Physical Design using Differentiable Learned Simulators

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Abstract
Designing physical artifacts that serve a purpose—such as tools and other functional structures—is central to engineering as well as everyday human behavior. Though automating design has tremendous promise, general-purpose methods do not yet exist. Here we explore a simple, fast, and robust approach to inverse design which combines learned forward simulators based on graph neural networks with gradient-based design optimization. Our approach solves high-dimensional problems with complex physical dynamics, including designing surfaces and tools to manipulate fluid flows and optimizing the shape of an airfoil to minimize drag. This framework produces high-quality designs by propagating gradients through trajectories of hundreds of steps, even when using models that were pre-trained for single-step predictions on data substantially different from the design tasks. In our fluid manipulation tasks, the resulting designs outperformed those found by sampling-based optimization techniques. In airfoil design, they matched the quality of those obtained with a specialized solver. Our results suggest that despite some remaining challenges, machine learning-based simulators are maturing to the point where they can support general-purpose design optimization across a variety of domains.

1. Introduction
Humans are creators. Our ancestors created stone tools which led to innovations in hunting and food consumption, aqueducts and irrigation systems which revolutionized farming and urban habitation, and more recently, airplanes which let us cross the globe in hours. Automatically designing objects to exhibit a desired property—often referred to as inverse design—promises to transform science and engineering, including aerodynamics (Eppler, 2012), material design (Butler et al., 2016), optics (Colburn & Majumdar, 2021), and robotics (Gupta et al., 2021; Xu et al., 2021).

Despite its promise, widespread practice of inverse design has been limited by the availability of fast, general-purpose simulators. In science and engineering, many methods rely on specialized “classical” solvers, which are handcrafted to simulate a particular physical process. While accurate and reliable, these solvers can be quite slow, may not provide gradients, and are narrow in their applicability (Cranmer et al., 2020). In robotics and reinforcement learning, simulators are often learned, but accumulate errors over long time horizons and often struggle to generalize beyond their training data (Janner et al., 2019; Talvitie, 2014; Venkatraman et al., 2015), making them unsuitable for design optimization without further finetuning.

Recently, a class of learned physics simulators based on graph neural networks (GNNs) has been proposed (Pfaff et al., 2021; Sanchez-Gonzalez et al., 2020). These models have shown success in general-purpose physical prediction, exhibiting high accuracy and generalization ability. However, this may still be insufficient for inverse design problems, as optimizers can exploit regions in the state space where model predictions are unreliable (Lutter et al., 2021). Models must therefore be more than just accurate overall: they must also be robust and smooth. It is unknown whether GNN-based physics simulators exhibit these properties.

In this paper, we optimize physical designs by performing gradient descent through pretrained, GNN-based, state-of-the-art learned simulators. We use this approach to perform successful inverse design without requiring further finetuning of the simulator. Across two high-dimensional fluid manipulation tasks (2D FLUID TOOLS and 3D WATER-COURSE) and a design task from aerodynamics (AIRFOIL), we show that learned simulators: (1) produce high-quality designs across diverse physical tasks with complex particle- or mesh-based physics, while using the same underlying GNN architecture; (2) generalize sufficiently to permit designs far outside their training data; (3) support gradient-based optimization over hundreds of time steps, through states with thousands of particles, in tasks with up to 625
design parameters (and as a result, produce better designs than sampling-based optimization using a classical simulator); and (4) can be much faster than specialized simulators used in engineering, while generating designs of similar quality. Overall, our results are a proof-of-concept for how state-of-the-art learned simulators can be used at scale to optimize designs for different physical tasks.

2. Background

Solving inverse problems with physical simulators has a long history in science and engineering, spanning data assimilation for weather modeling (Navon, 2009), system identification in robotics (Guevara et al., 2017; Seita et al., 2020), and tomographic and geophysical imaging (Cui et al., 2016; Pascual-Marqui, 1999). Inverse design can be framed as an inverse problem in which the objective is to optimize design parameters to produce some desired target property. Simulation-based inverse design has been studied in a variety of disciplines, including nanophotonics (Molesky et al., 2018), material science (Dijkstra & Luijten, 2021), mechanical design (Coros et al., 2013), and aerodynamics (Anderson & Venkatakrishnan, 1999; Rhie, 1983).

Classical numerical solvers used for inverse design can be highly accurate, but are often inefficient (making sampling-based inference methods infeasible for high-dimensional designs) and domain-specialized (prohibiting general-purpose inverse design across domains (Choi et al., 2021)). Differentiable simulators (Freeman et al., 2021; Hu et al., 2019; Schenck & Fox, 2018) have recently garnered attention, as they allow for more sample-efficient gradient-based optimization. However, like classical solvers, they are still typically narrow in application scope, as many simulation techniques are hard to express as a differentiable program (e.g. constraint dynamics and multiphysics coupling).

The last few years have seen increased interest in using machine learning to accelerate inverse design across a variety of applications (e.g. Challapalli et al., 2021; Christensen et al., 2020; Gómez-Bombarelli et al., 2018; Forte et al., 2022; Hoyer et al., 2019; Kumar et al., 2020; Li et al., 2020a; Liu et al., 2018; Sha et al., 2021; Zheng et al., 2021). These methods can provide impressive speedups over classical approaches by using learned generative models to propose designs (thus calling an expensive simulator fewer times), or by learning a scoring function which maps designs to target values (replacing the simulator entirely). However, they are based on components with limited out-of-domain generalization, restricting new designs to configurations near the training data or requiring model refinement.

We instead propose to replace classical simulators with learned simulators. Like learned scoring functions, learned simulators can be faster than classical simulators (Kochkov et al., 2021; Stachenfeld et al., 2021) and are differentiable when parameterized as a neural network. Furthermore,
learned simulators mimic the underlying physical dynamics independent of the design task and are therefore more likely to generalize. Physics simulators have been successfully implemented as learned, differentiable models of complex dynamics such as fluids, rigid-body interactions, and soft-body systems (Bhatnagar et al., 2019; Li et al., 2020b; Mrowca et al., 2018; Rudy et al., 2017; Thuerey et al., 2020; Ummenhofer et al., 2020; Wang et al., 2020). Graph-based models in particular are promising candidates for design problems, having demonstrated high accuracy, stability, efficiency, and generalization performance (Belbute-Peres et al., 2020; Pfaff et al., 2021; Sanchez-Gonzalez et al., 2020).

However, high-quality forward models do not necessarily translate into better downstream task performance (Hamrick et al., 2020; Lutter et al., 2021). While learned simulators have been used successfully for planning and control in low-dimensional state spaces (up to 21 degrees of freedom (DOF), e.g. Bharadhwaj et al., 2020; Sanchez-Gonzalez et al., 2018; Wang et al., 2019) or more complex domains with small action spaces (6 DOF, e.g. Li et al., 2019), they require replanning at every timestep to avoid error accumulation. It is not known whether learned simulators can support gradient-based, high-dimensional inverse design which demands high accuracy, well-behaved gradients, long-term rollout stability, and generalization beyond the training data.

Here we study inverse design in non-rigid, graph-based physical systems with up to 625 design dimensions and 2000 state dimensions, over 50–300 timesteps, without replanning after the design period. We show that using gradient-based optimization with learned, general-purpose simulators is an effective choice for inverse design.

3. Problem Formulation

Consider the design task depicted in Figure 1, in which the goal is to direct a stream of water (shown in blue) into two “pools” (shown in purple) by designing a “landscape” (shown in green) parameterized as a 2D height field. Here, an ideal design will create ridges and valleys that direct fluid into the two targets. In the next sections, we formalize what it means to find and evaluate such a design and discuss our choices for simulator and optimizer.

3.1. Learned simulators

To demonstrate the utility of learned simulators for finding physical designs, we rely on the recently developed MeshGraphNets model (Pfaff et al., 2021), which is an extension of the GNS model for particle simulation (Sanchez-Gonzalez et al., 2020). MeshGraphNets is a type of message-passing graph neural network (GNN) that performs both edge and node updates (Battaglia et al., 2018; Gilmer et al., 2017), and which was designed specifically for physics simulation. Here, we briefly summarize how the learned simulator works, and refer interested readers to the original papers for details.

We consider simulations over physical states represented as graphs \( G \in \mathcal{G} \). The state \( G = (V, E) \) has nodes \( V \) connected by edges \( E \), where each node \( v \in V \) is associated with a position \( u_v \) and additional dynamical quantities \( q_v \). These graphs may be either meshes (as in MeshGraphNets) or particle systems (as in GNS). In a mesh-based system (such as AIRFOIL), \( V \) and \( E \) correspond to vertices and edges in the mesh, respectively. In a particle system (such as 2D FLUID TOOLS), each node corresponds to a particle and edges are computed dynamically based on proximity. Under this framework, we can also consider hybrid mesh-particle systems (such as 3D WATERCOURSE). See Appendix B for model implementation details, and Appendix C for details on the representation used for each domain.

The simulation dynamics are given by a “ground-truth” simulator \( f_S : \mathcal{G} \rightarrow \mathcal{G} \) which maps the state at time \( t \) to that at time \( t + \Delta t \). The simulator \( f_S \) can be applied iteratively over \( K \) time steps to yield a trajectory of states, or a “rollout,” which we denote \((G^{t_0}, ..., G^{t_K})\). Using MeshGraphNets, we learn an approximation \( f_M \) of the ground-truth simulator \( f_S \). The learned simulator \( f_M \) can be similarly applied to produce rollouts \((\tilde{G}^{t_0}, \tilde{G}^{t_1}, ..., \tilde{G}^{t_K})\), where \( \tilde{G}^{t_0} = G^{t_0} \) represents initial conditions given as input. We note that a learned simulator allows us to take much larger time-steps than \( f_S \), permitting shorter rollout lengths: in our running example, one model step corresponds to 200 internal steps of the classical simulator. See Figure 1a for an illustration of simulation using a learned model.

3.2. Optimizing design parameters

To optimize a physical design, we leverage the pipeline shown in Figure 1: (1) transform design parameters into an initial scene, (2) simulate the scene using \( f_M \) or \( f_S \), (3) evaluate how well the simulation achieves the desired behavior, and (4) adjust the design parameters accordingly.

**Design parameters** To produce the initial state \( G^{t_0} \), we introduce a differentiable design function \( f_D : \Phi \times A \rightarrow \mathcal{G} \) which maps design parameters \( \phi \in \Phi \) and other initial conditions \( \alpha \in A \) to an initial state: \( G^{t_0} = f_D(\phi, \alpha) \). In our landscape design task (Figure 1), \( \phi \) is the 2D height field of the mesh, while \( \alpha \) is the non-controllable objects in the scene like the initial position of the fluid.

**Maximizing reward** The reward function \( f_R : \mathcal{G} \times \Theta_R \rightarrow \mathbb{R} \) maps the final state of a length-\( K \) trajectory \((G^{t_K} \text{ or } \tilde{G}^{t_K})\) and parameters \( \Theta_R \in \Theta_R \) to a scalar value. In our running example, the reward function is defined as the Gaussian likelihood of each fluid particle under the closest “pool”, averaged across particles (Figure 1a).
We define the full objective under the ground-truth simulator \( f_S \) as \( J_S(\phi) := f_R(f_S^{(K)}(f_D(\phi, \alpha); \theta_R)) \), where \( f_S^{(K)} \) indicates \( K \) applications of the simulator. We want to find the design parameters that maximize \( J_S \), i.e. \( \phi^* = \arg\max_\phi J_S(\phi) \). We can approximate this optimization using a learned simulation model instead by maximizing 
\[ J_M(\phi) := f_R(f_M^{(K)}(f_D(\phi, \alpha); \theta_R)) \].

**Optimizers** Optimal design parameters \( \phi^* \) can be found using any generic optimization technique. Given the differentiability of the learned simulator \( f_M \), we are particularly interested in evaluating gradient-based optimization, which requires fewer function evaluations and scales better to large design spaces than sampling-based techniques (Bharadhwaj et al., 2020). We focus on the Adam optimizer (Kingma & Ba, 2015), which we use to find \( \phi^* \) by computing the gradient \( \nabla_\phi J_M(\phi) \). This involves backpropagating gradients through the reward function \( f_R \), length-\( K \) rollout produced by \( f_M^{(K)} \), and design function \( f_D \).

As a baseline, we consider the cross-entropy method (CEM) (Rubinstein & Kroese, 2004), a gradient-free sampling-based technique that is popular in model-based control (Chua et al., 2018; Wang et al., 2019). CEM can be used with any simulator and works by sampling a population of candidates for \( \phi \) and evolving them to maximize the reward. However, CEM requires multiple evaluations of \( f_M \) or \( f_S \) per optimizer step (depending on the population size, which is 20–40), whereas Adam considers only a single candidate \( \phi \) and thus only a single evaluation per step.

Across our design tasks (Section 4) we compared: gradient descent with the learned simulator (GD-M), CEM with the learned simulator (CEM-M), and CEM with the ground-truth simulator (CEM-S). In all tasks, \( f_S \) is non-differentiable, preventing a comparison to GD-S. However, in the special case of AIRFOIL, we compare to DAFOam (He et al., 2020), a specialized solver which computes gradients with the adjoint method.

**Evaluation** Unless otherwise noted, we always evaluate the quality of an optimized design \( \phi^* \) using the ground-truth objective \( J_S(\phi^*) \), regardless of whether \( \phi^* \) was found using the learned model \( f_M \) (as in CEM-M and GD-M) or the ground-truth simulator \( f_S \) (as in CEM-S). We also use rollouts from \( f_S \) to produce visualizations in the figures.

### 4. Design Tasks

We formulated a set of design tasks across three different physical domains with high-dimensional state spaces and
To evaluate higher-dimensional inverse design, we created a 2D FTOOLS dataset released by Sanchez-Gonzalez et al. (2020) which contains scenes with 1–4 straight line segments; no curved lines or large numbers of obstacles are shown. The learned model uses the solver in Hu et al. (2018) to generate trajectories. Lift and drag coefficients are computed by running an aerodynamics simulation on a 4158 node mesh. The learned model \( f_M \) is trained on data generated from the simulator in Bender & Koschier (2015).

4.3. AIRFOIL

Shape optimization in aerodynamics is one area where gradient-based optimization is routinely applied using traditional simulators (Buckley et al., 2010). Here, we consider the well-studied task of drag optimization of a 2D airfoil profile (Figure 4, Section C.3). In this task, a wing is defined as optimizing the wing shape to minimize drag under certain constraints such as constant lift and bounds on the shape to prevent degenerate (e.g. infinitely thin) configurations. Lift and drag coefficients are computed by running an aerodynamics simulation on a 4158 node mesh. The learned model \( f_M \) is trained on data generated from the ground-truth simulator \( f_S \), for which we use the OpenFOAM solver (OpenCFD Ltd, 2021).

4.4. Model learning and optimization

While each domain has a different state space structure and uses a different ground-truth simulator \( f_S \), the learned simulators \( f_M \) all share the same architecture (with identical

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**Figure 3.** 3D WATERCOURSE results found with GD-M or CEM-M and evaluated with \( f_S \). Simulations use up to 2000 particles and 625 design parameters. The heightmap of a 2D landscape is optimized to redirect the fluid towards the purple targets; birds eye views of heightmaps are shown in the upper right corner of each subplot. In this high-dimensional task domain, GD finds designs with high reward (a-c), while CEM fails to find meaningful solutions for 2 Pools (d) as well as the other tasks (Figure A.5a). Each subplot reports the mean reward (\( r \)) and bootstrapped [lower, upper] 95% confidence intervals for the corresponding optimizer and task (averaged over 10 randomized initial designs for each task variation, see Section C.2).
5. Results

Our results show that learned simulators can be used to effectively optimize various designs despite significant domain shift and long rollout lengths. The same underlying model architecture is used for each domain, highlighting the generality of learned simulators for design. Examples of designs found with our approach are available at: https://sites.google.com/view/optimizing-designs. Performance is always evaluated using the ground-truth simulator \( f_S \) (see Section 3.2).

Here we discuss these results, and compare the capabilities of gradient descent with learned simulators over classical simulators and sampling-based optimization techniques.

5.1. Overall results

We first asked whether a learned simulator combined with gradient descent (GD-M) could produce good-quality designs at all. This approach might fail in various ways: accumulating model error, vanishing or exploding gradients (Bengio et al., 1994), or domain shift (Hamrick et al., 2020). However, as the following results show, GD-M produced high-quality designs across all three domains.

Figure 2a shows qualitative results for 2D FLUID TOOLS (Section 4.1), where our approach (GD-M) produces intuitive, functional designs to contain (Contain), transport (Ramp), or funnel (Maze) the fluid to a target location. On average, GD-M outperforms CEM-M by 16.1–118.9%, indicating a substantial benefit of gradient-based optimization. GD-M also outperforms CEM-S by 3.9–37.5%, despite using a learned simulator rather than the ground-truth. However, these design spaces are still relatively small (between 16 and 36 dimensions). In 3D WATERCOURSE (Section 4.2), we substantially increase the dimensionality to a 625-dimensional landscape. Here, GD-M produces robust designs, creating ridges to re-route water in particular directions or valleys to direct water into pools (Figure 3a-c). In comparison, CEM-M cannot solve any of these tasks, with performance 30–85× worse than GD-M.

In AIRFOIL (Figure 4b), GD-M recovers the characteristic S-curve shape for a low-drag airfoil under a small angle of attack, and matches the design obtained with DAFOam, an adjoint aerodynamics solver which computes close-to-optimal designs for this task. Specifically, the design obtained with DAFOam yields a drag coefficient of 0.01902, while GD-M finds designs with drag between 0.01898–0.01919 depending on ensemble size (see Section 5.4). Importantly, DAFOam’s solver and optimizer are highly specialized for the particular task of airfoil design, while our approach is more general-purpose in that it requires only trajectory data for training and a generic gradient-based optimizer.

5.2. Model stability & gradient quality

We investigated accuracy over long timescales by measuring the effect of rollout length on design quality in 2D FLUID TOOLS (Figure 5a and A.10). Longer rollouts can in principle allow for higher reward in this task as they give the fluid time to settle; however, with learned models, they can also be unstable due to error accumulation (Talvitie, 2014; Venkatraman et al., 2015). Nevertheless, we find that the learned simulator does not seem to be severely impacted by this problem. Specifically, the quality of designs found by GD-M increases up to 225 steps (Figure 5a), indicating that the learned simulator’s accuracy and gradients remain stable for a surprisingly long time. Across the episode lengths eval-
5.3. Generalization

Deep networks often struggle to generalize far from their training data (Geirhos et al., 2018). This poses a problem for design: to produce in-distribution training data, we would already need to know what good designs look like, thus defeating the aim of wanting to find new designs. However, we find that the GNN-based simulators studied here overcome this issue. As noted in Section 4.1, the learned simulator for 2D FLUID TOOLS was trained on a pre-existing, highly simplified dataset where only one to four straight line segments interact with a fluid (Appendix C). In contrast, the design tasks studied here involve highly articulated, curved obstacles (Contain, Ramp) or a larger number of obstacles (Maze); yet, GD-M still discovers effective designs without requiring any finetuning (Figure 2). We suspect this is because the model is trained to learn local collision rules, making it more robust to global distribution shift.

Using a learned simulator trained in a relatively simple environment has another unexpected advantage. In rare cases, classical simulators suffer from degenerate behavior around certain edge cases. For example, with the classical simulator for 2D FLUID TOOLS, particles can get stuck in between joint segments, especially when there are a large number of parts or joints (Figure A.9). However, since the learned simulator was trained on simpler data where these effects are unobserved, it picks up only on the appropriate collision performance and not the unrealistic edge cases. Thus, the learned simulator produces more plausible rollouts than the classical simulator in these cases, and might therefore be a better candidate for producing designs that would transfer to the real world.

5.4. Improving accuracy with ensembles

In engineering tasks like AIRFOIL, simulators must be especially accurate, as small differences in the predicted pressure field can cause large errors in lift and drag coefficients. While GD-M (without ensembling) can produce designs close to DAFoam’s, we notice a slightly rounder wing front (Figure 4b, bottom) causing a small increase in drag (0.01919 versus 0.01902 in DAFoam).

To further improve performance, we implemented an ensemble of learned simulators trained on separate splits of the training set. Ensembles are a popular choice for training transition models for use in control (Chua et al., 2018), as they can provide higher quality predictions and are more resistant to delusions—a particularly problematic issue for accuracy-sensitive domains such as airfoil design. During optimization, we make predictions with all models in the ensemble, each trained on a different data split, and average the gradients. As shown in Figure 4b-c, larger ensembles yield designs with significantly lower drag ($\beta = -5.3 \times 10^{-5}$, $p = 0.0003$, where $\beta$ is a linear regression coefficient) and higher overall reward ($\beta = 5.6 \times 10^{-5}$, $p = 0.0001$), and are able to produce designs very close to the solution found by DAFoam, with a drag coefficient of 0.01898 (size-5 ensemble). Thus, with ensembles, we are able to achieve performant designs with a general-purpose learned simulator, indicating that we can use learned models for design optimization in spaces traditionally reserved for specialized solvers like DAFoam.
5.5. Scalability to larger design spaces

In larger design spaces, sampling-based optimization procedures quickly become intractable, especially with relatively slow simulators. We hypothesized that gradient descent with fast, learned simulators could overcome this issue, especially as the size of the design space is increased. We therefore compared different optimizers on 2D FLUID TOOLS as a function of the dimensionality of the design space (the number of tool joints in Contain or the number of tools in Maze) and on the higher-dimensional 3D WATERCOURSE.

For Contain (Figure 5b), the performance of GD-M increases with the number of joints as increasingly fine grained solutions are made possible. In contrast, for both CEM-M and CEM-S, design quality deteriorates with more joints as high quality solutions become harder to find with random sampling. In the highest dimensional Contain task with 48 tool joints, GD-M outperforms CEM-M by 154.9% and CEM-S by 126.5%. When CEM does find solutions (Figure 2b), they lack global coherence and appear more jagged than solutions found with GD-M. Similarly, for Maze (Figure 5c), the performance of GD-M is largely unaffected by the number of tools, while the performance of CEM-M and CEM-S both degrade as the design space grows. For the highest dimensional Maze problem with 36 joints, GD-M outperforms CEM-M by 207.4% and CEM-S by 133.7%.

In the 625-dimensional 3D WATERCOURSE task, CEM-M performs 30–85× worse than GD-M (Figure 3) despite extensive hyperparameter tuning. This trend held across all tasks (Figure A.5a), and even persisted when using fewer control points in the design space (Figure A.6). This is due not only to 3D WATERCOURSE’s larger design space, but also because this problem requires a globally coherent solution: modifying small areas independently is unlikely to have much effect on the global movement of the fluid.

5.6. Model speed and sample efficiency

Learned simulators can provide large speedups over traditional simulators in certain domains by learning to compensate for coarser sub-stepping and making optimal use of hardware acceleration. In AIRFOIL, although we use a very simple GD setup, our approach is able to find very similar designs as DAFOam’s specialized optimizer. Moreover, our approach requires only 21s (single model) to 62s (size-5 ensemble) on a single A100 GPU, compared to 1021s for DAFOam run on an 8-core workstation, despite requiring 10× more optimization steps.

In 2D FLUID TOOLS, the ground-truth simulator runs at a similar speed to the learned model (see Sanchez-Gonzalez et al., 2020), but is non-differentiable and therefore depends on more expensive gradient-free optimization techniques which require more function evaluations. We use 20–40 function evaluations per optimization step of CEM, compared to a single evaluation with GD (which is about 3× more costly, due to the gradient computation). Thus, GD with a differentiable learned model can be much more efficient than using the ground truth simulator with a sampling-based method.

6. Discussion

We used state-of-the-art learned, differentiable physics simulators with gradient-based optimization to solve challenging inverse design problems. Across three domains and seven tasks, which involved designing landscapes and tools to control water flows or optimizing the shape of an airfoil, we demonstrated that gradient descent with pre-trained simulators can discover high-quality designs that match or exceed the quality of those found using alternative methods. This approach succeeds in a variety of interesting and surprising ways: it permits gradient backpropagation through complex physical trajectories for hundreds of steps; scales to tasks with large design and state spaces (100s and 1000s of dimensions, respectively); and successfully generates designs which require the learned simulator to generalize far beyond its training data. In the classic aerodynamics problem of airfoil shape optimization, our approach produces a design comparable to that of a specialized solver using only simple, general-purpose strategies like model ensembling.

While our results have exciting implications for inverse design, they also open up possibilities for explaining everyday human behavior like tool invention—a longstanding puzzle in cognitive science (Allen et al., 2020; Osiurak & Badets, 2016; Shumaker et al., 2011). With general-purpose learned simulators, we have the potential not just to create highly specialized tools in engineering domains, but also to model everyday tool creation, such as creating a hook from a pipe cleaner, building a blanket fort, or folding a paper boat.

Our approach has limitations that should be visited in future work. Gradient descent is inappropriate for design spaces with regions of zero gradients, such as in 2D FLUID TOOLS tasks where the fluid may not always make contact with the tool (Figure A.8). Many interesting design tasks also have variably-sized or combinatorial design spaces that cannot easily be optimized with gradient descent, such as computer-aided design (CAD) approaches to 3D modeling. An exciting future direction will be to integrate general-purpose learned simulators with hybrid optimization techniques such as those used in material science and robotics (Chen & Gu, 2020; Toussaint et al., 2018). As learned simulators continue to improve, we could also use them to do even broader cross-domain, multi-physics design. While challenges remain, our results represent a promising step towards faster and more general-purpose inverse design.
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Zheng, L., Kumar, S., and Kochmann, D. M. Data-driven topology optimization of spinodoid metamaterials with seamlessly tunable anisotropy. *Computer Methods in Applied Mechanics and Engineering*, 383:113894, 2021. Cited on page 2.
A. Optimizer hyperparameters

Optimization hyperparameters were chosen to reflect good performance for each optimizer in each domain. We therefore performed sweeps for major hyperparameters of each optimizer for each domain, with those used for experiments in the paper shown in the table below.

CEM maintains a population of samples and uses these to estimate the mean $\mu$ and standard deviation $\sigma$ of a Gaussian distribution over design parameters. To optimize $\mu$ and $\sigma$, it takes the top performing fraction, deemed the “elite portion,” from the current step. The initial standard deviation of this distribution is given by “Initial $\sigma$”, and the initial mean is set to 0. We found that for CEM, the population sample size had a significant effect on overall optimization quality (Figure A.1). Due to computational considerations, we picked the smallest value for this hyperparameter that performed within 1 standard deviation of the optimal sample size. Both the elite portion and initial $\sigma$ parameters were chosen as the best performing values on a set of held-out random tasks for each domain.

For GD, we only performed a hyperparameter sweep over the learning rate, which was the only parameter to significantly affect performance. For the 2D FLUID TOOLS tasks, we introduced gradient clipping to eliminate the effect of rare gradient spikes over the course of optimization. However, not using gradient clipping still produced qualitatively and quantitatively similar results. For additional Adam parameters, we used the default values for the exponential decay rates that track the first and second moment of past gradients of $b_1 = 0.9$ and $b_2 = 0.999$ (Hessel et al., 2020).

|        | 2D FLUID TOOLS | 3D WATERCOURSE | AIRFOIL |
|--------|----------------|----------------|---------|
|        | Contain Ramp Maze Direction Pools |
| GD     |                |                |         |
| Learning rate | 0.005 | 0.005 | 0.01 | 0.01 | 0.01 |
| Momentum term $b_1$ | 0.9 | 0.9 | 0.9 | 0.9 | 0.9 |
| Momentum term $b_2$ | 0.999 | 0.999 | 0.999 | 0.999 | 0.999 |
| Gradient clip | 10 | 10 | 10 | — | — |
| CEM    |                |                |         |
| Sampling size | 20 | 20 | 20 | 40 | 40 | — |
| Elite portion | 0.1 | 0.1 | 0.1 | 0.1 | 0.1 | — |
| Initial $\mu$ | 0 | 0 | 0 | 0 | 0 | — |
| Initial $\sigma$ | 0.5 | 0.5 | 1.5 | 0.1 | 0.1 | — |
| Evolution smoothing | 0.1 | 0.1 | 0.1 | 0.1 | 0.1 | — |
| Optimization steps | 1000 | 1000 | 1000 | 200 | 200 | 200 |

Figure A.1. For CEM, increasing population size, while more computationally expensive, can lead to improvements in performance. (a) In the 3D WATERCOURSE domain, CEM benefits from large sample sizes, although returns are diminishing for sizes beyond 40 (Direction task, 36 design parameters). (b) In the 2D FLUID TOOLS domain, CEM benefits from larger sample sizes, although returns are diminishing for sizes beyond 20 (Contain task, 40 design parameters).
B. Model architecture and training

For each task domain, we train a GNN for next-step prediction of the system state. For the domains considered in this paper, we unify the approaches of GNS (Sanchez-Gonzalez et al., 2020) and MESHGRAPHNETS (Pfaff et al., 2021): In the AIRFOIL domain, we encode/decode mesh nodes and mesh edges as a graph as described in the aerodynamics examples of MESHGRAPHNETS, while for particle-based fluids, edges are generated based on proximity as in GNS. In the case of 3D WATERCOURSE, both particles (fluid) and a mesh (the designed obstacle) are present; hence, edges are generated based on proximity (for fluid-fluid and fluid-obstacle interaction) or from the landscape mesh. As the landscape does not have any internal dynamics, we did not find it necessary to distinguish between world- and mesh edges, and use a single edge type.

Once encoded as a graph, the core model and training procedure is largely identical between GNS and MESHGRAPHNETS, and we refer to the above papers for full details on architecture and model training. Briefly, we use an Encode-Process-Decode GNN with 10 processor blocks. All edge and node functions are 2-layer MLPs of width 128, with ReLu activation and LayerNorm after each MLP block. The model is trained with Adam and a mini-batch size of 2, with training noise, for up to 10M steps. We implemented this model in JAX (Bradbury et al., 2018). In addition to the different encoding procedures for mesh vs. particle systems, the parameters for training noise and connectivity radius have to be set per-domain, to account for differences in particle size/mesh spacing. These details are described in Appendix C.

Gradient computation In our experiments, we pass gradients through long model rollouts of up to 300 steps. As it is prohibitive to store all forward activations for the backwards pass, we use gradient checkpointing (Chen et al., 2016) to store activations only at the beginning of each step of the trajectory during the forward pass, and recompute the intermediate activations for each step as needed when the backwards pass walks the trajectory in reverse. Gradient calculation using this method has roughly 3 times the time cost of a pure forward simulation: forward dynamics have to be computed twice for activations for each step as needed when the backwards pass walks the trajectory in reverse. Gradient computation using this method has roughly 3 times the time cost of a pure forward simulation: forward dynamics have to be computed twice for each step, in addition to the computation of the backwards pass itself.

C. Task domains

C.1. 2D FLUID TOOLS

Tasks in 2D FLUID TOOLS are procedurally generated from templates specified in Table A.1. The simulation domain is a 2D box, with the lower left corner specified as [0, 0], and upper right corner specified as [1, 1]. Fluid particles are initialized as a box of size Initial fluid box with bounding boxes given in format [x_min, y_min, x_max, y_max]. Certain task parameters were varied for ablation experiments in Figure 5 (rollout length, # joints (Contain), # tools (Maze)); Table A.1 contains default values used unless otherwise specified.

Design space A “tool” in this task domain is a 2D curve composed of several line segments connected by joints. For a large number of joints, a tool can thus approximate a smooth curve (Figure A.2). Each task’s design space consists of the relative joint angles controlling the tool’s shape. We consider tasks with a single, multi-segment tool (Contain, Ramp) and a task with multiple, single-segment tools (Maze). For each tool, relative angles are calculated by moving from the anchor point on the left, along the tool segments to right, such that angle_i = angle_{i-1} + \phi_{joints} for the i\textsuperscript{th} joint from the anchor. We also experimented with two additional design space parameterizations: (1) jointly optimizing the joint angles and a global position offset [x, y] for each tool, and (2) changing the parameterization of angles to be absolute (such that angle_i = \phi_{joints} directly). We discuss the effects of these alternate parameterizations in Section D.2.

Simulation and objective Both fluids and tools are represented as particles with different types, and simulated with the learned model for 150 steps (with the exception of the ablation experiment on rollout length). Scenes consist of N = 100 . . . 1000 fluid particles. For ground-truth evaluation of the designs, we simulate particle dynamics with an MPM solver (Hu et al., 2018). Task reward is calculated using the Gaussian likelihood of the final particle positions after rollout (u_v from G^{\theta_R}). That is, for a task with reward parameterized with mean \mu and spherical covariance \sigma (\theta_R = [\mu, \sigma]), the reward is calculated as

f_R := \text{mean}_v \mathcal{N}(u_v; \mu, \sigma).

Contain For this task, the center of the goal region \mu is sampled uniformly from a rectangular reward region in the lower-middle section of the 1 \times 1 simulation domain ([0.4, 0.6] \times [0.2, 0.4]). A tool protruding to the right is initially placed below the fluid rectangle. By optimizing a single tool’s relative joint angles, successful solutions must “contain” the fluid in...
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|                           | Contain | Ramp    | Maze (nxn) |
|---------------------------|---------|---------|------------|
| Environment size          | 1x1     | 1x1     | 1x1        |
| Rollout length            | 150     | 150     | 150        |
| Initial fluid box         | [0.2, 0.5, 0.3, 0.6] | [0.2, 0.5, 0.3, 0.6] | [0.2, 0.75, 0.8, 0.8] |
| Reward sampling box       | [0.4, 0.1, 0.6, 0.3] | [0.8, 0, 1, 0.2] | [0.1, 0.1, 0.9, 0.2] |
| Reward σ                  | 0.1     | 0.1     | 0.1        |
| Design parameter          | joint angles | joint angles | rotation |
| # tools                   | 1       | 1       | $n^2$      |
| # joint angles            | 16      | 16      | 1          |
| Tool position (left)       | [0.15, 0.35] | [0.15, 0.35] | —          |
| Tool domain box (3x3)      | —       | —       | [0.14, 0.3, 0.65, 0.6] |
| Tool domain box (4x4)      | —       | —       | [0.14, 0.3, 0.71, 0.6] |
| Tool domain box (5x5)      | —       | —       | [0.14, 0.3, 0.75, 0.6] |
| Tool domain box (6x6)      | —       | —       | [0.14, 0.25, 0.77, 0.65] |
| Tool Length               | 0.8     | 0.8     | —          |
| Tool length (3x3)          | —       | —       | 0.72       |
| Tool length (4x4)          | —       | —       | 0.64       |
| Tool length (5x5)          | —       | —       | 0.65       |
| Tool length (6x6)          | —       | —       | 0.63       |

Table A.1. Task Parameters for 2D FLUID TOOLS tasks. Boxes are described as $[x_{min}, y_{min}, x_{max}, y_{max}]$.

![Figure A.2. Visualization of the design space parameterization for the 2D FLUID TOOLS task. Each red dot corresponds to the anchor points (Contain and Ramp) and center of rotation (Maze) being optimized.](image)

the region by creating a cup or spoon.

**Ramp** The fluid and tool are initialized as in Contain, and $\mu$ is sampled from a region lower and further to the right than in Contain ($[0.8, 1] \times [0, 0.2]$). By again optimizing a single tool’s relative joint angles, successful solutions will create a “ramp” from the initial fluid position to the goal location in the bottom right.

**Maze** The goal is sampled from a long region near the bottom of the domain ($[0.1, 0.9] \times [0.1, 0.2]$). By optimizing the rotation angles of a grid of rigid, linear tools, successful solutions will create a directed path from the top of the screen to the goal location at the bottom.
Model training We trained the learned simulator on the WATERRAMPs datasets released by Sanchez-Gonzalez et al. (2020). This dataset consists of 1000 trajectories featuring a single large block of water falling on one to four randomized straight line segments (see image below for examples). Model architecture and hyperparameters are described in Appendix B, with a training noise scale of $6.7 \times 10^{-4}$ and connectivity radius of 0.015.

![Image of trajectories from the WATERRAMPs dataset](image)

*Figure A.3. Four examples of trajectories from the WATERRAMPs dataset released by (Sanchez-Gonzalez et al., 2020) used as training data for the supervised prediction model.*

C.2. 3D Watercourse

Design space This domain has a design space $\phi_{map}$ of 625 parameters, which determine the y coordinate offset to nodes of a $25 \times 25$ square mesh centered at $c = (0.5, 0.5, 0.5)$ in the simulation domain. While we could directly mapping the parameters to coordinates, we use the design function $y_i = \gamma_H \tanh(\phi_{map})$ ($\gamma_H = 0.3$ for all tasks) to prevent trivial task solutions (i.e. obstacles which touch the floor).

Simulation The simulation consists of an inflow pipe located at (-0.5, 1.0, 0.5) above the landscape which continually emits a stream of liquid, represented as particles. These particles are then redirected by the designed landscape, and finally removed once they hit the floor at $y = 0$. In our experiments we observed up to 2084 particles present in the scene at one time. We unroll the learned simulation model for trajectories of 50 time steps, and store the final particle positions $y_i$ which has hit the floor, we assign it to its closest pool $\hat{p}$, i.e.

$$R(p) := \text{mean}(\mathcal{N}(\mathbf{u}_p^D; \mu_p, \sigma)) - \gamma_R \text{mean}(\nabla \phi_{map})$$

where $\sigma = 0.4$ and a regularization term as above.

Rewards can be in 8 different directions, spaced between 0 and 180 deg. We collapse across directions for reporting reward means and confidence intervals for each optimizer.

2 Pools and 3 Pools In these tasks, we define two and three pools, respectively, with center $\mu_p$ on the floor. For each particle which has hit the floor, we assign it to its closest pool $\hat{p}$, and define the reward as the Gaussian probability under $\mu_p$, i.e.

$$R_pools := \text{mean}_{\mathcal{N}(\mathbf{u}_p^D; \mu_p, \sigma)} - \gamma_R \text{mean}(\nabla \phi_{map})$$

with $\sigma = 0.4$ and a regularization term as above.

To showcase different ways of splitting the water stream, we consider one positioning of the pools for the two pool case, and two for the three pool case. In the two pool case, pools are placed at [1.49, -0.35] and [1.49, 1.35]. In the three pool case, pools are placed either at [1.6, -0.45], [1.85, 0.5], and [1.6, 1.45], or at [0.5, -0.5], [1.7, 0.5], and [0.55, 1.5]. These were selected to ensure the task was solvable – pools directly beneath the landscape, or too far away from the landscape, would not be reachable even with dramatically warped surfaces.
The accuracy for AIRFOIL is within a single standard deviation across all ensemble sizes, while the predictions in 3D...
Figure A.4. (a) Aerodynamics are computed on a large 4158 node mesh centered around the airfoil, with the closeup regions around the airfoil in (b, c) marked as a white square in the center. (b, c) Pressure predictions and control points (orange) for the initial and final optimized wing design (Ensemble-5 model).

**WATERCOURSE** match very closely for both the high performing designs (GD-M) and low performing designs (CEM-M). However, we do notice some discrepancies in the predicted and ground truth reward for the 2D FLUID TOOLS domain, particularly the **Maze** task. As mentioned in the main text, the ground truth solver sometimes produces unrealistic rollouts for this domain (see Figure A.9), with fluid particles becoming stuck between the different tools. Despite this issue, we find that the model is sufficiently similar to the ground truth to produce designs that still achieve high reward overall.

Figure A.5. (a) For the 3D WATERCOURSE domain, reward predicted by the learned model (GD-M eval w/ M, CEM-M eval w/ M) is very close to the ground-truth simulator evaluation (GD-M, CEM-M) for all tasks. (b) This is also true for the 2D FLUID TOOLS domain, though the reward is slightly overestimated when using the model. This effect is amplified in **Maze**, where the ground-truth dynamics sometimes struggles to correctly simulate “sticky” bottlenecks (see Figure A.9). (c) Model predictions of drag (GD-M eval w/ M) are relatively close to the ground-truth simulator evaluation (GD-M), particularly for larger ensemble sizes.

### D.2. Effects of design parameterization

In this section, we study how different parameterization choices for the design space affect both gradient-based and sampling-based optimizers.

First, in the 3D WATERCOURSE domain, we investigate a parameterization of the design space that uses interpolation to minimize the number of control points on the 2D heightfield (Figure A.6). Control points are placed evenly across the grid, and bi-linearly interpolated onto the $25 \times 25$ mesh. We vary the number of control points from $2 \times 2$ up to $14 \times 14$, and find that while CEM-M performs similarly to GD-M when very few control points are allowed, its performance quickly drops as more control points are added.
Figure A.6. Performance on the 3D WATERCOURSE Direction task with variable design space resolution: GD performs well for large design spaces, while CEM performance quickly drops with increased number of design parameters. (b) Examples of CEM designs at two different design space resolutions. (C) Examples of GD designs at two different design space resolutions.

Second, in the 2D FLUID TOOLS domain, we investigate what happens when we change the design space to use absolute joint angles rather than relative ones. When using relative joint angles, changes to joints near the tool’s pivot (left side) affect the global properties of the tool. We hypothesized that this could be selectively benefiting the sampling-based approaches, as this makes the effective design space much lower dimensional. We therefore change the design space to be absolute, with a tool’s joint angles calculated directly: \( \text{angle}_i = \phi_i \).

As hypothesized, this change does dramatically decrease the performance of the sampling-based technique (see Figure A.7). Perhaps more surprisingly, the gradient-based optimizer is almost completely unaffected by this reparameterization. While the qualitative solutions it finds differ (with tools now containing “kinks” to prevent the motion of the fluid rather than curves, Figure A.7 top), the overall reward achieved is similar.
D.2.1. Failure Modes of Gradient Descent

Other parameterizations of the design space can badly affect the performance of gradient-based optimizers. In particular, gradient-based optimizers suffer when there are regions of zero gradients. In the AIRFOIL and 3D WATERCOURSE domains, this is not normally a problem, as the design always interacts with the physical system on which reward is being measured. But in the 2D FLUID TOOLS domain, we can manipulate this.

In particular, for this experiment we changed the design space for 2D FLUID TOOLS to include a global position offset \([x, y]\) for the tool. Making this simple change often has no effect on the discovered designs, but occasionally the gradient-based optimization procedure can move the tool such that it no longer interacts with the fluid (Figure A.8). Once the tool has been moved out of the range of the fluid, there is no longer any way to affect the reward, and therefore there is no gradient signal to recover. To overcome this problem, future work would need to consider more sophisticated hybrid optimization techniques (Toussaint et al., 2018).
D.3. Failure modes of the MPM solver

One of the advantages of using a learned simulator over a classic simulator is learned simulators can be trained in regions of the state and action space that are known to exhibit regularized, smooth behaviors. For example, as mentioned in Section D.1, the MPM solver (Hu et al., 2018) we use for evaluation in 2D FLUID TOOLS shows surprising irregularities with “sticking” behavior when there are a large number of different tools. In the Maze task, this is particularly prevalent, as fluids often become stuck stochastically in some funnels but not others of similar sizes (Figure A.9).

Since the learned simulator was trained on much simpler scenarios where this effect is not observed, it only learns the smooth behavior of the fluid’s movement, which makes the resulting trajectories look more realistic. This may enable better generalization to real world scenarios.

D.3.1. Further designs found in 2D Fluid Tools

In the figures below, we demonstrate the range of found solutions for different solvers in tasks in the 2D FLUID TOOLS domain.

Figure A.8. Failure mode of the GD optimizer: in some instances where the translation of the tool is included in the design, the tool may end up outside of the scope of the fluid. In this cases, the optimization can no longer recover as it will get zero gradients from there on.

Figure A.9. Left: MPM simulation (Hu et al., 2018) of a problem with many separate solid objects. As highlighted in the red circle, the MPM solver struggles with water movement between obstacles, often creating artificially sticky bottlenecks. Right: Learned model rollout for the same setup. The model rollout looks significantly more plausible, without any “stickiness” artifacts. Please see https://sites.google.com/view/optimizing-designs for videos demonstrating this effect clearly.
Figure A.10. Example solutions for each optimizer across the range of rollout lengths sampled for Contain in Figure 5.

Figure A.11. Example solutions for each optimizer across the range of joint angle numbers sampled for Contain in Figure 5.
Figure A.12. Example solutions for each optimizer across the range of grid sizes sampled for Maze in Figure 5.

Figure A.13. Example solutions on Contain task for each optimizer across 6 random seeds.
Figure A.14. Example solutions on *Ramp* task for each optimizer across 6 random seeds.

Figure A.15. Example solutions on *Maze* for each optimizer across 6 random seeds.