Effect of Flow Separation Control with Suction Velocity Variation: Study of Flow Characteristics, Pressure Coefficient, and Drag Coefficient

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Received December 28, 2019; Revised April 5, 2020; Accepted April 19, 2020

Abstract  Flow separation is expected to have the effect of increasing aerodynamic drag due to decreased pressure distribution at the rear of the vehicle. The faster the flow separation occurs, the lower the pressure distribution is in the area, thereby reducing vehicle performance. Therefore, flow modification is needed with expected effects on the separation delay and the reduction in wake and vortex formation. This modification can be done through the application of suction active control in the separation area. The research is intended to analyze the effect of suction active control on flow characteristics, pressure distribution and aerodynamic drag on vehicle models with suction velocity variations. The test model used is an Ahmed model modified by changing the orientation of the flow. The study used a numerical computational approach with a standard k-epsilon turbulence model at 19.4 m/s upstream velocity. Results revealed that the use of flow active control was able to reduce wake and vortex formation through separation delay and to increase the minimum pressure coefficient by 73% on the model with Usc2 suction velocity of 0.5 m/s, gaining the highest drag coefficient reduction of 10.897% in the same model.

Keywords  Active Control, CFD, Drag Coefficient, Flow Characteristics, Flow Separation, Pressure Coefficient

1. Introduction

Flow separation is a condition where flow passing through the surface of an object which no longer adheres to the object’s surface. Steadily moving flow will split when separation occurs, resulting in a decrease in pressure distribution and the emergence of aerodynamic drag [1]. Reduced distribution of pressure on the rear window and the surface of the vehicle is a major cause of aerodynamic drag, where both locations contribute 90% of the total resistance and 80% of it occurs at the rear of the vehicle [2]. Flow separation also results in a pressure difference between the front and rear of the vehicle [3]. The faster the separation occurs, the formation of wake and vortex on the back of the vehicle will be even greater. This has the impacts on reducing vehicle velocity, increasing fuel consumption and leading to lower vehicle performance efficiency [4].

Theoretically, flow separation is a form of the viscous effect of fluids [5]. The viscous effect causes the flow of fluid that reaches the top end of the back of the vehicle to lose kinetic energy so it does not have enough momentum to flow in the boundary layer toward the lower edge of the rear of the vehicle. The large difference in velocity between the wake area and the boundary layer creates a backflow phenomenon [5]. Consideration of flow characteristics in the form of a vortex structure around the vehicle needs to be considered to evaluate the stability of the vehicle while moving [6].

The backflow at the rear of the vehicle triggers suction rearward. Rearward suction is caused by large pressure differences between the front and the vacuum area that occurs at rear parts of the vehicle [7]. This affects the vehicle's performance due to the effects of aerodynamic drag. Some of the efforts that can be done to increase the pressure coefficient while reducing aerodynamic drag are reducing or delaying flow separation and minimizing the recirculation zone [8,9].

Reduction of wake formation can be done by identifying the location of the flow separation and then applying a suction active control device in the initial location where the separation occurred [10]. Wessen and Thiele [11]
investigated the effect of using active controls in the form of small holes placed along the upper edge of the rear of the vehicle model. The results showed the use of active controls can increase the average pressure on all parts of the rear wall of the vehicle model and reduce aerodynamic drag by 9.4%.

Gerop and Odenthal [9] also examined the effect of active controls placed at the back of the test model. The results showed that the use of active controls was able to increase base pressure by almost 50%, eliminate the recirculation area and reduce aerodynamic drag by 10%.

Roumeas et al. [12] researched a numerical computational approach employing the k-epsilon turbulence model, applied in an Ahmed model with a slant angle of 25°. The Reynolds number used was $2.8 \times 10^6$ at 40 m/s upstream velocity. The suction active control was placed on the back sloping side. The results showed the use of suction active control could eliminate flow separation and produce reduction in aerodynamic drag close to 17%.

Moussa et al. [13] conducted a study related to the effect of using suction active control on aerodynamic drag of a vehicle model through a numerical computational approach. The results showed the application of suction active control placed on the back side of the vehicle model was effective to reduce the drag coefficient by 9%.

2. Methods

The test model used in this study is a modified Ahmed model (reversed Ahmed body) by changing the direction of flow with 0.17 dimension ratio to the original model. Modification of Ahmed models with the application of active control has been extensively researched to describe the flow phenomena occurring around real vehicles [14,15]. The test model dimension has a length of $l = 174$ mm, height $h = 48$ mm and width $w = 64.83$ mm with a slope of the front geometry of 25°. Active controls in the form of apertures with 7 mm diameters are placed on the upper back side of the model. The active control actuation amounts to five apertures with 10.81 mm distance between them, and designated as suction 1 (sc1), suction 2 (sc2), suction 3 (sc3), suction 4 (sc4) and suction 5 (sc5).

Variations of suction velocity are defined as $U_{sc1}$, $U_{sc2}$, $U_{sc3}$, and $U_{sc4}$. The simulation was carried out at an upstream velocity of 19.4 m/s. Details of the test model and the definition of suction active control are shown in figure 1 and figure 2. The ratio of upstream velocity ($U_0$) to suction velocity ($U_{sc}$) is shown in Table 1.

![Figure 1. Test model](image1)

![Figure 2. Definition of active controls](image2)

| $U_0$ (m/s) | Aperture | $U_{sc1}$ (m/s) | $U_{sc1}/U_0$ | $U_{sc2}$ (m/s) | $U_{sc2}/U_0$ | $U_{sc3}$ (m/s) | $U_{sc3}/U_0$ | $U_{sc4}$ (m/s) | $U_{sc4}/U_0$ |
|-------------|---------|----------------|--------------|----------------|--------------|----------------|--------------|----------------|--------------|
| 19.4        | sc1     | 1.0            | 0.051        | 0.5            | 0.026        | 1.0            | 0.051        | 1.0            | 0.051        |
|             | sc2     | 1.0            | 0.051        | 0.5            | 0.026        | 0.5            | 0.026        | 0.5            | 0.026        |
|             | sc3     | 1.0            | 0.051        | 0.5            | 0.026        | 1.0            | 0.051        | 1.0            | 0.051        |
|             | sc4     | 1.0            | 0.051        | 0.5            | 0.026        | 0.5            | 0.026        | 0.5            | 0.026        |
|             | sc5     | 1.0            | 0.051        | 0.5            | 0.026        | 1.0            | 0.051        | 1.0            | 0.051        |
The location of the pressure field data collection is determined at five different points according to the number of controls in the ratio of the width of the grid to the width of the model, \( z/w = -1/2, z/w = -1/4, z/w = 0, z/w = 1/4, \) and \( z/w = 1/2. \) On models unequipped with active control at each \( z/w \) location, data were taken at 5 (five) points along the axis of imaginary lines that represent the ratio of the height of the grid to the height of the model, respectively at \( y/h = 0.83, y/h = 0.67, y/h = 0.50, y/h = 0.30, \) and \( y/h = 0.17. \) For models with active controls, data were taken at each location corresponding to all values of \( z/w \) along 4 (four) axis lines of \( y/h = 0.67, y/h = 0.50, y/h = 0.30, \) and \( y/h = 0.17. \) Overall, data obtained are at 25 points for models without active control and at 20 points for models with active controls for each suction velocity. Details of the location for taking pressure field data are shown in figure 3 and figure 4.

In this study, the computational approach employed CFD Fluent 6.3 software on standard k-epsilon turbulence model. Initially, the vehicle model is defined into the computational domain and then through the mesh stages using the gambit program. Display mesh and computational domain are shown in figure 5 and figure 6. The boundary conditions applied in this computational simulation incorporated the upstream velocity of 19.4 m/s, air density of 1.225 kg/m\(^3\) and air viscosity of \( 1.7894 \times 10^{-5}, \) where the air is assumed to be at steady-state and incompressible flow.

**Figure 3.** Location of \( C_p \) data acquisition for model without active controls

**Figure 4.** Location of \( C_p \) data acquisition for model with active controls
Calculation of the pressure coefficient is written in the relationship of the pressure coefficient with the pressure on the surface of the model \( P \) and the upstream pressure \( P_0 \), as shown in equation (1) [5]:

\[
C_p = \frac{P - P_0}{\frac{1}{2} \rho U^2}
\]  

(1)

The calculation of aerodynamic drag is written in the relationship between the drag coefficient \( C_d \) and drag force \( F_d \) as denoted in equation (2) [16]:

\[
C_d = \frac{F_d}{\frac{1}{2} \rho U^2 A}
\]  

(2)

3. Results and Discussions

3.1. Flow field

With the help of FLUENT using the velocity relative to the cell zone motion, the pathlines can be displayed. The concept behind the pathlines generation is the over time integration of particle trace from some starting location and time to some later time step. In this simulation, a 0.01 m step size of 500 steps is used for node values and relative velocity pathlines. Velocity pathlines for models without active controls and models equipped with active controls Usc1, Usc2, Usc3, and Usc4 at an upstream velocity of 19.4 m/s are shown in figure 7, 8, 9, 10, and 11.
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Figure 8. Velocity pathlines of the model with active control Usc₁

Figure 9. Velocity pathlines of the model with active control Usc₂

Figure 10. Velocity pathlines of the model with active control Usc₃

Figure 11. Velocity pathlines of the model with active control Usc₄
Figures 7, 8, 9, 10, and 11 show the flow separations generated at the back of the vehicle model which results in the formation of wake and vortex. For models unequipped with active controls, the formation of wake and vortex is considerably large and long, due to the separation that occurs right near the back wall of the test model. A fairly large wake phenomenon will cause a rearward suction effect due to a decrease in the pressure coefficient. Wake is also caused by a longitudinal vortex that forms at the top side edges of the back area of the model. Longitudinal vortexes are formed due to differences in flow velocity between the wake area and the side area of the test model [17]. The longer the wake formed, the greater the effect on aerodynamic drag was [18].

On models that use flow control Usc1 suction velocity, the formation of wake and vortex tends to be smaller than that of the models without active controls. This is due to the delay of separation on the back side of the model. Formation of flow separation tends to be on greater distance from the back wall of the model and the flow velocity on the middle side increases when compared to the model without flow active control.

As for models with Usc2 suction velocity, the formation of wake and vortex also tends to be smaller than the one of models without flow control and models with Usc1 suction velocity. The generation of a wake that tends to be small is also caused by delays in flow separation and reduced recirculation areas. The formation of the small wake is seen from the flow pattern formed on the right side of the model shown by the arrow where the wake formation for the Usc2 velocity has a smaller gap and tends to be thin.

For models with Usc3 suction velocity, the formation of wake and vortex is smaller compared to that of models with no active control and models with Usc1 suction velocity. However, it is smaller than that occurring on the model with Usc2 suction velocity. This can be seen from the example of wake formation which has a smaller gap compared to the model without active controls and the model with the Usc1 suction velocity. The same situation happened in the Usc3 suction velocity model, where the wake and vortex formation tends to be smaller than that of the model without active control and that with Usc1 suction velocity, but it is greater than that of the models with Usc2 and Usc3 suction velocity. A similar study conducted by Harinaldi et al. [19] revealed that suction active control application on the back of the model van had the effect of reducing wake and vortex. This reduction in wake and vortex gives effect on improving overall vehicle performance [20].

### 3.2. Pressure Coefficient ($C_p$)

Minimum pressure coefficients obtained for models without flow control and those equipped with flow control varied in suction velocities Usc1, Usc2, Usc3, and Usc4 are shown in tables 2, 3, 4, 5, and 6.

For models without flow active control, the minimum pressure coefficient is on the upper back side of the model or at the ratio of grid height with model height $y/h = 0.83$ at all positions for the ratios of grid width to model width, $z/w$ as shown in figure 12. The minimum pressure coefficient value is written down at $z/w = -1/2$ of -0.724, $z/w = -1/4$ of -0.566, $z/w = 0$ of -0.439, $z/w = 1/4$ of -0.566 and $z/w = 1/2$ of -0.566, the minimum pressure coefficient. This happens because at the position $y/h = 0.83$, it is the starting point of flow separation and the onset of a backflow phenomenon resulting the pressure coefficient tends to be low in the area. Similar research was also carried out by Anderson [1], finding that the pressure coefficient will be lower in the area where flow separation starts to occur.

**Table 2. Minimum pressure coefficients on the vehicle model without active controls**

| Position | Minimum $C_p$ | $y/h$ position |
|----------|---------------|----------------|
| $z/w = -1/2$ | -0.724 | 0.83 |
| $z/w = -1/4$ | -0.566 | 0.83 |
| $z/w = 0$ | -0.439 | 0.83 |
| $z/w = 1/4$ | -0.566 | 0.83 |
| $z/w = 1/2$ | -0.566 | 0.83 |

**Figure 12.** Comparison of minimum $C_p$ values to $y/h$ positions on models without active control

For models with Usc1 suction velocity, the minimum pressure coefficient is on the lower side of the back of the vehicle model or in the ratio of grid height to model height, $y/h=0.17$ in all positions of ratios of grid width to model width $z/w$. Respective values of pressure coefficients are $z/w = -1/2$ of -0.182, $z/w = -1/4$ of -0.182, $z/w = 0$ of -0.325, $z/w = 1/4$ of -0.268 and for the position $z/w = 1/2$ of -0.239. The results obtained indicate that there has been a delay in flow separation and minimization of backflow. This can be seen from the increase in the minimum pressure coefficient on the top side of the back of the model and the minimum pressure coefficient moving to the bottom side of the vehicle model. Figure 13 shows the abovementioned result.
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Table 3. Minimum pressure coefficients on the model with Usc1

| Position | Minimum Cp | y/h position |
|----------|------------|--------------|
| z/w=-1/2 | -0.182     | 0.17         |
| z/w=-1/4 | -0.182     | 0.17         |
| z/w=0    | -0.325     | 0.17         |
| z/w=1/4  | -0.268     | 0.17         |
| z/w=1/2  | -0.239     | 0.17         |

Figure 13. Comparison of the minimum Cp value to the position of y/h on the model using active control, Usc1

The minimum pressure coefficient for models without controls is obtained at y/h=0.83 at all z/w positions as it is the starting position of the flow separation process resulting in a decrease in pressure distribution.

The same phenomenon occurs in models with Usc2 suction velocity, where the use of suction active control can increase the minimum pressure, greater than that of models without active control. The minimum pressure coefficient is at the bottom of the back of the vehicle model (y/h=0.17), as depicted in figure 14. where the respective values of each position are z/w=-1/2 of -0.168, z/w=-1/4 of -0.168, z/w=0 of -0.195, z/w=1/4 of -0.168 and z/w=1/2 of -0.168. An interesting finding is that the increase in the minimum pressure coefficient on the model with the Usc2 suction velocity is also higher than that of the model with the Usc1 suction velocity at all positions of the ratio of the grid width to the model width, z/w.

Table 4. Minimum pressure coefficients on the model with Usc2

| Position | Minimum Cp | y/h position |
|----------|------------|--------------|
| z/w=-1/2 | -0.168     | 0.17         |
| z/w=-1/4 | -0.168     | 0.17         |
| z/w=0    | -0.195     | 0.17         |
| z/w=1/4  | -0.168     | 0.17         |
| z/w=1/2  | -0.168     | 0.17         |

Figure 14. Comparison of the minimum Cp value to the position of y/h on the model using active control, Usc2

The model with Usc3 suction velocity showed an increase in the minimum pressure coefficient when compared to the model with no active control and the model with Usc1 suction velocity. But the increase in the minimum pressure coefficient of the model with the Usc3 suction velocity is not higher than that of the model with the Usc2 suction velocity. The lowest pressure, as shown in figure 15, is also at the bottom of the back of the model (y/h=0.17) with each value of z/w=-1/2 of -0.180, z/w=-1/4 of -0.179, z/w=0 of -0.210, z/w=1/4 of -0.217 and at position z/w=1/2 of -0.177.

Table 5. Minimum pressure coefficients on the model with Usc3

| Position | Minimum Cp | y/h position |
|----------|------------|--------------|
| z/w=-1/2 | -0.180     | 0.17         |
| z/w=-1/4 | -0.179     | 0.17         |
| z/w=0    | -0.210     | 0.17         |
| z/w=1/4  | -0.217     | 0.17         |
| z/w=1/2  | -0.177     | 0.17         |

Figure 15. Comparison of the minimum Cp value to the position of y/h on the model using active control, Usc3
For models with Usc₄ suction velocity the minimum pressure coefficient also shifts at the bottom of the rear of the vehicle model (y/h=0.17) which is shown on figure 16. The increase is higher compared to the model with the Usc₁ suction velocity, but it is lower compared to the model with the Usc₂ and Usc₃ suction velocity. The values of pressure coefficient for the position of the ratio of the width of the grid to the width of the model are z/w=−1/2 of -0.181, z/w=−1/4 of -0.181, z/w=0 of -0.215, z/w=1/4 of -0.221 and for the position z/w=1/2 of -0.186.

Table 6. Minimum pressure coefficients on the model with Usc₄

| Position | Minimum Cp | y/h position |
|----------|------------|--------------|
| z/w=−1/2 | -0.181     | 0.17         |
| z/w=−1/4 | -0.181     | 0.17         |
| z/w=0    | -0.215     | 0.17         |
| z/w=1/4  | -0.221     | 0.17         |
| z/w=1/2  | -0.186     | 0.17         |

Figure 16. Comparison of the minimum Cp value to the position of y/h on the model using active control, Usc₄

On all vehicle models with suction active control, the increases in the minimum pressure coefficient are omnipresent. Obtained results indicated a delay in flow separation and minimization of the recirculation wake zone at the back area of the vehicle model as shown in figures 7, 8, 9, 10, and 11 in regard to the flow characteristics. These results are in line with previous research, where the use of flow active control that is positioned at the top edge of the rear of the vehicle model can increase the base pressure [21].

The comparison of the minimum pressure coefficients on the without active control model and the models equipped with active controls with variations suction velocity of Usc₁, Usc₂, Usc₃, and Usc₄ are shown in table 7.

Table 7. Comparison of minimum pressure coefficients

| Models   | Minimum Cp | z/w position | Increased Cp (%) |
|----------|------------|--------------|------------------|
| Without control | -0.724     | -1/2         | -                |
| Usc₁     | -0.325     | 0            | 55.1             |
| Usc₂     | -0.195     | 0            | 73.1             |
| Usc₃     | -0.217     | 1/4          | 70.0             |
| Usc₄     | -0.221     | 1/4          | 69.5             |

3.3. Drag Coefficient (Cd)

The comparison of the drag coefficients obtained for models without flow control and using flow control with variations in the suction velocity of Usc₁, Usc₂, Usc₃, and Usc₄ are shown in table 8.

Table 8. Comparison of drag coefficients

| Models   | F_d (N) | Cd    | Reduction (%) |
|----------|---------|-------|---------------|
| Without control | 0.7012 | 1.4499 | -              |
| Usc₁     | 0.6278  | 1.2982| 10.463        |
| Usc₂     | 0.6248  | 1.2919| 10.897        |
| Usc₃     | 0.6251  | 1.2925| 10.856        |
| Usc₄     | 0.6252  | 1.2928| 10.835        |

The highest drag coefficient was recorded on the model unequipped with active control at a value of 1.4499. This happens because the flow separation that occurs at the back of the vehicle model causes the formation of backflow right near the rear wall of the model so that the backward suction phenomenon appears as shown in figure 7. This is in line with research conducted by Gopal and Senthilkumar [22] who found that the main cause aerodynamic drag is the flow separation on the rear wall of the vehicle. For models with Usc₁ suction velocity, the drag coefficient of 1.2982 occurs, giving a reduction of 10.463% lower than to the model without active control. On the model with Usc₂ suction velocity, a drag coefficient of 1.2919 is obtained or occurred reduction of 10.897% is obtained. For models with Usc₃ and Usc₄ suction velocity, each drag coefficient of 1.2925 and 1.2928 or occur red reductions of 10.856% and 10.835% are obtained, respectively. The results revealed that the use of flow active control is able to reduce the drag coefficient, where the greatest reduction was recorded on the model with Usc₂ suction velocity of 10.897%. This is in line with the results obtained on the flow pattern characteristics and the increase in the pressure coefficient which shows that the model with the Usc₂ suction velocity is the model that produces the smallest wake and vortex as shown in figure 9 and the highest increase in the minimum pressure coefficient as shown in table 7. These results are in line with research conducted by
Tarakka et al. [23] which found that the use of active controls on vehicle models is capable in reducing aerodynamic drag. The application of flow separation control is also able to improve overall vehicle performance [24].

4. Conclusions

A. The application of suction active control can reduce the formation of wake and vortex, where the greatest reduction is obtained on the model with U_{sc2} suction velocity of 0.5 m/s.

B. The use of suction active control is also able to increase the minimum pressure coefficient, where the greatest increase in the minimum pressure coefficient of 73.1% is obtained on the model with an U_{sc2} suction velocity of 0.5 m/s.

C. The use of suction active control gives an effect on the reduction of the drag coefficient, where the highest reduction in the drag coefficient of 10.897% is obtained on the model with an U_{sc2} suction velocity of 0.5 m/s.

Acknowledgements

We are very grateful to the Head of Fluid Mechanics Laboratory and Faculty Members of Mechanical Engineering Department, Hasanuddin University for their support and financial assistance.

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