Research on Brownian Thermal Noise Limit of a Cylindrical Ultra-Stable Cavity with Support Pads

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Abstract: Brownian thermal noise (TN) of ultra-stable cavities (USCs) imposes a fundamental limitation on the frequency stability of ultra-narrow linewidth lasers. This work investigates the TN in cylindrical USCs with the four support pads in detail through theoretical estimation and simulation. To evaluate the performance of state-of-the-art ultra-narrow linewidth lasers, we derive an expression of the TN for a cylindrical spacer according to the fluctuation–dissipation theorem, which takes into account the front face area of the spacer. This estimation is more suitable for the TN of the cylindrical USC than the previous one. Meanwhile, we perform detailed studies of the influence of the four support pads on the TN in cylindrical USCs for the first time by numerical simulations. For a 400 mm long cylindrical USC with an ultra-low expansion spacer and fused silica substrates, the displacement noise contributed from the four support pads is roughly four times that of the substrates and the GaAs/AlGaAs crystalline coating. The results show that the four support pads are the primary TN contributors under some materials and geometries of USCs.

Keywords: ultra-narrow linewidth laser; ultra-stable cavity; Brownian thermal noise; support pads

1. Introduction

Ultra-narrow linewidth lasers play a key role in several fields, such as high-precision spectroscopy [1,2], time and frequency standards [3–7], gravitational wave detection [8], fundamental physics tests [9,10], and transfer of optical frequencies by fiber networks [11,12]. Generally, ultra-narrow linewidth lasers are based on lasers that are stabilized to USCs using the Pound–Drever–Hall locking technique and the optical length stability of USCs determines the frequency stability of ultra-narrow linewidth lasers [1,2,13,14]. Using thermal isolation, vacuum, and mechanical isolation, the influence of pressure, temperature, and vibration on the USCs can be well suppressed [14].

Nevertheless, a fundamental issue is that TN causes microscopic length changes in the USCs. TN is composed of a variety of noise sources, such as Brownian, thermo-refractive, and thermo-elastic noise. The main TN is Brownian noise. To decrease this noise, researchers have invested in significant work [15–28]. Y. Levin [15] proposed an effective method for calculating TN, which is based on the fluctuation–dissipation theorem (FDT). To estimate the TN of a spacer, mirror substrates, and mirror coating of cylindrical USCs, Numata et al. [16] provided several equations using this method. A few strategies are being applied to reduce the TN in the USC, such as low mechanical loss of substrate and coating, low-temperature technologies, and increasing the length of USCs [18,19,24–28]. At present, the ultra-narrow linewidth laser with the best performance shows that the frequency stability has achieved a $10^{-17}$ order of magnitude at 1s [24–26], and its linewidth is about 8 mHz [24]. As a result, the spacer and the support structure contribution are non-negligible [17]. Therefore, to evaluate the performance of state-of-the-art ultra-narrow...
linewidth lasers using the cylindrical USC, it is important to propose a more reasonable analysis approach of the TN in cylindrical USCs with support pads.

In this work, theoretical estimation and finite element simulation (FEA) of the TN in cylindrical USCs with support pads are presented in detail. First, according to FDT, we derive the formula for estimating the TN of a cylindrical spacer. Under the same conditions, the new estimation method and FEA are compared. The two results are consistent. The new estimation formula is suitable for spacers made of various materials and with any geometric dimension. We present the analysis and interpretation of the discrepancies between the FEA and theoretical estimation. Second, the influence of the support pads on the TN of USCs is studied in detail. The results show that for a long cylindrical USC made of ultra-low expansion (ULE) spacer and fused silica (FS) substrates and GaAs/AlGaAs crystalline coating, the contribution of the support pads to the total TN cannot be ignored, and may even be the main factor. This point should be taken into consideration for the ultra-narrow linewidth laser with the design objective of the fractional frequency stability $10^{-17}$ order and realized by a long cylindrical USC.

The work is arranged as follows: The TN principle framework in the USCs is discussed in Section 2. A typical cylindrical USC design is simulated by FEA in Section 3.1, and the distribution characteristics of TN in the USC are discussed. In Section 3.2, towards a better understanding of the differences between the FEA and theoretical calculation, the spacers of TN with various conditions are investigated including the USC materials and the USC length. In Section 4, this work focuses on the influence of the support pads on the TN in USCs. Finally, we present the conclusions of this paper in Section 5.

2. TN Principle Framework

2.1. Fundamental Principle

Based on Y. Levin’s proposed direct approach [15], Numata et al. [16] investigated the influence of TN caused by Brownian motion on the USC length stability. This theory is illuminated below. The distance between two highly reflective mirrors of USCs is $x$. While a USC is subjected to a force with an $F_0$ amplitude and an $f_0$ frequency, the power spectral density (PSD) $G_x(f_0)$ of the fluctuations of USCs length along the $x$-axis can be written as [14,16]:

$$G_x(f_0) = \frac{4 k_B T U_{\text{max}} \phi}{\pi f_0 F_0^2}$$

where $k_B = 1.381 \times 10^{-23}$ J/K and $T$ are the Boltzmann constant and the temperature, respectively. $U_{\text{max}}$ and $\phi$ denote the maximum elastic strain energy (ESE) and the USC loss angle, respectively.

Generally, a USC is made up of a spacer and two mirrors which consist of substrate and coating. We use $G_x(f_0)_{\text{spacer}}$, $G_x(f_0)_{\text{substrate}}$, and $G_x(f_0)_{\text{coating}}$ to express the PSDs of length fluctuations for the spacer, substrates, and coatings, respectively. It is assumed that all TN sources are not related, the total PSD $G_x(f_0)$ can be presented as [16]:

$$G_x(f_0)_{\text{total}} = G_x(f_0)_{\text{spacer}} + 2(G_x(f_0)_{\text{substrate}} + G_x(f_0)_{\text{mirror}})$$

$$= \frac{4 k_B T}{\pi f_0 F_0^2} (U_{\text{spacer}} \phi_{\text{spacer}} + 2U_{\text{substrate}} \phi_{\text{substrate}} + 2U_{\text{coating}} \phi_{\text{coating}})$$

where $U_{\text{substrate}}$, $U_{\text{spacer}}$, and $U_{\text{coating}}$ are the ESE of the substrates, spacer, and coatings, respectively. $\phi_{\text{substrate}}$, $\phi_{\text{spacer}}$, and $\phi_{\text{coating}}$ are the loss angles of the substrates, spacer, and coatings, respectively. In this work, theoretical estimations of $U_{\text{substrate}}$ and $U_{\text{coating}}$ are referred to ref. [16]. For ultra-narrow linewidth lasers, when the length of USC is $L$, the Allan deviation [29] $\sigma_y$ of the fractional frequency fluctuations can be expressed as:

$$\sigma_y = \sqrt{\frac{2 \ln(2) f_0 G_x(f_0)}{L^2}}$$
2.2. **TN Principle of a Cylindrical Spacer**

The cylindrical USC supported by four elastic support pads is shown in Figure 1. At the end of the cylindrical USC, there are two high-reflection mirrors, which are optically contacted on the USC. There is a diameter $2r$ through a hole in the light. At the outer diameter of the USC, two grooves are cut along the light direction to serve as the support surface, and the USC has a mushroom-shaped cross-section. The four supporting pads are symmetrically placed on the supporting surfaces on both sides.

![Figure 1. Geometrical model of the cylindrical USC supported by four elastic support pads.](image)

An estimation of the TN of the spacer in the USC is calculated by averaging the cross-sectional area. Based on Levin’s direct approach, pressure is uniformly distributed across the front faces of the spacer. It leads to an ESE $U_{\text{spacer}}$ and the PSD of length fluctuations $G_x(f_0)_{\text{spacer}}$ can be expressed as

$$
U_{\text{spacer}} = \frac{LE_{\text{spacer}}}{\pi R^2 (R^2 - r^2 - w_c h_c)} \left\{ \frac{1}{b} \phi_{\text{spacer}} \right\}
$$

$$
G_x(f_0)_{\text{spacer}} = \frac{4h_y T}{\pi L_{\text{spacer}} \pi R^2 (R^2 - r^2 - w_c h_c)} \left\{ \frac{1}{b} \phi_{\text{spacer}} \right\}
$$

where $E_{\text{spacer}}$ is the modulus of elasticity of the spacer. Equation (4) takes into account that the areas of the light hole and the cut parts for supporting the USC reduce the cross-sectional area of the spacer. We use the formula for the position fluctuations of a free elastic bar end for the derivation of the above formulas. When the length fluctuation of the USC is calculated, it is necessary to apply opposite forces at both ends simultaneously based on Y. Levin’s direct method.

2.3. **TN Principle of the Support Pads**

For modern USCs, generally, vibration-insensitive mountings are used to achieve fractional frequency stabilities below $\sigma_y = 10^{-15}$ [1,2,26,30–32]. The horizontal USCs are widely supported at the Airy points by four elastic pads [1,26,30,31]. As the loss angle of the pad materials $\phi_{\text{support}}$ is relatively high, the pads’ contribution to the TN may not be ignored [17]. It is assumed that the pads are fixed and the spacer is simplified as an elastic bar [33]; the ESE $U_{\text{support}}$ and the PSD of length fluctuations $G_x(f_0)_{\text{support}}$ for the four support pads are provided by [17]:

$$
U_{\text{support}} = \frac{2L_{\text{support}} G_{\text{support}}^2}{\sqrt{3} d_{\text{support}} L_{\text{support}} R^2} U_{\text{spacer}}
$$

$$
G_x(f_0)_{\text{spacer}} = \frac{2L_{\text{support}} G_{\text{support}}^2 c^2_{\text{support}}}{\sqrt{3} d_{\text{support}} L_{\text{support}} R^2 \phi_{\text{support}}} G_x(f_0)_{\text{ spacer}}
$$

where $G_{\text{support}}$ is the shear modulus of the support pads. $G_{\text{support}}$ can easily be converted by the elastic modulus $E_{\text{support}}$ and Poisson ratio $\sigma_{\text{support}}$ of the pads $G_{\text{support}} = E_{\text{support}} / (2(1-\sigma_{\text{support}}))$. 
+ 2\sigma_{\text{support}}) \text{[34]. Note that the Airy points are located at } L_{\text{sup}} / L = 1/(2\sqrt{3}) \text{ according to } \text{ref. [33]} \text{ in Equation (5).}

According to Equations (1) and (2), 0.1 nJ ESE corresponds to a \( G_x(f_0) = 0.858 \times 10^{-35} \text{ m}^2/\text{Hz} \) at 1 Hz and a \( \epsilon_T = 3.44 \times 10^{-17} \) at 1s for an all-ULE USC with 100 mm length \([17,23]\).

3. Estimation and Simulation

The TN of the USC is proportional to ESE according to Equation (1). In this work, we use ESE to present our results for research convenience in this Section. The ESE in the spacer is simulated by the FEA software. As shown in Figure 2, the red wide line represents the laser beam, whose profile corresponds to the pressure distribution \( p(r) \) loading on the two mirrors. \( p(r) \) can be expressed as \([17]\):

\[
p(r) = \pm \frac{2F_0}{\pi w_0^2} e^{-2(\frac{r}{w_0})^2}
\]

where a radius \( w_0 \) of the Gaussian laser beam is 250 \( \mu \text{m} \).

![Figure 2. Schematic diagram of the Gaussian laser beam acting on the two mirrors of the USC.](image)

According to FDT, the position of the force \( F_0 \) acting on the two USC mirrors is consistent with the laser beam. When the relationship between stress and strain of an elastic-mechanical system is linear, the amplitude of force \( F_0 \) is not strictly required \([23]\). For the convenience of calculation and simulation, a force \( F_0 \) of 1 N is used in this work.

The important parameters are presented in Tables 1 and 2 for the estimation and simulation. In this work, the materials of the USCs include room-temperature materials (\( T = 300 \text{ K} \)) and low-temperature materials (\( T = 4 \text{ K} \)). For the USC, to reduce the influence of temperature change on its length, it needs to be placed in the temperature environment of its zero coefficient of thermal expansion (CTE). Generally, the difference between low-temperature materials and room-temperature materials is mainly determined by the temperature of zero CTE. When the zero CTE point of the material is at room temperature, it is a room-temperature material. The same is true for a low-temperature material.

| Table 1. Parameters used in the calculation estimation and FEA simulation. |
|-----------------|-----|-----|-----|-----|-----|-----|
| Parameters (mm) | Value | \( R \) | \( r \) | \( r_{\text{sup}} \) | \( w_c \) | \( h_c \) | \( d_{\text{sup}} \) |
|-----------------|-----|-----|-----|-----|-----|-----|
|-----------------|-----|-----|-----|-----|-----|-----|
|-----------------|-----|-----|-----|-----|-----|-----|
|-----------------|-----|-----|-----|-----|-----|-----|
|-----------------|-----|-----|-----|-----|-----|-----|

| Table 2. Material properties used for the estimation and simulation. |
|-----------------|-----------------|-----------------|-----------------|-----------------|-----------------|-----------------|-----------------|-----------------|
| Material Properties | ULE [11] | Fused Silica [18] | Silica <111> [25] | Zerodur [19] | Sapphire [35] | GaAs/Al GaAs [19] | Viton [34,36] |
|-----------------|-----------------|-----------------|-----------------|-----------------|-----------------|-----------------|-----------------|
| Elastic modulus (GPa) | 67.6 | 73 | 187.5 | 91 | 400 | 100 | \( 8 \times 10^{-4} \) |
| Poisson ratio | 0.17 | 0.16 | 0.23 | 0.24 | 0.29 | 0.32 | 0.27 |
| Loss angle | \( 1.67 \times 10^{-5} \) | \( 10^{-6} \) | \( 10^{-7} \) | \( 3 \times 10^{-4} \) | \( 3 \times 10^{-9} \) | \( 2.5 \times 10^{-5} \) | 0.33 |
3.1. Distribution Characteristics of ESE

Figure 3 shows that the majority of the ESE in the USC is mainly located in the two mirrors, which is similar to the USCs of other shapes [16,17,23]. The mirror’s deformation reflects the Gaussian distribution. In the simulation, a radius \( w_0 \) of the Gaussian laser beam is 2 mm for illustration. The maximum ESE in the spacer locates around the light hole, which is in close contact with the two mirrors. The simulation results show that there are large deformations at the boundary between the two mirrors and the spacer. Uniform ESE density distribution is obtained at distances beyond this critical depth. According to Equation (2) in ref. [16], the analytical estimation of the spacer is not satisfied for the entire spacer length because it ignores the local deformation caused by the uneven pressure distribution on the cross-sectional area.

![Figure 3. Plot of the ESEDEN of the USC, in which the ESEDEN denotes the ESE density. The logarithms of ESEDEN and the color coding are corresponding. The USC length \( L \) is 80 mm. For the demonstration, the deformation amplified by a factor of \( 1 \times 10^6 \) has been adopted.](image)

3.2. Influence of USC Geometric and Materials on ESE

To better investigate the difference between FEA and analytical estimate, we studied the TN in spacers under different conditions, such as the length and materials of spacers.

3.2.1. Room-Temperature Materials

For a room-temperature material USC shown in Figure 4, as the length \( L \) of the USC increases, the ESE calculated by FEA, Equation (4) in this work, and Equation (2) in ref. [16] just gets bigger. As shown in Figure 4, for the 2 spacers with a length \( L \) of 300 mm, the FEA results are approximately 1.3 times larger than that of Equation (4) and 5 times larger than that of Equation (2) in ref. [16], respectively. For the ULE and zerdur spacers, the ESE divergences between the simulation and Equation (4) are near-constant deviations with approximately 0.8 nJ and 0.7 nJ, respectively. However, gaps between the FEA simulation and estimation by Equation (2) in ref. [16] widen with the increase in length \( L \). Taking the two spacers with a length \( L \) of 500 mm as an example, the gaps of ESE between Equation (2) in ref. [16] and FEA are roughly 5 nJ and 3.6 nJ, respectively. Compared with the results calculated by Equation (2) in ref. [16], the estimation by Equation (4) in this work is more fit with the simulation results.
3.2.2. Low-Temperature Materials

For the silica and sapphire USCs shown in Figure 5, the ESE estimated by Equation (4) is closer to the FEA results than that calculated by Equation (2) from ref. [16], which is similar to ULE and zerodur. For the two spacers with a length $L$ of 500 mm, the FEA results are roughly 1.2 times larger than that of Equation (4) and 4 times larger than that of Equation (2) in ref. [16]. When the length $L$ of the spacer increases, for the low-temperature materials USCs, the divergence of the ESE between simulation and Equation (2) tends to be approximately 0.25 nJ and 0.12 nJ, respectively. The gap is similar to USCs made of room-temperature materials.

4. Effect of Support Pads on TN

4.1. Comparison between Simulation and Estimation

To verify Equation (5), the TN of the support pads under various parameters is calculated and simulated, including materials and USC lengths $L$. The thickness and diameter of the two mirrors are 6.3 mm and 25.4 mm, respectively. The support pad material is Viton rubber. The geometric dimensions and the material properties are based on Tables 1 and 2, respectively. The position of the four support pads is arranged according to the “Airy point” estimated in ref. [31].

As shown in Figure 6, Equation (5) can estimate the TN of the support pads with the USC length $L$. With the increase in the length $L$, the gap between the TN of the support pads made of room-temperature materials increases.
pad estimated by Equation (5) and that of the simulation also increases. For different USC materials, this gap is discrepant, and the primary reason may be that the mechanical properties of USC materials are different. For room-temperature materials, when the USC length is 400 mm, the simulation result is about three times the estimate. For low-temperature materials, the result of the all-silica USC simulation is roughly three times the estimate, while for the sapphire USC, this is approximately four times. For cylindrical USCs with a length L less than 100 mm, the TN of the support pads estimated by Equation (5) is basically consistent with the simulation.

![Figure 6](image-url)

**Figure 6.** TN of the four support pads as functions of the spacer length L. The red solid line with squares and the green dashed line with squares denote the FEA simulation results and the estimated results of the all-ULE USC, respectively. The blue solid line with circles and the light blue dashed line with circles denote the FEA simulation results and the estimated results of the all-zerodur USC, respectively. The red solid line with stars and the green dashed line with stars denote the FEA simulation results and the estimated results of the all-silica USC, respectively. The blue solid line with triangles and the light blue dashed line with triangles denote the FEA simulation results and the estimated results of the all-sapphire USC, respectively. (a) Room-temperature materials. (b) Low-temperature materials.

### 4.2. Contribution Ratio of Components to TN

#### 4.2.1. Room-Temperature Materials

Figure 7 shows the contribution ratio of the four parts of the room-temperature cylindrical USC to the total TN including the spacer, the two mirror substrates, the coating, and the four support pads. The coating material is GaAs/AlGaAs with a thickness of 6.83 µm. The support pad is Viton.

![Figure 7](image-url)

**Figure 7.** Percentage of contributions to the TN of the optical length $Gx(f_0)$ as functions of the length L. The coating is GaAs/AlGaAs crystalline with the thickness $d_{coa} = 6.83 \, \mu m$ cited from ref. [19]. The solid red line with circles denotes the spacer results. The dashed green line with squares denotes the substrate results. The dot blue line with triangles denotes the coating results. The dash-dot cyan line with stars denotes the four support pads’ results. (a) All-ULE USC. (b) ULE spacer and FS substrates. (c) All-zerodur USC.
As shown in Figure 7a, for all-ULE USCs, the main contribution of TN comes from the two mirror substrates. With the increase in the USC length $L$, the TN contribution of the two mirror substrates decreases. For instance, compared with the length of 50 mm, the TN contribution of the mirror substrates of the length of 400 mm USC decreases by about 20%. The TN contribution of the mirror coating is the smallest, and its TN contribution remains unchanged with the increase in the USC length and can be ignored. The TN contributions of the four support pads and the spacer increase with the increase in the USC length. For example, compared with the USC length of 50 mm, the TN contributions of the support pads and spacer of the USC length of 400 mm increase by approximately 10%.

According to Figure 7b, for the USC with ULE spacer and FS mirror substrates, with the increase in USC length $L$, the contribution of the support pads to TN increases. However, for the mirror substrates and coating, it decreases. When the USC length is 400 mm, the TN contributions of the spacer and the four support pads are roughly 40%. The interesting phenomenon is that when the USC length increases, the contribution of the spacer to the total TN increases first and then decreases. When the USC length $L$ is approximately 200 mm, it reaches the maximum, about 45%.

As shown in Figure 7c, for the all-zeroindex USC, the main contribution of TN comes from the mirror substrates, which is similar to the all-ULE USC. With the increase in the USC length $L$, the contribution of the TN of the spacer gradually increases, and the contribution of the mirror substrates gradually decreases, while the contributions of the coating and the support pads are unchanged, close to zero, which can be ignored. When the USC length increases to about 400 mm, the two mirror substrates are still the main contributor to the TN, with a value of approximately 80%, and the contribution of the spacer is about 20%.

According to Figure 7, the contribution of the four support pads to the total TN cannot be ignored in some cases, even the main factor, which is mainly reflected in the relatively long cylindrical USC. For instance, when a 400 mm ULE spacer and FS mirror substrate are used, the contribution of the four support pads has risen to the primary position.

4.2.2. Low-Temperature Materials

The contributions of the four parts of the low-temperature cylindrical USC to the total TN are shown in Figure 8. The relevant parameters are identical to those of the room-temperature USC.

Figure 8. Percentage of contributions to the TN of the optical length $Gx(f_0)$ as functions of the length $L$. The coating is GaAs/AlGaAs crystalline with the thickness $d_{\text{coa}} = 6.83 \, \mu\text{m}$ cited from ref. [19]. The solid red line with circles denotes the spacer results. The dashed green line with squares denotes the substrate results. The dot blue line with triangles denotes the coating results. The dash–dot cyan line with stars denotes the four support pads’ results. (a) All-silica USC. (b) All-sapphire USC.

As shown in Figure 8a, for the all-silica USC, with the increase in USC length $L$, the contribution of TN of the support pads gradually increases, whereas the contributions of mirror substrates and coating decrease, and the spacer’s contribution remains unchanged,
which is almost negligible. When the USC length \( L \) is 50 mm, the main contribution is the coating, accounting for about 80\%, and the contributions of the support pads and the spacer is close to zero. When the USC length \( L \) is 400 mm, the main contribution is the four support pads, accounting for approximately 70\%.

According to Figure 8b, for the all-sapphire USC, as the USC length \( L \) increases, the TN contribution of the four support pads increases, and the TN contribution of the coating decreases, which is similar to the all-silica USC. The TN contributions of the mirror substrates and spacer are close to zero, which is almost negligible. When the USC length \( L \) is 50 mm, the main contribution is the coating, accounting for about 100\%. When the USC length \( L \) is 400 mm, its contribution value decreases to approximately 40\%, and the main contribution becomes the four support pads, accounting for roughly 60\%.

For a low-temperature cylindrical USC, under some conditions, the contribution of the four support pads to the TN of the USC cannot be ignored, which needs to be taken into consideration.

5. Conclusions

This work uses theoretical estimation and FEA simulation to study the TN of cylindrical USCs with the four support pads in detail. According to the FDT theory, an estimation formula of the TN of the spacer is derived. The calculation results of the estimation formula are qualitatively consistent with the simulation results. In comparison with the previous estimation, this estimation formula is well suited for the TN estimations of cylindrical USCs. We provide the reasons for the results gap between the theoretical estimation and simulation.

The support pads must be used for cylindrical USCs. The effect of the support pads on the TN of cylindrical USCs is investigated in detail. In previous relevant studies, this impact on TN has not been considered in detail. This study is the first investigation of the support pads influenced by FEA simulations.

For a 400 mm length ULE cylindrical USC with FS substrates, the \( Gx(f_0) \) of the four support pads is approximately four times that of the substrates and the GaAs/AlGaAs crystalline coating. The simulation results show that the four support pads are the major TN contributors under certain materials and dimensions of USCs. In the future, when studying TN in other USCs with different geometric shapes, the support pads should be a consideration, especially the transportable USCs [37]. The results in this work provide guidance on design and minimizing TN for long cylindrical USCs.

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