Once the planets have been successfully captured, the resonant angle will execute bounded oscillations (see the after mass-loss panel of Figure 4). In isolation, this orbital configuration is stable on long timescales, as discussed in Duncan & Lissauer (1998). However, the inflated orbits and weakened gravitational pull from the reduced solar mass render this new configuration more vulnerable to perturbations from stellar flybys.

The large amplitude ($\sim 100^\circ$) of the libration seen in the after mass-loss panel of Figure 4 is indicative of a MMR system near the separatrix (i.e., the MMR boundary, where the resonant angle changes from libration to circulation). For roughly 30 Gyr after solar mass loss, the system remains stable. Given sufficient time, however, a close encounter ($R_{min} < 500$ au) from a stellar flyby will create a perturbation large enough to perturb Jupiter and Saturn into the chaotic region of the 5:2 MMR (see Morbidelli 2002). This event triggers chaotic diffusion as shown in the bottom panel of Figure 4, leading to large-scale instabilities in the outer solar system. In most cases the Jupiter/Saturn resonance angle will repeatedly switch from circulation to libration and back, pumping up the eccentricity of Uranus, Neptune, and Saturn until they are ejected from the solar system. This process takes place over $\sim 10$ Gyr after the onset of accelerated chaos. Jupiter is usually the last planet remaining, but this ordering is not the only possible outcome. In one case Saturn was the final remaining planet and in another Neptune survived the tumultuous instability of the inner gas giants. However with this large-scale instability
played out, one planet remains orbiting the Sun for an extended period of time in most of the simulations.

3.4. The Last Planet Standing

In all but one simulation we found that a single planet remains orbiting the Sun for about 50 Gyr. (In the exceptional case, orbital diffusion left Jupiter with an eccentricity greater than \( e \sim 0.9 \), leading to a swift ejection, 200 Myr later.) With an absence of additional planets, the surviving planet lacks a direct mechanism to attain positive energy. The only remaining source of energy exchange is through interactions with passing stars. Significantly, the large-scale instability that led to the ejection of the other three gas giants leaves the final planet with a heightened eccentricity (typically in the range \( e \sim 0.2 \text{–} 0.5 \)). As shown by Li & Adams (2015), the dynamical cross-section required for ejection is an exponential function of eccentricity. As a result, the expected timescale for ejection of the post-instability gas giant is decreased by roughly a factor of two (compared to the planetary orbit before the onset of instability).

Since flyby encounters are rare (entering the 10,000 au sphere once every 23 Myr), and most interactions will have small dynamical effects on the remaining planet, the process of ejection can in principle occur steadily (e.g., through incremental increases in orbital eccentricity and semimajor axes). On the other hand, given sufficient time, it is also possible that an extremely close encounter will independently liberate the final planet. The underlying mechanism for the removal of the final planet thus represents a competition between these two processes. In other words, will the final planet be ejected by a single major event or many small energy exchanges?

To intuitively understand the timescales over which these possible outcomes take place, we can crudely describe the process as a random walk. The follow derivation is merely an order of magnitude calculation, similar to the dynamical relaxation time derived by Binney & Tremaine (2008). The cumulative change in velocity of \( \Delta U \) is given by

\[
\Delta U \sim \Delta u \sqrt{N},
\]

where \( \Delta u \) is the change in the velocity of the orbiting planet from a single interaction and \( N \) represents the number of interactions. The planet must attain a positive energy in order to be disassociated from the system. To leading order, this requirement can be written in the form

\[
\left( \frac{\Delta U}{2} \right)^2 \approx \frac{GM_\odot}{2a}.
\]

Assuming all interactions follow the impulse approximation, we can express \( \Delta u \) as

\[
\Delta u = \frac{2GM_\odot}{vb},
\]

where \( v \) is the velocity of the flyby star and \( b \) is the impact parameter for the stellar encounter. In reality, most interactions are much weaker and produce smaller changes in the planets orbital velocity. As a result, this calculation will provide a limiting constraint. Recalling Equation (2), we note that the
number of interactions can be calculated as \( N = \Gamma t \), where \( t \) is the expected time needed to achieve \( N \) interactions. Finally, we can use Equation (8) to solve for the time required to eject the planet, i.e.,

\[
t = \frac{v^2}{4\pi GM_s a \left< n_s \right> (v)}. \tag{11}
\]

Note that this equation is independent of the flyby impact parameter.\(^9\) Under the assumption that the impulse approximation is valid, we find that a single planet is equally likely to be ejected by a series of distant encounters or by a single close-encounter flyby. However, this approximation breaks down for adiabatic interactions, where the gravitational interaction timescale \( (T_{\text{enc}}) \) is much greater than the planet’s orbital period \( (P) \),

\[
T_{\text{enc}} \sim \frac{2b}{v} \gg P. \tag{12}
\]

In typical cases, the star will enter the 10,000 au sphere of influence with a velocity of the order of 40 km s\(^{-1}\) and an impact parameter of the order of 7000 au. These values indicate an interaction time of \( T_{\text{enc}} \sim 1800 \) yr, which is nearly five times greater than Neptune’s Phase II period (350 years). Most interactions that occur after the large-scale instability has isolated a single planet will thus be adiabatic. In these cases, where perturbations are effectively secular, semimajor axis growth will be suppressed (Batygin et al. 2020) relative to predictions from the impulse approximation. In other words, the distant encounters will be weaker than the limiting case of Equation (10). As a result, a single extreme encounter is more likely to liberate the final planet than the accumulated effects of many distant perturbers. Nonetheless, both the cumulative effects of many distant encounters and the impact of rare close encounters are likely to play a role.

In Figure 5 we show how the eccentricity and semimajor axis of our simulated final planet changes before eventual disassociation. From the onset of isolation, nearly all cases show slow orbital diffusion due to weak gravitational interactions. Providing consistency with the above discussion, four of the simulations show the eventual disassociation of the final planet by a single extremely close flyby encounter (with \( R_{\min} < 200 \) au). In the remaining simulations, a close encounter significantly modifies the orbit, making the planet far more susceptible to perturbations from subsequent stellar flybys. After surviving this initial interaction, the final planet is ejected within a few gigayears. In both scenarios, we find that both major and minor energy exchanges play a role in the removal of the final planet. However, the majority of the liberating energy comes from a single close encounter.

Figure 5. The eccentricities and semimajor axes of the final planet, which remains bound after all of the other planets have been ejected from the system. The slow rise in eccentricity and semimajor axis are due to energy exchanges with multiple stellar flybys, but an important mechanism for disassociation of the final planet is an extremely close flyby encounter (with \( R_{\min} \leq 200 \) au).

3.5. Caveats

3.5.1. Binary Encounters

Within the current study we assume all flybys are single star encounters, in reality, however, observations indicate that roughly half of all nearby stars are part of a binary system (e.g., Raghavan et al. 2010; Duchêne & Kraus 2013). Previous dynamical studies, considering these binary interactions, have found that the two stars will appear as independent perturbers when the relative flyby velocity \( (v) \) of the binary system is sufficiently large, increasing the interaction cross-section by roughly a factor of 2 (Laughlin & Adams 2000; Li & Adams 2015). However, when the orbital velocity of the interacting planet and the velocity between binary members are near the relative flyby velocity, the gravitational influence is enhanced. In this scenario, the extended interaction time allows motion from the binary system orbit to increase the gravitational cross-section (by a factor greater than 2), thereby increasing the likelihood of a significant perturbation.

The 40 km s\(^{-1}\) expected flyby velocity used for this study is quite large compared to the orbital speeds of the gas giants.

\(^9\) A similar calculation can be achieved through a random walk process of the orbital energy, under the assumption of the impulse approximation, culminating in a solution that deviates by only a factor of 4 from Equation (11).
large-scale instability. The planet’s increased eccentricity and expanded semimajor axis enhances the subsequent dynamical interactions due to stellar encounters, which lead to the expeditious disassociation of the final gas giant.

4.1. Historical Context

It is worth noting that speculation regarding large-scale instabilities within the solar system that are driven by the 5:2 resonant interaction between Jupiter and Saturn dates back to the work of Newton (1687, 1713, 1726), as well as the pioneering development of perturbation theory by Laplace (1799–1825). Accordingly, let us contextualize our results against the backdrop of this remarkable saga.

The earliest data on the orbits of Jupiter and Saturn date back to Ptolemy in 228 BC (Saturn) and 240 BC (Jupiter) and were not recorded again until 1590 (nearly 1800 yr later) when Tycho Brahe measured their orbital positions (see the account of Laskar 1996). When Kepler set about mapping the elliptical orbits of our solar system, in 1625, he was unable to reconcile the data collected by Ptolemy, using the model derived from Tycho Brahe’s observations. He noted that Ptolemy’s observation required a slower mean motion for Jupiter and a faster mean motion for Saturn (Wilson 1985). In an effort to understand this discrepancy, Halley (1676) extrapolated the semimajor axes of planets over this period of time and determined Jupiter was slowly moving inward while Saturn’s orbit was moving outward (Laskar 1996). At face value, the data implied that given sufficient time Jupiter would collide with the Sun and Saturn would expand out into deep space. When Newton (1687) announced the universal law of gravity, he indicated that such deviations from the invariant elliptical orbits were due to mutual gravitational interactions. His belief was that such perturbations, lacking divine intervention, would eventually lead to the demise of the solar system (Laskar 1996). However, Newton was not able to directly explain these observations since perturbation theory had not yet been developed.

Laplace (1799–1825) finally explained this phenomenon, which became known as the “great inequality,” by showing this was not a secular effect, but rather a periodic modulation of Jupiter and Saturn’s semimajor axes by the 5:2 near resonance. As seen in the top panel of Figure 4, this nearly 900 yr circulation leads to a cyclical expansion and contraction of the Jovian and Saturnian orbits, and appears to be stable for the remainder of the Sun’s life as a main-sequence star.

Our simulations show that the mechanism responsible for the eventual disintegration of the outer solar system is keenly related to the great inequality. That is, the expanded orbits during Phase II (after solar mass loss) allow the planets to be captured in a 5:2 MMR, producing large-scale instability when perturbed by stellar flybys. As a result, Jupiter and Saturn appear to be responsible for the ultimate demise of the solar system, only at a much later time than originally prophesied by Newton.

4.2. Free-floating Planets

Once liberated, the outer solar system planets will independently roam through the Galaxy, becoming free-floating planets. Current estimates, using gravitational microlensing, suggest there are less than about 0.25 free-floating planets for every main-sequence star in the Galaxy (Mróz et al. 2017). The

Note that if the binary fraction is 1/2, then the factor by which the cross-section is increased would be 3/2.
exact origin of these abundant planets remains unclear, but it is probable that a large fraction of these free-floating planets were once bound to a host star and disassociated via dynamical instability (Ford et al. 2003).

Here, we have shown that the outer gas giants of the solar system will eventually be ejected and contribute to this reservoir of free-floating planets. Given that all stars will experience some amount of mass loss over their lifetimes, it is likely that other planetary systems—that survive stellar evolution—will also eventually experience large-scale instability (e.g., Veras et al. 2011). We can therefore conclude that as the Galaxy, and the stars that reside in it, continue to age the number of free-floating planets will increase as a function of time.

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