Abstract

The wall shear stress is a quantity of profound importance for clinical diagnosis of artery diseases. The lattice Boltzmann is an easily parallelizable numerical method of solving the flow problems, but it suffers from errors of the velocity field near the boundaries which leads to errors in the wall shear stress and normal vectors computed from the velocity. In this work we present a simple formula to calculate the wall shear stress in the lattice Boltzmann model and propose to compute wall normals, which are necessary to compute the wall shear stress, by taking the weighted mean over boundary facets lying in a vicinity of a wall element. We carry out several tests and observe an increase of accuracy of computed normal vectors over other methods in two and three dimensions. Using the scheme we compute the wall shear stress in an inclined and bent channel fluid flow and show a minor influence of the normal on the numerical error, implying that the main error arises due to a corrupted velocity field near the staircase boundary. Finally, we calculate the wall shear stress in the human abdominal aorta in steady conditions using our method and compare the results with a standard finite volume solver and experimental data available in the literature. Applications of our ideas in a simplified protocol for data preprocessing in medical applications are discussed.

Keywords: fluid flow, numerical methods, the lattice Boltzmann method, LBM, wall shear stress, WSS

1. Introduction

Computer simulations of the blood flow in the human circulatory system allow physicians to visualize the internal flow structure, a capability that might have a significant impact on diagnosis and medical treatment of arterial diseases (for review see [1, 2, 3]). Digital simulations have been applied in many scenarios and for various purposes e.g. to study the role of hemodynamic forces in the development of atheromatous plaques in human carotids atherosclerosis [4], in a clinical study on the rupture risk assessment in the cerebral aneurysm [5] or in a discovery of a shear-driven mechanism of platelet aggregation in arterial thrombosis [6]. Direct results of hydrodynamical simulations (velocity and pressure fields) are not practical in medical analysis. To better understand the impact of the blood flow dynamics on the complex biochemical phenomena associated with the development of vascular diseases, a wall shear stress (WSS, see Fig. 1) with its gradient (WSSG), a hoop stress or an oscillatory shear index (OSI) are often computed. WSS is a tangential force exerted on the unit of the wall surface by the flow and OSI describes its deviation from the average direction [8]. WSS has been a subject of intensive research for many years, including studies on its role in development of human atherosclerosis [8, 10], the growth of an intracranial aneurysm [11] and development of the dissecting aneurysm in the human aorta [12]. WSS determines the structure, function and gene expression of an endothelial cells [4, 13]. It is now well established that a low shear stress is correlated with rapid development of various vascular malfunctions [14, 15, 16, 17, 18]. These facts motivate physicians to find applications of the WSS in diagnosis [19, 20] and to seek efficient and practical protocols for its measurement.

In order to compute WSS for patient-specific data one has to acquire the organ geometry, convert and import it to the Navier–Stokes solver, compute the velocity field and take gradients of its components at the wall. The process may be iterative and further time-averaging might be necessary if the flow is transient. The major difficulty of this procedure is the prohibitively long pre-processing and computation time which limits its applicability in a real patient diagnosis. The pre-processing involves mainly the conversion of digital data from the computer tomography scans into a three-dimensional mesh required by standard finite element/finite volume solvers [21, 22, 23]. High quality of the output mesh is required by flow solvers to work properly and computer-human interaction during the meshing process is often necessary. The flow solver runtime depends on
many factors e.g. solver type, mesh resolution, flow properties and hardware used. It may be a bottleneck of the whole procedure as well, particularly for geometries of high resolution, time dependent flows and problems involving fluid-structure interaction. It is widely believed that the run time of fluid solvers could be significantly reduced by using efficient parallel algorithms on massively parallel architectures, e.g. graphics processing units (GPUs) [24].

Time issues mentioned above might be addressed with the lattice Boltzmann method (LBM), a recently developed method for solving fluid flow problems which is based on the kinetic theory of gases [25]. The method operates directly on voxel data and thus does not require meshing at the level of accuracy required by standard solvers. It also proved excellent parallel scalability on supercomputers [25, 27] as well as on emerging parallel architectures [28, 29, 30, 31]. The method was validated against analytical solutions in a two-dimensional oscillatory channel flow and its second order accuracy in space and time was confirmed [32, 33]. In medical applications, LBM was tested in solving time-dependent three-dimensional flows in models of the human abdominal aorta [34] and compared against finite element solvers in the superior mesentric artery [35]. Palabos, a parallel open source LBM implementation, was used to model blood flow in cerebral aneurysm and tumor development were recently discussed [36].

Non-Newtonian LBM models also exist e.g. the Carreau–Yasuda fluid model applied in cerebral aneurysm problem [37]. Palabos, a parallel open source LBM implementation, was used to model blood flow in cerebral aneurysm and compared to a finite volume flow solver [37, 38] and the difference between the velocity fields obtained in both methods was estimated to be around 10%. Application of LBM to modeling aneurysm and tumor development were recently discussed [39]. He et al. compared LBM combined with the level-set method to an OpenFOAM solver in patient-specific geometry [40].

Despite the second order accuracy of the LBM stress tensor in bulk fluid being recently reported [41, 42], the WSS in LBM suffers from inaccuracy introduced by a staircase boundary. The problem may be eased with subgrid conditions [43] or the recently developed unstructured mesh LBM formulation [44], but both of them introduce additional complexity into the solver code, making it more difficult to maintain and parallelize. Recently Stahl et al. showed that a few lattice nodes away from the wall the error is significantly decreased in the standard LBM formulation [45]. He suggests to calculate normal vectors required by this procedure from the velocity field as the fluid flowing along the noslip wall the tangential component of the velocity vanishes and the no slip boundary condition holds. However, in a vicinity of the wall the tangential component of the velocity does not vanish; the corresponding gradient of the tangential
velocity component along the wall normal generates the wall shear stress, a force that is exerted by the fluid on the wall’s surface. WSS may be expressed as the difference between the Cauchy stress on a plane and its projection on the plane normal (see Appendix A). For an incompressible, Newtonian fluid the WSS in the lattice Boltzmann method can be calculated in terms of the non-equilibrium distribution function $f^{neq}$:

$$\tau_i = \frac{\mu \alpha}{c_s^2 \rho} f^{neq}_{\alpha} c_{ij} n_j (c_{\alpha i} - c_{\alpha k} n_k),$$

where $c_s$ is the lattice sound speed, $\mu$ is the dynamic viscosity, $\rho$ is the fluid density, and $\alpha$ is the LBM relaxation parameter. $c_{\alpha i}$ and $n_i$ denote lattice and normal vector components, i.e. $n_i$ is the $i$-th component of the wall normal vector $n$ and $c_{\alpha i}$ is the $i$-th component of the lattice vector $c_{\alpha}$. Einstein summation convention is implied. A detailed derivation of Eq. (2) is given in Appendix A with its algorithmic form in Appendix B.

### 3.1. Wall orientation

To calculate WSS with Eq. (2) the wall orientation and the corresponding wall normal $n$ must be known. A typical three-dimensional LBM simulation works on a discrete, uniform grid built of adjacent cubic cells. If a fluid node has a no-slip boundary neighbour then there is a boundary facet between them at which the no-slip condition of zero velocity is fulfilled. The facet is perpendicular to the line that joins node centers whereas its physical location depends on the LBM relaxation time. For a three-dimensional model there are six possible boundary normals: $\pm e_x = (\pm 1, 0, 0)$, $\pm e_y = (0, \pm 1, 0)$, and $\pm e_z = (0, 0, \pm 1)$.

In order to calculate $n$ we suggest taking an average of normal vectors of the individual fluid/no-slip boundary facets in a spherical vicinity of the cell. For this we find all boundary facets whose centers lay in the sphere of radius $r$ centered at a given cell (Fig. 2a). If $b_i$ is the boundary normal of the $i$-th facet then the above procedure may be expressed as the arithmetic mean:

$$n = n_b \frac{1}{n_b-1} \sum_{i=0}^{n_b-1} b_i,$$

where $n_b$ is the number of facets within the neighborhood. The same procedure is repeated for each boundary cell (Fig. 2 right).

### 3.2. Weighted scheme

In the application of Eq. (3) an unacceptable averaging of normals that belong to separate or opposite fragments of the object’s surface may take place if $r$ is taken to be too large. In the hemodynamical scenario, this may occur in small arteries or bifurcation regions where object radii are smaller than $r$ and details may be lost due to averaging normals to opposite facets. Moreover, if $r$ is too small, $n$ is influenced by the staircase approximation of the surface and may change rapidly from node to node. To deal with these issues we suggest to weight the mean in Eq. (3):

$$n = \left( \sum_i w_i \right)^{-1} \sum_i w_i b_i,$$

where the sum goes over neighbour facets. We choose $w_i = 1/(1 + d_i)^{\gamma}$, where $d_i$ is the distance between the center of the node for which the average is computed and the center of the node with normal $b_i$.

The exponent $\gamma$ defines how strongly concentrated the weighting is, with $\gamma = 0$ corresponding to uniform weights (reducing to Eq. (3)); the strength of weighting increases with the value of $\gamma$.

### 3.3. Numerical tests for normal vectors

We test the above procedure in three-dimensional ball and tube geometries for which exact normal vectors are known. First, a boolean array of size $L^3$ is generated such that it holds 0 if the voxel (three dimensional pixel) center lays inside the object and 1 otherwise. Boundary cells are represented by voxels equal to 1 having at least one adjacent zero neighbour. The object (ball or tube) radius is $R = L/2$ and we test three resolutions: $L = 25^3, 50^3$ and $100^3$. We calculate normals on the circle created by boundary nodes at half vertical position. As an example, in Fig. 3 the geometry of a ball for $L=25$ is visualized with normal vectors calculated using our procedure. Fig. 4 depicts the maximum error between geometric and exact normals. Results are compared to dynamic normals at various distances from the wall [45]. Geometric normals are found to be more accurate. To test the influence of the averaging radius on the accuracy of the normal for various exponents $\gamma$ we vary $r$ and
calculate the error on a ball and a tube of radii $R$. The resulting error is a non-monotonic function of $r$ with a minimum at $r/R = 1$ (see Fig. 5). At $r/R > 1$ the mean error for $\gamma = 0$ grows rapidly whereas for $\gamma \neq 0$ it remains constant (with slight favour to $\gamma = 1/2$). A large error for $r/R > 1$ for $\gamma = 0$ comes from the fact that the averaging area covers the entire object, the resulting normal is 0, and the information on the object geometry vanishes. If $\gamma \neq 0$, the problem is avoided because normal vectors defined on facets on the boundary lying on the opposite side of the object have much lower weights.

4. Wall shear stress in a channel flow

In this section we test the accuracy of the WSS equation (2) combined with geometric normals in an inclined and bent channel flows.

4.1. Inclined channel

A two-dimensional channel of an initial resolution $120 \times 20$ lattice units (l.u.) was inclined at angle $\alpha$ to the x axis so that the radii of the inlet and outlet were kept constant. The analytical velocity profile (Poiseuille flow) was imposed on the inlet and outlet boundaries. The lattice viscosity was $\nu = 0.01$ and the maximum velocity in the channel center was $u_{\text{max}} = 0.01$ (lattice units). Using deviations from the analytical velocity field we found the steady state in less than $10^4$ iterations. We ran the flow for $\tan \alpha \in [0; 2]$ and calculated dynamical and geometrical normal vectors in a strip of length 80 l.u. (centered horizontally). In Fig. 6 the maximum angle between the numerical and exact vectors is given as a function of the channel inclination angle. The error of the geometric procedure decreases with $r$ and is smaller than for the dynamic scheme. The most accurate results were obtained for $\tan \alpha = 0$ and $\tan \alpha = 1$.

Next, we compute WSS using Eq. (2) and show that in an inclined channel WSS is essentially independent of the choice of the normal vector type (Fig. 7). This implies that the largest contribution to the error comes from the velocity field near the staircase boundary. This fact is also visible in Fig. 6 where the large error of dynamic normals is caused by the velocity field inaccuracy. The maximum relative error of numerical WSS calculated at a single node is much larger than the error of the mean taken over 80 nodes. Therefore one could use an equation similar to Eq. (4), where a spatial average of WSS in a vicinity of the
wall element is computed:

$$\tau = \left( \sum_{i} w_i \right)^{-1} \sum_{i} w_i \tau_i,$$

(5)

where $w_i$ are weights of the same form as used for normals. However, we leave applicability of Eq. (5) to more complex flows as an open problem.

### 4.2. Bent channel

Next, we investigate the flow in a two-dimensional bent channel geometry defined by its inner and outer radii ($R_1$ and $R_2$, respectively). For the mesh of resolution $W \times H$ a site placed at $(i, j)$ is of fluid type if $R_1 < |(i - W/2, j) - R_2$, otherwise it is a no-slip site (or it is excluded). We take $R_1 = 10$ and $R_2 = 20$ and the mesh resolution $45 \times 25$ l.u. and run the steady state simulation with $\nu = 0.01$ and analytical inlet/outlet boundary conditions.

We calculated normals using both procedures and compare them visually against exact normals (Fig. 8). To understand the influence of the flow field near staircase wall on dynamic normals the flow field is shown using streamlines (Fig. 8 a). We determined the angle between each individual pair of numerical and exact vectors using the data from Fig. 8 and found the maximum error of the dynamic scheme to be around $25^\circ$. For the geometric scheme the error is kept below $10^\circ$. The average errors are calculated as $10.8^\circ$ and $3.9^\circ$ for dynamic and geometric vectors, respectively.

Then we computed WSS at the inner wall at various angular positions and plotted its components against exact solutions (Fig. 9). The numerical data follow the analytical solutions. Regions of larger deviations may be recognized (e.g. $\tau_x$ at $\alpha \in (20, 40)$) mostly because WSS was calculated at the first node next to the boundary. An interesting observation may be done for the central region ($\alpha \in (80, 100)$) where introduction of the geometric normals slightly decreased the error for $\tau_z$ (similarly for $\tau_x$). The reason could be that the central part of the bent channel is horizontally flat in the grid representation (see Fig. 8) and the velocity field follows the geometry. Therefore, the resulting dynamic normals are almost perpendicular to the surface whereas they are not in the original geometry (a bent inner circle). This results in a large systematic deviation of the $x$ component of WSS which is slightly decreased if more accurate geometric normals are used.

### 5. A flow through the human abdominal aorta

Finally, we test the procedures described in previous sections in a real situation and compute WSS in an abdominal aorta model based on patient-specific data. For this purpose we utilize data from the Virtual Family (VF) [51] – a freely available library of high resolution anatomical human body models based on MRI measurements. We use Eq. (2) to compute the geometric normals and Eq. (3) for WSS.

#### 5.1. Data preparation

From several models provided by the VF we chose Duke – a 34 year old man. Using the software included in VF we manually selected a part of his abdominal aorta 17.8 cm over and 5.2 cm under the bifurcation point. We exported voxel data as raw files that encode a tissue type with integers. Then we used the CVMLCPP library [52] ray marching algorithm to approximate the artery surface. The resulting mesh was of very poor quality, therefore further smoothing was done using MeshLab [53]. At this point we either voxelized the mesh to get various grid resolutions for the LBM or continued meshing until we obtained a three-dimensional unstructured volumetric mesh using NETGEN [54]. Next we applied the boundary conditions and exported the mesh in the neutral NETGEN format for further use in FVM software.

#### 5.2. Simulation

First, we ran the steady state LBM simulation using the Saffish library adjusted for our purposes. We imposed a flat velocity profile on the inlet and zero pressure at both outlets of the model. Due to the tortuous geometry of the aorta the flat inlet velocity causes a macroscopic flow with a slightly lower effective flowrate. We used the following parameters: $u_{0,\alpha} = 0.0764$ m/s (the velocity at inlet), $R_{0,\alpha} = 0.03$ m (the diameter of the inlet), $\nu = 3 \cdot 10^{-6}$ m$^2$/s (kinematic viscosity.
of the fluid), \( \rho_p = 1000 \text{ kg/m}^3 \) (the fluid density). Thus, the characteristic time \( t_{0,p} = \frac{l_{0,p}}{u_{0,p}} = 0.393 \text{ s} \) and the Reynolds number \( Re=300 \) (based on an effective flowrate). The model was subdivided into \( 461 \times 108 \times 163 \) voxels (5mm resolution). The number of grid nodes along the inlet diameter was \( N = 60 \), thus \( \delta x = 0.0166 \). We chose \( u_{lb} = 0.125 \), for which \( \delta t = \frac{u_{lb} \cdot \delta x}{\delta t} = 0.00208 \) was found. The final lattice viscosity \( \nu_{lb} = \frac{\delta t}{\delta x^2} \cdot \frac{1}{Re} = 0.0098 \) was found. We continued the simulation until the steady state was reached (41 000 time steps) based on the observation of the velocity and pressure field convergence. The resulting distribution function that describes the steady state was used in Eq. (2) to get WSS at 2 l.u. away from the boundary. We used our geometric normals in Eq. (2) as well as to move away from the boundary nodes. We first computed a dimensionless counterpart of \( \tau_{lb} \), namely \( \tau_d = \frac{\delta t}{\delta x^2} \cdot \tau_{lb} = 63.7 \tau_{lb} \). Then, the stress in [Pa] was obtained using \( \tau_p = \frac{\rho_p \cdot l_{0,p}}{l_{0,p}} \cdot \tau_d = 5.84 \tau_d \).

Second, we ran the same steady flow using finite volume Semi Implicit Method for Pressure Linked Equations (SIMPLE) [55]. We used simpleFoam solver from OpenFOAM [56] with previously generated mesh built of 496 656 tetrahedral elements. We applied a constant flowrate condition at the inlet with the flowrate \( Q=2.1206 \times 10^{-2} \text{ [m}^3/\text{s}] \) corresponding to \( Re = 300 \). The zero pressure and zero velocity gradient was applied at both outlets. We stopped the simulation after 1 second and confirmed that the convergence criteria was met (the initial and final residuals of pressure and velocity was smaller than \( 10^{-5} \)). Then, we computed WSS using standard OpenFOAM tools.

The resulting WSS in the abdominal aorta model computed with both methods is visualized as a color map (Fig. 1 and Fig. 10). The resulting ranges of WSS were \( (6.74 \times 10^{-5}, 0.32) \) for LBM and \( (0, 0.42) \) for FVM. We found that the discrepancy in maximum values is due to the extreme values at the very boundary conditions, thus we encoded the WSS magnitude in the range of WSS \( \in (0, 0.15) \text{ [Pa]} \) only for better visualization.

The profile of WSS (Fig. 10 a) is taken along the white dashed line (see Fig. 10 b) from a gray scale projection of the results onto a plane. The peak value in the bifurcation area is around 0.12 [Pa] as observed in two branches behind the bifurcation point where the wavy profile of WSS may be easily recognized. The wavy profile of WSS seems to be mainly an effect of the geometry: the smaller profile diameter, the larger WSS. The quantitative comparison of WSS between LBM and WSS (Fig. 10a) shows an excellent agreement in the area not influenced by boundary conditions.

Next, we compare the WSS magnitudes obtained for both methods with the experimental data measured using the laser photochromic dye tracer technique [57]. There, for similar conditions (the abdominal aorta of the healthy 35-year-old subject...
at rest and the steady flow) and the Reynolds number $Re = 227$ (based on the inlet flowrate) the measurement values of WSS were in the range $\tau \in (0.06, 0.3)$ [Pa]. We found our numerical computations agree well with the experimental data. The remaining difference in the minimum value of WSS is most probably an effect of the patient-specific geometry as we found the minimum of WSS in Fig.10 at $y = 16$cm, where the enlargement of the cross-section of our aorta model is noticeably.

6. Discussion

Comparison of normal vectors computed for three-dimensional objects with two different methods given in Sec. 5.1 indicates an improvement in accuracy of our procedure against the dynamic scheme based on the velocity field. The improvement is significant e.g. close to the wall for a small lattice the error for dynamic normals is close to 90$^\circ$ but it remains below 10$^\circ$ in our method (Fig.6). Additional tests in the flow through an inclined (Fig.5) and bent (Fig.8) channel flows confirm higher accuracy of geometric normals. Application of the geometric normals gives practically the same WSS as when its dynamic or exact counterparts are used in an inclined channel.

In the bent channel, however, some differences are visible if separate stress components are analyzed (see Fig.9). We found a region of an increased WSS accuracy based on the geometric normals (central part of the inner wall). We believe this observations favorize the use of our scheme as it is simpler, more efficient (fewer algebraic operations) and provides the sense of the normal. We are aware that our method is nonlocal which may lead to complications in parallel implementation, this, however, might be solved e.g. by using algorithms that utilize fast shared memory of GPU processors.

Our tests in a real patient-based geometry show an excellent agreement with a standard finite volume solver. The WSS in the fluid flow through the abdominal aorta is both qualitatively and quantitatively the same in LBM and FVM (see Fig.10). The slight difference between both methods at inlets and outlets is an effect of differences in the boundary conditions used. The data in Fig.10a) give an impression of what is a typical length scale at which the imposed inlet boundary conditions play a role in hemodynamical simulations (e.g. we got around 10 cm from the inlet for the given setup).

To get acceptable matching of the results in Fig.10 we moved the computation 2 nodes away from the wall (as suggested in [45]). This procedure has, however, some limitations. First, while producing color-maps of WSS we noticed some missing nodes (see discontinuity in Fig.10) just above the bifurcation point. The reason for it is the following: if we start from two neighbouring sites and move some distance along their normals, the destination nodes no longer have to be neighbours (especially if normals were pointing in different directions) and empty nodes at the surface may be visible. This artifact does not influence WSS values obtained. To produce decent visualization one could additionally compute WSS for those empty sites, e.g. in the Fig.1 we visualize WSS at a distances $d=2$ and at $d=0$ which practically eliminated the problem of empty sites. Second, moving away from boundary may be unwanted in general because one could argue the quantity is no longer the wall shear stress but rather a near-the-wall shear stress. To make the use of the information at the wall surface only, we suggested averaging of WSS at the wall in the neighbourhood of a site for which WSS is calculated (see Eq. 5). Averaging decreased the error of WSS at almost each site of the inclined channel flow by more than 50% (see Fig.10). To make the WSS averaging procedure a part of a complete protocol one should further test its behavior for smaller averaging radii and various three-dimensional geometries.

We found that for a single model the preparation of FVM mesh described in Sec. 5.1 takes around an hour without data segmentation (the Virtual Family implements this). In contrast, LBM could utilize the segmented MRI/CT data directly and skip the meshing process. We can imagine that, if the gray level was decoded correctly into some of physical variables used in LBM simulation (e.g. endothelial wall roughness) then these data could be used directly without segmentation. In this way one could turn the largest weakness of the LBM method (staircase approximation of the boundary) into its strongest point in hemodynamical applications (no need for tedious data preprocessing), where our simple procedure for normals would be of high importance.

7. Conclusions

There are several reasons why one would want to calculate the wall orientation in LBM using the geometric procedure described here. First, our method gives a gain in accuracy over dynamic normals. Second, the computation overhead is much smaller as the averaging is done using a few simple algebraic operations rather than by solving an eigenvalue problem. Third, we showed that our procedure gives practically the same WSS as if exact normals were used. Fourth, the computation of geometric normals is – by definition – independent of the Reynolds number whereas dynamic normals are dependent on Re, which is unphysical. Fifth, it is suitable for the time dependent flows. Sixth, the given scheme provides the sense of normals that might be crucial in the calculations that require determination of separated wall shear stress components e.g. oscillating shear index and wall shear stress gradient.

8. Acknowledgement

The publication has been prepared as part of the project of the City of Wroclaw, entitled – “Green Transfer” – academio-business knowledge transfer project co-financed by the European Union under the European Social Fund, under the Operational Programme Human Capital (OP HC): sub-measure 8.2.1 (MM). ZK was supported by MNiSW grant No. N 519 437939. We acknowledge support and discussions from Dominik Szcerba, Jakub Kominarczuk, Ziemowit Malecha, Michał Januszewski, Jonas Latt and Bastien Chopard. Additionaly to the software mentioned in the text we used also Palabos, Paraview and Techdig. We gratefully acknowledge hardware donation from Nvidia.
Appendix A. The stress on arbitrary plane in LBM

The stress tensor in fluids is a sum of pressure and viscous terms:

$$\sigma_{ij} = -p\delta_{ij} + \sigma'_{ij}. \quad (A.1)$$

The Cauchy formula gives the overall stress on the wall with a normal vector \( n \) (Fig. A.11):

$$T^{(n)}_i = \sigma_{ij} n_j. \quad (A.2)$$

The shear stress may be computed as the difference between the overall stress and its projection onto the normal:

$$\tau_i = \sigma_{ij} n_j - (\sigma'_{kj} n_k) n_j. \quad (A.3)$$

We insert Eq. (A.1) into Eq. (A.3), and since \( \delta_{jk} n_k = 1 \), the equation for WSS reads:

$$\tau_i = \sigma'_{ij} n_j - (\sigma'_{kj} n_k) n_j. \quad (A.4)$$

For the incompressible, Newtonian fluid the viscous stress \( \sigma'_{ij} \) is a linear function of the rate of strain \( \varepsilon_{ij} \):

$$\sigma'_{ij} = 2\mu \varepsilon_{ij}. \quad (A.5)$$

Combination of Eq. (A.4) and Eq. (A.5) gives:

$$\tau_i = 2\mu \left( \varepsilon_{ij} n_j - (\varepsilon'_{kj} n_k) n_j \right). \quad (A.6)$$

A definition of the rate of strain tensor in LBM, which uses the non-equilibrium part of the distribution function reads:

$$\varepsilon_{ij} = \frac{\omega}{2} \sum_a f^\text{neq}_a c_{ai} c_{aj}, \quad (A.7)$$

where \( f^\text{neq}_a = f_a - f^\text{eq}_a \) and \( f^\text{eq}_a \) is an equilibrium Maxwell-Boltzmann distribution. Finally, we combine Eqs. (A.6) and (A.7) and get Eq. (2).

Appendix B. WSS algorithm

Algorithm 1 Calculation of WSS from Eq. (2).

```plaintext
dim ← 3
C ← (\mu \omega)/(c_i^2 \varphi)
for i = 0 \rightarrow dim - 1 do
  \text{sum}_i ← 0
  for all \alpha do
    \text{sum}_k ← 0
    for k = 0 \rightarrow dim - 1 do
      \text{sum}_k ← \text{sum}_k + c_{ai} n_j n_k
    end for
    for j = 0 \rightarrow dim - 1 do
      \text{sum}_j ← \text{sum}_j + C \cdot \text{sum}_i \cdot n_j
    end for
  end for
end for
WSS ← |\text{sum}_i|
```

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