Prediction of the Cyclic Life of Pieces with Macrocracks by Thermographic Method

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Abstract

To improve the accuracy for prediction of cyclic life of pieces with macrocracks we propose to use a new thermographic method. Traditionally this question is solved on the basis Paris formula which connects the speed of crack growth (SCG) with Stress intensity factor $K$. However parameter $K$ is not identical to the SCG because $K$ doesn’t consider non-linear processes at the top of crack (TC). That is why the using $K$ gives the considerable error. For overcoming this problem we proposed instead of $K$ to connect SCG with another diagnostic parameter, such as $\Delta S^{(i.e.)}$—increment of specific entropy for cycle (ISE) at the TC, which can be calculated with sufficient accuracy through passive temperature field on the surface of tested object. Parameter ISE can be obtained both simultaneously with building of a kinetic fatigue diagram and on the basis of measuring of temperature under exploitation of piece. In both cases the prediction of cyclic lifetime is much higher than with the help parameter $K$. Besides parameter ISE allows to follow the crack development inside tested object. This means that suggested parameter ISE is more universal and convenient than traditional parameter $K$.

Keywords

Change of Temperature, Speed of Crack Growth, Specific Entropy, Accuracy of Prediction

1. Introduction

Under cyclic deforming of pieces with macrocrack and certain conditions some plastic domain in top of crack is appeared. As a result the crack begins its movement. As it is known, under plastic deforming of metals the most part of mechanical energy is transformed in the heat energy. Therefore, a heat source
arises at the top of a growing crack. Because of high heat conductivity of metals these heat sources form a passive thermal field on the surface of testing object, which characterizes the irreversible changing in the material and has a lot of information about damaging processes. This information can be received without contact with investigated object by using up-to-date infrared equipment allows to fix the kinetics of temperature distributions near top of crack with high precision. Thermographic method, based on that phenomenon, has some advantages in comparison with traditional approaches. These advantages consist in a higher accuracy and universality, because thermographic method sufficiently enlarged range of tested pieces [1]-[7]. Note also, that Stress intensity factor $K$ is attribute of linear continuum and doesn’t consider non-linear processes at the top of crack (TC) [1] [3] [6] [7].

2. Objectives and Method of Research

The main problem in the thermographic method is correct choice of damage parameter. Such natural parameter is temperature. But there is essential moment. It is necessary to calculate not itself temperature in the domain of damage, but its change for sufficiently small interval of time, let us say, for one cycle of oscillation. In that case influence of background’s temperature is practically excepted.

As direct damage’s parameter we use $\Delta S^{(s)}$—increment of specific entropy in the domain for cycle. Entropy is function of state, which more full reflects all irreversible changing in domain of damaging. Besides $\Delta S^{(s)}$ calculated through increment of temperature on the surface of investigated object registered by up-to-date thermographic equipment allowing to measure it without contact with high accuracy. That is why, the thermographic method essentially increases precision of cyclic life prediction.

Traditionally this question is solved on the basis of Paris formula [6]-[12]

$$\frac{dl}{dn} = B \cdot \Delta K^{m_1}$$

(1)

or its variety

$$\frac{dl}{dn} = v_0 \cdot \left( \frac{K_{\text{max}}}{K_*} \right)^{m_2},$$

(2)

which connect the speed of crack growth $\frac{dl}{dn}$ with the maximum value of stress intensity factor (SIF) $K_{\text{max}}$ or its increment $\Delta K = K_{\text{max}} - K_{\text{min}}$ for one cycle.

In the Formulas (1) and (2) $B$, $m_1$, $m_2$—empirically defined parameters of material, $v_0 = 10^{-7}$ m/cycle—given speed of crack growth, $K_*$—parameter corresponding to $v_0$.

Kinetic fatigue diagrams, built by testing of some samples under different loads on the basis Formulas (1) or (2), are characterized the material of tested samples.
Integration of Formula (2)

\[ n - n_0 = \int_0^l K_{\text{cr}} \frac{dl}{V \cdot K_{\text{max}} (l)} \]  

(3)

gives dependence \( n(l) \) allowing to calculate the lifetime of the piece \( n - n_0 \), which corresponds to the crack growth at a critical length \( l_c \).

It must be noticed that using SIF as criterion of crack growth is often criticized at last years. SIF is attribute of linear elastic medium and using it as parameter of destruction not takes into account of many factors which have influence on the events near top of crack. It leads to essential, often unpredicted errors in cyclic life calculation.

Besides we notify the some limitations on using formula (3). These limitations connect with necessity of receipt function \( K_{\text{max}}(l) \). It leads to using of standard details for which function \( K_{\text{max}}(l) \) is certain. In case of cyclic life prediction for non standard pieces, that function can be received by experiment. But in that case it is necessary overcomes essential difficulties, connected with measuring length of growing crack especially if the crack grows inside detail.

These difficulties to some extent are softened and overcame under using as damage parameter \( \Delta S^{(v)} \) —increment that part of specific entropy, produced in damage domain, which caused it’s direct heating [13] [14]:

\[ \Delta S^{(v)} = c_v \ln \left( \frac{T_2}{T_1} \right). \]  

(4)

here \( c_v \)—specific heat capacity of material, \( T_1 \) and \( T_2 \)—temperatures of domain at the beginning and end of given cycle.

Our investigations show that under using thermographic method it is conveniently to keep structure of (2). Corresponding formula lead to

\[ \frac{dl}{dn} = V \cdot \left[ \frac{\Delta S^{(v)} (n)}{\Delta S^{(v)}} \right]^m \]  

(5)

For prediction of cyclic life Formula (5) must be integrated and after that critical growth of crack \( \Delta l_c \), corresponding cyclic life \( n_c \), is determined:

\[ \Delta l_c = \int_0^{n_c} V \cdot \left[ \frac{\Delta S^{(v)} (n)}{\Delta S^{(v)}} \right]^m \, dn. \]  

(6)

We tested some samples, produced from various materials under different loads and built kinetic fatigue diagrams (Figure 1) on the basis Formula (5). These diagrams have qualitatively the same form as corresponding diagrams built by Paris Formula (2), but our diagrams allowed to define the parameters of materials not only tested samples.

On the Figure 1 such diagrams are built after testing 10 samples, loaded bending oscillations (Figure 2). As a result we defined parameters of tested steel 20. Under \( p = 0.5 \) and \( m = 5.6 \): \( V = 10^{-8} \) m/cycle, \( \Delta S^{(v)} = 1.4 \int (m^3 \cdot K) \).
Fig. 1. Diagrams $\frac{dl}{dn} = f(\Delta S^{(c)})$ for steel 20 under various probabilities $p$.

Fig. 2. Sample for testing. $H = 15$ mm, $b = 10$ mm (thickness of sample), $L = 260$ mm, $l = 0.75$ mm.

These values are taken as middle point on the linear middle part of diagram, which practically defined the cyclic life of samples.

Increment of crack $\Delta l$ for some loading cycles $n$ is fixed by microscope “MBC-11” with net on the objective. The temperature at the top of growing crack is measured with help of thermo visor “Rubin MT”, having sensibility 0.01˚C under turned off scanning mechanism and took down by automatic apparatus “Endim-621.02”. $\Delta S^{(c)}$ is calculated by Formula (4).

Dependence $\Delta S^{(c)}(n)$ presents monotonously increasing function (Fig. 3) and it can be represent as
Figure 3. Diagram $\Delta S^{(1c)} = f(n)$, built for one sample under oscillation's amplitude of free end $f = 3$ mm.

$\Delta S^{(1c)} = \beta n^a$

(7)

where $\beta$ and $a$—empirical coefficients.

Dependence (7) under constant oscillation’s amplitude can be received by two methods:

- direct from diagram (Figure 1); in that case function must be considered as characteristic of material and prediction of cyclic life will be probable;
- by testing givens ample under some small cycles of loading (so keep the cyclic life), or by observing for detail under its exploitation; in that case $\Delta S^{(1c)}(n)$ will be characteristic of given sample and prediction of cyclic life will be individual.

3. Results of Experiments

All tested samples are brought to destruction, and as result its actual cyclic life $n_f$ is defined. Cyclic life, calculated from experiment, is defined by first method ($n_{1f}$) and by second one ($n_e$). After that it is counted divergence $\delta$ between $n_{1f}$, $n_e$ and $n_f$.

The results both experiment and calculation for one sample are showed in Table 1.

For all samples tested under one-stage loading, divergence $\delta = n_e - n_f$ was less on 20% - 25% than $\delta = n_{1f} - n_f$. It is accounted, because under calculation $n_e$ the sample gives information about only itself cyclic life.

Thermographic method allows predict cyclic life and many-stages loading. In that case Formula (6) become as

$$\sum \int_{n_{1f}}^{n_e} \frac{v_i}{(\Delta S^{(1c)})^m} \left[\Delta S^{(1c)}(n)\right]^m dn = \sum \Delta l_i.$$  

(8)
Table 1. Calculating and actual cyclic life for one sample (f = 2 mm).

| Method of calculation | α    | β     | ΔL, mm | n₀, cycles | n₁, cycles | nₑ, cycles | δ,%     |
|-----------------------|------|-------|--------|------------|------------|-------------|---------|
| Probable              | 0.44 | 0.0046| 6.61   | 6.21 × 10⁵ | 7.67 × 10⁵ | -           | 24.9    |
| Individual            | 0.495| 0.00223| 6.61   | 6.21 × 10⁵ | -           | 7.4 × 10⁵   | 19.2    |

Table 2. The results of cyclic life prediction for 10 samples produced from steel 20.

| № sample | First step of loading | Second step of loading |
|-----------|-----------------------|------------------------|
|           | f, mm | n₀·10⁻⁵, cycles | nₑ·10⁻⁵, cycles | δ,% | f, mm | n₀ cycles | nₑ cycles | δ,% |
| 1         | 2     | 6.21           | 7.40             | 19.2 | -     | -         | -         | -   |
| 2         | 2     | 6.00           | 7.13             | 18.8 | -     | -         | -         | -   |
| 3         | 2.5   | 3.20           | 2.92             | 8.8  | -     | -         | -         | -   |
| 4         | 3     | 2.31           | 2.61             | 13.1 | -     | -         | -         | -   |
| 5         | 3.5   | 0.347          | 0.43             | 29.6 | -     | -         | -         | -   |
| 6         | 2     | 2.82           | -                | -    | 3     | 1.1 × 10⁵ | 0.79 × 10⁵ | 28.2|
| 7         | 2.5   | 2.03           | -                | -    | 3     | 2·10⁴    | 1.54 × 10⁴ | 23  |
| 8         | 2.5   | 2.65           | -                | -    | 3.5   | 10⁴      | 1.31 × 10⁴ | 31  |
| 9         | 2     | 3.11           | -                | -    | 2.5   | 2 ·10⁴   | 1.73 × 10⁴ | 13.5|
| 10        | 3     | 2.25           | 2.70             | 20   | -     | -         | -         | -   |

With multistage loading, it is possible to consider various tasks for determining a cyclic resource. In particular, the task can be put as: the numbers of cycles \( n₁ \) and crack increment on the previous stages of loading are known; it is necessary to calculate the number of cycles to destruction \( nₑ \) on the last stage.

Peculiarity of calculation by Formula (8) consists in necessity to count over again the actual number of cycles from previous stage of loading on equivalent ones on the next stage, proceeding from the same increment of crack.

Let us examine that peculiarity at the example of cyclic life calculation for one sample on the second stage of loading (Table 2).

First stage: \( f = 2 \text{ mm}, n₀ = 2.82 \times 10^5 \text{ cycles}, \Delta L = 0.65 \text{ mm} \).

Second stage: \( f = 3 \text{ mm}, n₀ = 1.1 \times 10^5 \text{ cycles} \) (destruction), \( \Delta L = 2.31 \text{ mm} \).

Let us calculate cyclic life at second stage \( nₑ \) and compare it with \( n₀ \).

Using Formula (7) under \( f = 3 \text{ mm} \) (Figure 3) by method of least squares we receive:

\[ \alpha = 0.43; \beta = 0.0092. \]

Now we calculate equivalent number of cycles \( n₁ \), corresponding crack increment to \( \Delta L = 0.65 \text{ mm} \) under \( f = 3 \text{ mm} \) by Formula (6):

\[
\int_{\bar{n} = 1.45 \times 10^5} \frac{10^{-8}}{1.4^{0.8}} \left(0.0092\pi^{0.43} n^{0.43}\right)^{5.6} dn = 0.65 \times 10^{-3},
\]

whence it follows \( \bar{n}_1 = 1.45 \times 10^5 \text{ cycles} \).

Now let us define the number of cycles \( \bar{n}_2 \) under \( f = 3 \text{ mm} \), corresponding crack increment
\[ \Delta l = \Delta l_1 + \Delta l_2 = 2.86 \text{ mm} \int_0^\infty \frac{10^{-8}}{1.4^{5.6}} (0.0092n^{0.43})^5 \text{ dn} = 2.86 \times 10^{-3} \text{ , whence it follows } \overline{\Delta l} = 2.24 \times 10^3 \text{ cycles.}

Calculated cyclic life on the second stage \( n = n_2 - n_1 = 0.79 \times 10^5 \) cycles. Error of prediction \( \delta = \frac{n_{f2} - n_{f2}}{n_{f2}} \times 100\% = 28.2\% \).

Similar calculation under two-stages loading fulfills for samples № 6-9 and one-stage loading for samples № 1-5, 10 (Table 2). As it follows from Table 2, error of cyclic life prediction practically not changes.

4. Analysis of the Results and Conclusions

Elaborated thermographic method for prediction cyclic life of pieces with macrocracks [1] [2] [3] [14] allows:

1) To check crack development through temperatures kinetics on the surface of tested piece without contact with investigated object by using up-to-date infrared equipment. This gives possibility to calculate new diagnostic parameter \( \Delta S^{(c)} \)—increment of specific entropy for cycle (ISE) at the top of crack with high accuracy. Parameter ISE can be obtained both simultaneously with building of a kinetic fatigue diagram and on the basis of measuring of temperature under exploitation of object. In both cases the prediction of cyclic lifetime is better than with the help parameter \( K \).

2) To use (ISE) instead Stress intensity factor \( K \) decreases the error of prediction in some times (not more then 30%) in comparison with Paris method, having error which can exceed 100%.

3) Considerably widen the range of objects for control, in particular, for pieces, having some difficulties for observation at crack development.

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