Synchronizing a single-electron shuttle to an external drive

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Abstract

The nanomechanical single-electron shuttle is a resonant system in which a suspended metallic island oscillates between and impacts at two electrodes. This setup holds promise for one-by-one electron transport and the establishment of an absolute current standard. While the charge transported per oscillation by the nanoscale island will be quantized in the Coulomb blockade regime, the frequency of such a shuttle depends sensitively on many parameters, leading to drift and noise. Instead of considering the nonlinearities introduced by the impact events as a nuisance, here we propose to exploit the resulting nonlinear dynamics to realize a highly precise oscillation frequency via synchronization of the shuttle self-oscillations to an external signal. We link the established phenomenological description of synchronization based on the Adler equation to the microscopic nonlinear dynamics of the electron shuttle by calculating the effective Adler constant analytically in terms of the microscopic parameters.
1. Introduction

Micro- and nanoscale resonators [1, 2] currently receive much attention both in applied and basic research. Their extreme sensitivity and design flexibility have enabled, for instance, mass sensing on the single ad-atom level [3–5], force sensitivity below the attonewton range [6–8], and access to the ultimate quantum mechanical limits of motion [9–11]. The nonlinear dynamical behavior in nanomechanical oscillators has received new attention recently, including studies of nonlinear damping [12], the use of buckling modes for mechanical information storage [13], noise reduction due to nonlinearities [14], the choice of optimal operation points [15], and signal amplification at bifurcation points [16].

One particularly important example of a nanomechanical system that can become highly nonlinear is the electron shuttle. Its nonlinear response has been demonstrated, most recently in a system of two coupled electron shuttles [17]. Nanomechanical electron shuttles [18] have the potential to administer current with single-electron accuracy. This ultimate limit of current control would allow to redefine the current standard in terms of fundamental constants [19–22], and to close the so-called ‘metrological triangle’ by relating voltage and resistance defined in terms of fundamental constants.

A nanomechanical electron shuttle [23–30] consists of a nanomechanical resonator carrying a metallic island which oscillates between two metallic electrodes (figure 1). The island defines a quantum dot which, if operated in the Coulomb blockade regime [31, 32], can be consistently charged with a known number of charge carriers down to the single electron level. The advantage of nanomechanical shuttles over other approaches is their inherent suppression of unwanted co-tunneling due to tunnel contact with only one electrode at a time. Given a sufficiently large dc voltage, the shuttle enters into Coulomb attraction-driven self-oscillation, where Coulombic forces drive the charged island to impact with and re-charge on subsequent electrodes [26, 27, 30], realizing a nanoscale version of Benjamin Franklin’s ‘lightning bell’ [33].

Synchronization dynamics in nanomechanical systems has only recently been discussed both for synchronization to an external drive [34] as well as for synchronization between two nanomechanical oscillators [34–38]. In this paper, we show that synchronization to an external drive can be exploited to provide a stable and reproducible frequency for an oscillating single-electron shuttle. In combination with charge quantization [28, 29], this will ensure the desired reproducibility and precision of the shuttle current for metrological applications. We establish the requirements for this synchronization regime by analyzing the nonlinear impact dynamics of the shuttle, i.e. a dynamical regime that has received comparatively little attention so far. In doing so, we go beyond the dynamical effects of the alternating recharging of the shuttle at the electrodes [39]. Our results can be cast in the form of an effective equation (of the ADLER type), where we discuss the strength of the frequency locking as a function of microscopic parameters. This will serve as a guideline for experiments that attempt to exploit the effect predicted here.
2. Model

We model the shuttle as a nanomechanical harmonic oscillator with eigenfrequency $\omega_0$, effective mass $m$, and quality factor $Q = \omega_0 / \Gamma$ using nondimensional coordinates ($\bar{z} = x/L$) and time ($\tau = \omega_0 t$). Two electrodes are located at positions $x = \pm L$ and restrain its oscillatory motion such that inelastic impact events occur: then the velocity reverses sign and is rescaled by an impact damping parameter $\eta \leq 1$ as kinetic energy is lost. Simultaneously, the shuttle is charged to a value $q = \pm q_{cv}$ and is accelerated towards the other electrode by a force $F = qV/2L \propto V^2$. The dimensionless electrostatic force term is $\mathcal{V}^2 = (V/V_s)^2$ with $CV_s^2/2 = m\omega_0^2 L^2$; it is half the ratio between the charging energy and the oscillator’s potential energy at the impact point and carries a sign depending on which electrode has been contacted last (‘+’ for $z = -1$). Crucial for our discussion is an additional external driving, e.g. by a periodic force $\mathcal{F}$. The equation of motion between the electrodes reads

$$\ddot{z}(\tau) + Q^{-1} \dot{z}(\tau) + z(\tau) = \mathcal{F} \cos (\Omega \tau) \pm \mathcal{V}^2. \tag{1}$$

Our model is described by five dimensionless parameters: (i) the mechanical quality factor $Q$, (ii) the dimensionless amplitude $\mathcal{F}$ of the external driving force, (iii) its external driving frequency $\Omega$, written in units of the oscillator resonance frequency $\omega_0$, (iv) the impact damping parameter $\eta$ and (v) the dimensionless voltage $\mathcal{V} \equiv V/V_s$. Figure 1 displays a typical trajectory with impact at the electrodes, under simultaneous driving.

We note that any extra intrinsic nonlinearities, while possibly present, do not alter the essential results below, since the main nonlinearity is provided by the impacts at the electrodes. The synchronization we analyze is a robust phenomenon, insensitive in its main aspects to many details of the model.

All the physical quantities of interest may be expressed by the dimensionless parameters and a few dimensional quantities. For example, the average shuttle current can be obtained from the transported charge $CV$ and the frequency: $I = CV\omega_0 \bar{\tau}^{-1}$. Here $\bar{\tau}^{-1}$ is defined as the averaged inverse (dimensionless) time for a one-way trip between the electrodes.

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7 Here we assume the charge $q$ to be linear in the voltage $V$. At low temperatures, in the Coulomb blockade regime, the charge shows discrete plateaus as a function of $V$. Then $\mathcal{V}^2$ in equation (1) and in all of the subsequent formulas and figures must simply be replaced by $QV / (2m\omega_0 L^2)$.

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**Figure 1.** (a) A nanomechanical electron shuttle oscillating between and impacting on two electrodes and (b) a typical trajectory.
2.1. Self-oscillations without external driving

When a sufficiently large dc voltage is applied to the electrodes, the acceleration by the electric field overcomes frictional losses due to impact and the shuttle can execute self-oscillations even in the absence of external resonant driving [18]. The properties of this well-known regime are crucial to our analysis as they form the reference state for a subsequent perturbative approach. In this regime, the shuttle shows regular motion, where the velocity before impact is always the same. We first assume intrinsic oscillator losses to be absent (i.e. $Q = \infty$, but $\eta \neq 0$). The energy lost upon impact is obtained via the relation $\eta' = \sqrt{\eta} v$ between the speeds before ($v$) and after ($v'$) impact: $(v^2 - v'^2)/2 = v^2 (1 - \eta)/2$. This must equal the energy gained by acceleration in the electric field, $2V^2$ in our units. Equating the two yields the velocity before impact:

$$v^2 = 4V^2/(1 - \eta). \quad (2)$$

Now we can obtain the one-way travel time $\tau_0$. We assume the last impact was at $z(0) = -1$, with $z(0') = v'$. Based on the solution $z(\tau) = V^2 + v' \sin(\tau) - (1 + V^2) \cos(\tau)$, we demand $z(\tau_0) = +1$. This yields $\tau_0$ via $\sin\tau_0 = \left[-B + \sqrt{B^2 - 4AC}\right]/2A$, with $A = v^2 + (1 + V^2)^2$, $B = 2v' (V^2 - 1)$, and $C = -4V^2$.

Despite the impact damping, self-oscillations are possible down to the lowest voltages for $Q = \infty$, because the grazing impact of the shuttle minimizes the loss: impact velocity $v \to 0$ for $V \to 0$. In this limit, the one-way travel time is just half the intrinsic oscillator period ($\tau_0 = \pi$), i.e. the impact happens exactly at the turning point of the oscillatory motion, with the shuttle barely touching the electrode. Consequently, $\tau_0$ then turns out to be independent of the impact damping parameter $\eta$. For small voltages, we obtain $\pi - \tau \approx 2\sqrt{V(1 + \sqrt{\eta})}/\sqrt{1 - \eta}$. This relation represents the decrease of the travel time from its zero-voltage value, to first order in $V$, and it could serve to extract the impact damping $\eta$. At high voltages ($V \gg 1$), one would observe $I \propto V^2$, as higher accelerations lead to shorter travel times. However, current experimental parameters [30] indicate that typically $V^2 \lesssim 10^{-2}$, which suggests that the high-voltage regime is not reached.

We now reconsider the intrinsic oscillator damping (thus $Q < \infty$), where a minimum voltage has to be applied for self-oscillations [18, 32] in order to overcome the frictional losses. This voltage can be obtained by demanding the total loss (both by impact damping and mechanical friction) during one half-cycle to equal the energy gain by electrostatic acceleration. This then defines the threshold dc voltage for the shuttle’s self-oscillations in the absence of driving:

$$V^2_{\text{thr}} = \frac{\pi}{4} Q^{-1}. \quad (3)$$

Numerical calculations for the experimentally relevant parameters suggest that apart from the appearance of this threshold voltage there is no appreciable modification of the shuttle dynamics above threshold as long as $Q$ is large [30].
2.2. General nonlinear map in the presence of driving

Much more complex nonlinear behavior may arise under sufficiently strong additional oscillatory driving. For a given impact event at $z = -1$ and time $\tau^{(n)}$, the pair $(v_n', \varphi_n')$ of the initial velocity $v_n$ and the phase $\varphi_n = \Omega \tau^{(n)}$ of the external oscillating force, completely determine the shuttle evolution towards the next impact at $z = -1$ (usually with an intermediate impact at $z = 1$). The new values $\varphi_{n+1}$ and $v_{n+1}$ follow from the unique mapping (cf figure 2):

$$\varphi_{n+1} = \varphi' \left( \varphi_n, v_n \right),$$  

(4)

$$v_{n+1} = v' \left( \varphi_n, v_n \right).$$  

(5)

Note that in the absence of an external drive, there would be only a one-dimensional relation $v_{n+1} = v' \left( v_n \right)$ with a simple fixed-point $\tilde{v} = v' \left( \tilde{\varphi}, \tilde{v} \right)$. That is the regular shuttling already discussed above.

3. Synchronization of an impacting shuttle to an external drive

Although it is known that such two-dimensional nonlinear mappings can generate both complex attractors with a period larger than one cycle and chaotic dynamics our analysis focuses on the case of greatest potential impact for applications. This is synchronization to an external drive (injection locking), which allows to operate a nanomechanical shuttle as a rectifier [40]; it may also be exploited to lock the shuttle dynamics to a very precise external frequency source even for a rather weak drive. Thus the nonlinear dynamics can be turned from a complication into a useful tool. In addition, the framework presented here can form the basis for discussing higher-order fixed-points and chaotic motion in shuttles and similar impacting systems in general.

3.1. Numerical simulation of the phase dynamics

At small external driving strength, there is no simple fixed-point. This is because the phase just proceeds according to $\varphi_{n+1} \approx \varphi_n + 2\Omega \tau_0$, with $\tau_0$ the one-way travel time in the absence of
driving. Generally, $2\pi\tau_0$ is not an integer multiple of $2\pi$, and so there is no definite phase relation between the external driving and the shuttling. The combination of the shuttle’s impact dynamics and the weak incommensurate driving makes the shuttle motion lose its exact periodicity. As a result, the shuttle’s round-trip time $2\tau_0$ effectively changes slightly from step to step. Figure 3(a) depicts the phase increment $\phi_{n+1} - \phi_n$ of the external drive between two impacting events is shown as a color-code. As time progresses (vertical axis, measured in shuttle cycles), the phase increment either displays a slow beating pattern, or it may lock to a value of $2\pi$, signalling synchronization. This happens inside a certain interval, as the dc shuttle voltage (or any other parameter) is varied, shifting the shuttle’s eigenfrequency. ($\eta = 0.2$, $Q = 10^3$, $F = 10^{-3}$, $\Omega = 1.054$) (b) The effective, time-averaged shuttle frequency $\omega = \pi/\bar{\tau}$ is plotted versus an external parameter (the voltage). Inside the synchronization interval, the shuttle becomes robust against noise (brown curve)—the curve shown here was obtained for fluctuations $\Delta \Omega = 0.02$ in the driving frequency (see main text) and is evaluated after 6000 impact events. The fluctuations decrease for longer averaging times.

![Figure 3](image_url)

Figure 3. Synchronization of an impacting electron shuttle to an external drive. (a) The phase increment $\phi_{n+1} - \phi_n$ of the external drive between two impacting events is shown as a color-code. As time progresses (vertical axis, measured in shuttle cycles), the phase increment either displays a slow beating pattern, or it may lock to a value of $2\pi$, signalling synchronization. This happens inside a certain interval, as the dc shuttle voltage (or any other parameter) is varied, shifting the shuttle’s eigenfrequency. ($\eta = 0.2$, $Q = 10^3$, $F = 10^{-3}$, $\Omega = 1.054$) (b) The effective, time-averaged shuttle frequency $\omega = \pi/\bar{\tau}$ is plotted versus an external parameter (the voltage). Inside the synchronization interval, the shuttle becomes robust against noise (brown curve)—the curve shown here was obtained for fluctuations $\Delta \Omega = 0.02$ in the driving frequency (see main text) and is evaluated after 6000 impact events. The fluctuations decrease for longer averaging times.

3.2. Adler equation for a nanomechanical shuttle

Mathematically, this corresponds to the generation of a stable period-one fixed point $\tilde{\phi} = \phi' (\tilde{\phi}, \tilde{v})$, $\tilde{v} = v' (\tilde{\phi}, \tilde{v})$. We now examine the appearance of this phenomenon both numerically and analytically. Numerical observations indicate that the system, once started anywhere in the $(\phi, v)$-plane, quickly relaxes to a 1d manifold that can be described by the phase coordinate alone (figure 4(a)). Therefore, we focus on the phase map and expand it as

$$q_{n+1} = q_0 + 2\Omega \tau_0 + K \sin (q_0 - q_0) + \ldots$$

Here the values of $K$ and $q_0$ depend on the detailed microscopic parameters, with the coupling $K$ growing from zero upon increasing the drive, and the omitted terms are higher harmonics.
Let us denote the deviation of the drive frequency from the undriven shuttling frequency as \( \delta \Omega = \Omega - \pi / \tau_0 \). If this is small enough, then we obtain a fixed point \( \phi_{\tilde{\tau}} \), with some phase lag \( \tilde{\phi} \) between shuttling and drive, where \( 2 \tau_0 \delta \Omega = K \sin (\tilde{\phi} - \phi_0) \). This requires \( |2 \Omega \tau_0 - 2 \pi| = |2 \delta \Omega \tau_0| < K \), which defines the synchronization interval, i.e. the permissible deviation \( \delta \Omega \). Outside that interval, the phase drifts across the full range \([0, 2\pi]\), with a varying velocity. Near synchronization, we can assume small phase increments, and the phase map of equation (6) can be written as a differential equation by replacing \( \phi_{\tau+1} = \phi_0 - 2 \pi \approx 2 \tau_0 \delta \phi / d\tau \), introducing the time-dependent phase shift \( \phi (\tau) \). This yields a generic equation proposed by Adler to treat phase-locking phenomena [41, 42]. Integrating the Adler equation exactly one finds periodic dynamics for the phase increment in each cycle, with a period diverging as \( [2 \tau_0 \delta \Omega - K]^{-1/2} \) near synchronization. Direct simulations (figure 3) of the nonlinear shuttle dynamics under drive and dc voltage confirm this picture showing, outside the synchronization interval, a periodic modulation with the period diverging near the onset of synchronization.

### 3.3. Microscopic derivation of the Adler constant

While Adler-type equations appear generically in synchronization physics, they become predictive only when we are able to connect the coupling \( K \) to the microscopic parameters of the specific model. To this end we repeat the calculation of the one-way travel time for the driven oscillator, obtaining the correction \( \delta \tau \) to the travel time \( \tau_0 \) in the leading order of the driving \( F \). This also involves finding the change in the impact velocity by a new self-consistency equation, where we exploit that the velocity quickly relaxes to the limit cycle while the phase dynamics are much slower. We defer the rather lengthy calculation to the appendix and just state that \( \delta \tau \)
becomes a harmonic function of $\varphi$. The corresponding correction to the phase increment is 

$$\delta \varphi = 2\Omega \delta \tau \left( \varphi \right)$$

and the Adler coupling constant $K$ reads

$$K \left( \mathcal{F}, \Omega \right) = \frac{\Omega \mathcal{F}}{1 - \Omega^2} v_0 \left[ e^{i \Omega \zeta_0} - 1 \right] \left( Z \left( \tau_0, \Omega \right) + S_f \left( \tau_0, \Omega \right) \right).$$

(7)

Here $v_0 = 2V/\sqrt{1 - \eta}$ is the impact velocity without drive (see above). $Z \left( \tau_0, \Omega \right)$ and $S_f \left( \tau_0, \Omega \right)$ are harmonic functions of their arguments, depend on the impact damping parameter $\eta$ and the velocity $v_0$, and are defined in the appendix. Equation (7) is given under the simplifying but realistic assumptions of large $Q \gg 1$ and accurately matches the numerical results for sufficiently low drive $\mathcal{F}$.

The synchronization interval $\delta \Omega < K/(2\tau_b)$ grows linearly with drive strength and increases for lower voltages (while shifting towards lower frequencies; cf figure 4(c)). At very small voltages, modifications occur due to intrinsic damping (finite $Q$), which destroys the self-oscillations.

3.4. Stability against noise

The existence of an extended synchronization interval implies an increased robustness of the shuttling against noise. Within this interval, the Adler equation describes a phase particle trapped in a local minimum of a washboard potential (figure 4(b)). There the effective shuttle frequency is completely insensitive to a slow drift of system parameters (voltage, effective capacitance, damping, etc). In contrast, time-dependent noise may give rise to phase slips by lifting the particle over the barrier. However, this process is exponentially suppressed when moving towards the middle of the synchronization plateau [43]. We have confirmed this by direct simulations of noise: a noisy impact parameter $\eta$ does not show any effect on the synchronization range both in a numerical and a leading order perturbative analysis. For a Gaussian distributed noisy driving frequency $\Omega$ (which we assume constant between the impact events) we observe phase slips which drive the effective shuttle frequency $\omega$ away from $\langle \Omega \rangle$. They are numerous for voltages at the edges of the former synchronization interval. However, in the center of that interval (e.g. $V = 0.053$ in figure 3), for a noise value of $\Delta \Omega/\Omega = 0.02$, we observe no phase slips within 300,000 iterations, resulting in a frequency imprecision of at most $10^{-5}$ in this example. The remaining fluctuations of the effective time-averaged frequency $\omega$ decay with time as $1/\sqrt{\tau}$, as expected. Deviations of $\omega$ from the drive $\Omega$ are exponentially suppressed in the middle of the interval.

While this numerical example illustrates the potential for noise-resistant operation, we note that a full analysis of the frequency fluctuations as well as suitable optimization of parameters is beyond the scope of the present work. The sources of noise, which would have to be characterized carefully in actual experiments, include: thermal mechanical Langevin noise acting on the shuttle (which can be obtained by measuring the shuttle’s mechanical quality factor at small oscillations), any fluctuating momentum transfer on impact, possible coupling to other mechanical modes of oscillation due to the nonlinear impact events, charge noise (e.g. 1/f noise from nearby impurities), and mechanical frequency fluctuations and drift (e.g. due to temperature fluctuations) and intrinsic fluctuations of the voltage source.

Applications of single electron devices in metrology demand both a high accuracy in measuring the transmitted charge as well as operation at a sufficiently high current output. For
improving the (‘phenomenological’) constants describing the three quantum electrical effects represented in the quantum metrology triangle, i.e. the Josephson effect, quantum Hall effect and single electron transport, a minimal frequency imprecision of $10^{-7}$ is required, and $10^{-8}$ for testing the consistency of the metrological triangle [22]. In order to serve as a future current standard, a single electron shuttling device must produce a current output clearly beyond 100 pA at a white noise level of $1 \text{fA}/\sqrt{\text{s}}$ or less [22]. For example, current single electron transfer experiments reach a level of frequency imprecision of almost $10^{-6}$ at a current of 150 pA [44].

In our numerical example, the goal for the optimization of device parameters then would be to reduce the effective frequency noise to less than the $10^{-6}$ [44] that are currently attainable. This should be reachable, e.g., by increasing the external drive strength sufficiently which increases the effective barriers in the washboard potential, preventing phase slips.

4. Possible experimental realization

Figure 1(a) depicts a possible experimental realization that would strongly suppress cotunneling events. Here, doubly-clamped, tensile-stressed silicon nitride string resonators with linear quality factors of several 100,000 [45, 46] can be efficiently and controllably driven to sufficiently high (impacting) amplitudes of several 10 nm using dielectric gradient field actuation [47] (acoustic [25] or capacitive drive [23, 24] can be used in similar contexts). Thus, during tunneling, the island is always tens of nanometers away from at least one electrode. The requirement that the charging energy dominates $k_BT$ can be satisfied with an island-electrode capacitance in the attofarad range, imposing experimentally accessible typical island cross sections of tens of nanometers. With these prerequisites within reach [25, 48], the ideas presented here would enable quantitatively reliable current measurements with a mechanical shuttle.

5. Conclusions

In this paper we have analyzed nanomechanical single electron shuttles in the impact regime. We predict that synchronization to an external driving frequency can occur and can be exploited to achieve a very precise shuttle current. This has immediate applications in the active experimental research on mechanical electron shuttles.

In the future, it might also be interesting to study alternative configurations like two nearby electron shuttles which can be coupled (mechanically or capacitively) and whose oscillations can synchronize. This will lead to nontrivial current correlations.

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Appendix A. Microscopic derivation of the Adler constant

The Adler equation has been introduced as an effective ansatz for the phase map (6). The phase map relates the total phase increment during one full cycle of an externally driven motion to the sum of two terms:

\[ \Delta \phi = \phi_{n+1} - \phi_n = 2\Omega \tau_0 + K \sin \left( \phi_n - \phi_0 \right). \]

The first term describes the (trivial) propagation of the phase under the driving frequency. The second term represents a periodic function of the phase itself and accounts for the nonlinearity in the phase evolution. Note that any non trivial periodic function would form a nonlinear term in the phase map. As there is no a priori criterion for picking a particular periodic function, a harmonic function is normally chosen as the simplest ansatz. The constant of proportionality, the so-called Adler constant \( K \), is usually treated as an effective (fitting) parameter. In this article, however, we directly calculate the Adler constant from the underlying microscopic dynamics—to our knowledge for the first time.

We approach the phase increment \( \Delta \phi (F) = \Omega \Delta \tau (F) \) by evaluating the variation of the total travel time \( \Delta \tau (F) \) of the oscillator caused by the external driving during one full oscillation cycle. This includes the forward and backward trajectory and two impact events (at the left and the right contact). We work to leading order in the driving strength \( F \) which resembles the force of a linear response calculation. It will turn out that a periodic function of the phase variable and, in consequence, Adler’s constant \( K \) can be simply read off. In the following, we collect various contributions to the leading order variation in \( F \).

A.1. Variation of the trajectory by external driving

We take the solution of the undamped and undriven oscillator \( z_0 (\tau) \) as the reference trajectory. The variation of the trajectory under driving is \( \delta z (\tau) \). With the ansatz \( z (\tau) = z_0 (\tau) + \delta z (\tau) \) equation (1) reduces to

\[ \delta \ddot{z} (\tau) = -\delta z (\tau) + F \cos (\Omega \tau). \] (A.1)

We solve this reduced equation of motion using the ansatz

\[ \delta z (\tau) = A \cos (\tau + \xi) + \frac{F}{1 - \Omega^2} \cos (\Omega \tau + \Phi). \] (A.2)

As there is no change to the impact position we keep the spatial boundary conditions fixed \( \delta z \big|_{\tau=0} = 0 \) but allow for changes of the velocity \( \delta \dot{z} \big|_{\tau=0} = \delta v'(0) \). From these boundary conditions we determine the missing constants \( A \) and \( \xi \) of the ansatz

\[ A \cos (\xi) = -\frac{F}{1 - \Omega^2} \cos (\Phi) \] (A.3)

\[ -A \sin (\xi) = \delta v'(0) + \frac{F}{1 - \Omega^2} \Omega \sin (\Phi). \] (A.4)

Inserting these expressions into the ansatz gives the following solution for the variation of the trajectory. We define \( g (\tau) \) for later reference.
\[\delta z (\tau; \Phi) = \mathcal{F} v_0 f (\tau, \Phi) + \delta v' (0) \sin (\tau) \]  
\[\delta \dot{z} (\tau; \Phi) = \mathcal{F} v_0 \dot{f} (\tau, \Phi) + \delta v' (0) \cos (\tau) \]  
\[f (\tau, \Phi) = \frac{1}{(1 - \Omega^2) v_0} \left[ \cos (\Omega \tau + \Phi) - \cos (\Phi) \cos (\tau) + \Omega \sin (\Phi) \sin (\tau) \right] \]  
\[g (\tau) = \sin (\tau) / v_0 (\tau_0). \]

\section*{A.2. Leading order correction to the one-way travel time}

Any variation in the trajectory may lead to a variation of the one-way travel time between two impacts \(\tau_{\text{impact}} = \tau_0 + \delta \tau\). We determine this variation for the one-way travel time around the undriven travel time \(\tau_0\). Note that this variation consists of two parts, one directly related to the variation of the trajectory under drive, the other originating from a change of the velocity of the oscillator. This velocity change will be discussed self-consistently. Yet until that point, effectively two (dependent) variational parameters \(dv\) and \(\mathcal{F}\) will be considered.

\[\delta \tau = - \frac{\delta z (\tau_0)}{\dot{z} (\tau_0)} \equiv - \mathcal{F} f (\tau_0, \Phi) - \delta v' (0) g (\tau_0) \]  

Within this approximation, we replace the perturbed velocity by the unperturbed one \(\dot{z} (\tau_0; \Phi) = v_0 (\tau_0)\).

\section*{A.3. Leading order corrections to the velocity at the second impact}

The velocity shortly before the second impact is expanded around the corresponding velocity of the undriven case up to linear order in the variation w.r.t. the one-way travelling time and the velocity

\[\ddot{z} (\tau_0 + \delta \tau) \equiv \dot{z}_0 (\tau_0 + \delta \tau) + \delta \dot{z} (\tau_0 + \delta \tau) \]

We insert the expressions for \(\delta \tau\) and \(\delta \dot{z} (\tau_0)\) to obtain the velocity at impact

\[\delta v (\tau_0 + \delta \tau) = \mathcal{F} R_f (\Phi) + \delta v' (0) R_v \]  

The driving \(\mathcal{F}\) and the initial velocity variation \(\delta v' (0)\) can be regarded as forces in a linear response approach. For convenience, we denote the response functions

\[R_f (\Phi) = v_0 \dot{f} (\tau_0, \Phi) - \dot{z}_0 (\tau_0) f (\tau_0, \Phi) \]  
\[R_v = v_0 (\tau_0) g (\tau_0) \]  

The one-way travel time \(\tilde{\tau}\) and the velocity variation \(\delta \dot{v}\) on the backward path can be obtained analogously with the replacements \(\delta v' (0) \rightarrow \delta v' (\tau_0 + \delta \tau) = \sqrt{\eta} \delta v (\tau_0 + \delta \tau)\) (initial velocity variation), \(\Phi \rightarrow \Phi = \Phi + \Omega (\tau_0 + \delta \tau)\) (propagated phase), and \(\mathcal{F} \rightarrow -\mathcal{F}\) (force acting towards the opposite direction).
\[ \delta \tau = \mathcal{F} \left( \tau_0, \Phi \right) - \delta v \left( \tau_0 + \delta \tau \right) \sqrt{\eta} g \left( \tau_0 \right) \]
\[ = \mathcal{F} \left( \tau_0, \Phi \right) - \mathcal{F} g \left( \tau_0 \right) \sqrt{\eta} R_f \left( \Phi \right) - \delta v' \left( 0 \right) g \left( \tau_0 \right) \sqrt{\eta} R_v \]
\[ \delta v \left( T \right) = - \mathcal{F} R_f \left( \Phi \right) + \sqrt{\eta} \delta v \left( \tau_0 + \delta \tau \right) R_v \]
\[ = - \mathcal{F} R_f \left( \Phi \right) + \mathcal{F} \sqrt{\eta} R_f \left( \Phi \right) R_v + \delta v' \left( 0 \right) \sqrt{\eta} R_v^2. \]

**Steady state velocity.** We assume that the velocity relaxes quickly to the steady state value. In the following we only consider cycles with exactly two impact events but no other turning points, i.e. the full period of the motion is \( T \approx 2 \tau_0. \) Then the steady state velocity can be calculated self-consistently from the condition
\[ \delta v' \left( T \right) = \delta v' \left( 0 \right). \]
Inserting the above result for \( \sqrt{\eta} \delta v \left( T \right) \) we solve for \( \delta v'. \) The steady state velocity is a harmonic function of the phase \( \Phi \)
\[ \delta v' = \mathcal{F} \sqrt{\eta} \frac{-R_f \left( \Phi \right) + \sqrt{\eta} R_f \left( \Phi \right) R_v}{1 - \eta R_v^2}. \quad \text{(A.13)} \]

**A.4. Total phase shift during one cycle**

The total phase shift during one full cycle is
\[ \delta \Phi / \Omega = \delta \tau + \delta \tilde{\tau} \]
\[ = \mathcal{F} \left[ f \left( \tau_0, \tilde{\Phi} \right) - f \left( \tau_0, \Phi \right) - g \left( \tau_0 \right) \sqrt{\eta} R_f \left( \Phi \right) \right] - \delta v' \left( 0 \right) \left( 1 + \sqrt{\eta} R_v \right) \]
\[ = \mathcal{F} \left[ f \left( \tau_0, \tilde{\Phi} \right) - f \left( \tau_0, \Phi \right) \right] + \mathcal{F} g \left( \tau_0 \right) \sqrt{\eta} \frac{R_f \left( \Phi \right) - R_f \left( \Phi \right)}{1 - \sqrt{\eta} R_v}. \]

**A.5. Analytic expression for Adler constant**

An analytic expression for the Adler constant \( K \) is most easily extracted using a complex representation of the phase dependent terms. We write \( \cos \left( x \right) = \Re \left[ e^{ix} \right], \sin \left( x \right) = \Re \left[ -ie^{ix} \right] \) and use the linearity of the real part to express
\[ f \left( \tau_0, \Phi, \Omega \right) = \Re \left[ e^{i \phi} \frac{Z \left( \tau_0, \Omega \right)}{(1 - \Omega^2) v_0} \right] \quad \text{(A.14)} \]
\[ Z \left( \tau_0, \Omega \right) = e^{i \Omega \tau_0} - \cos \left( \tau_0 \right) - i \Omega \sin \left( \tau_0 \right) \quad \text{(A.15)} \]
\[ R_f \left( \Phi, \Omega \right) = \Re \left[ e^{i \phi} v_0 \frac{Z \left( \tau_0, \Omega \right) - z_0 \left( \tau_0 \right) Z \left( \tau_0, \Omega \right)}{(1 - \Omega^2) v_0} \right] \quad \text{(A.16)} \]
\[ S_y(\tau_0, \Omega) = \frac{g(\tau_0)}{1 - \sqrt{\eta} \sqrt{\eta}} \left[ v_0 \dot{Z}(\tau_0, \Omega) - \ddot{z}_0(\tau_0)Z(\tau_0, \Omega) \right]. \]

Together with the previously obtained solution for the case without driving
\[ v_0 = 2\sqrt{V(1 - \eta)} \]
\[ \ddot{z}_0(\tau_0) = -v_0\sqrt{\eta} \sin(\tau_0) + (1 + \sqrt{V})\cos(\tau_0), \]
we can write the phase advance during one cycle as a harmonic function of the phase \( \Phi \)
\[
\delta \Phi = \frac{F\Omega}{(1 - \Omega^2)v_0} \Re \left[ e^{i\Phi} (e^{i\Omega \tau_0} - 1) (Z(\tau_0, \Omega) + S_y(\tau_0, \Omega)) \right] \\
= \frac{F\Omega}{(1 - \Omega^2)v_0} \cos(\Phi - \varphi(\tau_0, \Omega)) \left| (e^{i\Omega \tau_0} - 1) (Z(\tau_0, \Omega) + S_y(\tau_0, \Omega)) \right| \tag{A.17}
\]
\( \varphi(\tau_0, \Omega) \) represents a phase shift caused by the nonlinear response of the system. We read off the Adler constant
\[
K(\mathcal{F}, \Omega) = \frac{\Omega\mathcal{F}}{(1 - \Omega^2)v_0} \left| (e^{i\Omega \tau_0} - 1) (Z(\tau_0, \Omega) + S_y(\tau_0, \Omega)) \right|. \tag{A.18}
\]

**A.6. Approximate solution**

To obtain an approximate solution we assume that the undriven oscillation frequency does not deviate from the eigenfrequency \( \omega = 1 \) of the oscillator. This means that we neglect the effect of the voltage on the one-way travel time and set \( \tau_0 \approx \pi \). In consequence, \( S_y(\pi, \Omega) = 0 \). Expanding \( \Omega = 1 + \delta \Omega \) then results in
\[
K(\mathcal{F}) = \pi\mathcal{F}/2v_0.
\]
Note that in this approximation \( K \) is independent of the external driving frequency \( \Omega \). Yet the initial approximation did not completely eliminate its dependence on the voltage as the velocity \( v_0(\sqrt{V}) \) remains a function of the voltage.

Applying the approximate solution we give a rough estimate of the synchronization range based on the following argument: In the steady state, the phase increment must be a multiple of \( 2\pi \), i.e. \( \delta \varphi = 2\pi n = 2\Omega \tau_0 + K \sin(\varphi - \varphi_0) \). With \( |\sin(\varphi)| < 1 \) a constraint on the minimal driving strength follows:
\[
\mathcal{F} > 4v_0|\Omega - \omega| = \mathcal{F}_{\text{min}}(\delta \Omega).
\]
If the driving frequency matches the eigenfrequency of the shuttle (or one of its higher harmonics) synchronization trivially holds. For small detunings \( \delta \Omega \) driving above the minimal driving strength is capable of synchronizing the oscillator.

If the full \( \Omega \)-dependence of \( K(\mathcal{F}, \Omega) \) is considered in the general solution, the minimal driving strength shows a nonlinear dependence on the detuning. However, for small detunings, the qualitative picture still holds.
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