The natural frequencies of masonry beams

Abstract  The present paper aims at analytically evaluating the natural frequencies of cracked slender masonry elements. The problem is dealt with in the framework of linear perturbation, and the small oscillations of the structure are studied under loaded conditions, after the equilibrium for permanent loads has been achieved. A masonry beam element made of no-tension (masonry-like) material is considered, and some explicit expressions of the beam’s fundamental frequency as a function of the external loads and the amplitude of imposed deformations are derived. The analytical results are validated via finite-element analysis.

Keywords  Nonlinear dynamics · Slender masonry structures · Linear perturbation

1 Introduction

The measurement of ambient vibrations has become a standard procedure in Civil Engineering. In fact, these vibrations contain precious information on both the structural behavior and the health status of buildings. Moreover, experimental frequencies and mode shapes can be introduced in model updating procedures [22] and allow estimating the mechanical properties and boundary conditions of such structures, while long-term measurements can help revealing the onset of structural damage, through damage detection procedures. In fact, modal properties are damage-sensitive features [2,24,29].

With regard to heritage masonry structures, the assumption of linear elasticity, which usually underlies the study of their ambient vibrations, may lead to errors. In fact, such structures are unable to withstand large tensile stresses and are usually affected by crack patterns. These nonlinear effects have in general a non-negligible influence on the structural stiffness and should not be disregarded in the analysis. The dynamic behavior of these structures should be analyzed taking into account the existing cracks.

A great deal of effort has been devoted to numerically simulating the effects of damage on structural vibrations. With regard to masonry buildings, a common approach consists of simulating the cracks actually observed on the structure, by reducing the stiffness of the elements of the finite-element model which belong to the damaged parts [26,28]. In [7] the dynamic properties of some masonry structures at different damage levels are investigated via the discrete element method. In [13] a linear perturbation numerical procedure is presented, implemented in the NOSA-ITACA program [6] to take into account the presence of cracks in the calculation of a masonry structure’s dynamic properties. The procedure applies the constitutive equation of masonry-like materials [10], and the paper proves that the problem is governed by the global tangent stiffness matrix, used in place of the linear elastic one to evaluate the structure’s modal properties. Some example applications are shown in [25], where linear perturbation is carried out to model reinforcement operations on...
the Mogadouro tower in Portugal, and in [19], where the procedure is employed to reproduce the results of some laboratory tests conducted on a masonry arch subjected to settlements of one support.

Similar problems also arise in the study of the dynamic properties of prestressed concrete cracked elements; the dependence of such structures’ modal properties on the external loads and prestressing force level has long been debated, as shown in [1,14,23].

In the present paper, an analytical approach is presented to evaluate the natural frequencies of masonry beams. The small oscillations of such structures are studied under loaded conditions, after the equilibrium for permanent loads has been achieved. In the framework of the Euler–Bernoulli beam theory, an explicit expression of the beam’s fundamental frequency as a function of the external loads and initial deformations is derived, by using a constitutive relationship between the generalized deformation $\chi$ and the generalized stresses $N$ and $M$ acting on the beam’s sections.

Many empirical formulations have been proposed to evaluate the fundamental frequency of masonry towers and buildings [4,17]. Nevertheless, the simple case presented in the paper turns out to be of interest, since at the best of the author’s knowledge the literature does not furnish yet explicit expressions to estimate the effects of cracking on the modal properties of masonry structures.

The paper is organized as follows: In Sect. 2 the problem is introduced. Section 3 briefly recalls the constitutive equation of masonry-like beams, and in Sect. 4 some example applications are presented. Finally, Sect. 5 is devoted to testing of the analytical results via finite-element analysis.

2 Small oscillations of a loaded masonry beam

Let us consider a rectilinear beam with length $L$ and rectangular cross section with height $h$ and width $b$, subjected along its axis to a constant axial force $N$. Let us denote by $E$ and $\rho$ the Young’s modulus and the density of the material, respectively, and by $J = bh^3/12$ the moment of inertia of the beam’s cross section. Let $x$ be the abscissa along the beam’s axis and $t$ the time.

The curvature of the beam is denoted by $\chi$ and, under the Euler–Bernoulli hypothesis, the bending moment $M(\chi) : R \rightarrow R$ is a continuous differentiable function, whose second derivative is assumed to be piecewise continuous.

We assume the effects of both the shear strain and the rotary inertia to be negligible; as pointed out also in [16], this hypothesis turns out to be acceptable for slender elements (the interval $1/6 \leq h/L \leq 1/20$ could be suggested for masonry beams [15]). The proposed formulation is moreover limited to the first frequency, whose mode shape involves the whole beam’s length and is thus less affected by this approximation.

We limit ourselves to considering situations in which the transverse displacement $\phi(x,t)$ and its derivative $\phi_t(x,t)$ are small, so that we can neglect the effects of the axial force on the dynamic equilibrium of the beam and write

$$\chi(x,t) = -\phi_{xx}(x,t). \quad (2.1)$$

If we introduce the function

$$f(\chi) = \frac{M(\chi)}{\rho bh}, \quad (2.2)$$

the motion of the beam is expressed by

$$\phi_{tt} - (f(\chi))_{xx} = q(x,t), \quad (2.3)$$

where

$$q(x,t) = \frac{p(x,t)}{\rho bh} \quad (2.4)$$

and $p(x,t)$ is the transverse load per unit length.

Let us consider, at $t = 0$, a load $\tilde{q}(x)$ inducing in the beam an initial deformation $\tilde{\phi}(x)$ and curvature change $\tilde{\chi}(x)$. The beam reaches equilibrium under the load $\tilde{q}$, and thus,

$$- (f(\tilde{\chi}))_{xx} = \tilde{q}(x). \quad (2.5)$$

We are interested in studying the small oscillations $\delta\phi$ of the beam around $\tilde{\phi}$

$$\delta\phi_{tt} - \left( \frac{\partial f}{\partial \chi} \right)_{\tilde{\chi}} = 0, \quad (2.6)$$
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where we use the approximation

\[
    f(\bar{\chi} + \delta \chi) \simeq f(\bar{\chi}) + \frac{\partial f}{\partial \chi} \bigg|_{\bar{\chi}} \delta \chi.
\]  

(2.7)

In case of a simply supported beam, since we limit ourselves to studying the fundamental frequency of the beam, we can approximate the small oscillations \( \delta \phi \) as follows:

\[
    \delta \phi \simeq \delta a \sin \left( \frac{\pi}{L} x \right) u(t),
\]

(2.8)

where \( \delta a > 0 \) represents the amplitude of the beam’s oscillation around \( \bar{\phi} \). In view of (2.8), (2.6) becomes

\[
    \sin \left( \frac{\pi}{L} x \right) \ddot{u} - \frac{(\pi/L)^2}{2} \left( \frac{\partial f}{\partial \chi} \bigg|_{\bar{\chi}} \sin \left( \frac{\pi}{L} x \right) \right)_{xx} u = 0.
\]

(2.9)

Let us now use the Galerkin method, multiply (2.9) by \( \sin \left( \frac{\pi}{L} x \right) \) and integrate over the beam’s length,

\[
    \ddot{u} \int_0^L \sin \left( \frac{\pi}{L} x \right)^2 dx - \frac{\pi^2}{L^2} \left( \int_0^L \sin \left( \frac{\pi}{L} x \right) \left( \frac{\partial f}{\partial \chi} \bigg|_{\bar{\chi}} \sin \left( \frac{\pi}{L} x \right) \right)_{xx} dx \right) u = 0.
\]

(2.10)

Using the boundary conditions \( \sin \left( \frac{\pi}{L} L \right) = 0 \), integrations in (2.10) can be performed by parts and give

\[
    \ddot{u} + \frac{2\pi^4}{L^5} \left( \int_0^L \sin^2 \left( \frac{\pi}{L} x \right) \frac{\partial f}{\partial \chi} \bigg|_{\bar{\chi}} dx \right) u = 0.
\]

(2.11)

The fundamental frequency can be directly obtained from Eq. (2.11), once the constitutive equation \( f(\chi) \) is given. Equation (2.11) holds under the assumption of small displacements and rotations. Therefore, the formulas proposed below are expected to be reliable for small values of external loads and initial deformations acting on the beam.

3 A constitutive equation for masonry-like beams

In this section we briefly recall a simple constitutive equation for one-dimensional masonry-like elements with rectangular cross section. Despite its simplicity, this constitutive equation effectively describes the mechanical behavior of masonry slender elements subjected to axial force and bending moment. The beam’s cross section is fully compressed since the vertical load falls inside the central nucleus of inertia, and reaches the limit elastic behavior when the vertical load acts on the perimeter. Many authors have made use of this simplified equation to find analytical and numerical solutions to the static and the dynamic behavior of masonry pillars, arches and towers (some examples can be found in [3,9,18,30]). In [11] and [12] some explicit approximate solutions are proposed for modeling the free and forced oscillations of masonry beams.
The constitutive equation $M - \chi$ for a rectangular section with infinite compressive strength and zero tensile stress is given by:

$$f(\chi) = \begin{cases} 
  c^2 \chi & \text{for } |\chi| \leq \alpha, \\
  c^2 \alpha \text{Sign}(\chi)(3 - 2 \frac{a}{|\chi|}) & \text{for } |\chi| > \alpha,
\end{cases}$$  \hspace{1cm} (3.1)

where

$$\alpha = -\frac{2N}{Ebh^2}$$  \hspace{1cm} (3.2)

is the curvature corresponding to the elastic limit and $c^2 = EJ/\rho bh$ is the elastic constant of the beam.

Function (3.1) is plotted in Fig. 2. In the nonlinear region, for $|\chi| > \alpha$, increasing values of $\chi$ correspond to a rapid decrease in the section’s stiffness, and the bending moment tends toward its limit value $|N|h/2$. Moreover, $M(\chi)$ is continuous with its first derivative, while the second derivative undergoes a jump at $|\chi| = \alpha$.

Equation (3.1) is used in the following in conjunction with (2.11) to explicitly evaluate the fundamental frequency of masonry beams subjected to different loading conditions. The first derivative of $f$ with respect to $\chi$, which appears in (2.11), is:

$$f'(\chi) = \begin{cases} 
  c^2 & \text{for } |\chi| \leq \alpha, \\
  c^2 \sqrt{\frac{a}{|\chi|}} & \text{for } |\chi| > \alpha.
\end{cases}$$  \hspace{1cm} (3.3)

### 4 Example applications

#### 4.1 Case 1: constant curvature

Let us consider the case of a masonry beam in an initial deformed shape with constant curvature $\bar{\chi}$. This is the simple, but meaningful case of a beam subjected to a vertical load acting with constant eccentricity $e$ along the axis. In this case, the bending moment acting on the beam assumes the constant value

$$M = N \cdot e,$$  \hspace{1cm} (4.1)

and nonlinear behavior of the structure is attained for the eccentricity

$$|e| = \frac{h}{6}.$$  \hspace{1cm} (4.2)
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Thus, from (2.11) and (3.3) the fundamental frequency $\omega$ of the beam (in radians per second) is:

$$\omega^2 = \frac{\pi^4}{L^2} c^2 = \omega_{el}^2 \quad \text{for} \quad |\bar{\chi}| \leq \alpha, \quad (4.3)$$

$$\omega^2 = \frac{2\pi^4}{L^5} \int_0^L \sin^2 \left(\frac{\pi x}{L}\right) \sqrt{\frac{\alpha^3}{|\bar{\chi}|^3}} \, dx \quad \text{for} \quad |\bar{\chi}| > \alpha, \quad (4.4)$$

and $\bar{\chi}$, given the bending moment (4.1), can be deduced from (3.1)

$$|\bar{\chi}| = \frac{4 \alpha^3 c^4}{(|f| - 3 \alpha c^2)^2}. \quad (4.5)$$

Equation (4.3) coincides with the fundamental frequency of a linear elastic beam [8] and can be deduced by means of (2.11), per $|\chi| = \alpha$. When the eccentricity reaches the border of the section, $|e| = h/2$, the frequency of the beam tends to zero.

It is worth noting that, under the current hypothesis of small displacements and rotations, the value of the linearized frequency (4.4) does not depend on the normal force $N$, but only on the eccentricity $e$. In fact, using (4.1) and (4.5), after simple calculations, the frequency assumes the following expression:

$$\omega = \frac{\pi^2}{L^2} c = \omega_{el} \quad \text{for} \quad |e| \leq \frac{1}{6}, \quad (4.6)$$

$$\omega = \frac{3 \pi^2 c}{4 L^2} \sqrt{6 \left(1 - 2 \frac{|e|}{h}\right)^3} = \frac{3}{4} \omega_{el} \sqrt{6 \left(1 - 2 \frac{|e|}{h}\right)^3} \quad \text{for} \quad \frac{1}{6} < |e| \leq \frac{1}{2}. \quad (4.7)$$

Equations (4.6) and (4.7) are plotted in Fig. 3.

4.2 Case 2: uniform transverse load

Let us consider the case of the beam subjected to a uniform transverse load $p$ and axial force $N$.

The beam enters the nonlinear field when $p \geq 0$ reaches the value

$$\bar{p} = \frac{4 |N| h}{3 L^2}. \quad (4.8)$$

For greater values of the transverse load, the beam’s axis is divided into three regions: The central part of the beam is in the nonlinear field, while two lateral regions, symmetric with respect to the midsection, still exhibit linear behavior (Fig. 4). Let us denote by $x_0$ the abscissa of the beam’s section with curvature $|\chi| = \alpha$, $0 \leq x_0 \leq L/2$. Function $x_0$ can be easily deduced:
In view of (2.11), the expression for the fundamental frequency becomes, for $p > \bar{p}$

$$\omega^2 = \frac{4\omega_{el}^2}{L^2} \left( \int_0^{x_0} \sin^2 \left( \frac{\pi x}{L} \right) \, dx + \int_{x_0/L}^{1/2} \sin^2 \left( \frac{\pi y}{L} \right) \sqrt{\frac{\alpha^3}{|\tilde{\chi}|^3}} \, dx \right).$$

(4.10)

with $|\tilde{\chi}|$ given by (4.5). After simple calculations, Eq. (4.10) can be expressed in the following form

$$\omega^2 = \frac{4\omega_{el}^2}{L^2} \left( \int_0^{x_0/L} \sin^2 (\pi y) \, dy + 8 \int_{x_0/L}^{1/2} \sin^2 (\pi y) \left| \frac{p}{\bar{p}} (y - y^2) - \frac{3}{4} \right|^3 \, dy \right).$$

(4.11)

with $y = x/L$. Thus, ratio $\omega/\omega_{el}$, as shown in Fig. 5, turns out to be dependent only on the ratio $p/\bar{p}$.

The midsection of the beam reaches the limit bending moment for the transverse load

$$\bar{p} = \frac{4|N|h}{L^2} = 3 \bar{p}.$$ 

(4.12)

Accordingly, abscissa $x_0$ attains the value

$$x_0 = \frac{L}{2} - \frac{L}{\sqrt{6}}.$$ 

(4.13)

which is independent of the transverse load and the normal force acting and corresponds to about one tenth of the beam’s total length. In this case, Eq. (4.11) reduces to

$$(\omega/\omega_{el})^2 = 4 \left( \int_0^{1/2} \frac{1}{\sqrt{6}} \sin^2 (\pi y) \, dy + 8 \int_{1/2}^{1/2} \frac{1}{\sqrt{6}} \sin^2 (\pi y) \left| 3 (y - y^2) - \frac{3}{4} \right|^3 \, dy \right) \approx 0.05.$$ 

(4.14)
4.3 Case 3: Initial deformed shape

In this last case, let us impose on the beam an initial deformation defined by the following equation:

$$\tilde{\phi}(x) = A \sin \left( \frac{\pi x}{L} \right)$$

(4.15)

with $A \geq 0$. Accordingly, curvature $\tilde{\chi}$ assumes the value

$$\tilde{\chi}(x) = \frac{A\pi^2}{L^2} \sin \left( \frac{\pi x}{L} \right),$$

(4.16)

and, setting $\tilde{\chi} = \alpha$, abscissa $x_0$ becomes

$$x_0 = \frac{L}{\pi} \arcsin \left( \frac{\alpha L^2}{A\pi^2} \right).$$

(4.17)

Function $x_0$ is $L/2$ for

$$A_m = \alpha L^2/\pi^2,$$

(4.18)

and tends to zero for $A \to \infty$.

In this case, Eqs. (2.11) and (3.3) yield:

$$\omega^2 = \frac{2\pi^4 c^2}{L^3} \left( 2 \int_0^{x_0} \sin^2 \left( \frac{\pi x}{L} \right) dx + \int_{x_0}^{L-x_0} \sin^2 \left( \frac{\pi x}{L} \right) \sqrt{\frac{\alpha^3 L^6}{A^3 \pi^6 \sin^3 \left( \frac{\pi x}{L} \right)}} dx \right) =$$

$$= \frac{2\pi^4 c^2}{L^3} \left( x_0 - \frac{L^3 \alpha}{A^3 \pi^3} \sqrt{1 - \frac{L^4 \alpha^2}{A^2 \pi^4}} + \frac{L^3 \alpha}{A^3 \pi^3} \int_{x_0}^{L-x_0} \sqrt{\frac{\alpha}{A} \sin \left( \frac{\pi x}{L} \right)} dx \right) =$$

$$= 2 \omega_{el}^2 \left( \frac{x_0}{L} - \frac{A_m}{A} \frac{1}{\pi} \sqrt{1 - \frac{A_m^2}{A^2}} + \int_{x_0}^{L-x_0} \sqrt{\frac{A_m}{A}} \sin(\pi y) dy \right),$$

(4.19)

where the first two terms regard the unfractured part of the beam, while the third is for the fractured part. For $A = A_m$ frequency (4.19) reduces to the linear elastic value $\omega_{el}$, while for $A \to \infty$ the frequency tends to zero. Function $x_0/L$ depends solely on the variable $A_m/A$. Thus, as for the previous cases, expression (4.19) can be expressed in the non-dimensional form shown in Fig. 6.
5 Numerical solutions

The analytical solutions have been tested against finite-element analysis. To this end, we adopt a conforming beam element equipped with cubic polynomial shape functions (Hermite polynomials), thereby guaranteeing continuity of both the transverse displacements and rotations of the beam’s axis \([5,21]\). Accordingly, the beam element has two degrees of freedom for each node: one for the beam’s transverse displacement \(\phi\) and the other associated with its derivative \(\phi_x\).

The first application of this element to masonry-like beams is reported on in \([18,27]\), where the element is described in detail. Here the main ingredients of the calculation are briefly recalled. The core of the analysis is calculation of the element tangent stiffness matrix, which for constitutive Eq. (3.1) takes the explicit form

\[
K_T^e = \frac{EJ}{c^2} \int_e \Psi'' \Psi'^T f' d\xi
\]  

(5.1)

where \(\Psi\) is the vector of the shape functions, \(f'(\chi)\) is given by Eqs. (3.3), and relation (2.1) holds true. The vector of the element’s internal forces is

\[
f_i^e = \frac{EJ}{c^2} \int_e \Psi'' f d\xi,
\]  

(5.2)

where \(f\) is given by (3.1), while the vector of the external loads acting on the element and the mass matrix of the element assume the usual expressions

\[
f_e^e = \int_e \Psi p d\xi,
\]  

(5.3)

\[
M^e = \rho bh \int_e \Psi \Psi' d\xi.
\]  

(5.4)

The numerical procedure adopted here follows the method shown in \([13]\) and is known as linear perturbation. This method consists of the following steps:

**Step 1** Model the structure via finite elements.

**Step 2** Apply external loads and solve the nonlinear equilibrium problem through an iterative scheme. Extract the tangent stiffness matrix calculated in the last iteration before convergence.

**Step 3** Perform a modal analysis about the equilibrium solution, using the tangent stiffness matrix in place of the linear elastic one.

Herein the finite element (5.1)–(5.4), implemented in the Mathematica® environment \([20]\), is used to model the masonry beam. For each load case, the tangent stiffness matrix is calculated via a Newton–Raphson scheme, and the corresponding value of the fundamental frequency is evaluated. This value is then compared to the analytical solutions reported in Sect. 4.

The numerical procedure is applied to a masonry-like beam–column with the following geometric characteristics:

\[
L = 6 \text{ m},
\]

\[
h = 0.4 \text{ m},
\]

\[
b = 1 \text{ m},
\]

\[
\rho = 1800 \text{ kg/m}^3,
\]

\[
E = 3 \cdot 10^9 \text{ Pa},
\]  

(5.5)

The beam is simply supported at the ends and discretized into 30 elements (different numbers of elements have been also tested to check software performance). The example applications presented regard the three cases discussed in Sect. 4.
5.1 Case 1: constant curvature

Beam (5.5) is subjected to increasing values of the eccentricity. The normal force acting on the model is $N = -500000$ N. Different tests, not reported on here, have been performed to check the influence of the normal force on the beam’s fundamental frequency given eccentricity $e$: The numerical value of the frequency turned out to be independent of the normal force, as predicted by the analytical solution.

Figure 7 shows the beam’s fundamental frequency $f_{fun}$ versus the eccentricity $e$ of the normal force. The analytical solution (4.6), (4.7) is plotted as a continuous line, while boxes are for the finite-element simulation.

5.2 Case 2: uniform transverse load

Beam (5.5) is subjected to a transverse load $p$ of increasing amplitude, with a normal force $N$ acting along the beam. Different values of $N$ are tested. The beam’s fundamental frequency is shown in Fig. 8, where lines represent the analytical solution (4.10), while boxes represent the results of the finite-element analysis.

5.3 Case 3: initial deformed shape

Beam (5.5) is in an initial deformed shape defined by Eq. (4.15). Increasing values of the maximum amplitude $A$ are evaluated for different values of the normal force $N$ acting along the beam. The beam’s fundamental
frequency is shown in Fig. 9, where lines represent the analytical solution (4.19), while boxes represent the results of the finite-element analyses.

Conclusions

An analytical approach to determine the fundamental frequency of masonry beam–columns in the presence of cracks has been presented. The study has been carried out under the hypothesis of masonry-like material. Some explicit expressions have been obtained, with the frequency depending on the external loads acting or the initial deformation imposed on the beam. The analytical method has been validated via finite-element analysis, and such a comparison reveals the good agreement found between the numerical and analytical results.

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