The Research on the Damping of Prestressed Membrane Structure Subjected to the Impact Load

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Abstract: The damping ratio plays a main role in the vibration of membrane structures. In order to study the damping force of air application to membrane structures, this present paper investigated the vibration response of a membrane structure subjected to impact loads. Eight experiments with the application of different tension forces to a tension membrane structure were conducted, and the impact load was simulated using a rigid bullet with a certain velocity. The displacement data were obtained using a laser displacement meter. FEM was used to simulate the vibration, and the results had good agreement. The results show that the effect of air applied to a prestressed membrane was equivalent to viscous damping, and the damping force was determined using the air. The damping ratio was proportional to the density of the air over the density of the membrane. The parameter of the coefficient could be determined using the geometry of the structure.

Keywords: prestressed membrane; impact load; air damping; experimental study; numerical simulation

1. Introduction

Prestressed membrane structures are widely used by architects due to their beautiful shape, small ratio of weight over strength and other advantages [1–3]. Being different to traditional structures, their stiffness is mainly provided by tension force and geometry curvature, which leads to prestressed membrane structures being very sensitive to dynamic loads, such as wind, rain, hail and other loads [4–7]. Prestressed membranes are sensitive to shock loads. If the large displacement under the dynamic load of a prestressed membrane structure is not controlled well, the membrane structure will be damaged.

At present, there are some studies that have considered the vibration of prestressed membrane structures via analytical methods, numerical simulations and experiments [8–10]. Many researchers have derived theoretical formulas for membrane structures subject to dynamic loads. Zheng Zhoulian et al. [11,12] used the large deflection theory of thin shells to derive the free vibration control equation of orthogonal anisotropic thin membranes. Zhang Lin et al. [13,14] established a new method for the study of orthogonal heterosexual rectangular membrane structures under impact loads, and they obtained an analytical solution which was verified through an experiment. However, these theoretical methods often fail to account for complicated conditions. Liu Changjiang et al. [15–18] studied the nonlinear vibration response of an orthogonal anisotropic plane and a curved prestressed membrane under a concentrated impact load. Kapoor et al. [19] studied the dynamic response of highly flexible membrane structures under explosive loads and established an FEM model with a square chamber structure. Meanwhile, they calculated the response of displacement and stress. After a discussion, they revealed that as the Young’s modulus increases, deflection decreases. Furthermore, damping is crucial for vibration, and membrane structures are sensitive to wind load. The question of how to analyze the response...
of membrane structures under wind loads or other dynamic loads is very important, and the damping ratio is the key parameter for conducting this analysis. Liu Ping [20–22], etc., proposed a new analytical method to obtain some rules on the damping force of air applied to a membrane structure on the basis of no vortex assumption.

The present paper considered the damping force of air applied to a membrane structure which can enhance the sustainability of the structure, and it is the first study to discuss the influence of a damping ratio applied to a membrane structure via both a numerical simulation and experiment.

2. Test Settings

2.1. Experiment Characteristic

The membrane material used in the experiment was PVDF-1050P, which was manufactured by Shanghai Weibo Advanced Membrane company [23]. It was composed of a base and yarns, and there were 12 yarns in each warp and weft direction. The density of these yarns was called fabric density, and the total mass of the fabric and base was called the membrane density; both are shown in Table 1. The ultimate strength of it in the warp direction was 4200 N/5 cm (84 kN/m) and 4500 N/5 cm (90 kN/m) in the weft direction. The density of the fabric and the membrane was 270 g/m² and 1050 g/m², respectively.

### Table 1. PVDF-1050 membrane material parameters.

| Strength (N/dtex) | Warp Count (yarn/cm) | Density of Fabric (g/m²) | Density of Membrane (g/m²) | Tensile Strength (N/5 cm) | Tear Strength (N) | Seam Strength (N/cm) |
|------------------|----------------------|--------------------------|---------------------------|--------------------------|------------------|---------------------|
| 1100             | 1212                 | 270                      | 1050                      | 4200/4500                | 550/500          | >120/5              |

The membrane in the test was a square shape with the size of 1320 mm width and 1320 mm length. The details are shown in Figure 1. Each end of the specimen was clamped with rope so that it could be fixed with a steel clamping panel, and the bolt used was M10 × 1.50 × 50. The details are shown in Figure 2.

![Figure 1. The details of the membrane material specimen (mm).](image-url)
Figure 2. The layout of clamping characteristics (mm).

Due to different geometry stiffnesses, the tension force was applied along both the warp and weft directions. The tension force was set to 1.0, 2.0, 3.0, 4.0, 5.0, 6.0, 7.0 and 8.0 kN/m, all of which were no more than the 10% of the ultimate strength according to the methods of Ying Xiao-Ming [24,25]. To simulate the impact load, a small steel ball with a mass of 30 g was launched by a catapult. Additionally, we tested the velocity of the small ball using a speedometer near the membrane, which is illustrated in Figure 3b, and the velocity of the ball is listed in Table 2.

Figure 3. FEM model. (a) Structure of bracket. (b) The details of catapult. (c) The speedometer. (d) The manual windlass. (e) The data acquisition card. (f) Laser displacement sensor. Non-English word in the figure mean the care belong to school.
was fixed to was 640 mm in width and 640 mm in length. To produce the impact load, a prestressed membrane structure was not considered a constant value of material but was noted that the bullet had the same mass as the ball in the test. In order to improve the calculation efficiency, the ball was set as a rigid body. Based on the fact that the ball could not be deformed, the ball was simulated with a shell element. The interaction between the small ball and the membrane was set as an Automatic Surface to Surface model. The mesh is shown in Figure 4.

### 3. FEM Model

#### 3.1. The Simulation Model without Air

In this paper, LS-DYNA nonlinear finite element software was used to simulate the vibration process of a prestressed membrane structure subjected to an impact load considering air damping. The membrane was simulated with SHELL163 element, and the bullet was simulated with a small rigid ball to reduce the calculating time. It should be noted that the bullet had the same mass as the ball in the test. In order to improve the calculation efficiency, the ball was set as a rigid body. Based on the fact that the ball could not be deformed, the ball was simulated with a shell element. The interaction between the small ball and the membrane was set as an Automatic Surface to Surface model. The mesh is shown in Figure 4.

![Figure 4. Mesh division of ball and membrane.](image)

The membrane was simulated as a linear elastic material, the mechanical properties of which [20] are shown in Table 3. The distance between the center of the ball to the membrane was 20 mm. The mesh size of the sphere was 0.5 mm. The mesh in the collision area was refined, and the mesh is shown in Figure 4. In the experiment, the damping of the prestressed membrane structure was not considered a constant value of material but was

### Table 2. The speed of the steel ball.

| Case        | Case 2 (m/s) | Case 3 (m/s) | Average (m/s) |
|-------------|-------------|-------------|---------------|
| Velocity    | 10.21       | 9.95        | 10.35         |
| Velocity    | 15.12       | 15.32       | 15.16         |
|             |             |             | 10.17         |
|             |             |             | 15.20         |
related to the tension force. The damping coefficient obtained from the experiment was applied to the membrane without air, and the damping parameters are shown in Table 4.

### Table 3. PVDF membrane and air material parameters.

|                        | Membrane | Ball | Case 1 | Case 2 |
|------------------------|----------|------|--------|--------|
| Young’s Modulus/Mpa    | 200      | 2e6  | /      | /      |
| Density/(kg/m³)        | 1050     | 7850 | 1.29   | 1.29   |
| Poisson’s ratio        | 0.11     | 0.3  | /      | /      |
| Bulk modulus/Mpa       | /        | /    | 0.101  | 0.101  |

### Table 4. Damping results of experiment.

| Tension Force (kN/m) | 1  | 2  | 3  | 4  | 5  | 6  | 7  | 8  |
|----------------------|----|----|----|----|----|----|----|----|
| Damping              | 3.34 | 4.39 | 6.02 | 7.73 | 8.77 | 9.99 | 10.86 | 11.89 |
| Damping ratio        | 0.098 | 0.111 | 0.108 | 0.122 | 0.128 | 0.122 | 0.134 | 0.133 |

3.2. The Simulation Including Air

Our research team believes that the vibration response of membrane structures is significantly influenced by air, and the damping force comes from the interaction between the membrane and air, which is different to other structures. Additionally, it plays a main role in the damping force. In the present paper, it was assumed that air was an elastic body, and the fluid–solid coupling problem between the membrane and air was transformed into an interaction problem between structures, which not only simplified the numerical model, but also obtained effective results. The Z degree of both the membrane and air was coupled within the simulation, and the other degrees were free. In Figure 5, the green part of the diagram is the coupling node. The material parameters of air are shown in Table 4 [20]. Air was an elastic fluid constitution, and the mesh of air is shown in Figure 5a. The surface on which air was in contact with the prestressed membrane was meshed in the same manner as the prestressed membrane, and the air as a whole was a free mesh. This paper shows one half of the air grid, with the other half being symmetrical with the air grid on display.

![Figure 5. The mesh details in different cases. (a) The mesh of case 1. (b) The mesh of case 2.](image)

In order to simulate the effect of air, in the first model, we set the damping ratio to simulate it. In the other model, the air was included in the FEM model directly. The air model was simplified as an elastic material based on the small velocity of membrane vibration. Thus, the FSI problem became a structure problem. Additionally, the mesh is shown in Figure 4. The material properties are shown in Table 3. We assumed that the fluid could not transfer from one side to another side, so we used half of the model to simulate this problem.
Finally, the boundary was set as the same as the experiment, and the tension force was generated using the temperature method. The temperature value was calculated as shown in Equation (1).

\[ t = \frac{T}{\alpha} \]  

(1)

where \( t \) is the temperature, \( T \) is the tension force and \( \alpha \) is the thermal coefficient of the membrane.

4. Results

4.1. Vibration Results

Eight prestressed membrane specimens with different tension forces were simulated and analyzed in a numerical simulation, and the results were obtained using LS-PREPOST. The initial stress of the prestressed membrane is shown in Figure 6. It can be seen that the highest stress was at the corners, which is the typical effect of stress concentration. Figure 7 shows the velocity contour of the air when the ball impacted the membrane. It can be seen that the higher the prestress was, the lower the membrane velocity was, indicating that the pretension force was a key factor in the energy transfer between the ball and the membrane.

![Figure 6. Initial stress of prestressed membrane structure. (a) The initial effective stress at T = 6 kN/m. (b) The initial effective stress at T = 7 kN/m.](image)

Eight experiments with different tension forces were carried out. The effective stress contours of the membrane subjected to the tension force are plotted in Figure 6. It is shown that the max effective stress was located on the four corners, and the effective stresses are shown in a circular shape. Figure 7 shows the velocity contour of the air when \( t = 0.02 \) s. It can be seen that the max velocity was about 9.7 m/s and 8.1 m/s with respect to the tension forces of 6 kN/m and 7 kN/m. This is because the membrane could absorb more energy when the tension force was higher. Additionally, we can see that the velocities that appeared on the membrane were at impact points, and there were no responses to the other fields.

Figure 8a shows a stress contour with air when the small ball impacted the membrane at 0.04 s. It can be seen from the diagram that when comparing the stress of the blue area in the figure, the result in the simulation with no air was greater than that when air was present, because when the model vibrated, the presence of air lowered the stress. The difference in stress in other regions is due to the software calculation.
Figure 7. The velocity contour of air velocity in case 1. (a) The velocity contour of air at $T = 6 \text{ kN}$ (time = 0.02 s). (b) The velocity contour of air at $T = 7 \text{ kN}$ (time = 0.02 s).

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Figure 8b shows the effective stress contour of a membrane respective to the simulation without air and including air. The present results show that the stress in the case with no air was higher than that with air. This is because in the model including air, air had an effect. When the membrane vibrated, the air would apply some pressure on the surface of the membrane, which would reduce the inner stress in the membrane.

Figure 8. The contour of effective stress of membrane at 0.04 s with case 1 and case 2. (a) The contour of effective stress in case 1 at time = 0.04 s. (b) The contour of effective stress in case 2 at time = 0.04 s.

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4.2. The Results of Displacement in Case 1

The displacement data of the impact points under different tension forces are shown in Figure 9. It can be seen that the numerical results of the displacement data are in good agreement with the experimental results. When the ball collided with different prestressed membranes, the displacement vibrations at the point of impact were significantly different. With the increase in tension force, the displacement and vibration time was shortened, as shown in Table 5. This is because higher tension force produced higher frequencies, and the energy of vibration would disperse faster. So, the speed of vibration attenuation was accelerated, and the maximum displacement of the impact point was decreased, as shown in Table 6.

| Tension Force (kN/m) | 1    | 2    | 3    | 4    | 5    | 6    | 7    | 8    |
|----------------------|------|------|------|------|------|------|------|------|
| Case 1               | 1.390| 0.836| 0.558| 0.438| 0.334| 0.411| 0.228| 0.226|
| Experiment           | 1.568| 0.804| 0.580| 0.473| 0.314| 0.250| 0.236| 0.245|

Table 6. Numerical and experimental maximum vibration displacement.

| Tension Force (kN/m) | 1    | 2    | 3    | 4    | 5    | 6    | 7    | 8    |
|----------------------|------|------|------|------|------|------|------|------|
| Case 1               | 7.35 | 5.52 | 5.05 | 4.22 | 4.19 | 2.56 | 2.38 | 1.97 |
| Experiment           | 7.20 | 4.96 | 4.80 | 4.16 | 4.00 | 2.56 | 2.40 | 2.88 |

We compared the displacement results of the simulation to the results of the test. Figure 9 shows the comparison of displacements with respect to the experiments with tension forces of 1 kN/m, 2 kN/m, 3 kN/m, 4 kN/m, 5 kN/m, 6 kN/m, 7 kN/m and 8 kN/m.
Figure 9. Numerical and experimental displacement time history curves. (a) The displacement curve at 1 kN/m. (b) The displacement curve at 2 kN/m. (c) The displacement curve at 3 kN/m. (d) The displacement curve at 4 kN/m. (e) The displacement curve at 5 kN/m. (f) The displacement curve at 6 kN/m. (g) The displacement curve at 7 kN/m. (h) The displacement curve at 8 kN/m.
4.3. The results of Displacement in Case 2

Based on the numerical model of case 1, the model of case 2 was obtained by removing the air mesh in case 1. On the basis of the damping coefficient obtained from the above test results, the damping coefficient was added to the model. Additionally, the other parameters were the same as in case 1. The results of displacement are shown in Figure 10. Figure 10 shows the comparison of the displacement of case 1 and case 2. It can be seen that the comparison of the numerical simulation results is basically in agreement.

Table 7 shows the numerical and experimental results of the vibration frequencies of prestressed membranes under impact loads. It can be seen from Table 7 that the frequencies were almost the same. The experimental and numerical results show the same trend, and the vibration frequency increased with the increase in tension force.

Table 7. The main vibration frequencies of numerical simulations and tests.

| Tension Force (kN/m) | 1    | 2    | 3    | 4    | 5    | 6    | 7    | 8    |
|----------------------|------|------|------|------|------|------|------|------|
| Case 1               | 12.99| 15.63| 21.74| 25.00| 26.32| 29.41| 31.25| 35.71|
| Case 2               | 11.63| 17.24| 22.72| 25.00| 27.78| 27.78| 31.25| 35.71|
| Test                 | 12.87| 15.10| 21.37| 24.15| 26.28| 29.76| 31.15| 34.33|
| mean                 | 12.50| 16.00| 21.94| 24.72| 26.80| 28.99| 31.22| 35.25|

![Figure 10](image-url)
Figure 10. The comparison of displacement in case 1 and case 2. (a) The displacement curve at 1 kN/m. (b) The displacement curve at 2 kN/m. (c) The displacement curve at 3 kN/m. (d) The displacement curve at 4 kN/m. (e) The displacement curve at 5 kN/m. (f) The displacement curve at 6 kN/m. (g) The displacement curve at 7 kN/m. (h) The displacement curve at 8 kN/m.

5. Discussion

The wave peak was obtained from the displacement data. The frequency of each wave peak was calculated to obtain the average value, and then, the damping ratio formula could be obtained, as shown in Formula (2):

\[
\begin{align*}
\{ u &= u(x, y)e^{-\eta \omega t} \sin(\omega t) \\
\nu &= -\eta \omega e^{-\eta \omega t} \sin(\omega t) + \omega e^{-\eta \omega t} \cos(\omega t)
\end{align*}
\]

(2)

where \( \omega \) is the main frequency of the vibrating membrane and \( u(x, y) \) is the function of the displacement to the position of the membrane. Thus, the total energy of the membrane is the kinetic energy at the beginning \( (t = 0) \). So,

\[
E_a = \frac{1}{2} \rho_s \omega^2 \oint u(x, y)^2 ds
\]

(3)

Additionally, after some time, the energy transferred to the air, so, the vibration of air deduced by the membrane propagated to a far field using the rule of Equation (4), where
we assumed the damping ratio $\eta$ was very small, so we did not consider the first term in the velocity expression of Equation (2).

$$v = \frac{1}{r} v_s(t - r/c) = \frac{\omega}{r} e^{-\eta \omega (t - r/c)} \cos(\omega(t - r/c))$$  \hspace{1cm} (4)

Thus, the total energy of air after vibrating time $t_s$ is shown as Equation (5),

$$E_a = \rho_a \omega^2 \oint u(x, y)^2 ds \int_{r_0}^{c \cdot t_s} \left[ \frac{1}{r} e^{-\eta \omega (t - r/c)} \cos(\omega(t - r/c)) \right]^2 \frac{4\pi r^2 dr}{4}$$  \hspace{1cm} (5)

Therefore, the total energy is equal to $E_a$ when time $t_s$ is infinite, which means the vibration lasted for a long time and all energy from the membrane was transferred to the air. Additionally, also we know that

$$\lim_{t_s \to \infty} E_a = \pi \rho_a c \omega^2 \frac{1 + 2\eta^2}{\eta \omega (1 + \eta^2)} \oint u^2 ds$$  \hspace{1cm} (6)

Thus,

$$E_a = \frac{1}{2} \rho_a \omega^2 \oint u^2 ds = \pi \rho_a c \omega \frac{2\eta^2 + 1}{\eta (\eta^2 + 1)} \oint u^2 ds$$  \hspace{1cm} (7)

From Equation (7), we can obtain the expression of $\eta$ shown as Equation (8),

$$\eta \approx \frac{1}{\pi} \frac{\rho_s}{\rho_a}$$  \hspace{1cm} (8)

Table 8 shows the numerical and experimental results of the damping ratio of prestressed membranes under impact loads. It can be seen from Table 8 that the damping ratio increased with the increase in tension force, and the damping ratio of the numerical simulation is basically consistent with the experimental results.

Table 8. The damping ratio of numerical simulations and tests.

| Tension Force (kN/m) | 1     | 2     | 3     | 4     | 5     | 6     | 7     | 8     |
|----------------------|-------|-------|-------|-------|-------|-------|-------|-------|
| Case 1               | 0.110 | 0.123 | 0.145 | 0.120 | 0.143 | 0.114 | 0.141 | 0.146 |
| Case 2               | 0.097 | 0.115 | 0.118 | 0.103 | 0.129 | 0.105 | 0.144 | 0.139 |
| Test                 | 0.098 | 0.111 | 0.108 | 0.122 | 0.128 | 0.122 | 0.134 | 0.133 |
| mean                 | 0.102 | 0.116 | 0.124 | 0.115 | 0.133 | 0.114 | 0.140 | 0.140 |

Figure 11 shows the results of the damping ratio vs. tension force. The damping coefficient method could be used to simulate the vibration response of a prestressed membrane under an impact load in air. The corresponding damping coefficient can range from 0.1 to 0.15.

Because of the error of the assumption of the real motion of air and Equation (4), the coefficient of the damping ratio, expressed as Equation (7), can be expressed as Equation (8), where the coefficient of $\alpha$ should be determined as the geometry of the membrane structure, including the shape, dimension, materials, etc. If no more experiment data are used, it can be accepted as 0.10–0.15, as shown in Equation (9) in this paper.

$$\eta = \alpha \cdot \rho_a / \rho_s$$  \hspace{1cm} (9)
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