SRG X-ray orbital observatory
Its telescopes and first scientific results

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ABSTRACT

The orbital observatory Spectrum-Roentgen-Gamma (SRG), equipped with the grazing-incidence X-ray telescopes Mikhail Pavlinsky ART-XC and eROSITA, was launched by Roscosmos to the Lagrange L2 point of the Sun–Earth system on July 13, 2019. The launch was carried out from the Baikonur Cosmodrome by a Proton-M rocket with a DM-03 upper stage. The German telescope eROSITA was installed on SRG under an agreement between Roscosmos and the DLR, the German Aerospace Agency. In December 2019, SRG started to perform its main scientific task: scanning the celestial sphere to obtain X-ray maps of the entire sky in several energy ranges (from 0.2 to 8 keV with eROSITA, and from 4 to 30 keV with ART-XC). By mid-June 2021, the third six-month all-sky survey had been completed. Over a period of four years, it is planned to obtain eight independent maps of the entire sky in each of the energy ranges. The sum of these maps will provide high sensitivity and reveal more than three million quasars and over one hundred thousand massive galaxy clusters and galaxy groups. The availability of eight sky maps will enable monitoring of long-term variability (every six months) of a huge number of extragalactic and Galactic X-ray sources, including hundreds of thousands of stars with hot coronae. In addition, the rotation of the satellite around the axis directed toward the Sun with a period of four hours enables tracking the faster variability of bright X-ray sources during one day every half year. The chosen strategy of scanning the sky leads to the formation of deep survey zones near both ecliptic poles. The paper presents sky maps obtained by the telescopes on board SRG during the first survey of the entire sky in triaxial orientation mode and deep scanning of selected celestial fields with an area of up to 150 square degrees. These modes of operation were tested during the performance verification phase. Every day, data from the SRG observatory are dumped onto the largest antennas of the Russian Deep Space Network in Bear Lakes and near Ussuriysk.

Key words. X-rays: general – space vehicles: instruments – surveys – X-rays: galaxies: clusters – quasars: general – X-rays: binaries

1. Introduction

The Spectrum-Roentgen-Gamma (SRG) orbital observatory¹ was launched into a halo orbit around the Lagrange L2 point of the Sun–Earth system on July 13, 2019, from the Baikonur Cosmodrome by a Russian Proton rocket with a DM-03 booster (Fig. 1). The Navigator platform (total mass 2712 kg), developed by NPO Lavochkin (N POL) in Khimki near Moscow, carries a scientific payload (total mass 1170 kg) consisting of two X-ray telescopes with grazing-incidence optics (Figs. 2 and 3): extended Roentgen Survey with an Imaging Telescope Array (eROSITA, Predehl et al. 2021), developed in the Max Planck Institute for extraterrestrial Physics (MPE), Germany, and the Mikhail Pavlinsky ART-XC (Astronomical Roentgen Telescope – X-ray Concentrator), developed in Russia (Pavlinsky et al. 2021b). This scheme is implemented in accordance with a memorandum signed in 2007 between the Russian Space Agency (at that time) Roscosmos and the German Aerospace Agency, Deutsches Zentrum für Luft- und Raumfahrt (DLR).
The eROSITA telescope is sensitive to X-rays in the energy range from 200 eV to 8 keV, and ART-XC is sensitive at 4–30 keV.

The Navigator platform has been developed as a universal medium-class platform for scientific and meteorological missions to be launched into various orbits. Since January 2011, the Navigator platform has been used in the three Elektro-L meteorological satellite missions, as well as in the scientific Spektr-R mission (RadioAstron, Kardashev et al. 2013), which was launched in 2011 and operated until 2018.

The SRG spacecraft is operated by NPOL’s Control Center, while the responsibility for operation of the X-ray telescopes lies with the MPE in Garching near Munich (eROSITA) and the Space Research Institute (IKI) of the Russian Academy of Sciences in Moscow (ART-XC). The downlink and uplink of spacecraft telemetry and commanding data is performed by the Centers of Deep Space Communications in Bear Lakes (64 m diameter antenna) near Moscow and Ussuriysk (70 m diameter antenna) in the Russian Far East as well as Baikonur (12 m antenna) placed in Kazakhstan (Fig. 4).

The daily command uplink and scientific data downlink take about one hour for ART-XC and from 2.5 to 4 h for eROSITA (including health checks of the telescopes' detectors and subsystems). The uplink of commands to the spacecraft and telescopes is performed using the antennas mentioned above or additional 12 m diameter antennas in Baikonur and Bear Lakes.

The SRG orbit allows observations to be conducted around the clock with just short breaks for uploading and implementation of operating commands for the Navigator platform and the telescopes. The Navigator platform and both telescopes have onboard mass memory, allowing them to store the data accumulated over several days and dump them to the receiving stations during daily communication sessions. During downlink, the telescopes can (in most cases) continue observations and accumulation of data.

The eROSITA and ART-XC telescopes each have seven independent mirror systems and seven independent positionally sensitive detectors in their focal planes (see Fig. 39 for ART-XC and Fig. 42 for eROSITA), six of which are coaligned on a hexagon surrounding an identical central mirror system and detector. To focus X-rays, both telescopes use grazing incidence optics (Wolter 1952a,b). Figure 40 demonstrates the Wolter I optical scheme of each of the seven ART-XC telescope modules, consisting of 28 pairs of coaxial parabolic and hyperbolic mirror shells, and a detector at the focus of each.

During the 100-day flight of SRG to the L2 point, a thorough check-up of subsystems of the Navigator platform and both telescopes as well as calibrations of the telescopes were carried out. An extensive program of observations of known point and diffuse X-ray sources with various spectral and timing characteristics was carried out during the flight to L2 to calibrate the alignment of the optical axes of the 14 mirror systems of the telescopes (and their alignment with the axes of the satellite, of the optical star trackers, and of the eROSITA and ART-XC telescopes) and to calibrate the imaging and spectral characteristics.
Observation modes: 
- a) all-sky survey
- b) scanning (up to 150 sq. deg)
- c) triaxial pointing

| Table 1. Overview parameters of the SRG observatory and its telescopes. |
|---------------------------------------------------------------|
| **Observatory**         | eROSITA | ART-XC |
| **Orbit**               | Halo orbit around L2 point of the Sun–Earth system |
| **Total mass**          | 2712 kg |
| **Power consumption**   | 1470 W  |
| **Telemetry rate**      | 512 Kbit s⁻¹ |
| **Pointing accuracy**   | 4°      |
| **Observation modes:**  | a) all-sky survey, b) scanning (up to 150 sq. deg), c) triaxial pointing |
| **Planned mission duration** | 4 yr all-sky survey and >2 yr pointed/scanning obs. |
| **Telescope**           | eROSITA | ART-XC |
| **Energy range**        | 0.2–8 keV | 4–30 keV |
| **Energy resolution**   | ~80 eV at 1.5 keV | 9% at 13.9 keV |
| **FoV (FWZI)**          | ∇ = 62' | ∇ = 36' |
| **Angular resolution**  | 30'' HEW | 53'' FWHM |
| (during survey)         |        |        |
| **Pixel angular size**  | 9.6''   | 45''   |
| **Sensitivity**         | <μCrab  | ~100μCrab |
| (4-yr all-sky survey)   |        |        |

Notes. FWZI – full width at zero intensity, HEW – half-energy width, FWHM – full width at half maximum.

...of extended X-ray sources and selected regions in the sky with an area of up to 150 square degrees will also be possible. Table 1 summarizes the main parameters of the SRG observatory and its telescopes. Further parameters of the ART-XC and eROSITA telescopes are provided in Table 4 in Sect. 6 and Tables 5 and 6 in Sect. 7, respectively.

The paper is organized as follows: in Sect. 2 we present the orbit and operation regimes of the SRG observatory. In Sect. 3 we demonstrate the sensitivities and capabilities of the two X-ray telescopes of the SRG and discuss the heterogeneity of the X-ray maps obtained during the first two scans of the entire sky. In Sect. 4 we describe the scientific goals and first results of the mission. In Sects. 5–7 we describe the key properties of the Navigator platform and of the ART-XC and eROSITA telescopes. In Sect. 8 we briefly describe the history of the SRG project in Russia and the responsibilities of the German and Russian consortia of scientists in the processing and scientific analysis of the SRG/eROSITA data coming from two complementary hemispheres of the sky. We finally draw our conclusions.

2. Orbit and operation regimes of the SRG observatory

Figure 5 shows the projection of the SRG orbit onto the ecliptic plane and the scheme of the flight to the L2 point (Eismont et al. 2020). Orbital corrections that were implemented before and after entry into the halo orbit around L2 are also indicated. During the all-sky survey, the satellite rotates around an axis close to the direction toward the Sun with a period of four hours. The rotation axis gradually shifts by approximately one degree per day following the motion of the Sun (see Fig. 6). As a result, the eROSITA telescope (with its 1 degree field of view, FoV) observes each point source in the sky six times for 30–40 s over a day, but typically only once every six months. The full FoV of ART-XC is 36 arcmin, so that a given celestial source is typically exposed four times per day for ~20 s, also every six months. The variability of the sources in the vicinity of the ecliptic poles could be monitored much longer. The SRG observatory observes the entire sky and builds its full map every six months (Fig. 7).

Star trackers are used for the precise orientation of the SRG observatory. Both telescopes use star tracker data to refine their pointing accuracy. Figure 8 shows the positions of the main star trackers that are installed on the two telescopes of the observatory, as well as the position of the radio antennas on the Navigator platform.

2.1. Inhomogeneity of the sky coverage during the survey

The adopted strategy of the SRG survey leads to the appearance of deep fields around the north and south ecliptic poles in the sky map (see Fig. 7), where the large circles of all individual scans cross. As a result, the exposure time depends on the ecliptic latitude θ as 1/ cos θ, that is, the exposure is shortest at the ecliptic equator and longest at the ecliptic poles.

During the first two all-sky surveys, additional inhomogeneity in the sky exposure results from the SRG elongated (in the ecliptic plane) halo orbit around L2. The distance to the Sun varies significantly over six months (see Fig. 5). The satellite moves ahead of Earth for three months and lags Earth for the next three months, which causes a significant variation in drifting rate of the scanning plane, from 0.7 deg day⁻¹ at the closest to the Earth segment of the orbit to 1.6 deg day⁻¹ at its farthest segment (see Fig. 9). Figure 10 shows the actual exposure map for the first SRG all-sky survey based on ART-XC data.
Fig. 5. Three projections of the SRG orbit onto the ecliptic planes and the scheme of the flight to the L2 point. The dotted line shows the lunar orbit. The points TCM1 and TCM2 denote major trajectory correction maneuvers of the orbit (see Sect. 5.4). The point TCM3 denotes the insertion into the quasi-stationary orbit around L2. The green and blue points indicate subsequent minor corrections of the orbit. Courtesy of the Keldysh Institute of Applied Mathematics of the Russian Academy of Sciences.

Fig. 6. Rotation of SRG with a period of four hours around the axis pointed at the Sun. This causes scans of large narrow circles in the sky. The survey plane moves slowly, following the Sun at approximately 1 degree per day.

Fig. 7. Beginning of the eROSITA all-sky survey. Initial test scans are visible, and a dark stripe shows when there were no observations. Individual scans cross at the north and south ecliptic poles. The inset shows a 2° × 2° region within the deep field near the north ecliptic pole, containing the cluster of galaxies A2255 and a bright quasar.

Fig. 8. SRG spacecraft. Positions of the key star trackers, two omnidirectional radio antennas and a medium-gain antenna for transmitting scientific data.
Fig. 9. Rotation speed of the Z-axis of the SRG spacecraft in the ecliptic plane during the first two sky surveys. The spacecraft spins around the Z-axis with a period of four hours, thus performing the all-sky survey. Due to limitations on orientation of the spacecraft, the angle between the Z-axis direction and the Sun and Earth must not exceed 13 and 24 degrees, respectively. This condition defines the speed with which the Z-axis has to follow the Sun and Earth at any point in the halo orbit around L2. Variation in the speed of the Z-axis causes variations in the exposure of the survey. Higher speed leads to lower exposures, and lower speed allows higher exposures of the corresponding sky regions. The vertical bars mark the dates of the gaps in the all-sky survey that are associated with orbit corrections and calibrations.

Fig. 10. Exposure map (in Galactic coordinates) of the first all-sky survey by the ART-XC telescope. Exposure time is given in seconds (see the color scale shown on the right). The white spots in the map correspond to the ecliptic poles. Exposure times for eROSITA are approximately three times longer than for ART-XC, in accordance with the instruments' FoV (36' and 1.03°, respectively).

2.2. Deep scanning of sky fields with a size up to 150 square degrees

During the final stages of the flight to the L2 point, long observations of a large number of point sources and extended Galactic and extragalactic sources were carried out for calibration (Cal) and performance verification (PV) of the SRG telescopes. In addition, scanning deep-survey observations were performed of a number of extended extragalactic fields, a region of the Galactic X-ray Ridge, and fields in the direction of nearby molecular clouds (to reveal X-ray diffuse emission between them and us). The targets for the PV phase have been chosen by the ART-XC science team and the science working groups of the German and Russian eROSITA consortia (see examples of the maps obtained during these observations in Sects. 4.1 and 4.2 below).

To perform deep surveys of extended (up to 150 square degrees) fields, NPOL and IKI proposed and implemented a novel fruitful method of scanning observations, described in Sect. 5 below and illustrated in Fig. 12. This method provides a much more uniform coverage of the observed field compared to the standard method, which is based on a grid of pointed observations, and enables obtaining high-quality maps of extended astrophysical objects.

2.3. Pointed observations

The Navigator platform also allows observations of chosen targets to be done with the triaxial orientation of the observatory. This regime was successfully tested during flight calibration and PV observations. The pointing precision is described in Sect. 5 below.

2.4. Instrumental background at L2

The particle background recorded by both instruments at L2 has been notably stable during the first 1.5 yr of operations, when the solar activity was very modest. This is illustrated in Fig. 13, where the eROSITA count rate in the 7–9 keV band is shown. In this energy range, the internal detector background dominates the astrophysical background. The horizontal dashed red line shows the typical quiescent background. There are rare spikes
Fig. 11. Distribution of eROSITA exposure (upper panel) and sensitivity in the 0.3–2.2 keV energy band (lower panel) after the first three all-sky surveys. The solid curve in the upper panel shows the area of the sky in which the eROSITA exposure exceeded the given value. The solid curve in the lower panel shows the area of the sky in the hemisphere analyzed by the Russian consortium in which the 3σ sensitivity is better than the given value.

Fig. 12. Sky visualization of a typical route of the X-axis in scanning mode.

Fig. 13. eROSITA count rate (per single detector) in the 7–9 keV energy band during the first ten months of the all-sky survey in 6000 s bins. In this energy band, the detector background dominates astrophysical sources. The horizontal dashed red line shows the typical quiescent background, which is globally very stable. Short spikes, whose amplitude rarely exceeds 10–20%, are plausibly associated with low-energy charged particles, which reach the CCD through the thin entrance filters.

Fig. 14. Sketch of ART-XC (dashed lines) and eROSITA (solid lines) astrophysical and internal detector backgrounds. The blue lines show the estimated sky background using the models of Lumb et al. (2002) and Gruber et al. (1999), convolved with the telescopes' responses averaged over the FoV. The red lines show the level of the detector internal background (fluorescent lines are not shown). For eROSITA, the astrophysical background dominates below ∼2 keV, while at higher energies, the detector background exceeds the sky background by an order of magnitude.

The spectrum of the eROSITA detector background is described in Predehl et al. (2021). Here, we show a simplified version of the background model for both instruments (see Fig. 14). In this figure, the blue lines show the (slightly modified) model of sky astrophysical background (Gruber et al. 1999; Lumb et al. 2002) convolved with the telescopes’ responses. For comparison, the red lines illustrate the level of the internal background of the detectors (excluding fluorescent lines). For eROSITA, the astrophysical background dominates below ∼2 keV, while at higher energies, the detector background is the factor that affects the sensitivity.
The eROSITA telescope optics (Wolter-I scheme) focuses photons that are scattered two times: first by parabolic shells, and then by hyperbolic shells. However, some photons can reach the detector after only one scattering or after two scatterings involving “wrong scatterings”, for instance, when they are scattered by the outer surface of a shell. All these photons are collectively called “stray light”. A special baffle was introduced to reduce the magnitude of the stray light by an order of magnitude (Friedrich et al. 2014; Predehl et al. 2021). Nevertheless, a faint and extended halo associated with stray light can be seen when extremely bright sources are observed, such as the X-ray binary Cygnus X-1 (Fig. 15). The role of stray light is more significant in the case of ART-XC (see Sect. 6.2 below and Pavlinsky et al. 2021b for details).

### 3. Sensitivity of the telescopes of the SRG observatory

Figure 16 compares the on-axis effective areas of the eROSITA and ART-XC telescopes as a function of the energy of registered photons. The eROSITA and ART-XC curves intersect near 5 keV. The telescopes nicely complement each other, detecting X-ray photons in adjacent energy bands. The results of the first all-sky survey have confirmed the preflight estimates of the effective area of the SRG telescopes.

Figure 17 shows the sensitivity of the eROSITA and ART-XC telescopes that was achieved during the first half-year survey as a function of energy. The sensitivity has been evaluated assuming an unabsorbed power-law spectrum with a photon index of 1.8 in the energy interval $\Delta E \sim E$. The in-flight background spectrum (both the intrinsic detector background and unresolved astrophysical sky emission, see Freyberg et al. 2020 for details) is integrated over this energy interval to predict the image surface brightness. Using this value, we can predict the expected distribution of fluxes (due to Poisson fluctuations of the number of counts) at a given position of an image convolved with the telescope point spread function (PSF). The sensitivity is then estimated as the flux associated with peaks that for a Gaussian distribution would correspond to $5\sigma$ deviations. The corresponding curves are labeled S1. In the survey, the exposure is shortest near the ecliptic equator, where it is 200 and 60 s for eROSITA and ART-XC, respectively. The difference in exposure is due to the difference in the solid angles subtended by the two telescopes at any given moment. The horizontal bars show the broadband sensitivities in the first survey, approximately converted into units of flux density.
Figure 18 shows the distribution over the sky of the ~600 sources detected by ART-XC in the 4–12 keV energy band during its first sky survey (December 12, 2019–June 10, 2020). Recently, an updated version of this map has been presented by Pavlinsky et al. (2021a), which is based on the sum of the first two sky surveys (December 2019–December 2020). A total of 867 sources (821 point sources and 46 extended sources) have been detected by ART-XC. The 750 sources of known or suspected origin in this catalog include 56% extragalactic sources (mostly AGN and also 52 rich low-redshift clusters of galaxies), and the rest are Galactic (X-ray binaries, cataclysmic variables (CVs), SNRs, etc.). From 114 sources, ART-XC has detected X-rays for the first time. Although the majority of them (~80) are expected to be spurious given the chosen detection threshold, there are expected to be ~35 newly discovered astrophysical objects. An ongoing program of optical follow-up observations of these sources has already led to the identification of several new AGN and CVs (Zaznobin et al. 2021a).

The achieved sensitivity to point sources after the first year of the ART-XC all-sky survey varies between $\sim 4 \times 10^{-12}$ erg s$^{-1}$ cm$^{-2}$ near the ecliptic plane and $\sim 8 \times 10^{-13}$ erg s$^{-1}$ cm$^{-2}$ (4–12 keV) near the ecliptic poles. The typical depth of the ART-XC survey is already comparable to that achieved in a similar energy band (4–10 keV) in the recent MAXI (Monitor of All-sky X-ray Image) all-sky survey (Kawamura et al. 2018) and is just slightly poorer than the sensitivity of the XMM-Newton Slew Survey in the 2–12 keV band (Saxton et al. 2008). However, the ART-XC survey greatly improves on the former in terms of angular resolution and provides full and regular sky coverage in contrast to the latter. The ART-XC survey sensitivity will be growing during the mission as a result of increasing exposure and probably also due to a decreasing flux of Galactic cosmic rays as the next solar maximum approaches. As many as ~5000 sources can be found by ART-XC by the end of the four-year all-sky survey.

As noted above, the eROSITA telescope is noticeably inferior in sensitivity to the ART-XC telescope at energies above 5–6 keV, but between 4 and 5 keV, eROSITA is more sensitive. As shown in Fig. 19, during the first two surveys of the sky, eROSITA detected over 600 X-ray sources in one hemisphere in the range from 4 to 8 keV. This number is comparable to the number of sources detected by ART-XC over the entire sky during one survey in the 4–12 keV range.

The eROSITA telescope has obtained the best map of the sky in the history of X-ray astronomy in the 0.3–2.3 keV energy band (Fig. 20) even after its first all-sky survey. The good angular resolution and high sensitivity of eROSITA allowed it to detect over a million compact sources and map about 20 thousand extended sources. This huge number of sources cannot be displayed in a single image; only the brightest of them are visible on the map as dots. After scanning the sky for just six months, eROSITA has constructed a map that is about four times more sensitive and contains almost eight times more sources than the previously best all-sky map that was obtained in 1990 by the ROSAT satellite. eROSITA has already nearly doubled the total number of sources detected by all orbital observatories over the ~60 yr of X-ray astronomy.

The eROSITA map reveals spectacular giant bubbles of hot gas with temperatures up to 10 million K expelled from the plane of the Galaxy. These are the outcome of hundreds of thousands of supernova explosions and/or the intermittent activity of the supermassive black hole in the center of our Galaxy (Predehl et al. 2020). These X-ray bubbles are seen above and below the midplane of the image that encloses the well-known Fermi bubbles, which are associated with gamma-rays emitted through the interaction of cosmic rays with the ambient gas (Ackermann et al. 2014).

About three-quarters of the objects in the SRG/eROSITA map are AGN that are powered by the accretion of matter onto supermassive black holes residing in their centers.
Fig. 18. Positions (in Galactic coordinates) of the X-ray sources detected by SRG/ART-XC during its first all-sky survey (typical exposure per point is just ~60 s). Nearly 600 sources have been detected in the 4–12 keV energy band, ~60% of which are Galactic (black holes, neutron stars, white dwarfs, coronally active stars, SNRs, etc.) and ~40% are extragalactic (AGN and a few dozen massive clusters of galaxies), as well as a number of newly discovered sources. The symbol size reflects X-ray brightness of the sources.

Fig. 19. Positions (in Galactic coordinates) of X-ray sources detected by SRG/eROSITA in the 4–8 keV energy band during the first two all-sky surveys on that half of the sky in which the Russian Consortium is responsible for the data analysis (the Galactic center is on the right side of the image). The image includes ~600 sources, and some of the brightest are labeled. They are located far beyond the Milky Way. The quasar CFHQS J142952+544717 at $z = 6.2$ (corresponding to the age of the Universe of 900 million years) is particularly interesting. It was detected in X-rays for the first time by eROSITA and proved to have the highest X-ray luminosity ($\sim 3 \times 10^{46}$ erg s$^{-1}$, or $\sim 10^{13}$ bolometric luminosities of the Sun) of the quasars at $z > 6$ (Medvedev et al. 2020, 2021). During the calibration and performance verification (CalPV) phase and in the course of scanning the entire sky, SRG/eROSITA has discovered a number of extremely luminous quasars at redshift $z > 5$ and a noticeable number at $z > 4$ (Khorunzhev et al. 2020; Dodin et al. 2020; Bikmaev et al. 2020; Wolf et al. 2021). In Fig. 21 we show the spectrum of such a quasar discovered by SRG/eROSITA at $z = 5.5$. It was obtained with the 6-meter BTA telescope in the North Caucasus (Khorunzhev et al. 2021). The majority of the ~20 000 extended objects in the eROSITA map are clusters of galaxies filled with dark matter and hot intergalactic gas, which shines in X-rays. Fewer than 50% of these clusters were previously known from optical surveys or due to detection of the Sunyaev–Zeldovich effect (Sunyaev & Zeldovich 1980) in their direction by the Planck spacecraft (Planck Collaboration XXVII 2016), the Atacama Cosmology Telescope (ACT, Hilton et al. 2021), or the South Pole Telescope (SPT, Bleem et al. 2015) sky surveys.

There will be much synergy and competition between different methods of observation of galaxy clusters over the next years through the continuation of sky surveys in different spectral bands, including the SRG survey in X-rays and ground-based surveys by ACT and SPT in microwaves. Not only will the results of these surveys be valuable for cosmological studies, but we should also expect the discovery of thousands of strong gravitational lenses due to the deep gravitational potential of clusters of galaxies.

Some two hundred thousand fairly close stars with coronae much more powerful than the corona of the Sun also contribute to the emission from the zones of relatively low temperature in the eROSITA sky map (Fig. 20). Interestingly, eROSITA has detected X-ray emission from 150 stars with known exoplanets.
This amounts to some 10% of all nearby stars with known planetary systems (excluding the more distant stars with exoplanets in the field that were explored by the Kepler satellite).

Figure 22 demonstrates the dramatic difference between images of the sky (one hemisphere) in the 0.3–0.7 keV and 0.7–2.3 keV energy bands. In the 0.3–0.7 keV band, the emission is fairly homogeneous, bright, and diffuse. This might be the background radiation of a huge number of soft X-ray sources located at cosmological distances and/or emission from the hot ($10^5 < T_e < 10^7$ K) gas in the halo of our Galaxy. An additional significant contribution to the sky brightness at these energies might be provided by relatively nearby sources such as the hypothetical Local Bubble that was created by supernova explosions in the relative vicinity of the Solar System. The strong absorption of soft X-rays by the cold atomic and molecular gas and dust in the Galactic plane is also noteworthy. A very different picture is observed in the 0.7–2.3 keV energy band. Here the image is dominated by many hundreds of thousands of extragalactic sources. The main contribution to the source counts and to the diffuse background is provided by AGN. The absorption by cold gas and dust in the Galactic plane is strongly reduced by the rapidly decreasing photoabsorption cross-section with increasing photon energy. Bright Galactic sources in the region of active star formation in the Cygnus constellation are clearly visible in both maps.

4.2. Examples of the results of deep surveys during the PV phase

4.2.1. Galactic center

During the CalPV phase, ART-XC performed deep surveys of a number of extended fields in the sky, in particular, of a large (∼40 square degrees) region in the center of our Galaxy.
Figure 23 shows a fragment of the image obtained in the 4–12 keV energy band, which reveals the high quality and richness of the data.

All in-flight characteristics of ART-XC have proved to be close to preflight expectations (see more details in Sect. 6.4). In particular, the PSF averaged over the FoV is better than 1 arcmin (half-power diameter, HPD) in survey mode. The good angular resolution of the telescope is evident from the image (Fig. 24) of the SNR RX J1713.7–3946 that is clearly resolved by ART-XC.

4.2.2. Galactic Ridge

Figure 25 demonstrates a large variety of astrophysical objects of our Galaxy that are accessible for observation by the SRG observatory. During scans of the Galactic Ridge (in a field ~20 degrees away from the Galactic center), eROSITA has detected stars with active X-ray emitting coronae, star-forming regions and clusters of young stars, X-ray pulsars (rapidly rotating magnetized neutron stars), and SNRs. In the latter, X-ray photons are emitted by gas that is compressed in shocks where the material of the exploded star collides with the surrounding interstellar matter. As is well known, hot gas occupies 80–90% of the volume near the Galactic plane, and just 10–20% of the volume is filled by dense clouds of cold molecular and atomic hydrogen.

4.2.3. Lockman Hole

Figure 26 demonstrates the richness of the extragalactic X-ray sky revealed during a long scan (about 8 ks per pixel) of the Lockman Hole zone. In this unique region, the absorption of X-rays by the interstellar medium of the Galaxy is close to its minimum in the entire sky. This allows studies of distant quasars and clusters of galaxies in unprecedented detail. In the ~20 square degrees field, eROSITA has detected over 8500 point X-ray sources, that is, ~400 sources per square degree. This number corresponds to ~16 million objects when extrapolated to the whole sky. The vast majority of these sources are AGN. According to photometric redshift estimates, the most distant of the quasars detected by eROSITA in the Lockman Hole are located at z ~ 5. Some ~200 clusters and rich groups of galaxies filled with hot gas were detected. In addition, several hundred active Galactic stars are also seen in the Lockman Hole field.

4.2.4. Coma cluster

One of the main targets of the CalPV phase was the Coma cluster (Fig. 27). SRG observations in the scanning mode are particularly well suited for a detailed mapping of this particular and nearby cluster well beyond its virial radius. The obtained ART-XC data enable measuring the temperature of the hot gas and estimating the contribution of cosmic rays to the observed X-ray emission, while eROSITA images provide detailed information about the merger of the Coma cluster with its less massive companion, the galaxy group NGC 4839. Figure 27 clearly shows the rich substructure that is generated by the merger, including shock waves and contact discontinuities extending over a few megaparsec. The data from the Planck observatory (distribution of brightness in the microwave band due to the SZ effect) provide additional information about the distribution of pressure in the space between galaxies (Planck Collaboration Int. X 2013). The comparison of the data from the three telescopes provides a detailed picture of the distribution of hot gas and dark matter in the cluster and its outskirts.

It is important to note that in the Coma cluster field, eROSITA has unveiled many dozen massive clusters of galaxies and thousands of quasars located at cosmological distances far beyond the Coma cluster. This is a consequence of the deep scanning of a field with a size of $3\times3^\circ$.

4.3. Source catalog

The source catalogs are maintained by the Russian (RU) and German (DE) consortia for the respective sky hemispheres. A preliminary version of the log(N) – log(S) distribution of compact sources in the extragalactic part of the RU sky is shown in Fig. 28.

4.4. X-ray spectroscopy

During the CalPV phase, ART-XC performed long pointed observations of a number of bright X-ray sources, including the well-known nearby AGN Circinus galaxy. The obtained spectrum (Fig. 29) shows a complex continuum in the 4–20 keV energy band along with a strong Fe-Kα emission line. The estimated energy resolution of ~1.3 keV at 6 keV matches the preflight estimates.

The excellent capabilities of the eROSITA telescope in spectroscopy of hot astrophysical plasmas (temperatures of some million degrees) in bright X-ray sources during the all-sky survey are clearly demonstrated (Fig. 30) by the spectrum of the central zone of the Cygnus Loop SNR obtained using an exposure of just 200 s. X-ray emission lines of a number of ions of various chemical elements are clearly seen. The spectroscopic capabilities of eROSITA in the 3–9 keV band are demonstrated in Fig. 31, which shows the spectrum of the Galactic microquasar SS 433 obtained during the SRG/eROSITA PV phase. Emission lines of highly ionized atoms (silicon, sulphur, iron, and nickel) corresponding to the approaching (blueshifted) and receding (redshifted) baryonic multitemperature jets are clearly resolved. The line positions are fully consistent with the expectations based on the kinematic precession model with the bulk velocity of the jets equal to a quarter of the speed of light.
4.5. X-ray timing

The ART-XC telescope has an excellent time resolution of 23 µs, which is determined by the resolution of time stamps assigned to incoming events by the reading electronics. This permits a detailed study of the time behavior of millisecond pulsars in pointing mode. The timing capabilities of the telescope were tested during observations of the Crab pulsar. Figure 32 shows the light curve of this pulsar in several energy bands as measured by ART-XC and folded with the Jodrell Bank radio ephemeris. The ∼33 ms pulsations are clearly detected up to 30 keV, and the energy dependence of the pulse morphology is as expected.

The nominal integration time of the eROSITA CCDs is 50 ms. This integration time is set by the requirement that smearing of X-ray images in the all-sky survey is to be avoided. With the nominal rotation rate (360 degrees in four hours), the sources in the telescope FoV shift by 3.5" in 50 ms, which is about three times smaller than the CCD pixel size. While this integration time precludes a systematic study of the variability on millisecond scales, the stability of the background and instrument characteristics provides an excellent opportunity for monitoring objects on longer timescales in three important intervals: 0.05–40 s (for sources crossing the FoV during a single scan), four hours to one day (during six intra-day scans of the same object), and on scales from six months to four years (over the entire SRG all-sky survey). In addition, the variability of sources located close to the intersections of individual scans near the ecliptic poles can be tracked nearly continuously on scales longer than four hours.
Fig. 25. Annotated X-ray RGB map of a 25 square degrees region in the disk of the Milky Way (the so-called Galactic Ridge) obtained by eROSITA in October 2019. In this field, thousands of Galactic X-ray sources are detected, as well as a number of quasars observed through the Galactic disk. In addition to many individual sources, the map also shows unresolved X-ray emission from hot gas and from a multitude of faint unresolved sources. Blue and green correspond to high photon energies emitted by a gas with a temperature of tens of millions of degrees, while red regions reveal colder gas of lower temperature. Adopted from Gilfanov et al. (in prep.). Absorption in the cold interstellar gas near the Galactic plane attenuates soft X-ray radiation, but allows hard X-ray photons to leak through because their absorption cross-section is smaller. A significant contribution to the apparently diffuse emission from the Galactic disk and bulge is provided by the superposition of emission from numerous accreting white dwarfs, hot coronae of low-mass stars, and flares on them (Revnivtsev et al. 2009). eROSITA cannot resolve individual contributions of these sources because of their great number and low luminosities.

4.6. ART-XC Galactic transients

In its daily scans (exposure of \(~60\) s per source), the ART-XC telescope reaches a sensitivity (\(\sim 5\sigma\)) of \(\sim 8 \times 10^{-12}\) erg cm\(^{-2}\) s\(^{-1}\) in the 4–12 keV energy band (about 0.6 mCrab). By surveying about 1% of the sky every day, ART-XC provides rapid alerts about new X-ray transients (e.g., Mereminskiy et al. 2020c,b; Schwope et al. 2020) or new outbursts from known or poorly studied historical sources (e.g., Mereminskiy et al. 2020d,a). These events are typically also detected by eROSITA, which provides a more accurate localization for newly discovered sources and detailed spectral information below \(\sim 2–8\) keV for a broadband spectral analysis.

The ART-XC telescope provides the unique possibility of studying the population of faint transients that would be otherwise missed because they are too weak for wide FoV telescopes and all-sky monitors (such as the INTErnational Gamma-Ray Astrophysics Laboratory (INTEGRAL)/IBIS, Swift/BAT, and MAXI). The relatively hard X-ray band of the ART-XC telescope also makes it less dependent on the source intrinsic or Galactic absorption and allows the detection of highly absorbed sources, which could be missed by soft X-ray instruments. For example, ART-XC transients often appear quite unremarkable in eROSITA data. Follow-up campaigns have already enabled establishing the nature of several ART-XC sources, such as the new microquasar SRG J043520.9+552226/AT2019wey (Yao et al. 2021b,a; Mereminskiy et al. 2021), the new nova-like CV SRGt J062340.2−265715 (Schwope et al. 2021), several new Be systems (Doroshenko et al. 2021; Lutovinov et al. 2021), and a few others.

4.7. Extragalactic transients and stellar flares

The approach of repeated sky surveys when every (typical) location on the sky is visited every six months has proven to be an efficient tool for studying the long-term variability of sources and for discovering various types of Galactic and extragalactic transients. Every 24 h, eROSITA detects between half a dozen and a dozen sources that have changed their luminosity by more than an order of magnitude compared to the previous visit six months earlier (e.g., Gilfanov et al. 2020; Sazonov et al. 2020; Medvedev et al., in prep.). About half of these sources are associated with Gaia stars, and the remaining half are presumably of extragalactic origin. The requirement of at least a tenfold flux
Tidal disruption events (TDEs) are being actively searched for among extragalactic eROSITA transients based on the optical and infrared properties of their hosts and on the shape of their X-ray spectra (Fig. 33). The Russian consortium identifies about one relatively bright TDE candidate every week (Khabibullin et al. 2020b,a; Gilfanov et al. 2020, 2021). They are followed up with optical spectroscopy on various telescopes in Russia (the 6 m BTA telescope in the Caucasus; the 1.5 m Russian-Turkish telescope, RTT-150, in Turkey; the 1.6 m AZT-33IK telescope of the Sayan Observatory; and the 2.5 m telescope of the Caucasus Mountain Observatory of the Sternberg Astronomical Institute of the Moscow State University). An active collaboration with the Zwicky Transient Facility (ZTF) team facilitates timely classification of detected transients and their optical follow-up on Palomar and Keck telescopes. During the period from June 10 to December 14, 2020, spanned by the second SRG/eROSITA all-sky survey, 16 TDEs have already been detected and optically confirmed in the 0 < l < 180° hemisphere (Sazonov et al. 2021, Gilfanov et al., in prep.). The first eROSITA TDEs have also been reported in the other half of the sky (Liu et al. 2021).

The SRG scanning strategy means that more rapid transients may be found that vary on a timescale of about one day, in particular, X-ray afterglows of gamma-ray bursts even without an observable gamma-ray trigger (Khabibullin et al. 2012). A systematic search for these events is continuously ongoing (e.g., Wilms et al. 2020).

5. SRG spacecraft

Design of the structure and configuration of the Spektr-RG spacecraft continued until 2008 (see Sect. 8 below about the history of the SRG project). At that time, the concept design was defined and accepted.

5.1. Navigator-SRG platform

The spacecraft (Figs. 2, 3, and 8) is based on the Navigator-SRG space platform (Fig. 34), which has been developed by NPOL for spacecraft of hydro-meteorological and scientific designation. By the time the SRG design was accepted, the flight models of the Navigator platform for the Spektr-R (Radioastron) and Electro-L No. 1 spacecraft had been produced and tested. In 2011 the platform started its flight qualification. This meant an opportunity to gather invaluable experience of the platform operation, which was considered during the creation of Spektr-RG. Modifications aimed at enhancing the mission reliability and achieving the required technical parameters of the spacecraft as a whole were made. A principal change made to the Navigator platform was the installation of a new X-band radiocomplex, induced by the necessity of maintaining ground contact with the spacecraft at large distances.

The Navigator-SRG platform includes the following units: an onboard control system, an onboard radiocomplex, solar panels with the orientation mechanism, a propulsion unit, a thermal control system, a telemetry system, an antenna-feeder system, a cable network, construction, a power supply system, and miscellaneous supplementary systems and units.

The onboard control system performs the following basic tasks: it controls operation of the scientific payload and service systems of the spacecraft, it controls the motion of the spacecraft around the center of mass and the motion of the center of mass, it controls the spacecraft orientation by choosing the correct omnidirectional antenna for linkage with ground stations, it controls the technical state and diagnostics of onboard systems with subsequent transfer of information to Earth via the onboard radiocomplex, it manages contingency cases, and performs automatic transition of the spacecraft into stand-by mode in contingency cases that cannot be managed by the onboard systems.

The onboard control system includes the following elements: an onboard computer, power automation units, a gyroscope angular velocity sensor, two SDP-1 solar position sensors, two POS 347K solar orientation devices, two SED26 star trackers, and four Agat-15M reaction wheel sets.

The main orientation control actuators of the onboard control system are reaction wheels (RWs). Three RWs are in operation, and one RW is in cold reserve. A set of three RWs ensures an angular velocity up to 0.07 deg s\(^{-1}\). For RWs desaturation (momentum dump), 16 stabilization thrusters are used (8 operating and 8 in cold reserve, 0.5 N each). These thrusters are
Fig. 27. X-ray image of the Coma cluster (Churazov et al. 2021) in the 0.4–2 keV band obtained by eROSITA in the course of the CalPV program (top left). The image is ~6 degrees on a side, corresponding to 10 Mpc at the distance of the cluster, with the logarithmic color-code spanning five orders of magnitude. The main cluster is in the process of merging with the NGC 4839 group (a bright blob in the bottom right corner from the Coma cluster). Flattened X-ray image of the Coma cluster field (bottom left). The labels schematically mark some of the features that are presumably associated with the merger with the NGC 4839 group. The dashed blue line is the suggested trajectory of the group, which enters the Coma cluster from the northeast direction, and is currently close to apocenter. The presumed positions of two shocks driven by the NGC 4839 group are shown with the red and purple curves. The shock closer to the center is driven by the displaced gas that settles back to hydrostatic equilibrium. This is the most salient feature that can be directly seen in the image at the surface brightness edge. The green line shows the faint X-ray bridge connecting NGC 4839 and the main cluster, which is a possible trace of the group passage through the Coma cluster. The yellow line shows the contact discontinuity, which is an interface between cold and hot gas patches with the same pressure. See Churazov et al. (2021) for details. ART-XC image of the Coma Cluster in the 4–12 keV energy band (top right). Planck spacecraft $\eta$-parameter map of the Coma cluster, based on the data in the microwave spectral band (bottom right). The morphology of this map, which reflects the distribution of the hot gas electron pressure, is strikingly similar to the eROSITA image in the X-ray band.

also used during the initial flight phase to establish the solar orientation and in contingency cases.

In addition to the stabilization thrusters, the propulsion unit also includes larger correction thrusters (5 N each) that are used for trajectory correction. All the thrusters are powered by hydrazine contained in two tanks, each of which is filled with 180 kg of fuel.

Electric power supply for the onboard service systems and scientific instruments is provided by solar panels. The total power consumption of all onboard systems does not exceed 1700 W, which is less than the expected capacity of the solar panels at the end of their service lifetime (1870 W). The power supply system also includes a 55 Ah battery that provides power for onboard systems during the initial phase after separation.
of the spacecraft from the launch vehicle or in contingency cases.

Thermal control of the spacecraft is established by axial and contour heat pipes, as well as by electrical heaters that are controlled by the onboard control system. In addition, both telescopes have their own thermal control systems.

The eROSITA and ART-XC telescopes are placed on a geometrically stable frame mounted on the top joint of the spacecraft base module. A number of electronic units of ART-XC and the gyroscopic angular velocity sensor device of the onboard control system are installed on a thermo-stabilized platform.

Fig. 28. Preliminary log(N) − log(S) distributions of compact sources in the extragalactic part of the hemisphere, where the Russian consortium of scientists is responsible for the data analysis and interpretation. The distribution of sources with and without Gaia stars among their optical counterparts are shown by blue and red histograms and are marked in the plot legend as Gaia stars and AGN, respectively. The distributions are derived from partial data of the first two sky surveys and do not use the full potential of the first-year data. No incompleteness correction was applied at the faint flux end. The coefficient A on the y-axis incorporates approximations and inaccuracies of the preliminary analysis. It is reasonably close to unity and may weakly depend on flux, which is not necessarily the same for Gaia stars and AGN.

Fig. 29. X-ray spectrum of the Circinus galaxy measured by ART-XC. The Fe-Kα line is prominent on top of the complex continuum. The exposure is about 50 ks. Flux units are arbitrary.

Fig. 30. 0.4–2 keV X-ray image of the SNR Cygnus Loop obtained by eROSITA during the first two SRG all-sky surveys (upper panel). The longest exposure time (per point) across the image is ∼400 s. The axes are labeled in degrees. X-ray spectrum of the entire Cygnus Loop (lower panel). The ions contributing to some of the brightest emission lines are indicated. The spectrum is normalized per square arcminute.

5.2. Orientation and operation regimes of the observatory

Except during the initial flight phase, the onboard control system operates in inertial orientation mode, which ensures triple-axis stabilization and spacecraft rotation into a programmed orientation with respect to the inertial frame. Orientation data from the star trackers are continuously coupled with the gyroscopic sensor data, resulting in an estimated orientation quaternion.

In the inertial orientation mode, the following operations can be performed: spacecraft rotation for pointing telescopes (SRG +X axis) to a predefined target on the celestial sphere and triple-axis stabilization with respect to the inertial frame; spacecraft rotation at a constant angular velocity (typically 0.025 deg s⁻¹) around the SRG +Z axis, in which the rotation axis is constantly rotated at a low angular velocity (∼1 deg day⁻¹) to ensure a smooth all-sky survey; and spacecraft triple-axis stabilization during trajectory-correction impulses.
Fig. 31. Spectrum of the Galactic microquasar SS 433 obtained during the SRG/eROSITA PV phase. The data are well fit by the model of baryonic multitemperature jets (Khabibullin et al. 2016), with contributions of the approaching (blueshifted) and receding (redshifted) jets shown in blue and red, respectively. The positions of the lines of hydrogen- and helium-like silicon, sulphur, iron, and nickel are marked.

Fig. 32. Crab phase histograms reconstructed in several energy bands. The ART-XC exposure is about 45 ks.

Several consequent spacecraft rotations can be combined into series to scan fields on a celestial sphere by predefined routes. In this mode, the onboard control system points the SRG X-axis to the required starting point of a scan field (with respect to the inertial frame) and performs a program of relative rotations around the Y- and Z-axes of the spacecraft. During the relative rotations, the coupling of the star trackers and gyro data does not interrupt the precise route tracking with respect to the inertial frame. The scan field size can be up to 12.5° × 12.5°. A typical scan sequence consists of several S-turns (Fig. 35). The main scanning turns are performed around the Z-axis and smaller auxiliary turns (steps) are performed around the Y-axis. Step rotation can be arbitrarily low and is usually chosen from the 4–12 arcmin range. The highest angular velocity of the scanning turns may be as high as 0.04 deg s⁻¹.

The spacecraft attitude-control error has been confirmed to be less than 10 arcsec. The angular velocity stabilization error does not exceed 0.72 arcsec s⁻¹. Figure 36 illustrates the main orientation restrictions on pointing and scanning observations of the SRG observatory.

5.3. Radiocomplex and data transmission

In addition to receiving commands and performing trajectory measurements, the onboard radiocomplex is also capable of combining scientific data and service telemetry into one data stream. The nominal rate of transmission of service telemetry data is 16 Kbit s⁻¹ (real-time transmission of TM-data) and 32 Kbit s⁻¹ (stored TM-data dump).
The nominal rate of scientific data transmission is 512 Kbit s\(^{-1}\), but it can also be performed at 64, 128, and 256 Kbit s\(^{-1}\). Telecommands from ground stations are uploaded at a rate of 500 bits s\(^{-1}\) (typical) or 125 bits s\(^{-1}\) (contingency cases).

The radiocomplex receives telecommands through an omnidirectional antenna. It transfers data through a medium-gain antenna with a beam width of ±24°.

The onboard telemetry system collects service telemetry sensor data and digital telemetry data from the onboard control system and writes it into the internal memory device. During ground contact, it is possible to dump the stored telemetry data or run live telemetry transfer mode.

**5.4. Launch and operational orbit insertion**

The SRG observatory operates in a quasi-periodic orbit around the L2 libration point of the Sun–Earth system. This orbit has a number of clear advantages for the mission related to the remoteness (1.5 million km, corresponding to a light travel time of 5 s) of the L2 point from Earth in the direction away from the Sun, in particular, long visibility periods from ground stations for control and receipt of scientific data, and the prevention of periodical entries of the spacecraft into Earth’s radiation belts.

The location of the Russian ground-control stations in the northern hemisphere required choosing a launch date that would ensure maximum daily radio visibility of the spacecraft. In addition, the choice of an optimal operational orbit had to minimize the characteristic velocity required to keep the spacecraft near the L2 point. The formal solution based on these two criteria implied the necessity of preparing separate flight plans for the upper stage for each possible launch date.

Taking into account the anticipated time frame of the readiness of the spacecraft segments, two possible launch windows were eventually chosen: June 21–22, 2019, and July 12–13, 2019. The launch of the SRG spacecraft from the Baikonur Cosmodrome successfully occurred on July 13, 2019, at 15:30:57 Moscow time.

The Proton-M launch vehicle placed the orbital unit (OU), including the upper stage (US), the adapter system, and the spacecraft, in a falling trajectory with an apogee altitude of \(\sim 200\ km\). The first ignition of the propulsion unit (PU) of the US placed it in a transfer orbit \(\sim 168 \times 2013\ km\). The second ignition placed the OU in the nominal flight orbit toward the L2 point \(\sim 500 \times 14500000\ km\). The spacecraft later separated from the US, followed by the deorbiting of the US. The placement into orbit took two hours. During this process, the US spent 34 min in the Earth’s shadow. The flight toward L2 lasted approximately 100 days without any shadow intervals.

To ensure placement of the spacecraft in the nominal orbit, three flight trajectory corrections on the way to the L2 point had been planned: on days 10, 20, and 40 of the mission. The corresponding reserve dates were also defined: on days 15, 25, and 45. A characteristic velocity budget of 100 m s\(^{-1}\) was allocated for these corrections. The first correction was executed in accordance with the schedule. The second correction was moved to the reserve date to acquire more orbit measurements after the first correction. By the time the third correction was due, the calculations indicated that it was not needed, so that there was no correction on day 40 (see Table 2).

The operational orbit of the SRG observatory was defined by the following constraints: the maximum distance of the spacecraft from the L2 point in the ecliptic plane \(\sim 9200000\ km\), the exit from the ecliptic plane \(\sim 700000\ km\) and \(\sim 550000\ km\) toward the north and south ecliptic poles, respectively. After placement of the spacecraft into the nominal orbit near the L2 point (on day 100), orbit corrections are executed approximately every 50 days during the mission (see Sect. 5.7 below).

**5.5. Spacecraft control**

Spacecraft operation planning is performed on three timescales (long term, medium term, and short term). The degree of detail is to be increased when proceeding to the next stage.

Long-term mission planning is determined by science tasks formulated by the scientific community. At the medium-term level, operations of the spacecraft and ground stations are planned for the following calendar month. Approximately two weeks before its initiation, NPOL performs a priori calculations of the spacecraft trajectory for this month, schedules preliminary intervals for spacecraft control sessions, and, if necessary, schedules technological operations for the spacecraft (which usually include only trajectory correction). NPOL delivers these data to the science operation and data center of IKI, where the month plan is completed with scientific operations in accordance with the long-term plan. The plan produced at this stage determines the sequence and parameters of the spacecraft orientation modes. The plan undergoes preliminary verification of consistency with the spacecraft and ground-stations operation limitations both on the side of IKI and NPOL. The spacecraft orientation limitations

![Fig. 36. Main orientation restrictions for observations during scans and pointings.](image-url)

![Fig. 35. Typical route of the X-axis in scan mode.](image-url)
Table 2. Trajectory-correction schedule for the SRG flight to the vicinity of the L2 point.

| No. | Date/time (Moscow) | Days in flight | Number of PU ignitions | Total characteristic speed m/s | Consumed propellant kg |
|-----|--------------------|---------------|------------------------|-------------------------------|------------------------|
| 1   | Jul. 22, 2019 17:30:00.000 | 10            | 2                      | 13.59                         | 15.98                  |
| 2   | Aug. 06, 2019 17:30:00.000 | 25            | 2                      | 3.49                          | 4.09                   |

are the main verification criteria. In case of changes in the ground stations or in the onboard systems’ condition, a monthly plan can be adjusted even during its implementation. In these cases, secondary verification and decision making usually takes no longer than three days.

At the short-term level, exact operations of the ground station and spacecraft during an upcoming ground-contact session are planned. The main task of short-term planning is to ensure necessary spacecraft operations starting from the beginning of a session until at least the next session. Operations scheduled in a month plan are detailed in the form of a ground-contact session program, which is a sequence of commands for ground stations, spacecraft service systems, and the telescopes. To compose a ground-contact session program, NPOL collects requests for commands from supervisors of the corresponding systems. In particular, a sequence of commands to ensure the telescopes’ operation (with specific command parameters and issue times or intervals) is generated at IKI based on requests of the ART-XC and eROSITA teams.

Commands for the spacecraft and scientific instruments may be executed immediately during a ground-contact session or stored in the onboard control system memory for execution at a preset time (time-tagged commands). NPOL performs a simulation of a prepared ground-contact session on the onboard control system’s logical testing bench. It provides means of precise verification of a program consistency with spacecraft operation limitations, and detects logical planning errors and incorrect commands. Simulation is performed for an interval from the start of a planned control session until the start of the next session. This technology ensures end-to-end simulation during the whole mission, which increases control reliability.

Provided the simulation results are positive and all the parties endorse a control session plan, it is executed. The execution of the commands is controlled using telemetry data. If necessary, the program may be adjusted during a session, including issuance of commands initially not planned by the ART-XC and eROSITA teams.

Ground-contact sessions take place daily. Each session lasts approximately 4–5 h, which makes it possible to dump all scientific and service telemetry data gathered throughout the day, conduct trajectory measurements by at least two ground stations, check the status of onboard systems and scientific instruments, and upload a time-tagged command sequence for a subsequent period of up to several days. During a control session, the full real-time data stream from the spacecraft is received by NPOL through ground stations and transmitted to IKI.

5.6. Mission ground-control complex

Spacecraft control is provided by the Russian ground-control complex of the SRG spacecraft, including the NPOL mission control center, ground stations, ballistic centers, and means of communication. The central authority responsible for the flight tests and spacecraft control is NPOL, which provides a platform for the cooperation of specialists participating in flight tests and the implementation of the scientific program.

Radio communication with the spacecraft is provided by radio-technical ground complexes located at Baikonur (TNA-57 antenna of 12 m diameter), Bear Lakes (TNA-1500 antenna of 64 m diameter), and Ussuriysk (P-2500 antenna of 70 m diameter). Currently, only the Bear Lake and Ussuriysk ground complexes are used for receiving scientific data. The radio-technical ground complexes had been modernized for the implementation of the SRG project and currently fully ensure the mission requirements. During periods of short daily radiovisibility intervals from the Russian Deep Space Network antennas (which
Table 3. Trajectory-correction schedule for the orbiting of SRG near the L2 point.

| No. | Date/time (Moscow) | Days in flight | Number of PU ignitions | Total characteristic speed $\text{m s}^{-1}$ | Consumed propellant kg |
|-----|-------------------|----------------|------------------------|-----------------------------------------------|------------------------|
| 1   | Oct. 21, 2019 19:00:00.000 | 100            | 1                      | 0.21                                          | 0.25                   |
| 2   | Dec. 10, 2019 19:00:00.000 | 150            | 1                      | 0.18                                          | 0.21                   |
| 3   | Jan. 30, 2020 19:00:00.000 | 201            | 1                      | 0.27                                          | 0.32                   |
| 4   | Apr. 01, 2020 19:00:00.000 | 263            | 1                      | 1.00                                          | 1.13                   |
| 5   | Jun. 16, 2020 21:00:00.000 | 339            | 1                      | 0.95                                          | 1.07                   |
| 6   | Aug. 05, 2020 20:00:00.000 | 389            | 1                      | 0.94                                          | 1.08                   |
| 7   | Oct. 05, 2020 19:00:00.000 | 450            | 1                      | 3.01                                          | 3.38                   |
| 8   | Nov. 23, 2020 19:00:00.000 | 499            | 1                      | 6.23                                          | 6.99                   |
| 9   | Feb. 28, 2020 17:00:00.000 | 588            | 1                      | 6.24                                          | 6.94                   |

Remaining propellant: 319.63 kg

usually happen in April and May), the Malargue, Cebreros, and New Norcia stations of the ESTRACK network are involved when necessary in receiving scientific data, according to the agreement between Roskosmos and ESA.

Ballistic maintenance of the mission is provided by two ballistic centers located at the Keldysh Institute of Applied Mathematics of the Russian Academy of Sciences and the Central Research Institute of Machine Building (TsNIIMash). All elements of the ground-control complex are united into one data system by means of a high-bandwidth data network, which ensures data exchange between the elements of the ground-control complex and the scientific ground complex.

5.7. Large corrections of the SRG orbit

After placement of the spacecraft into the nominal orbit near the L2 point (on day 100), station keeping corrections were executed approximately every 50 days until September 2020 (see Table 2). Because the launch date was chosen using many criteria, there is a gap of about a month every Spring in the radiovisibility of SRG in its nominal operational orbit from the Russian ground-control stations (see Fig. 37). In the Fall of 2020, a strategy of large manoeuvres began to be implemented (see Table 3) to alleviate this problem. This strategy was developed in accordance with Canalias & Masdemont (2004) and consists of one test manoeuvre (3 m s$^{-1}$) and ten manoeuvres (6 m s$^{-1}$ each). As demonstrated by calculations (see Fig. 38), it should enhance the visibility of SRG from the Russian Deep Space Network antennas during Spring in the next years. This will significantly improve the implementation of control operations with the Navigator platform and telescopes as well as the dumping of science data.

6. Mikhail Pavlinsky ART-XC

ART-XC on board the SRG spacecraft is an X-ray grazing-incidence mirror telescope array. It was developed by the

Space Research Institute (IKI) and the All-Russian Scientific Research Institute for Experimental Physics (VNIIEF). The NASA Marshall Space Flight Center (MSFC) produced the flight modules of the X-ray mirror systems. ART-XC is designed for conducting an all-sky survey in the 4–12 keV energy band and pointed observations of selected astrophysical objects in the 4–30 keV energy band.

6.1. Design

ART-XC consists of seven identical mirror systems (MS) paired with the unit of the Rontgen detector (URD) (Fig. 39). Each MS plus URD pair forms a telescope. All seven of them are pointing in the same direction.

The basic structure of ART-XC is a cone-shaped carbon fiber tube with a height of three meters. The MSs are mounted on top of this tube and focus X-ray photons onto their respective URDs. A sunshield protects the MSs from direct sunlight. The upper part of this tube is covered with copper shielding to reduce the

Fig. 39. Internal arrangement of the ART-XC telescope: cone-shaped carbon fiber tube with seven identical X-ray mirror systems at the top. The mirror systems focus X-rays onto seven detector units.
Fig. 40. Wolter-I optical scheme of ART-XC. $D_{fr_{28}}$ and $D_{fr_1}$ are the entrance apertures of the outermost and innermost shells, respectively, $l_h$ and $l_p$ are the heights of the paraboloids and hyperboloids, respectively, and $f_{W}$ is the focal length of the mirror system. Adapted from Pavlinsky et al. (2020).

stray-light background in the detectors. Each URD is equipped with a collimator to reduce stray-light background. The collimator includes a block of calibration X-ray sources ($^{55}$Fe + $^{241}$Am) for in-flight calibration. Heat pipes and a radiator are used to maintain the operating temperature of the detector at about $-20^\circ$C. The star tracker is mounted near the MSs.

The ART-XC MSs were produced and calibrated by MSFC (Gubarev et al. 2012, 2014; Krivonos et al. 2017). Each MS contains 28 Wolter-I (Fig. 40) nested-mirror shells. The shells were fabricated using an electroformed-nickel-replication technique and coated with a $\sim 10$ nm layer of 90% bulk density iridium. The shell thickness varies slightly with radius: the thickness of the outer shells is larger than the nominal thickness to make them stiffer and thus to improve the angular resolution of the module. The upper ends of the shells are glued to the supporting spider. The nominal focal length of the MSs is 2700 mm. They were defocused by 7 mm during installation on the telescope to provide a more uniform angular resolution across the FoV.

The ART-XC FoV is $\sim 36^\prime$ ($\sim 0.3$ sq. deg), within which the MSs provide an angular resolution better than $1^\prime$. The effective area (on-axis) is substantial up to energies $\sim 30$ keV. In addition, there are photons that are only once reflected from the MSs, and as a result, they can fall onto the detector with offset angles up to $\sim 50^\prime$. Figure 41 illustrates the effect of these singly reflected photons on the ART-XC FoV and their contribution to the FoV-averaged effective area. For this type of events, the ART-XC FoV is $\sim 2$ square degrees. Although true imaging cannot be done in this extended FoV and singly reflected photons generally cause an increase in the background, they can be used to measure the X-ray fluxes of bright sources. Therefore, the ART-XC telescope can also be exploited as a concentrator. In survey mode, ART-XC can therefore monitor bright transient sources for at least 28–32 h (Pavlinsky et al. 2019a).

The URD can register photons with energies up to $\sim 100$ keV, with an energy resolution of 9% at 13.9 keV. The detector strip size corresponds to an angular size of 45$''$. During a calibration campaign at the IKI X-ray test facility (Pavlinsky et al. 2018, 2019b,a), it was determined that the efficiency of the ART-XC spare URD reaches 50% in the 4.47–4.76 keV energy range and 90% in the 9.43–10.04 keV energy range.

6.2. Characteristics

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The main goal of ART-XC is to survey the whole sky in the 4–30 keV band and an angular resolution better than an arcminute. The ART-XC survey will thus be the most sensitive X-ray all-sky survey complements the all-sky survey performed by eROSITA (IAAT) therefore proposed the mission called A BRoad-band Imaging X-ray All-sky Survey (ABRIXAS).

Preliminary estimates show that during the four-year all-sky survey, ART-XC will detect ≥5000 X-ray sources, mostly AGN, including heavily obscured ones (with hydrogen column densities $N_H \geq 10^{21}$ cm$^{-2}$). This will provide a rich database for the exploration of the AGN population at $z \leq 0.3$.

ART-XC will also provide valuable information about the temperature of the intracluster gas in rich low-redshift clusters of galaxies. This will help tighten constraints on the cosmological parameters inferred from the eROSITA all-sky survey.

The ART-XC all-sky survey can make breakthroughs in studying various classes of Galactic X-ray sources, such as X-ray binaries and cataclysmic variables (CVs). As many as $\sim 10^3$ CVs can be found, compared to ≤100 known from previous X-ray surveys.

With its unique combination of broad energy coverage, good angular resolution, and wide FoV, ART-XC should make major advances in the study of SNRs and the Galactic Ridge X-ray emission.

Finally, ART-XC is well suited for discovering and monitoring transient and variable X-ray sources, such as X-ray and gamma-ray bursts, Galactic X-ray transients, and AGN. Although the probability of catching GRBs in the ART-XC FoV is not high, bright GRBs can penetrate the shielding material of the telescope and induce a signal on the detectors. For these events, ART-XC can provide precise timing information that can be used to localize them through triangulation with other space observatories.

In particular, in order to calibrate the ART-XC effective area, a series of observations of the Crab nebula was performed during the CallPV phase. The effective area proved to agree well with the results of simulations and ground calibrations (Pavlinsky et al. 2018, 2019b,a).

The concept of the eROSITA telescope is based on a long series of previous scientific and technological developments dating back to the very successful German, US, and UK ROSAT mission (1990–1999; Trümper 1982), which was developed and managed under the leadership of the Max Planck Institute for Extraterrestrial Physics (MPE). ROSAT carried out the first complete survey of the sky with an imaging X-ray telescope in the energy range between 0.1 and 2.4 keV, and it performed tens of thousands of pointed observations.

The following generation of large X-ray telescopes, the NASA-led Chandra Observatory and the ESA mission XMM-Newton, required the development of mirror systems with longer focal length (7.5–10 meters) in order to bring harder X-ray radiation into focus. These observatories, both launched in 1999, were only able to perform pointed observations, however, their limited FoV meant that it was not possible to perform large-area surveys. This led to the proposal of a hard X-ray imaging telescope with all-sky surveying capabilities. The Astrophysikalisches Institut Potsdam (AIP), the MPE, and the University of Tübingen (IAAT) therefore proposed the mission called A BRoad-band Imaging X-ray all-sky Survey (ABRIXAS).

From the very beginning, the ABRIXAS mission concept was developed through the coherent adaptation of mirror and detector technologies developed for XMM-Newton for a national, small-scale satellite mission. Consequently, the development timescale for the project was relatively short (about three years) and the overall costs moderate. The seven mirror modules shared in the focal plane an identical copy of the pn-CCD camera developed for XMM-Newton, and with their small focal length of 1.6 meters, they were ideal for a small satellite. Because of a design failure in the power supply, however, the satellite lost its main battery soon after launch, in April 1999, and could never be used for scientific purposes. The hard- and software developments enabled by the ABRIXAS mission have been extremely useful in the framework of a number of subsequent projects, however.

Despite the unfortunate outcome of the ABRIXAS mission, the appeal of the original scientific goal of an imaging hard X-ray all-sky survey remained high, with no other planned astrophysical mission with similar objectives on the horizon. With the help of the Semiconductor Laboratory (HLL) for the production of high-sensitivity detectors of the MPE, a new project to further develop the highly successful XMM-Newton pn-CCD technology was launched. In March 2002, all participating institutes proposed to ESA to accommodate the ROentgen Survey with an Imaging Telescope Array (ROSITA) telescope on an external platform of the International Space Station (ISS). The seven mirror modules would have been built identically to those of ABRIXAS, but each of them would have been equipped with its own newly developed, frame-store pn-CCD in the focal plane. In

### Table 4. ART-XC characteristics.

| Parameter                                | Value                                      |
|------------------------------------------|--------------------------------------------|
| Telescope mass                           | 350 kg                                     |
| Dimensions                               | $3.5 \text{ m} \times 0.9 \text{ m}$       |
| Power                                     | 150 watts                                  |
| Number of modules                        | 7                                          |
| Nominal focal length                     | 2700 mm                                    |
| Operational energy range                 | 4–30 keV                                   |
| FoV                                      | 0.3 sq. deg                                |
| Eff. area for pointed observations       | $385 \text{ cm}^2$ at 8.1 keV              |
| Grasp                                    | $43.8 \text{ cm}^2 \text{ deg}^2$ at 8.1 keV |
| Ang. resolution (FWHM) in survey         | 53″                                        |
| Energy resolution                        | 9% at 13.9 keV                             |
| Time resolution                          | 23 μs                                      |

6.4. In-flight performance

ART-XC has been operating in orbit for almost two years at the time of writing. The results obtained during the CalPV phase and the all-sky survey so far fully confirm the expected unique capabilities of the instrument.
September 2002, this proposal was supported by ESA with the highest scientific priority, and was recommended for phase A study. Launch would only have been possible from 2011 onward, however, because the external platforms of the ISS was occupied before then. Subsequently, it emerged that the ISS would not have been a viable option for ROSITA, first of all because NASA decided to terminate its shuttle flights to the ISS in 2010, and furthermore, because a contamination experiment on the ISS revealed that its environment was not safe from contamination for the sensitive X-ray mirrors and detectors of ROSITA.

At the turn of the millennium, the observations of supernovae type Ia by two independent groups (awarded with the Nobel Prize for Physics in 2011 for their discoveries) revealed that the expansion of the Universe is accelerating, a fact that may suggest the existence of a cosmological constant. The subsequent measures, taken by Boomerang and WMAP, of the tiny temperature fluctuations of the microwave background radiation out of which galaxies, clusters of galaxies, and the overall large-scale structure of the Universe originated, pointed toward a flat Universe, whereby the total matter plus energy content of the cosmos attains almost exactly the critical value. In fact, studies of the baryon fraction in X-ray selected (mainly from ROSAT) clusters of galaxies throughout the 1990s had already shown compelling evidence that the matter density was lower than unity (Schuecker et al. 2003).

The fact that a very large sample of clusters of galaxies can be particularly useful for precision cosmology stimulated many different groups to conceive dedicated large-area cluster surveys. In April 2003, members of the ROSITA team participated in the proposal of DUO, a NASA SMEX-mission, based on a modification of the ROSITA telescope design. Together with another four missions, DUO was selected among 36 competing proposals for a phase-A study, carried out in 2004. DUO was to survey an area of the sky of about 6000 deg², overlapping that explored by the optical Sloan Digital Sky Survey (SDSS). In this way, about 10,000 clusters of galaxies could have been detected, providing constraints on the dark energy density to an accuracy of within 10%. NASA did not select the DUO-project for further development, however, and the only mission of the five that was finally executed within the SMEX program was the hard X-ray focusing telescope NuSTAR, which was launched in 2012.

In February 2005, the Astronomy and Astrophysics Advisory Committee (AAAC), founded by the National Science Foundation (NSF), NASA, and the Department of Energy (DOE), and the High Energy Physics Advisory Panel (HEPAP), founded by NSF and DOE, established a Dark Energy Task Force (DETF) with the task of advising NSF, NASA, and DOE on the optimal strategies for the future of dark energy research. In particular, the DETF evaluated and compared various ground- and space-based instrumentation and observational methods. A dedicated white paper (Haiman et al. 2005) reached the conclusion that within available technologies at the time, it would have been possible to gather a sample of about 100,000 X-ray selected clusters of galaxies that would have provided very stringent constraints on the fundamental parameters of the cosmological model of the Universe. The construction of a sample of galaxy clusters of this magnitude is the primary objective of the eROSITA telescope. To achieve this goal, the ABRIXAS mirror design had to be modified to include 27 additional shells, doubling the diameter of each telescope module and increasing the effective area at low energies by a factor of five. With this fundamental rescope, eROSITA will likely be the first stage IV dark energy probe to be realized and will outperform the DUO-like stage IV probe originally considered by the DETF.

In June 2006, the funding proposal for eROSITA was submitted to the German Space Agency (DLR). Five German institutes (MPE, IAAT, AIP, the Hamburg Observatory, and the Dr. Remeis-Sternwarte in Bamberg, the Astronomical Institute of the Erlangen-Nürnberg University) agreed to work together to develop, build, and organize the scientific exploitation of the instrument and formed the German eROSITA Consortium. They were later joined by three more institutes: the Argelander-Institut für Astronomie at the University of Bonn, the Max-Planck-Institut für Astrophysik (MPA), and the Universitäts-Sternwarte
Table 5. Basic eROSITA instrument parameters in launch configuration.

| Instrument                  | 7 mirror assemblies | 7 camera assemblies |
|-----------------------------|---------------------|---------------------|
| Size                        | 1.9 $\times$ 3.5 m  |                     |
| Mass                        | 808 kg              |                     |
| Power                       | 522 W max.          |                     |
| Datarate                    | 600 MB/day max.     |                     |
| Diam. of outer shell        | 358 mm              | CCD image           |
| Number of shells            | 54                  | 2.88 $\times$ 2.88 cm$^2$ [1:03 $\times$ 1:03] |
| focal length                | 1600 mm             | Time Resol.         |
| HEW on axis (1.5 keV)       | 18$''$              | 75 $\mu$m $\times$ 75 $\mu$m [9.6 $\times$ 9.6] |
| HEW FoV averaged            | 26$''$              | Qu-Efficiency       |
| Mass                        | 808 kg              |                     |
| Power                       | 522 W max.          |                     |
| Datarate                    | 600 MB/day max.     |                     |
| Diam. of outer shell        | 358 mm              | CCD image           |
| Number of shells            | 54                  | 2.88 $\times$ 2.88 cm$^2$ [1:03 $\times$ 1:03] |
| focal length                | 1600 mm             | Time Resol.         |
| HEW on axis (1.5 keV)       | 18$''$              | 75 $\mu$m $\times$ 75 $\mu$m [9.6 $\times$ 9.6] |
| HEW FoV averaged            | 26$''$              | Qu-Efficiency       |

München (LMU). In March 2007, DLR approved the project and funded the eROSITA telescope, while a Memorandum of Understanding was signed between DLR and the Russian Space Agency (Roskosmos) to ensure that eROSITA would be launched in the framework of the Spektrum-Roentgen-Gamma mission. Soon afterward, the construction of the telescope began. In September 2008, Roskosmos came to a final decision on the payload orbit and launcher. eROSITA would be launched by a Zenit-Fregat launcher (later in 2016 to be replaced by a Proton-M and Block-DM combination) together with the Russian hard-X-ray telescope ART-XC on an L2 orbit to ensure maximum efficiency for a survey instrument. Eventually, additional funds from the Max Planck Society and DLR were secured for eROSITA in July 2009, to finance the cost-driving L2 orbit. A detailed agreement between Roskosmos and DLR was signed the following month.

In the following years, the teams in Germany and Russia were intensively kept busy in the development and construction of the instrument. The mirror modules, smaller versions of the *XMM-Newton* mirrors, were supposed to be built using the same technology and by the same industrial consortium. However, there were significant problems with the shorter focal length and the smaller mirror shells, as figuring errors have a significantly greater effect on the image quality. The mirror engineering model had only a 42 arcsec HEW, almost three times lower than required. Fixing these issues took more development than expected, but was ultimately successful. Another major challenge was the development of the electronics: after the decision for the L2 orbit, the basically completed design had to be changed to use radiation hard components. All qualification tests were carried out in the MPE test facilities or in the facilities of companies in the Munich area.

eROSITA was finally delivered to Russia in January 2017. A series of tests was then conducted at NPOL, before integration of the instruments into SRG could begin. This was then completed in April 2019 with the transport to the Baikonur Cosmodrome.

7.2. Instrument

eROSITA (Fig. 42) consists of seven identical and coaligned X-ray telescopes housed in a common optical bench in a hexagonal shape. A system of carbon fibre honeycomb panels connects the seven mirror assemblies on the front side with the associated seven camera assemblies on the focal plane side. The optical bench is connected to the S/C bus through a hexapod structure.

Each of the mirrors comprises 54 paraboloid or hyperboloid mirror shells in a Wolter-I geometry, with an outer diameter of 360 mm and a common focal length of 1600 mm (Friedrich et al. 2008; Arcangeli et al. 2017). The average on-axis resolution of the seven mirror modules as measured during the on-ground calibration is 16.1$''$ half-energy width (HEW) at 1.5 keV. The unavoidable off-axis blurring typical of Wolter-I optics is compensated for by a 0.4 mm shift of the cameras toward the mirrors. This places each telescope slightly out of focus and leads to a slight degradation of the on-axis performance (18$''$), but improves the angular resolution averaged over the FoV. Each mirror assembly has a CCD camera in its focus (Meidinger et al. 2021). The eROSITA CCDs are advanced versions of the EPICpn CCDs on *XMM-Newton*. They have 384 $\times$ 384 pixels in an image area of 28.8 mm $\times$ 28.8 mm, yielding a square FoV of 1:03 $\times$ 1:03. Each pixel corresponds to a sky area of 9$''$ 6 $\times$ 9$''$ 6. The nominal integration time for all eROSITA CCDs is 50 msec. The additional presence of a frame-store area in the CCD substantially reduces the amount of so-called out-of-time events, which are recorded during the CCD read-out, a significant improvement with respect to the PN camera on *XMM-Newton* (Strüder et al. 2001). For optimal performance during operations, the CCDs are cooled down to about $\sim$85$^\circ$ by means of passive elements (Fürmelt et al. 2008).

For calibration purposes, each camera has its own filter wheel with a radioactive $^{55}$Fe source and an aluminum–titanium target. This provides three spectral lines at 5.9 keV (Mn-K$\alpha$), 4.5 keV (Ti-K$\alpha$), and 1.5 keV (Al K$\alpha$).

The electronics for onboard-processing of the camera data is provided by seven sets of camera electronics (CE), each one
mounted and interfaced to the cameras. Each of the CEs provides the proper voltage control and readout timing of the associated camera and performs the onboard data processing within the time constraints of the camera integration time.

The Interface and Thermal Controller (ITC) receives the telemetry generated by each CE and stores it in the mass memory, commands each of the CEs, and controls the power distribution and the temperatures of the mirrors and cameras. Given its criticality, the ITC is a cold redundant unit (Coutinho et al. 2018).

Finally, two (redundant) star trackers are mounted on eROSITA for the determination of an accurate boresight. The dimensions of the telescope structure are approximately 1.9 m (diameter) × 3.2 m (height in launch configuration, with closed front cover). The total weight of eROSITA is 808 kg. Table 5 summarizes the basic eROSITA instrument parameters.

7.3. Early operations, first results, and performance

The commissioning phase of eROSITA had the objective of switching on all subsystems to verify they were functional following the launch and that they performed as expected to fulfill the scientific objectives. This phase served not only to test and commission the complete eROSITA telescope, but also for the teams in Khimky and Moscow (NPOL, IKI) and Garching (MPE) to learn and update the procedures on how to safely operate the spacecraft and the telescopes in space. The first mission-critical event was the switch-on of the ITC, which had to happen less than four hours after launch, to enable the thermal control of mirrors and electronics. The second event was the opening of the telescope cover, which occurred on July 22. Camera switch-on, instead, had to wait several days before being activated to avoid excess contamination from the first two spacecraft burns, which occurred on days 10 and 25 of the mission. In addition, camera cooling could not be performed without a minimum of 21 days of out-gassing following the cover opening. The commissioning of the cameras lasted about two months, including time necessary to perform a series of tests of the functionality of the camera electronics and thermal balance system.

All seven eROSITA X-ray telescope modules had been observing the sky simultaneously since October 13. Over the following eight weeks, eROSITA collected its first-light images and performed a series of observations designed to accurately calibrate the instruments and verify that the performance of the telescope met pre-launch expectations. The Russian and German science teams jointly defined this CalPV program, which included a combination of pointed observations, field scans, and full-circle scan tests.

Figures 43 and 44 show two examples of images captured by eROSITA during the CalPV and early survey phase. They highlight the main properties of this unique X-ray telescope, namely, its ability to take deep images, which are highly sensitive to both point-like and diffuse emission, over very large areas of the sky. As a preview of the eROSITA capabilities, a mini-survey called eROSITA Final Equatorial Depth Survey (eFEDS) was devised as part of the PV plan, imaging a 140 deg$^2$ patch of the sky to a depth comparable to that expected at the end of the all-sky survey (see Table 6). These data confirm the sensitivity of the X-ray telescope to its main target classes with high accuracy. This mini-survey revealed more than 20,000 point-like X-ray sources, about 80% of which are distant AGN harboring growing supermassive black holes. Most of the remainder are X-ray active stars.

Finally, on December 13, 2019, the all-sky survey began. It was completed on June 12, 2020, after 182 days of almost continuous scanning of the sky. Eight all-sky surveys are planned in total, each delivering an average exposure with eROSITA of about 200 s $\cos(lat)$, where $lat$ is the ecliptic latitude, while the area of 1 square degree around the ecliptic poles is revisited every four hours. This accumulates an exposure of about 30 ks per survey.

During the course of the all-sky survey, the spacecraft has been rotating continuously with a scan rate of 90 deg h$^{-1}$, giving
Table 6. Summary of performance characteristics of the eROSITA telescope and its survey sensitivity.

|                          | Energy range          |
|--------------------------|-----------------------|
|                          | Soft band             | Hard band            |
|                          | 0.2–2.3 keV           | 2.3–8 keV            |
| FoV averaged effective area [cm²] | 1.365 at 1 keV       | 139 at 5 keV         |
| Total background [10⁻³ cts s⁻¹ arcmin⁻²] | ≈3.7                 | ≈2.1                 |

Point source sensitivity eRASS:1

| Region                  | Total exposure [s] | Effective sensitivities [erg s⁻¹ cm⁻²] |
|-------------------------|-------------------|---------------------------------------|
| Ecliptic equatorial region | 200               | 5 × 10⁻¹⁴ erg s⁻¹ cm⁻², 7 × 10⁻¹⁳ erg s⁻¹ cm⁻² |
| Ecliptic polar region    | 4000              | 7 × 10⁻¹⁵ erg s⁻¹ cm⁻², 9 × 10⁻¹⁴ erg s⁻¹ cm⁻² |

Point source sensitivity eFEDS

| Region        | Total exposure [s] | Effective sensitivities [erg s⁻¹ cm⁻²] |
|---------------|-------------------|---------------------------------------|
| eFEDS field   | 2500              | 9 × 10⁻¹⁵ erg s⁻¹ cm⁻², 1.3 × 10⁻¹³ erg s⁻¹ cm⁻² |

Point source sensitivity eRASS:8 (predicted)

| Region        | Total/Effective exposure [s] | Effective sensitivities [erg s⁻¹ cm⁻²] |
|---------------|-------------------------------|---------------------------------------|
| Ecliptic equatorial region | 1600                        | 1.1 × 10⁻¹⁴ erg s⁻¹ cm⁻², 2.5 × 10⁻¹³ erg s⁻¹ cm⁻² |
| Ecliptic polar region      | 30 000                       | 2.5 × 10⁻¹⁵ erg s⁻¹ cm⁻², 4 × 10⁻¹⁴ erg s⁻¹ cm⁻² |

Notes. The background counts are based on the first all-sky survey data. For eRASS:1 and the PV eFEDS 140 deg² survey (Brunner et al. 2021), the flux sensitivity in each band has been computed by taking all sources detected above a likelihood of 8 (soft band) or 10 (hard band), and measuring the flux below which the logarithmic number counts start declining. For eRASS:8 the predictions are based on detailed simulations that include all instrumental effects and particle background intensity consistent with that measured at L2. For each field or region, we quote the total (unvignetted) exposure in seconds. The corresponding effective (vignetted) exposures can be computed by dividing the total exposure by 1.8 and 3.31 for the soft and hard band, respectively.

8. History of the SRG project in Russia

8.1. International X-ray astronomy projects in Russia

The history of international X-ray astronomy space projects in the USSR and Russia began with the Roentgen observatory on the Kvant module of the Mir space station. This project had been proposed by the Space Research Institute (IKI) of the USSR Academy of Sciences for implementation in the framework of the Soviet Intercosmos program. A number of European institutions were invited to this project: (i) the University of Birmingham (UK) and the SRON Netherlands Institute for Space Research, which built the TTM coded-mask-aperture X-ray telescope, sensitive in the 2–25 keV energy band, (ii) the MPE (Germany), which provided the HEXE hard X-ray spectrometer, sensitive in the 20–120 keV energy band, (iii) the European Space Research and Technology Center (ESTEC) of ESA, which built the GSPS high-pressure-gas spectrometer, IKI constructed the Pulsar X-1 hard X-ray detector for this project.

The Kvant module was put into orbit by a Proton rocket and successfully docked with the space station in April 1987. The main scientific result of the Roentgen observatory was the discovery in August 1987 of hard X-ray emission with an unusual spectrum from the very bright supernova SN 1987A, which had exploded five months earlier in the Large Magellanic Cloud (Sunyaev et al. 1987; Syunyaev et al. 1987). The detected photons were emitted as gamma-ray lines associated with the decay of radioactive cobalt 56. During the passage through the optically thick shell of the supernova, these photons experienced multiple Compton scatterings off relatively cold electrons and lost their energy due to recoil. As a result, an extremely hard power-law X-ray spectrum formed, and this spectrum was detected for several months by HEXE and Pulsar X-1. At energies below 20 keV, photoabsorption of photons by ions of heavy elements, first of all iron and cobalt, came into play. This explained the absence of a detectable signal by TTM.

In 1987–1995, TTM discovered many transient and persistent sources, which now wear the names “KS” (Kvant source, Sunyaev et al. 1991). The availability of instruments sensitive in different bands of the X-ray spectrum allowed broadband spectra in the 2–200 keV energy range to be constructed for very many bright transient and persistent sources in binary stellar systems.
systems, including black holes and neutron stars (X-ray pulsars and neutron stars with weak magnetic fields) (Syunyaev et al. 1991, 1994). It was demonstrated for the first time how strongly the spectra of X-ray binaries depend on the nature of the accreting object. TTM provided excellent timing observations of X-ray pulsars (Gilfanov et al. 1989) and high-quality X-ray images of the Galactic center region.

The second X-ray orbital observatory of IKI within the Intercosmos program was the GRANAT satellite, built by NPO Lavochkin and put into an elongated elliptical orbit around Earth with a four-day period by a Proton rocket. It operated in orbit from December 1989 to May 1999. The payload of GRANAT included the SIGMA hard X-ray and gamma-ray telescope with a germanium spectrometer, the PHEBUS gamma-ray burst detectors developed in France, the WATCH all-sky monitor (Denmark), and the ART-P instrument constructed by IKI and its subsidiary in Frunze (now Bishkek, Kyrgyzstan), sensitive in the 2–25 keV energy band. Both SIGMA and the ART-P telescopes had a coded mask and position-sensitive detectors, permitting construction of the images.

The main results of the GRANAT mission include the maps of the central region of the galaxy in hard X-rays by the SIGMA (40–100 keV) and ART-P (3–30 keV) telescopes (e.g., Churazov et al. 1994; Gilfanov et al. 1993; Pavlinsky et al. 1994). WATCH discovered the extremely interesting black hole X-ray binary GRS 1915–105 (“GRS” means that it is a GRANAT source, Castro-Tirado et al. 1992). The spatial correlation of the 8–22 keV X-ray emission observed by ART-P with the distribution of molecular gas in the Galactic center region became the first indication of the X-ray echo of the past Sgr A* X-ray activity (Sunyaev et al. 1993) that is scattered by H2 molecules in the dense gas surrounding it, according to the prediction by Vainshtein & Sunyaev (1980). PHEBUS observations helped to clearly demonstrate (Tkachenko et al. 2002) the existence of two types of gamma-ray bursts: short and normal, which, as we know now, is closely related to the parameters of the stellar objects that are responsible for their appearance (merger of neutron stars in the case of short bursts and collapse of massive stars in the case of normal bursts).

In 2002, according to the agreement between ESA and Roscosmos, a Proton rocket placed the International Gamma-ray Laboratory (INTEGRAL) into a high-apogee orbit. INTEGRAL continues to provide scientific data to this day. Russian scientists are granted 25% of the observing time of this mission. The prominent results of INTEGRAL include the detailed spectroscopy of the electron-positron annihilation radiation (Churazov et al. 2005) and the first detection of gamma-ray lines from the type Ia supernova SN2014J (Churazov et al. 2014).

8.2. Beginning of the SRG project in the USSR

In 1987, the thirtieth anniversary of the launch of the first artificial satellite of Earth (Sputnik) was celebrated. On this occasion, a large international meeting was held at IKI, where Intercosmos announced a competition for the payload of two orbital observatories. This resulted in a decision to support projects Radioastron and Spectrum-Roentgen-Gamma (SRG). The SRG project was supported by a number of renowned Soviet physicists, including Ya.B. Zeldovich, A.D. Sakharov, and the IKI director R.Z. Sagdeev. In addition to the High Energy Astrophysics Department of IKI, which proposed the SRG project, it was joined by Denmark, the UK, Italy, MPE (Germany), NASA (USA), Finland, Switzerland, Israel, and Turkey. The work on the project was in full swing (see Fig. 45), but huge changes in the USSR led to a slowdown in the works on the spacecraft and instruments in 2001–2002, after which the project was terminated.

In 2003, the Space Council of the Russian Academy of Sciences considered and supported a more modest project with a launch by a Soyuz rocket (instead of a Proton), a smaller number of instruments, and with a shift in the scientific goals toward cosmology, fine X-ray spectroscopy, and monitoring the whole sky in X-rays. The idea of attracting international cooperation to the project was also supported. The project was then considered by Roscosmos, and a key decision was made to elaborate it further. It was also decided to retain (for the sake of continuity) the former project’s name SRG, although the new project did not foresee the presence of instruments sensitive in the gamma-ray range.

Among the first invited to discuss a possible participation in the project were scientists from the MPE in Germany, with whom the High Energy Astrophysics Department of IKI has closely worked during the Roentgen mission on Mir/Kvant and on the first version of SRG. Moreover, MPE had developed and built the X-ray telescope for the extremely successful ROSAT satellite, which obtained wonderful X-ray maps of the sky in 1990.

MPE proposed using a modification of the telescope that was developed by MPE for the unfortunately failed ABRIXAS mission for observations of clusters of galaxies (the most massive objects in the Universe, which are of interest to the Russian side as well). This telescope and its detector were designed for exploring the sky in a significantly harder energy band (from 0.5 to 10 keV) compared to ROSAT (see a more detailed discussion in Sect. 7.1). By this time, the number of massive clusters of galaxies required for the detection of baryonic acoustic oscillations in their spatial distribution had been calculated at the Max Planck Institute for Astrophysics (MPA) and was proved to be close to 100 000 (Hütsi 2006). This would require finding virtually all massive clusters of galaxies in the observable Universe. This question was actively discussed at MPA in connection with searches for the baryonic acoustic oscillations predicted by theorists (Peebles & Yu 1970; Sunyaev & Zeldovich 1970) and with emerging hopes to find a large number of galaxy clusters through the Sunyaev–Zeldovich effect using ground-based telescopes and spacecraft operating in the microwave band. Therefore the question arose about a telescope with a higher sensitivity and working in a softer X-ray energy band compared to the existing modifications of the instrument for the ABRIXAS satellite, which could not provide the necessary sensitivity to fulfill this task, nor the majority of other tasks that were of interest for cosmodists at the time.
After long discussions, the Russian side agreed on drastic increases in the size, mass, (800 kg) and power consumption of the German instrument to fulfill the requirements (according to elaborations by G. Hasinger and P. Predel) posed by the planned scientific results of the sky survey. The parameters of the Russian ART-XC instrument were also growing in parallel.

On March 23, 2007, Roscosmos and DLR signed a memorandum about the inclusion of the eROSITA instrument into the payload of the SRG orbital observatory. This opened the way to intensive work on the project and the telescope.

8.3. Data sharing between the participants of the SRG project

All data obtained by the ART-XC telescope belong to the scientists of IKI, who have developed the telescope. The data within an area of 200 square degrees around the north ecliptic pole (where the sensitivity of the all-sky survey is highest) are processed together by scientists at IKI and the MSFC, in exchange for the delivery of the latter of a subset of the ART-XC X-ray mirrors.

According to the memorandum of 2007 between Roscosmos and DLR, the SRG/eROSITA data belong in equal parts to the German and Russian scientists. Because the primary goal of SRG/eROSITA consists of constructing all-sky X-ray maps and creating X-ray source catalogs, a decision has been made that the German scientists are responsible for the processing and publication of data in one hemisphere of the sky, and the Russian scientists provide this in the other hemisphere (see Fig. 46).

9. Conclusion

The SRG observatory has been operating in orbit for almost two years at the time of writing. The ART-XC and eROSITA telescopes have already scanned 40% of the entire sky for the third time. The experience of 25 months of operation of SRG allows us to say that the initial plans of the observatory are being successfully implemented and the number of extragalactic objects being discovered corresponds to the expectations before the launch. Using the new data obtained in the fourth scan of the sky, it is possible to search for sources whose X-ray brightness has strongly changed over the period of 6–18 months that passed since their previous appearances in the telescopesc’ FoV during the first three all-sky surveys. Every day, the observatory discovers about five to ten objects within a 360° × 1° strip (i.e., less than 1%) of the sky whose brightness has increased more than ten times over 6 months. The depth of the X-ray maps and the number of detected sources continues to increase with increasing exposure. The observatory is continuing its round-the-clock mission.

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