Study of Azimuth Angle and Elevation Angle Variations of IRNSS/NavIC Signals

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Abstract The IRNSS satellite network covers a large area of the India with 7 satellites, 3 geostationary and 4 that are geosynchronous. IRNSS signals are transmitted in the L5 and S-Bands. The L5 band frequencies range from 1164.45-1188.45 MHz and the S band frequencies range of 2483.5-2500 MHz. NavIC is the IRNSS satellite constellation network that can be employed to a variety of applications. The paper shows the analysis of the azimuth and elevation angle with respect to carrier to noise ratio. This study will be stepping stone for the collection of data using reflectometry. The different graphs were obtained and analyzed. Satellite tool kit is also used for the analysis of the elevation and azimuth angle.

Keywords RNSS, Elevation Angle, Azimuth Angle, TOWC

1. INTRODUCTION

The Indian Regional Navigational Satellite System (IRNSS) is owned by the Government of India which is developed by Indian Space Research Organization (ISRO). This was renamed later as Navigation with Indian Constellation (NAVIC). This provides service of 1500km range around India. The IRNSS provides location and time to the IRNSS receiver [1,2]. The three segments in IRNSS are User segment, Control segment and Space segment. The space segment has 8 satellites where currently satellite 1’s clock is not working in which it failed in orbit to give position information. Four are in geostationary orbit (GEO) whereas 3 are in geosynchronous orbit (IGSO). These satellites are positioned in 34°, 83°, 132°, 55° E and 111.75° E. The operation and maintenance of the IRNSS stations is done by the Ground segment. It consists of Data uploading stations, Master control stations and Base stations. The User segment consists of antennas and receivers to receive the IRNSS signals. Other than IRNSS signals it can also receive Galileo, GPS and GLONASS signals. The bandwidth in which the ISRO has filed is 24MHz in L5 Band (1164.45 – 1188.45 MHz) and 16.5 MHz in S1 Band (2483.5-2500 MHz) in which IRNSS receiver works and the frequency spectrum shown in Figure 1 and Figure 2 [4].

![IRNSS Band](image_url)
Figure 3 shows the representation flow of NavIC user receiver which follows: LNA, RF Front End, Correlators and demodulators, Navigation Processor and User Interface. This can receive both IRNSS and GPS data [14]. It has separate LNA for L5 and S1 band where the gain is 31 dB and 30 dB with corresponding noise figure is 0.9 dB for both. This receiver can operate in both modes i.e., dual frequency or single frequency. The services offered by the user receiver and the antenna are Standard Positioning Service (SPS) which is a free of cost service provided to the user and the Restricted Service (RS) which provides to authorized users. CDMA (Code Division Multiple Access) and BPSK (Binary Phase Shift Keying) modulation is used by the SPS signal. The navigation processor processes the navigation solution computation and the user interface interfaces the external device [3, 13].

2. Azimuth and Elevation Angles

When the apparent position of a satellite in an orbit is to be determined relative to a point of observation, we need angles namely Azimuth and Elevation. These together are the antenna look angles [5]. Information needed for calculating the look angles are: latitude and longitude of the earth station and the satellite orbital position. The azimuth and elevation angle settings of its antenna must be known to the earth station to communicate with the satellite [15, 16].

- **Azimuth angle**: It is the angle between a satellite and the North which is measured clockwise around the observer’s (antennas) horizon or earth station’s horizontal plane, in general referenced north (0°) or to south (180°) clockwise.
- **Angle of Elevation (EL)**: Vertical angle between the travelling direction from the earth station antenna to the satellite and the horizontal plane which is the earth’s horizon [6].
- In non-geostationary orbits the measures of these angles vary while for geostationary orbits these angles remain the same since satellites in GEO are in sync with earth’s rotation and seem stationary.

For any receiver to determine its position it needs three satellites with its position and the distance between each of them and receiver is also known. It needs a fourth satellite for providing time. Navigational information obtained from these satellites is primary and secondary navigational parameters. Satellites broadcast two types of data almanac and ephemeris which are used for positioning. Ephemeris data is unique for every satellite. All the GNSS systems,
GPS of The United States, GLONASS of Russia, GALILEO of Europe and BEIDOU of China[7], additionally there are regional navigation systems which are QZSS of Japan and India’s very own IRNSS - use extremely accurate Rubidium or Cesium stable atomic clocks, with error probability of 1 second in every hundred million years, while receivers back on earth have crystal oscillators for providing time, these are chosen for economic reasons and are not synchronized with on board atomic clocks of the navigational satellites. This asynchronization reflects in calculated position and range. There are many other factors that cause delays and errors: ionospheric delay, tropospheric delay and multipath error [9]. The Distance commuted is known as the product of speed and time taken for travel, here the distance between satellite and receiver is determined by multiplying the time taken for the signal to reach the receiver from the transmitter satellite with the speed of light [10].

3. Data Collection

Data used for this analysis was collected on October 12th 2018 at JAIN GLOBAL CAMPUS’s CRICKET GROUND. Two receivers namely, A_37 and A_38 are used. The arrangement is such that the receivers were 30 meters apart facing each other. There are four cases of this survey based on the direction at which the receivers were placed ((i) North and South, (ii) East and West) and depending on each of the receivers being static and dynamic. The dynamic receiver was moved towards the static receiver and readings were taken simultaneously. Throughout the survey we were able to observe variations in elevation and azimuth angles with respect to its TOWC of all 7 functional IRNSS satellites, using this information various graphs are plotted [11,12].

4. Elevation and Azimuth Angle Variations

The following are basic graphs obtained by the navigational data received by A_37 and A_38 receivers. Parameters TOWC, elevation and azimuth angles are used for plotting the graphs.

4.1. Case 1

Receiver A_37 is dynamic and receiver A_38 is static. Receivers were 30 meters apart facing each other in north and south direction. The dynamic receiver A_37 was moved towards the static A_38 and readings were taken accordingly (Figure 4 (a) and (b)) and the variation of satellite visibility during the survey period shown in Figure 5. Figure 6 shows the carrier to noise variation of all IRNSS satellites during the survey. It is observed that the variations are below 25 dBiHz represents the distractions during receiving the signals.

(a) A_37 receiver
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Figure 4. Comparison of both azimuth and elevation angles with time of the week count for the entire survey

Figure 5. Satellite visibility with respect to Dilution of Precession during the survey

Figure 6. IRNSS L5 band C/N0 during the survey
4.2. Case 2

In this set of survey, receiver A_37 is static whereas A_38 is dynamic; again facing towards each other at north and south directions. Here receiver A_38 is moved towards A_37 and the readings are taken accordingly (Figure 7(a) and (b)). Figure 8 shows the carrier to noise variation of all IRNSS satellites during case 2 survey.

![Figure 7. Plot for elevation and azimuth angles with respect to TOWC of receiver](image)

![Figure 8. IRNSS L5 band C/N0 during the survey when A38 is in static condition](image)

4.3. Case 3

As before, both receivers A_37 and A_38 are facing towards each other but now in east and west directions. First case is A_37 being static and A_38 being dynamic, A_38 is moved towards A_37. The roles are reversed that is, A_37 is dynamic and A_38, here A_37 is moved towards A_38 (Figure 9).
Figure 9. Plot of Azimuth and Elevation vs C/No

Figure 10. C/No vs Elevation angle

Figure 11. Graph plotted for both azimuth and elevations angles of A_37 and A_38 versus its TOWC.

Figure 9 shows the plot between the Azimuth, Elevation angle and Carrier to noise ratio. We can observe that there are drops in C/No ratio with respect to elevation and azimuth angle. When the azimuth angle is at 120.255478 degree the C/No is zero because at that particular angle the carrier cannot track the channel and hence the reject clock varies randomly and the lock time will reset to zero and starts counts. The same holds good for the elevation angle. This can also be observed in Figure 10. Where the satellite is visible but still, we could not able to track the elevation and azimuth angle of a particular satellite.
Figure 11. Shows the plots of elevation and azimuth angle with respect to TOWC. Here we can observe that if few places the angle is dropping to zero. Even though the visibility of the satellite is there but the azimuth and elevation angle cannot be retrieved because of issues in tracking the satellite which make the reject clock to vary abruptly and the lock time will restart.

4.4. Case 4

Here the receivers are facing towards each other positioned in east and west directions. In this case receiver A_37 is dynamic and A_38 is static which means that A_37 is moved from its position towards A_38. Following are graphs obtained from the readings taken (Figure 12) and the variation of satellite visibility during the survey period shown in Figure 13. Figure 14 shows the carrier to noise variation of all IRNSS satellites during the survey. It is observed that the variations are below 25 dBHz represents the distractions during receiving the signals.
5. Elevation and Azimuth Angle v/s Time and Distance

The below graphs are obtained with the help of satellite tool kit. They are comparisons of elevation and azimuth angles of all the seven functional satellites IRNSS-1B, IRNSS-1C, IRNSS-1D, IRNSS-1E, IRNSS-1F, IRNSS-1G and IRNSS-1I, whereas IRNSS-1A and IRNSS-1H aren’t functional. The purpose of using STK is to simulate the altitude and position of both GPS and IRNSS satellites in accordance with user location. Speed, angle of satellite movement, visibility of satellites, elevation and azimuth angle are determined using time and receiver as their reference [1,5,8]. The latitude and longitude values are entered by the user manually. The two line elements (TLE) format is used to feed the data of IRNSS and GPS satellite into the STK platform. TLEs can describe the trajectories only of earth orbiting objects. “.tle” files which are used to map, track, examine and understand the satellite movement and behaviour are generated by ‘CELESTRAK’ with the receiver information given by the user. Path of the seven functional satellites of IRNSS are shown in fig. 15. The figure of ‘8’ path of Geo synchronous satellites(1B, 1D, 1E, 1I) is clearly visible; which is shown in the Figure 15, it shows the visibility of all the seven satellites in a 2 dimensional view.

Figure 14. IRNSS L5 band C/N0 during the survey when A37 is in static condition

Figure 15. Path of Geo synchronous and Geo stationary satellites.
The start and stop considered for this graph is 12/10/2018-6:30AM and 13/10/2018-6:30AM, respectively. There are 3 parameters which are used to plot the graph. They are time, angle and distance. The graph describes the variations in angle and distance over a particular time period. We are analyzing the graph for a fixed duration of 24 hrs. The start time is 12th October at 6:30a.m to 13 October 6:30a.m, with a step of 60 seconds. In Figure 16 graph the elevation angle and azimuth angle variations are clearly visible. From Table 1 it can be observed that the minor variation in the positional values i.e., latitude, longitude and altitude when compared with IRNSS and GPS receivers during the survey.

Table 1. Positional comparison between IRNSS and GPS during survey

| TOWC (s) | NavIC/IRNSS | GPS |
|----------|--------------|-----|
|          | Latitude (deg) | Longitude (deg) | Altitude (m) | Latitude (deg) | Longitude (deg) | Altitude (m) |
| 456210   | 12.6385      | 77.4412        | 616.0310     | 12.6383      | 77.4413        | 615.8921     |
| 456211   | 12.6385      | 77.4412        | 616.0528     | 12.6383      | 77.4413        | 615.9574     |
| 456212   | 12.6385      | 77.4412        | 615.3750     | 12.6383      | 77.4413        | 615.9575     |
| 456213   | 12.6385      | 77.4412        | 615.0501     | 12.6383      | 77.4413        | 616.1058     |
| 456214   | 12.6385      | 77.4412        | 615.0161     | 12.6383      | 77.4413        | 616.6309     |
| 456215   | 12.6385      | 77.4412        | 615.3763     | 12.6383      | 77.4413        | 616.4206     |
| 456216   | 12.6385      | 77.4412        | 615.5405     | 12.6383      | 77.4413        | 616.1795     |
| 456217   | 12.6385      | 77.4412        | 615.3326     | 12.6383      | 77.4413        | 616.2107     |
| 456218   | 12.6385      | 77.4412        | 616.3370     | 12.6383      | 77.4413        | 615.9951     |
| 456219   | 12.6385      | 77.4412        | 615.9829     | 12.6383      | 77.4413        | 615.7548     |

6. Conclusion and Future Scope

The IRNSS satellite network covers a large area of the India with 7 satellites, 3 geostationary and 4 that are geosynchronous. IRNSS signals are transmitted in the L5 and S-Bands. The L5 band frequencies range from 1164.45-1188.45 MHz and the S band frequencies range from 2483.5-2500 MHz. NavIC is the IRNSS satellite constellation network that can be employed to a variety of applications. In this paper the elevation and azimuth angle are studied with respect to the Carrier to noise ratio at different places and at different instant of time. The elevation angle is very much important for the study of the soil moisture using reflectometry technique. At different point of time even though the satellite is visible we could not able to retrieve the information about azimuth and elevation angle of particular PRN. This study shows that it is very important to know the satellite position before
collect the data in the ground station.

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